

SP-166-D

January 10, 1965

Revision 1 - November 1, 1965

ELECTRICAL REFERENCE HANDBOOK

LAUNCH EQUIPMENT BRANCH

ELECTRICAL SUPPORT EQUIPMENT

LC-39

VOLUME I of II

Launch Equipment Branch Launch Support Equipment Engineering Division

SP-166-D

- (Filter)

Sec. 1º

January 10, 1965

Revision 1 - November 1, 1965

APPROVAL

ELECTRICAL REFERENCE HANDBOOK

FOR

LAUNCH EQUIPMENT BRANCH ELECTRICAL SUPPORT EQUIPMENT

LC-39

Griffin

Chief, Electrical Section EDV-155

R. T. Moore, Jr.

Chief, Launch Equipment Branch EDV-15

. Poppel

Director, Launch Support Equipment Engineering Division EDV-1



TABLE OF CONTENTS

٠

.

VOLUME I of II

Section			Page
I		INTRODUCTION	
	1-1.	SCOPE AND PURPOSE	1-1
	1-2.	RELATED DOCUMENTS	1-2
п		LAUNCH COMPLEX 39	
	2-1.	GENERAL DESCRIPTION	2-1
ш		ELECTRICAL SUPPORT EQUIPMENT (ESE)	
	3-1.	INTRODUCTION	3-1
	3-2.	SERVICE ARMS SUBSYSTEM	3-4
	3-7.	APOLLO ACCESS ARM SUBSYSTEM	3-14
	3-12.	OXYGEN CONDITIONING SUBSYSTEM	3-20
	3-16.	HYDRAULIC CHARGING UNIT SUBSYSTEM	3-24
	3–20.`	LAUNCH EQUIPMENT FIRING CIRCUITS SUBSYSTEM	3-30
	3-28.	TAIL SERVICE MASTS SUBSYSTEM	3-42
	3-33.	PNEUMATICS CONTROL AND DISTRIBUTION SUBSYSTEM	3-48
	3-38.	Q-BALL COVER REMOVAL SUBSYSTEM.	3-56
	3-43.	ESE POWER SUBSYSTEM	3-60
	3-47.	S-IC ENGINE SERVICING PLATFORMS SUBSYSTEM	3-72
	3-51.	PLATFORM TRANSPORTER SUBSYSTEM	3-80
	3-55.	S-II and S-IVB ENGINE SERVICING PLATFORM SUBSYSTEM	3-84
	3-59.		

TABLE OF CONTENTS (Continued)

Section			Page
IV		ESE STANDARD COMPONENTS	
	4-1.	GENERAL	4-1
	4-2.	STANDARD ELECTRICAL DISTRIBUTOR J75M07835	4–2
	4-7.	STANDARD ELECTRICAL DISTRIBUTOR E75M09808	4-4
	4-12.	PORTABLE DC POWER SUPPLY D75M51116	4-6
	4-17.	STANDARD 24 INCH VERTICAL RACK E64-KN-F-260-10	4-8
	4-22.	STANDARD PATCH DISTRIBUTOR F75M04681 (42 CONNECTOR)	4-10
	4-27.	STANDARD PATCH DISTRIBUTOR F75M04681 (54 CONNECTOR)	4-12
	4-32.	STANDARD PATCH DISTRIBUTOR D75M09313 (27 CONNECTOR)	4-14
	4-37.	STANDARD RELAY MODULE D75M50348	4-16
	4-42.	STANDARD DIODE MODULE D75M50364	4-18
	4-47.	STANDARD METER PANEL J75M10165	4-20
	4-52.	STANDARD FUSE PANEL J75M10163	4-22
	4-57.	STANDARD MODULE FRAME J75M10432	4-24
	4-62.	STANDARD 500 AMPERE POWER SUPPLY	4-26
	467.	STANDARD TEST LOAD E40M03971	4-28
	4-72.	STANDARD LOAD CONTROL PANEL E40M03535	4-30
	4-77.	STANDARD BATTERY CHARGER	4-32
	4-82.	STANDARD BATTERY PANEL J40M03605	4-34
	4-87.	STANDARD POWER MODULE TYPE IIA J75M12295	4-36
	4-92.	STANDARD RECEPTACLE DISTRIBUTOR TYPE I E75M12316	4-38
	4-97.	STANDARD BUS DISTRIBUTOR E75M12323	4-40

1

TABLE OF CONTENTS (Continued)

Section			Page
IV	4-102.	STANDARD CONSOLE E64-KN-F-261-4	4-42
	4-107.	INTEGRATION RACK	4-44
	4-112.	STANDARD LATCHING RELAY MODULE D75M05895	4-46
v		TEST EQUIPMENT, ESE	
	5-1.	GENERAL	5-1
	5-2.	LAUNCHER GROUND EQUIPMENT TEST SET E75M12386	5-2
	5-7.	MINIATURIZED UNIVERSAL TEST SET E75M13761	5-8
	5-12.	PORTABLE ARM CONTROL CONSOLE J75M07542	5-10
	5–17.	UNIVERSAL PATCHBOARD TEST SET E75M09032	5-12
	5-22.	MSFC SERVICE ARM/TAIL SERVICE MAST TEST SITE	5-14
VI		REFERENCE INFORMATION	
:	*6-1	ESE FAMILY TREE OF DRAWINGS	6-1
	6-2	REFERENCE SPECIFICATIONS	6-147
	6-3.	ESE EQUIPMENT LOCATIONS	6-161/162
	6-4.	ESE CABLE INTERCONNECT DIAGRAMS	6-169
	6-5.	REFERENCE DESIGNATION NUMBERS VS. ASSEMBLY DRAWING NUMBERS	6-220
Appendi	x A	GLOSSARY	A -1

* contained in Volume II

1

1420

1. W. 1

1

9

v/vi

LIST OF ILLUSTRATIONS

 $10^{10} dt^{1/2}$

1

a series

18%

1.18⁻²-4-1

Ľ

•

Number	Title	Page
2-1	Launch Complex 39	2-2
2-2	Typical Launch Pad, LC-39	2-3
2-3	Mobile Launcher	2-4
2-4	Electrical Support Equipment	2-5
2-5	ESE Subsystems	2-6
3-1	Service Arms Subsystem	3-5
3-2	Typical Inflight Service Arm and Test Installation	3-7
3-3	Typical Preflight Service Arm and Test Installation	3-7
3-4	LCC Control Panel, S-II INTERMEDIATE	3-8
3-5	LCC Summation Panel D75M14440	3-9
3-6	Local Control Unit - Service Arms	3-11
3-7	Arm Control Panel J75M07546-3, Portable Arm Control Console	3-12
3-8	Charging Control Panel J75M07546-1, Portable Arm Control Console	3-13
3-9	Apollo Access Arm Subsystem.	3-15
3-10	Apollo Access Arm Level (Access Hood Control)	3-17
3-11	Apollo Access Arm Level (Arm Control)	3-17
3-12	LCC Control Panel, Apollo Access Arm.	3-18
3-13	C/M Access Arm Overlay for Arm Control Panel, Portable Arm Control Console	3-19
3-14	C/M Access Arm Overlay for Charging Control Panel, Portable Arm Control Console	3-19
3-15	Oxygen Conditioning Subsystem	3-2 1
3-16	Oxygen Conditioning Console J75M13664	3-22
3-17	GOX Module J75M12341	3-2 3
3-18	Hydraulic Charging Unit Subsystem	3-2 5
3–19	LCC Panel, Hydraulic Charging Unit	3-27
3-20	Hydraulic Charging Unit and Main Switches	3-29

vii

1. A.

- 1 C

19 N 19 N

in Sur

4

	<u>"Estle</u>	Page
8-21	Rolddown Arms Firing and Monitoring Circuits	3-31
$\gamma_{\pm}22$	Lacacher Accessories Test Set No. 2	3-33
0-23	Typical Level, Service Arms Firing Circuit	3-34
3-244	LCC Panel Service Arms Firing Circuit	3-35
3-2-0	Gervice Arms Firing Circuit.	3-36
<u>^2</u> ₽	Typical Service Arm Control Switches	3-38
3-26	Typical Holddown Arm and Test Installation	3-38
3-27	Tower MI Firing Distributor No. 9083	3-39
3-28	Service Arm MI Firing Distantation (Typical)	3-40
3-29	MI Cabling, Typical Mobile Launcher Level	3-41
3 - 39	MI Cable, Typical Installation	3-41
3-31	Tail Service Masts Subsystem	3-43
ان میں اور	LCC Control Panel, Tail Service Mast	3-45
3- 35-	Face Panel, TSM Test Set E75M07658	3-46
3-34	Typical Installation, TSM Test Set E75M12020	3-47
S35	Pneumatics Control and Distribution Subsystem	3-49
Ω. <i>-36</i> ,	LCC Panel Pneumatic Distribution System	3-51
27	High Pressure Pneumatics System	3-52
	Typical Test Installation, Valve Panel 11.	3-52
3-39	Face Panel, Launcher Accessories TestSet No. 1 E75M14009	3-53
3-40	Typical Installation, Tower Test Set E75M14137	3-54
3-41	Face Panel, Tower Test Set E75M14137	3-55
3-42	Q-Ball Cover Removal Subsystem	3-57
See 20	Q-Ball Cover Mechanism, Level 360	3-59
0-14	Monitor Power for Service Arms	3-61
2-45	Monitor Power for Service Arms Patch Distributor	3-62
2 - 4 3	Monitor Power for Launcher Networks and Patch Distributors No. 6356 and No. 6651	3-63

₹r}1 [

T

44

100

1 - 9 (1) (1)

2498

1. A. 1. A. 1.

1.194.1

Number	Title	Page
3-47	Control Power for ESE Patch Distributors.	. 3-64
3-48	Control Power for Astrionics Signal Conditioning Equipment	. 3-65
3-49	Power for Hardwire and DDAS Transmission	
3-50	Power Supply Rack (Typical)	3-66
3-51	Power Distribution Rack (Typical)	3-67
3-52	Battery Rack (Typical)	3-68
3-53	Relay Rack (Typical)	
3-54	Arming Panel Rack (Typical)	3-70
3-55	LCC Power Panel (Astrionics)	3-71
3-56	S-IC Engine Servicing Platforms Subsystem	3-73
3-57	F-1 Engine Change	3-75
3-58	F-1 Engine Skirt Installation	3-75
3-59	Relay Distributor, S-IC Engine Servicing Platforms Subsystem	3-76
3-60	Electrical Equipment, S-IC Engine Servicing Platforms Subsystem	3-77
3-61	S-IC Engine Servicing Platforms Controller E75M13570	3-78
3-62	Typical Installation, Platform Controller	3-79
3-63	Servicing Platform Transporter Subsystem	3-81
3-64	Motor/Winch Assembly, Platforms Transporter Subsystem	3-83
3-65	S-II & S-IVB Engine Servicing Platform Subsystem	3-85
3-66	Motor/Winch Assembly B75M14676	3-87
3-67	Launch Equipment Control Area, LCC	3-89
4-1	Standard Assemblies for ESE	4-1
4-2	Standard Electrical Distributor J75M07835	
4-3	Standard Electrical Distributor J75M09808	
4-4	Portable DC Power Supply D75M51116	

ix

 $e_{1}^{-\frac{N}{2}-1}e_{1}^{-1}$

A. 1. 1. 1.

1, 11 ke

1975

意思

Ì

1.45 E.C.

a tig

Number	Title	Page
4-5	Standard 24 Inch Vertical Rack E64-KN-F-260-10	4-9
4-6	Standard Patch Distributor F75M04681 (42 Connector)	4-11
4-7	Standard Patch Distributor F75M04681 (54 Connector)	4-13
4-8	Standard Patch Distributor D75M09313 (27 Connector)	4-15
4-9	Standard Relay Module D75M50348	4-17
4-10	Standard Diode Module D75M50364	4-19
4-11	Standard Meter Panel J75M10165	4-21
4-12	Standard Fuse Panel J75M10163	4-23
4-13	Standard Module Frame J75M10432	4-25
4-14	Standard 500-Amp Power Supply A40M03709	4-27
4-15	Standard Test Load E40M03971	4-29
-16	Standard Load Control Panel E40M03535	4-31
4-17	Standard Battery Charger	4-33
4-18	Standard Battery Panel J40M03605	4-35
4-19	Standard Power Module, Type IIA J75M12295 and Standard Frame Assembly E75M12301	4-37
4-20	Receptacle Distributor E75M12316	4-39
4-21	Standard Bus Distributor E75M12323	4-41
4-22	Standard Console E64-KN-F-261-4	4-43
4-23	Integration and Cable Entrance Racks	4-45
4-24	Standard Latching Relay Module D75M05895	4-47/48
5-1	Ground Equipment Test Set E75M13286	5-3
5-2	GETS Auxiliary Equipment	5-4
5-3	GETS Control, Monitor and Simulation	5-5
5-4	Left Bay Front Panel, Launcher GETS	5-6
5-5	Right Bay Front Panel, Launcher GETS	5-7
5-6	Miniaturized Universal Test Set E75M13761	5-9

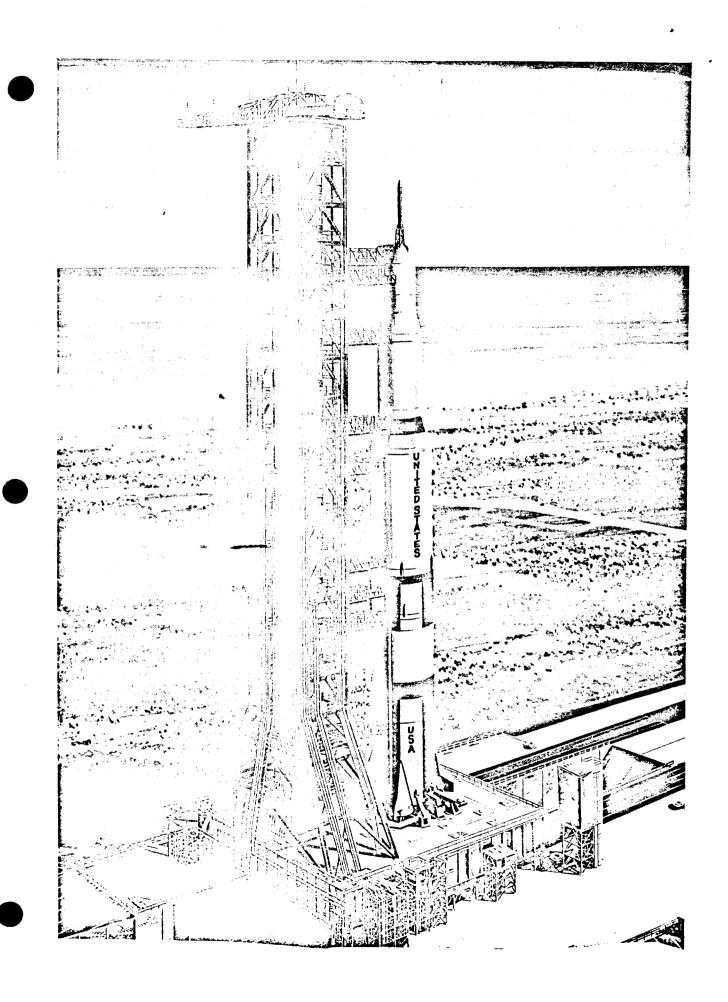
Number	Title	Page
5-7	Portable Arm Control Console J75M07542-1	5-11
5-8	Universal Patchboard Test Set E75M09032	5-13
5-9	MSFC Service Arms/Tail Service Masts Test Site	5-15
5-10	Typical Control Panel, Service Arms Test Site	5-17
5-11	ESE Equipment, Service Arms Test Site	5-18
5-12	Control and Distribution Racks, Service Arms Test Site	5-19
5-13	Tail Service Masts Test Site, MSFC	5-21
5-14	Tail Service Masts Test Site Control Panel	5-23/24
* 6-1	ESE Family Tree	6-1
6-2	ESE Equipment Locations, Launcher	6-163
6-3	Equipment Location Mobile Launcher, LC-39	6-164
6-4	Layout, Electrical Equipment and Cabling Arrangement, Mobile Launcher, LC-39	6-165/166
6-5	Mobile Launcher ESE Details	6-167/168
6-6	ESE Cable Interconnect Diagrams	6-170
6-7	Reference Designation Assignment Areas, LC-39	6-235/236

LIST OF TABLES

Number	Title	Page
6-1	Reference Specifications	6-147
6-2	Documents Discontinued from Launch Equipment Branch, Electrical Section, Use	6-159
6-3	Abbreviations, Cable Specifications	6-160
6-4	Reference Designation Numbers Versus Assy Drawing Numbers	6-220

* contained in Volume II

I



SECTION I

INTRODUCTION

1–1. SCOPE AND PURPOSE

This handbook describes that Electrical Support Equipment Equipment (ESE) which is the responsibility of the Launch Equipment Branch. The Launch Equipment Branch ESE comprises electrical subsystems which provide control, test, and monitoring of the service arms, command module access arm, tail service masts, pneumatic distribution panels, hydraulic charging and supply units, engine servicing platforms, Q-ball cover removal mechanism, and oxygen conditioning system. As the design of these systems is further refined, or as new systems are added to Launch Equipment Branch responsibility, this document will be revised and re-issued.

It should be noted that equipment also known as ESE is used in propellant loading, in certain vehicle-oriented systems on the Mobile Launcher, and in the integrated electrical networks for LC-39; no ESE except that named above is within the scope of this handbook.

In this document the ESE is described as an overall integrated system (section III), and again as individual panels and cabinets (section IV). It is also possible to determine the physical location of a particular item of Electrical Support Equipment installed in the Mobile Launcher or Launch Control Center. Detailed information about the equipments for which the ESE provides monitor and control, such as the tail service masts, may be obtained by reference to the document list, paragraph 1-2.

Section IV describes standard equipment such as panels, distributors, racks, test sets, and component modules employed throughout the ESE. These standard assemblies are the building blocks of present electrical ground networks and they should aid in the design of new systems.

. 1-1

Section V provides physical descriptions of the four ESE system test units. Their \cdot face panels and test functions are also given in section III in conjunction with the systems

be tested.

Reference material which applies to the entire ESE, such as the Mobile Launcher isometric drawing, the family tree of drawings, etc. is contained in section VI.

1-2. RELATED DOCUMENTS.

The following documents offer detailed information on systems controlled and monitored by the ESE:

- a. Apollo Systems Description, Volume III NASA TM X-882
- b. Tail Service Mast Design Report TM-23-0-D
 Launch Equipment Branch
 Launch Support Equipment Engineering Division
- c. Saturn V Service Arms Preliminary Engineering Report TR-4-4-2-D Launch Equipment Branch Launch Support Equipment Engineering Division
- d. LC-39 L/UT GSE Installation
 SP-95-D
 Launch Equipment Branch
 Launch Support Equipment Engineering Division
- e. Saturn V Service Arms Functional Description Launch Equipment Branch Launch Support Equipment Engineering Division
- f. Standard Cables, Saturn I LTM-4-4
- g. Standard Electrical Components, GSE Saturn Vehicles LTM-4-11
- h. Data Booklet, Saturn V Service Arms and Related Equipment, Complex 39
 KSC SP-4-50-D
 Launch Equipment Branch
 Launch Support Equipment Engineering Division
- Information Handbook, Saturn V Vehicle Service Arms and Access Arm, Launch Complex 39 SP-87-D
 Launch Equipment Branch
 Launch Support Equipment Engineering Division

SECTION II

LAUNCH COMPLEX 39

2-1. GENERAL DESCRIPTION.

· 4.15

The Saturn/Apollo space vehicle (Saturn V) will be launched from NASA's Launch Complex 39. Owing to the complexity and size of the space vehicle and to the need for maximum reliability in the Apollo program, the launch procedure for LC-39 departs from previous ones as follows:

a. Assembly and checkout of the space vehicle on its launcher, in the Vehicle Assembly Building (VAB) as shown in figure 2-1.

b. Transfer of the assembled and checked space vehicle on its launcher to the launch pad, figures 2-2 and 2-3.

c. Automatic checkout.

d. Remote control of actual launch operations from a distant Launch Control Center (LCC).

As the Saturn V is assembled on its launcher in the VAB, all vehicle and launch support systems are remotely exercised and checked out from consoles in the Launch Control Center, which adjoins the VAB. After transfer to a launch pad the Mobile Launcher and Saturn V are again checked out from these consoles. Finally, all events at launch are sequenced by computers in the LCC and Mobile Launcher and are monitored from the LCC consoles.

One group of equipment thus controlled from the LCC is the Launch Equipment (see figure 3-6), which is utilized to mate the vehicle to the Mobile Launcher during installation, checkout, and launch operations. Among the Launch Equipment are the service arms, tail service masts, and their associated hydraulic, pneumatic, and electrical units. These systems are located aboard various levels of the Mobile Launcher tower and base.

Electrical control for them is provided by the Electrical Support Equipment (ESE), figures 2-4 and 2-5, which is the central topic of this handbook.

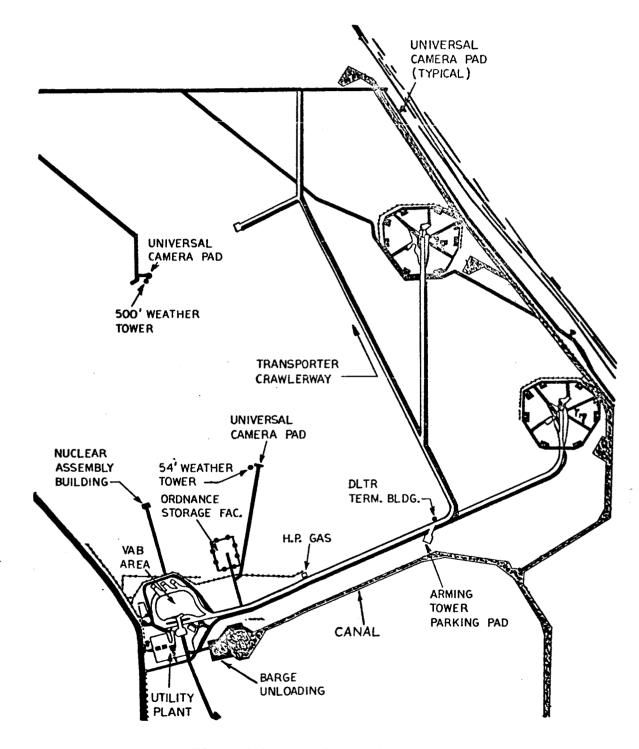


Figure 2-1 Launch Complex 39

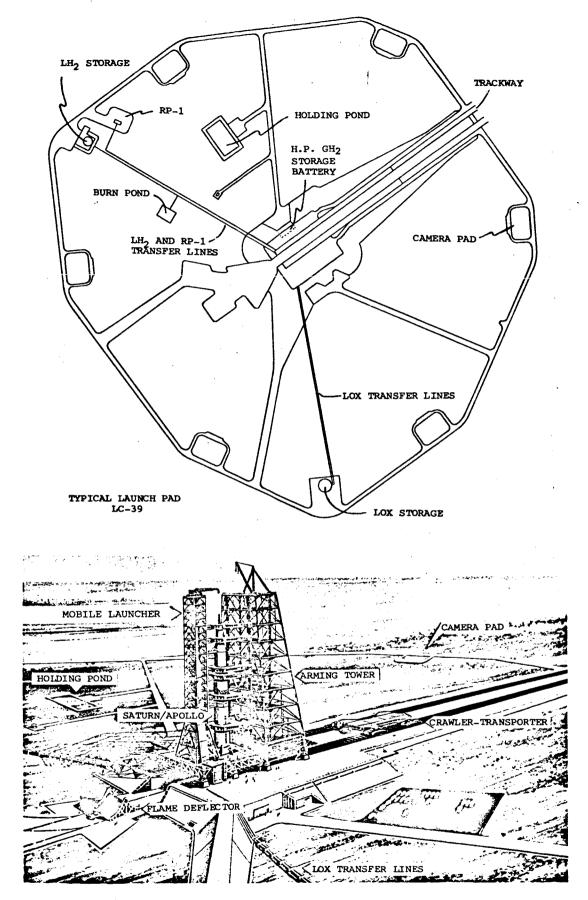


Figure 2-2 Typical Launch Pad, LC-39

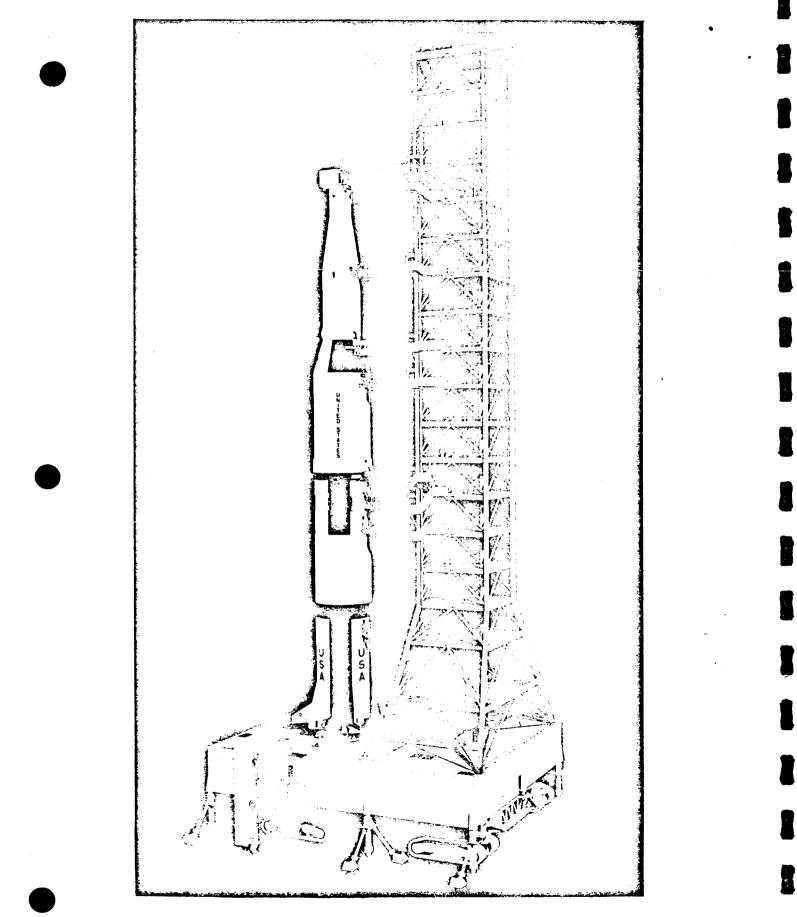


Figure 2-3 Mobile Launcher

2.Marc

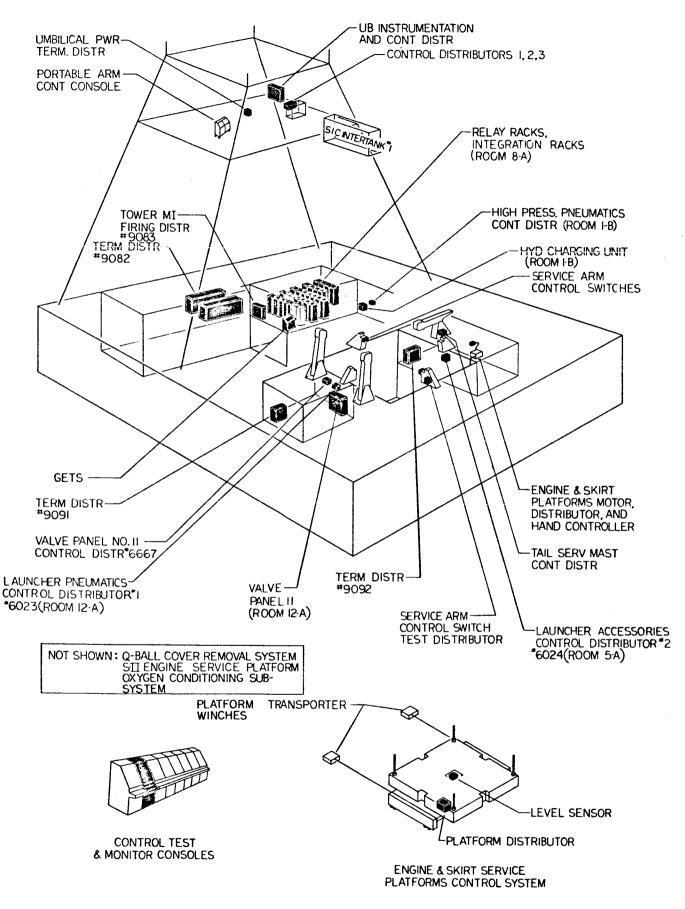
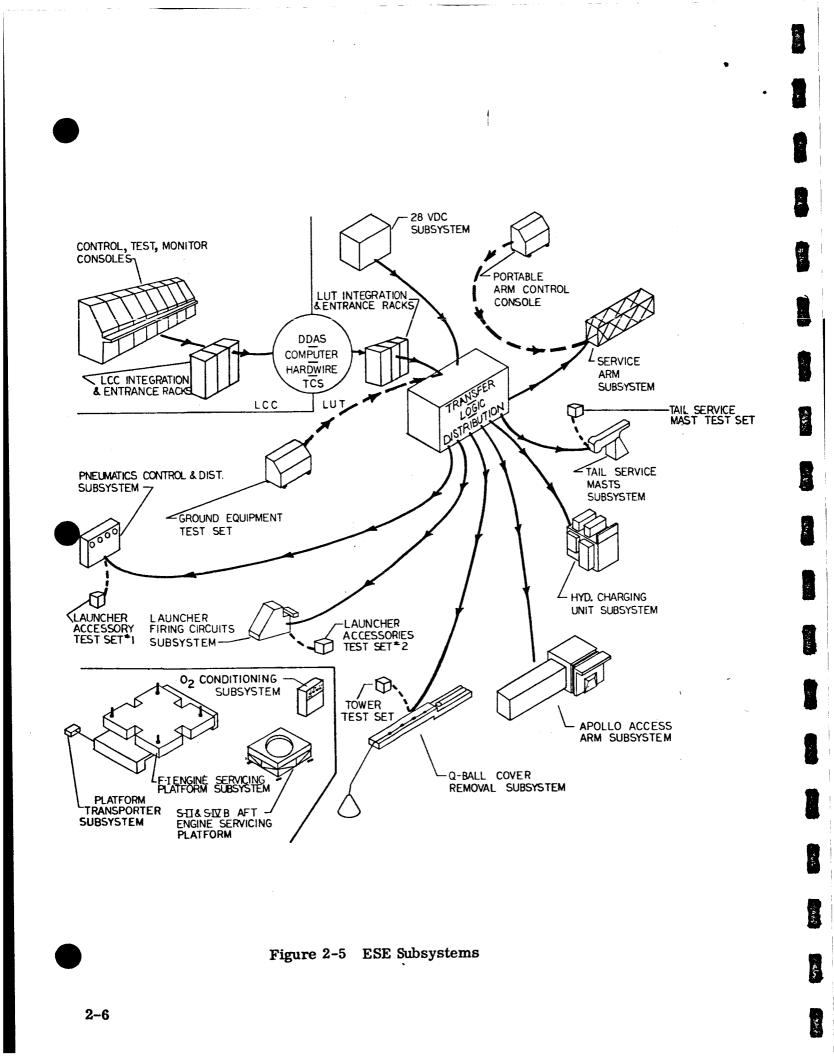


Figure 2-4 Electrical Support Equipment



SECTION III

ELECTRICAL SUPPORT EQUIPMENT (ESE)

3-1. INTRODUCTION.

The tail service masts and tower service arms are moved by pneumatic and hydraulic actuators. Local hydraulic and pneumatic accumulators and lines for these actuators are charged for check-out operations in the VAB, and are normally reduced to standby pressure only during Mobile Launcher transit to the pad.

For prelaunch operations these actuators and their electrical controls are remotely checked, operated and monitored from ESE consoles in the Launch Control Center, using 28 volts dc primary power. During countdown and launch the service arms, the tail service masts, and their hydraulic/pneumatic actuators are controlled by the ESE consoles in the LCC, by the Mobile Launcher and LCC computer complex, and partly by ESE sequencing circuits in the Mobile Launcher.

The twelve subsystems comprising the Mobile Launcher ESE are shown in figure 2-5 and are individually delineated in paragraphs 3-2 through 3-58. Among these, only four subsystems are independent of the LCC. They are the S-IC Engine Servicing Platform, the Platform Transporter, the S-II/S-IVB Engine Servicing Platform, and the Oxygen Conditioning System.

The remaining nine electrical subsystems are primarily controlled from the Launch Control Center via the DDAS-Computer-Hardwire-Terminal Countdown Sequencer (TCS) complex and the Transfer-Logic-Distribution Complex as in figure 2-5. The latter complex is comprised of relay racks which provide both interface separation and logic, and of distributors which provide a means for power and signal distribution.

The relay-rack portion of the Transfer-Logic-Distribution Complex also provides a central location for connecting the Launcher Ground Equipment Test

Sets (Launcher (GETS), which simulate either the LCC or any subsystem interfacing with the relay racks. The GETS (paragraph 5-2) can also test the relay racks themselves.

Another test unit shown in figure 2-5 is the Portable Arm Control Console. This unit allows an operator to check out or operate any service arm at the service arm level. See paragraph 5-12. In addition, test sets are provided to locally test launcher accessories and tail service masts.

In the following table some of the major LC-39 launch activities and events are given in chronological order. Those which make direct use of the Electrical Support Equipment are marked by an asterisk (*).

VEHIC LE ASSEMBLY BUILDING

L-58 to L-55 days	Offload S-IC and erect on mobile launcher.				
L-55 to L-49 days	* Install S-IC engine servicing platform;				
	connect ESE to LCC via data link.				
L-53 to L-49 days	* Install S-II, S-IVB, and IU; make umbilical				
	connections.				
L-48 to L-29 days	Check telemetry, navigation, power transfer,				
	launch vehicle systems, etc.				
L-28 to L-17 days	Install spacecraft and test spacecraft systems;				
	install light ordnance items.				
L-16 to L-15 days	* Perform interface and compatibility tests,				
	including vehicle-ESE; perform simulated				
	flight test, check LCC-pad data link with				
	simulators.				
	LAUNCH PAD				
L-12 to L-10 days	* Crawler transports vehicle and mobile				
	launcher to pad; mate launcher to ground				
	connections; power on (all systems "quick				
	look" to the LCC); position engine servicing				
	platform.				

L-10 to L-4 days	* Position arming tower; perform RF tests,			
	propellant loading tests, and limited			
	electrical subsystem verification tests			
	from LCC.			
L-3 days	Perform simulated flight test.			
L-2 to L-1 days	* Remove work platforms, test equipment,			
	and arming tower; retract Q-ball cover.			
L-0	Terminal countdown.			
T-7 hours	Begin LOX and LII ₂ loading.			
T-4 hours	Astronauts ingress via Mobile Launcher.			
T-105 min.	Conduct final systems tests.			
T-187 sec	* Launch support equipment preparation			
	complete and terminal countdown			
	sequencer start.			
T-60 sec	* Unlock & retract Apollo access arm.			
T-30 sec	* Apollo access arm retracted and latched;			
	unlock preflight service arms.			
T-22 sec	* Retract S-IC intertank reconnect mechanism.			
T-17 sec	* Retract S-IC Intertank & S-IC Forward service			
	arms; unlock inflight service arms.			
T-10 sec	* Pad water system on; tail service mast			
	purge on			
T-9 sec	* Launch support equipment ready for ignition			
T-7 sec	* Ignition command			
T-0 sec	* Launch commit - arm the lift off switches,			
	release holddown arms.			
T+0 sec	* Lift off, retract inflight service arms and			
	tail service masts			

.

٠

3-2. SERVICE ARMS SUBSYSTEM.

3-3. <u>Purpose of the Service Arms Subsystem</u>. The Service Arms Subsystem provides system control, test, and monitoring for service arm operation.

3-4. Equipment for the Service Arms Subsystem. The Service Arms Subsystem employs the following equipment:

- a. Service Arms Nos. 1 through 8, figure 3-1
- b. Service Arm Control Distributors (Control Console No.1)

#6062, #6064, Level 60 #6122, #6124, Level 120 #6142, #6144, Level 140 #6162, #6163, #6164, Level 160 #6202, #6203, #6204, Level 200 #6222, #6223, #6224, Level 220 #6262, #6263, #6264, #6265, Level 260 #6302, #6303, #6304, Level 300

c. "J" Boxes (Control Console No.2)

#6146, Level 140
#6206, Level 200
#6226, Level 220
#6266, Level 260
#6306, Level 300

d. Service Arm Racks, room 8-A. See Racks 9, 10, 11, 12, 13, 14, 15, 16, 17, figure 6-2.

e. Portable Arm Control Console.

- f. Control, Test, and Monitor Console, LCC.
- g. Integration Racks and Entrance Rack, room 8-A. See Racks 20, 21, and 22, figure 6-2.
- h. Terminal Distributor #9082, room 7-A.
- i. Power Racks, room 8-A. See Racks 5, 6, 7, 8, figure 6-2.
- j. UB Instrumentation & Control Distributors #9006, #9010, #9014, #9016, #9020, #9022, #9025, #9026, #9028 and #9030, figure 3-1.
- k. UB Power Distributors #9017, #9023, #9027, #9029, #9031, #9043, #9044, #9045, #9046.
 For locations see figure 3-44.
- Local Control Units #6067, #6127, #6147, #6170, #6207, #6227, #6267, #6307

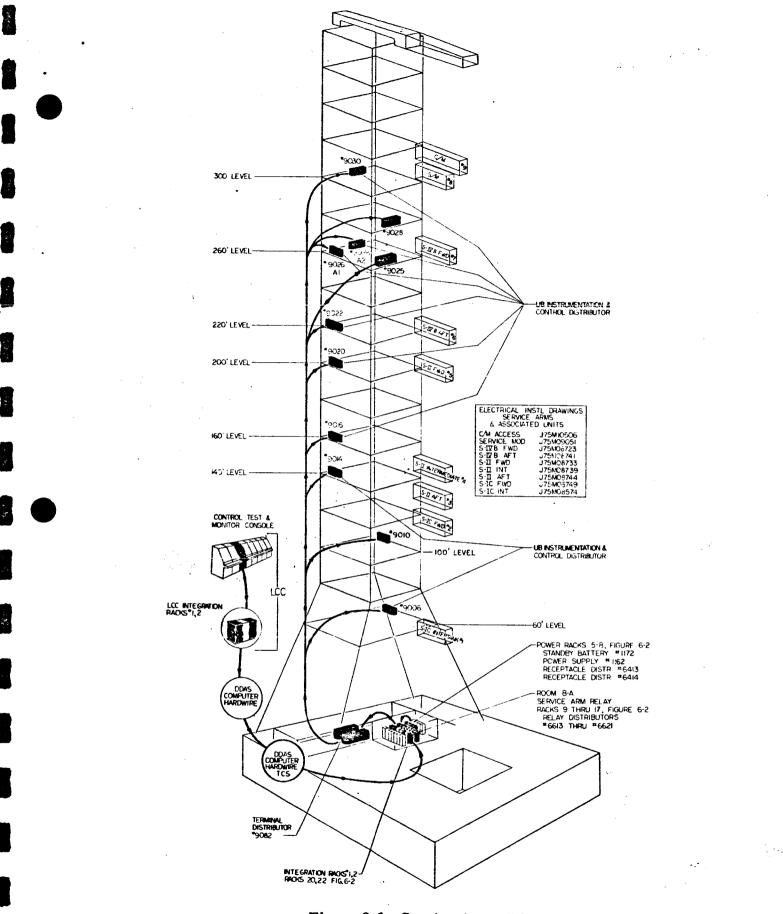


Figure 3-1 Service Arms Subsystem

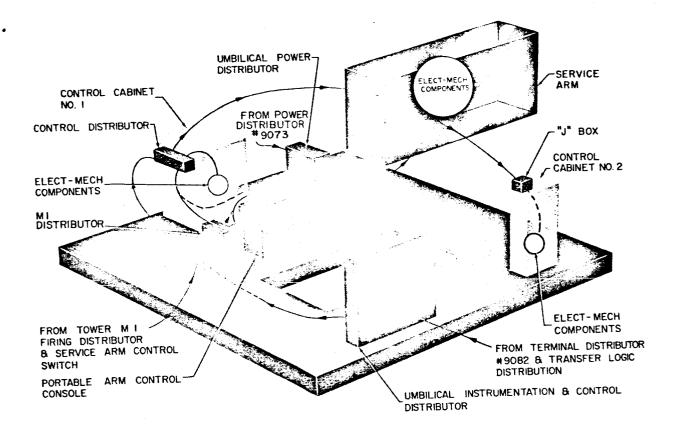
.

3-5. Description of the Service Arms Subsystem. Control and display functions for each of the eight service arms are provided by the eight LCC Control, Test, and Monitor consoles (figures 3-4, typical) and the LCC Status Panel (figure 3-5). All ESE consoles are shown in figure 3-67. These consoles provide visual display for critical system parameters and electrical control of solenoid-operated valves and pressure switches associated with the service arms. The LCC consoles interface with the seven Relay Distributors (#6613 through #6618 and #6621) via the DDAS-Computer-Hardwire complex, as in figure 3-1. These relay racks provide control logic and act as transfer units for the Service Arm Subsystem as follows.

Manual Operation of switches on the LCC Console provides command signals which cause relays in the relay racks to be actuated. Relay operation in turn provides 28 volts dc via the terminal and control distributors, figure 3-2, to solenoid-operated valves, limit and pressure switches, and analog transducers associated with each tower level. These components are shown for a typical preflight arm in figure 3-3 and a typical inflight arm in figure 3-2. Resulting mechanical movements of the arm and its control mechanisms are monitored by switches and transducers which return feedback signals to the relay racks along a path parallel to the command signal. In the relay racks these feedback signals actuate additional relays which provide visual display at the LCC Control, Test, and Monitor Console. At the same time, some of these feedback signals provide a stimulus for automatic sequenced operation of the service arm.

NOTE

During launch conditions five of the eight service arms are retracted at vehicle liftoff. Initiating commands for retracting these five arms are provided by the Launch Equipment Firing Circuits Subsystem, paragraph 3-20.





29.0.1

a,

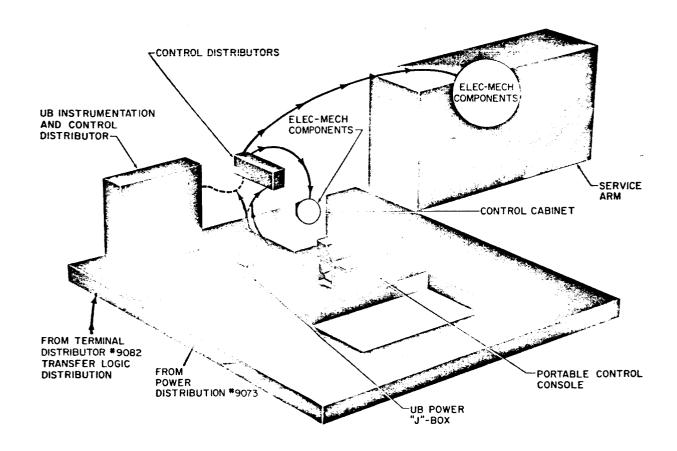
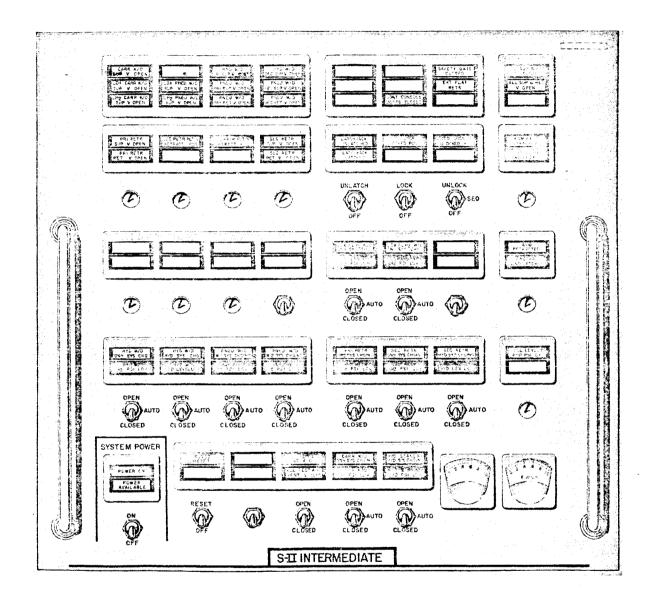


Figure 3-3 Typical Preflight Service Arm and Test Installation



a traph (B).

Naper-S

Sales in

Store State

- ANKE

ALC: N

- ANSAL

Safety -

STATE.

Figure 3-4 LCC Control Panel, S-II Intermediate

•	· .	• • • • • • • • • • • • • • • • • • • •		· · · · · · · · · · · · · · · · · · ·				
S-II ATTATA	S-20 ++0 5	S-D AT		3 <u>5</u> 7 40		-118 m		
				A ANY DAL ANY DAL ANY DAL ANY DAL ANY DAL		A A Mar Control	ANN LEVEL NEW COLOR	
			T59.1-2		4 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1	Contraction of the second		
							* PS 06	
							745 F(25) 784 (25) 784 (25)	
COUPLIER			· ·					
NUMER ON			STATUS					
		1	STATUS	PANEL				

. 24 M

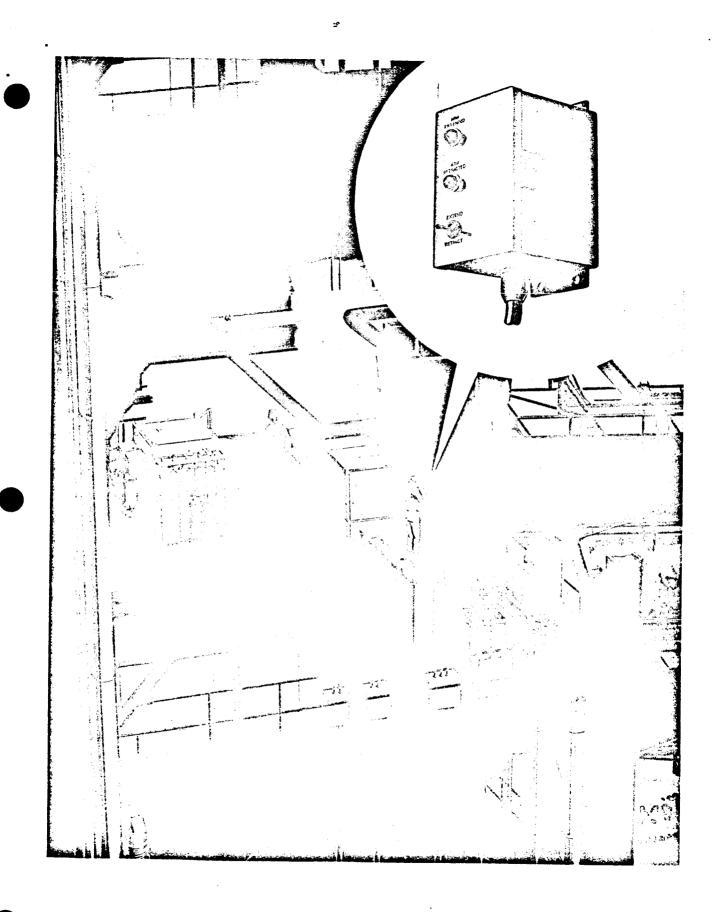
and a second

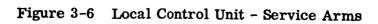
Figure 3-5 LCC Summation Panel D75M14440

3-6. Local Tests for the Service Arms Subsystem. In order to test and operate a service arm from the tower level with which it is associated the Portable Arm control Console (figures 3-2 and 3-3) is employed. After this console is positioned on the tower level, special test cables are connected to appropriate distributors and electrical-mechanical components, in place of the system cables.

Once the Portable Arm Control Console is connected, the electrically controlled hydraulic and pneumatic systems can be charged and the condition of all pressure systems monitored. Upon completion of the charging sequence, the service arm can be operated as a complete system, or any of its three major sequential functions can be exercised individually. Provisions are made in the Portable Arm Control Console to allow the operation of each critical component to be monitored. See figures 3-7, 3-8 and 5-7.

The service and access arms can also be extended or retracted by use of the local control unit shown in figure 3-6. One such unit is stored in each Control Cabinet No.1.





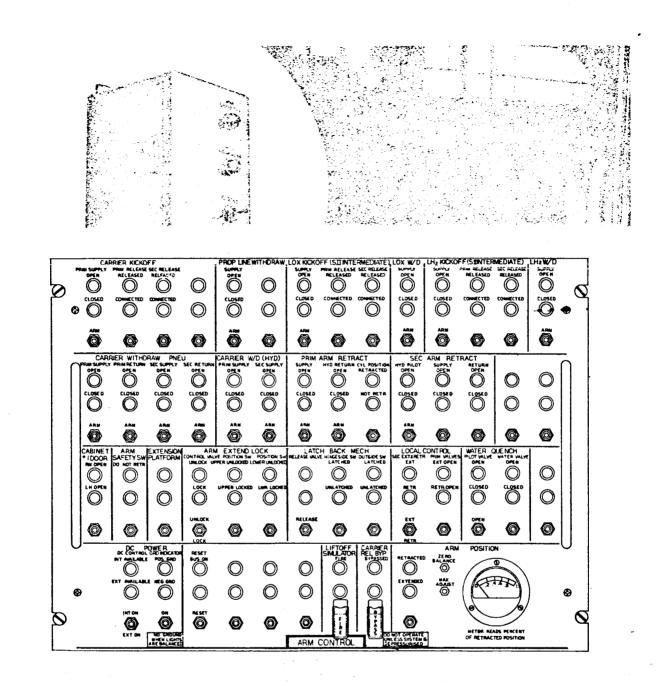


Figure 3-7 Arm Control Panel J75M07546-3, Portable Arm Console

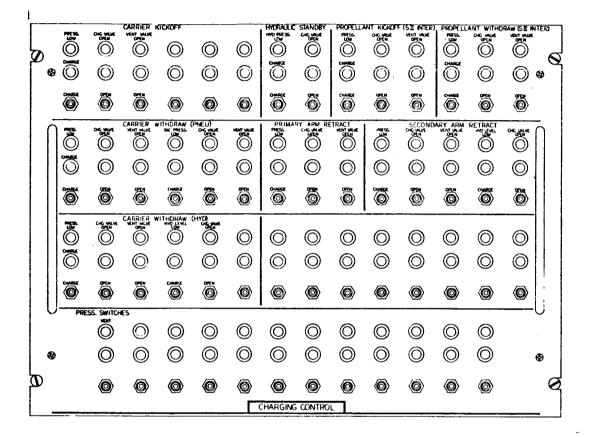


Figure 3-8

Charging Control Panel J75M07546-1, Portable Arm Console

3-7. APOLLO ACCESS ARM SUBSYSTEM.

3-8. <u>Purpose of the Subsystem</u>. The Apollo Access Arm Subsystem provides electrical control, test, and monitoring of the access arm and the hood positioning mechanism.

3-9. Equipment for the Subsystem. The Apollo Access Arm Subsystem employs the following equipment:

- a. Access Arm and associated electrical-mechanical components
- b. Hood positioning mechanism and associated electrical-mechanical components
- c. Apollo Access Arm Control Panel, LCC
- d. Relay Distributor #6619, Mobile Launcher room 8-A
- e. Power Rack, Mobile Launcher room 8-A
- f. Terminal Distributor #9082
- g. UB Instrumentation and Control Distributor #9032
- h. UB Power "J" Box #9047
- i. Control Cabinet No. 1 #6328
- j. Control Distributors #6322 and #6324
- k. Control Distributor #6357A55A2
- 1. Portable Arm Control Console
- m. Local Control Unit #6339

3-10. <u>Description of the Subsystem.</u> The Apollo access arm provides personnel access, environmental controls, and checkout service outlets for the Apollo Command Module. During launch conditions the arm is controlled from the Apollo Access Arm Control panel in the LCC, figure 3-12. When the arm control system is manually given READY TO FIRE status from the panel, the access arm disconnects itself from the vehicle and retracts approximately one minute before lift off, upon command from the Terminal Countdown Sequencer.

The command signals flow through the DDAS-Computer-Hardwire-TCS complex (figure 3-9) to Relay Distributor #6619 in the Mobile Launcher room 8-A. Relay closure provides 28 volt signals through Terminal Distributor #9082 to UB Instrumentation and Control Distributor #9032, at Mobile Launcher Level 320. The

1.00

No.

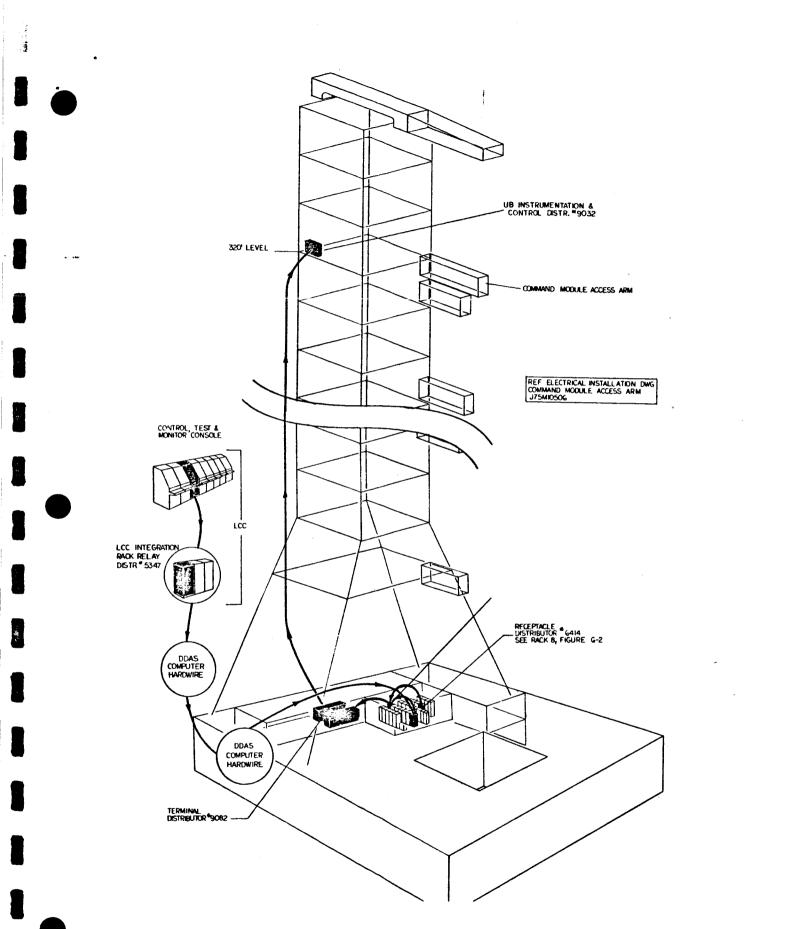


Figure 3-9 Apollo Access Arm Subsystem

control system then branches off into two operations, which are treated separately here.

The first operation, access hood positioning control, is illustrated in figure 3-10. At launch, command signals flow from the UB Instrumentation and Control Distributor #9032, through Control Distributor #6357A55A2, to various electrical-mechanical components in the vicinity of the access hood. These components control a pneumatic system which releases latching hooks, raises the hood, and retracts it several inches away from the Command Module.

dates.

H

1.1

Tradition of

(WAR AND A

The second operation, access arm control, is thereby automatically initiated. Signal flow as shown in figure 3-11 actuates electrical-mechanical components in the arm and in Control Cabinet No. 1. These components control the pneumatic and hydaulic system which swings the arm back and latches it to the tower.

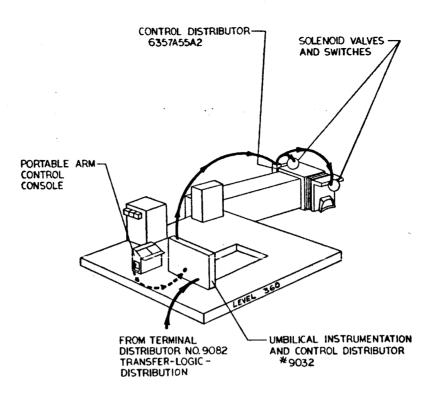
During launch conditions or prelaunch testing, various specific mechanical movements of the access arm and hood positioning mechanism can be controlled and monitored from the LCC.

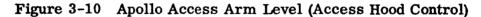
Thus at any time the arm can be extended and the access hood repositioned and latched to the space vehicle. See these functions in figure 3-12.

All major movements of the arm and its control elements are monitored by limit switches and analog transducers. Feedback signals from some of these components provide continuity for automatic operation. Others transmit monitor signals back to the LCC control panel.

3-11. Local Testing of the Access Arm. By use of the Portable Arm Control Console, the Apollo access arm can be locally tested and exercised. For this purpose the control console is outfitted with special overlay panels (figures 3-13 and 3-14) and is internally patchboard-programmed. It is then positioned on tower level 320 and connected to appropriate distributors and electricalmechanical components, replacing system cables.

Thus the Portable Arm Control Console can charge all necessary





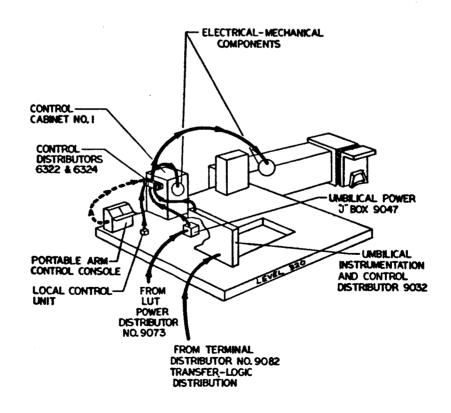


Figure 3-11 Apollo Access Arm Level (Arm Control)

by draulic and pneumatic accumulators, and can fully exercise the access arm and hood positioning mechanism.

The arm itself can also be extended or retracted by Local Control Unit #6339, which is stored in Control Cabinet No. 1. A typical Local Control Unit is shown in figure 3-6.

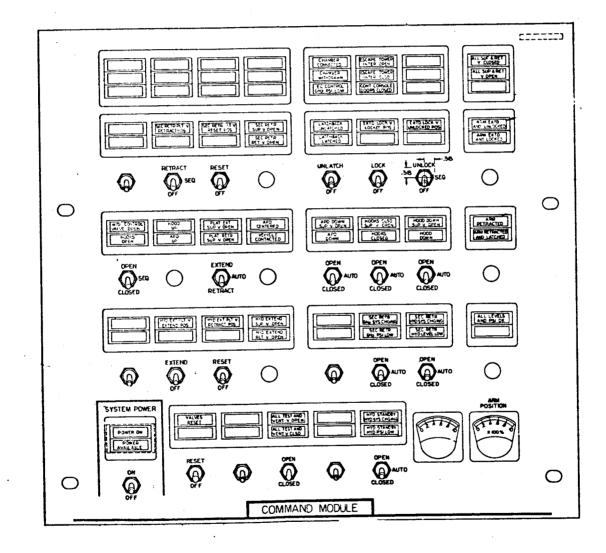


Figure 3-12 LCC Control Panel, Apollo Access Arm

-

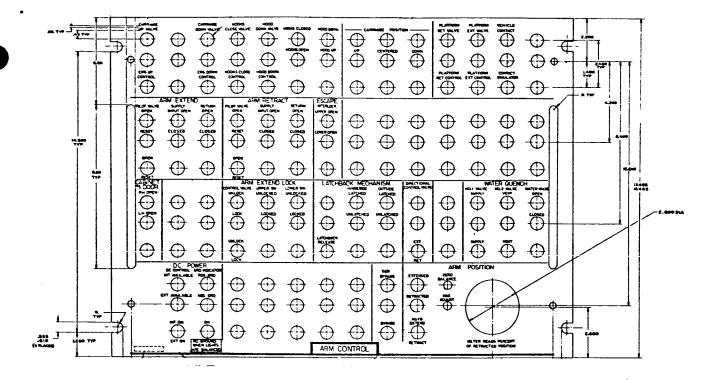


Figure 3-13 C/M Access Arm Overlay for Arm Control Panel, Portable Arm Control Console

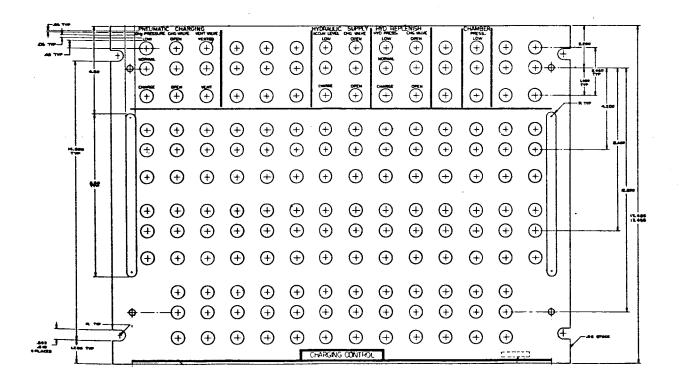


Figure 3-14 C/M Access Arm Overlay for Charging Control Panel, Portable Arm Control Console

3-12. OXYGEN CONDITIONING SUBSYSTEM.

P13. <u>Purpose of the Subsystem</u>. The Oxygen Conditioning Subsystem provides pressurized gases to the GOX Module, figure 3-15, as follows:

a. Gaseous Oxygen (GOX) is supplied at 125 psi and 45^o F, and is tapped for leak-testing the astronauts' suits, for cooling the interior of the Command Module, and for human consumption during pre-launch operations.

b. Gaseous helium and gaseous nitrogen are supplied at 125 psi for leaktesting the Command Module.

These pneumatic services are required only during testing and prelaunch checkout, and the Oxygen Conditioning Subsystem is shut down by tower technicians prior to launch.

3-14. <u>Equipment for the Subsystem</u>. The Oxygen Conditioning Subsystem employs the following equipment:

- a. Oxygen Conditioning Console #6326
- b. GOX Module #57A55A11

c. Pneumatic lines and electrical cabling between these two assemblies.

3-15. <u>Description of the Subsystem</u>. The Oxygen Conditioning Subsystem is a locally controlled electrical-pneumatic system. It prepares oxygen at the Oxygen Conditioning Console #6326 (figure 3-16) and delivers it under controlled pressure and temperature to the GOX Module (figure 3-17).

The Oxygen Conditioning Console #6326 receives gaseous oxygen from storage bottles on level 300. It heats or cools the GOX as required (see controls, figure 3-16), and delivers it through a water-glycol jacketed line to the GOX Module. Electrical-mechanical components required to condition the oxygen include switches, solenoid valves, pressure switches, a pump, a cooler, and a heater. Switching and logic are provided by a relay distributor on the console. The Oxygen Conditioning Console has two separate banks of GOX controls, either of which can automatically switch to the other in case of malfunction.

Pneumatic values and gauges are also provided on the console for supplying gaseous nitrogen and helium to the GOX Module. The GOX Module contains manual values, pressure gages, and indicating lamps as shown in figure 3-17. The entire Oxygen Conditioning Subsystem can be turned on from this Module.

14 V.

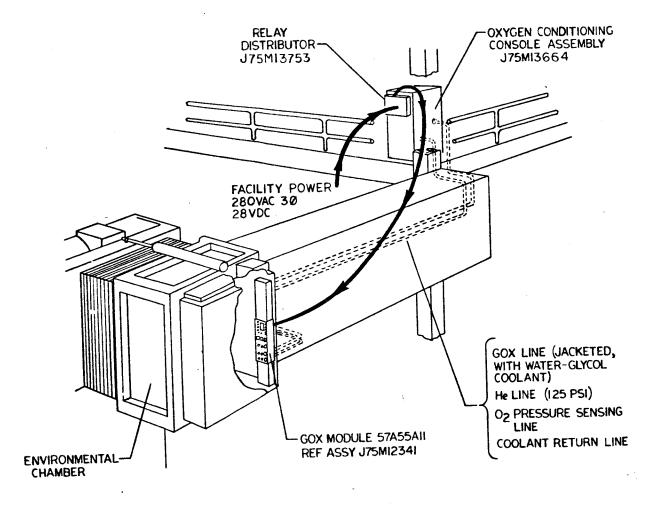
191-

- "W. -

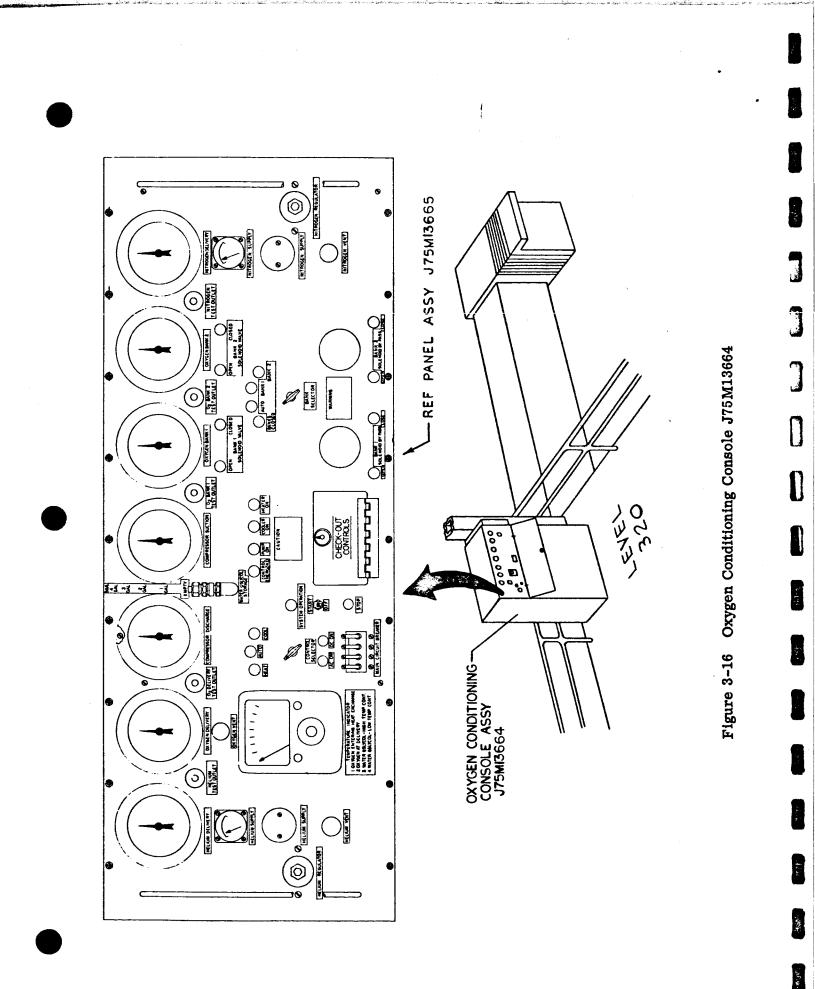
4.84 B

No.

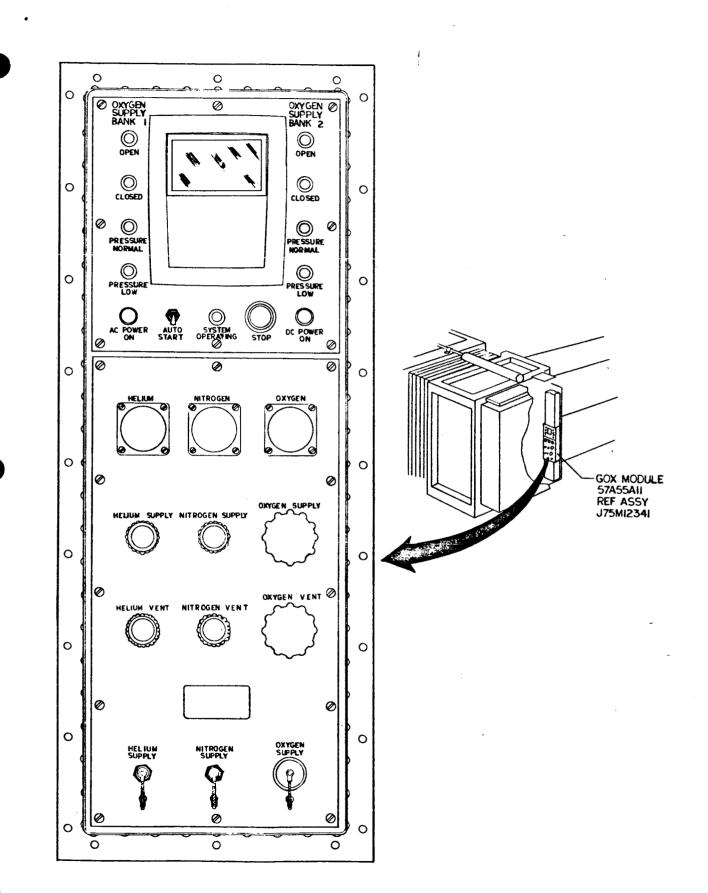
i a







Sec.



1884.50

1998 C

- Hilling

.) **h**hit

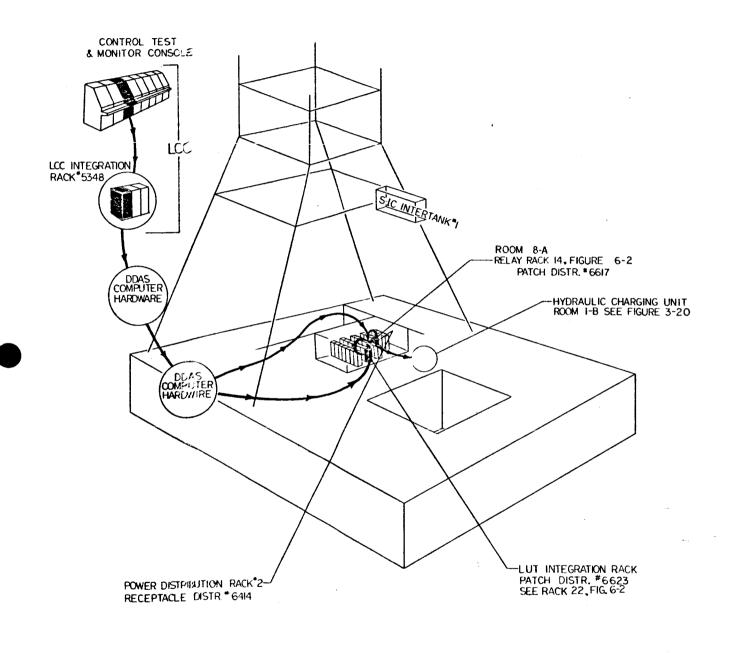
Figure 3-17 GOX Module J75M12341

3-16. HYDRAULIC CHARGING UNIT SUBSYSTEM.

•17. <u>Purpose of the Hydraulic Charging Unit Subsystem</u>. The hydraulic charging unit furnishes high pressure hydraulic fluid to initially charge and periodically replenish the service and access arm control system hydraulic accumulators. The control system hydraulic accumulators supply the high pressure hydraulic fluid to the service and access arm hydraulic cylinders to allow umbilical carrier withdrawal or extension, and service/access arm retraction or extension. Periodic replenishment of the hydraulic accumulators occurs during in-transit and pre-launch operations to offset hydraulic fluid demands of the hydraulic mechanisms in tracking vehicle motion caused by wind loading, fuel operations, thermal bending or operation of the mobile launcher enroute from the VAB to the launch site.

3-18. Equipment for the Subsystem. The hydraulic charging unit subsystem is composed of the following equipment:

- a. Motor Pump Units, #6426 and #6419. See figure 3-20.
- b. Motor Starter Units, #6424 and #6417. See figure 3-20.
- c. Control Distributors, #6423 and #6416. See figure 3-20.
- d. Local Control Panels, #6425 and #6418. See figure 3-20.
- e. Main Power Switches, #6415 and #6420. See figure 3-20.
- f. Pressure Transducers, #6423A2 and #6416A2.
- g. Pressure Switches, #6423A1 and #6416A1.
- h. Relay Rack, Room 8A, Mobile Launcher. See Rack 14, figure 6-2.
- i. LCC Control and Monitor Panel, #5419. See figure 3-19.



.

Figure 3-18 Hydraulic Charging Unit Subsystem

3-19. <u>Description of the Subsystem</u>. The hydraulic charging unit consists of a 500-gallon hydraulic fluid reservoir; two identical high pressure, low volume, parallel operated hydraulic pumping units; and separate, identical, electrical control systems.

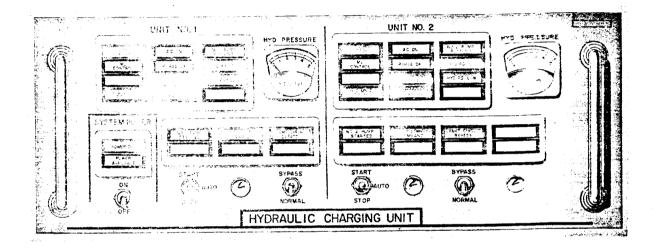
The control systems for both motor-pump units employ separate motor starters and control distributors which may be controlled locally by two independent control panels (#6425 and #6418) or at the launch site by the LCC control panel (#5419). Both motor-pump units serve the same hydraulic reservoir.

During transit of the Mobile Launcher from the Vehicle Assembly Building (VAB) to the launch site, local control panels are utilized to control motor-pump operation. In order to reduce transients in the Crawler-Transporter power generator, internal circuitry prohibits simultaneous operation of both pumping units during intransit operations of the Mobile Launcher.

Operation of the hydraulic charging unit at the launch site may be either through the use of the local control panels previously described or through use of the LCC control panel. Normal operation at the launch site employs the LCC panel.

The Control, Test, and Monitor Console housing the LCC control panel is electrically connected to the hydraulic charging unit control distributors by the Computer-Hardwire complex and Relay Distributor #6617. Command functions are electrically connected by both hardwire and the computer; discrete monitor functions are electrically connected only through the computer. See figures 3-18 and 3-20.

The electrical control system of the hydraulic charging unit is designed to sense a loss of electrical power to the motor-pump due to internal or external causes, or a loss of hydraulic fluid pressure to the service/access arm hydraulic accumulators. Either occurrence will automatically place the alternate motorpump on the line without manual supervision.



 $r \gg Z^{+},$

Ę

4

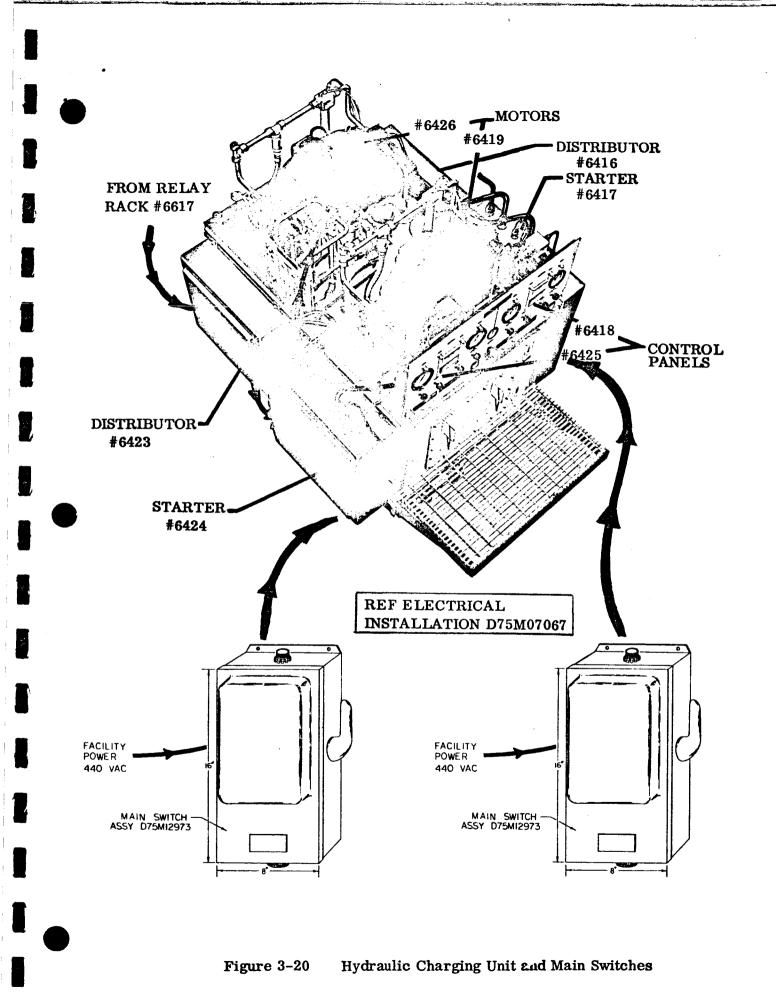
.

í

Figure 3-19 LCC Panel, Hydraulic Charging Unit

Hydraulic fluid pressure is indicated on the local and on the LCC control anels. Each panel contains electrical indicator lamps and meter circuits connected to the hydraulic supply line at the output of the main hydraulic pumps. Pressure transducers provide dynamic main-pump output pressures from 0 to 3000 psi, and fixed pressure switches provide discrete pressure indications (normal or abnormal pressure) at time of closure (2750 psi).

Should a hydraulic or electrical malfunction (with the exception of ac power loss) occur during critical launch countdown or mission abort, override capability of the start/control circuitry from the LCC control panel is provided. Use of the override capability will automatically cause a full-voltage start of the affected motor-pump, or, should the motor-pump be in a pump-running condition, continue operation of the affected pump until the override circuit is returned to the "normal" operating state.



3-20. LAUNCH EQUIPMENT FIRING CIRCUITS SUBSYSTEM.

3-21. <u>Purpose of the Subsystem</u>. The Launch Equipment Firing Circuits Subsystem initiates holddown arm release and inflight service arm retraction.

3-22. Equipment for the Subsystem. The Launch Equipment Firing Circuits Subsystem employs the following equipment:

a. Holddown Arms and Purge Valves - LCC Panel #5424

- b. Relay Distributor #6651 (Rack 19, figure 6-2)
- c. Terminal Distributor #9082
- d. Terminal Distributor #9091
- e. Terminal Distributor #9092
- f. Control Distributor #6023
- g. Control Distributor #6024
- h. Holddown arms 1, 2, 3 and 4
- i. Relay Distributor #6615 (Rack 12, figure 6-2)
- j. Firing Battery #6631
- k. Firing Battery #6632
- 1. Arming Panel #6661 (Rack 24, Figure 6-2)
- m. Arming Panel #6662 (Rack 24, figure 6-2)
- n. Liftoff switches (Holddown arms 2 & 4)
- o. Tower MI Firing Distributor #9083
- p. Service Arm Firing Distributors #9054, #9055, #9056, #9057, #9059

3-23. Description of the Holddown Arms Firing Circuit. The holddown arms firing circuit provides control, testing, and monitoring of the vehicle holddown arms. These functions are controlled from the LCC Firing Accessories Monitor and Test Panel, figure 3-21, or from the terminal countdown sequencer, and at launch, a signal is routed from the terminal countdown sequencer through relay distributor #6651 and then through electrical control distributor #6023 to two parallel solenoid-operated pneumatic valves. The operation of either of these valves allows high pressure GN₂ to release the holddown arms.

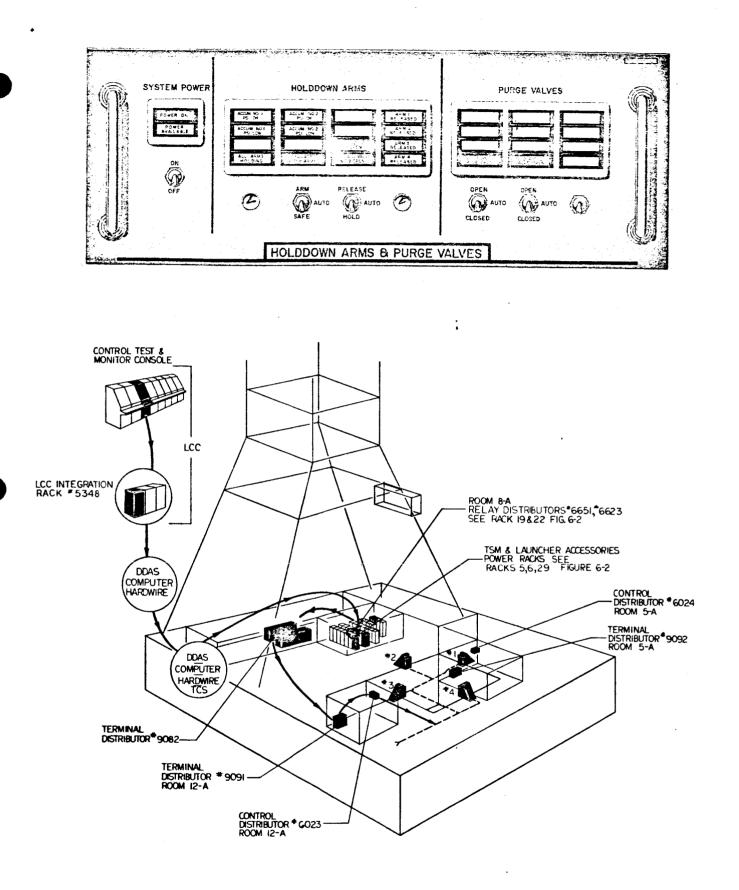


Figure 3-21 Holddown Arms Firing and Monitoring Circuits

When the holddown arms are fully retracted, limit switches return monitor signals to the LCC panel. The limit switches for holddown arms 1 and 4 are connected through distributors #6024 and #9092, and those for arms 2 and 3 are connected to distributors #6023 and #9091. The monitor signals flow back to the LCC through the DDAS-Computer-Hardwire Complex.

While in the manual mode, the LCC can also provide various test commands as shown in figure 3-21.

3-24. Local Test, Holddown Arms Firing Circuit. The holddown arms can be exercised by use of Launcher Accessories Test Set No.2, figure 3-22. It is connected to Control Distributor #6023 in place of the power and system cables.

3-25. <u>Description of the Service Arms Firing Circuit</u>. The Service Arms Firing Circuit provides 28 volts dc and automatic sequencing for initiation of umbilical carrier kickoff, umbilical carrier withdrawal, and service arm retraction for the five inflight service arms. The circuit is illustrated in figures 3-23 and 3-24, and is described in the following paragraphs.

The Service Arm Firing Circuit is first armed by the terminal countdown sequencer, which sends a command to relays in Relay Distributor #6615. These relays arm the firing circuit by closing contactors in Arming Panels #6661 and #6662, figure 3-54. Contactor closure applies 28 volts dc from firing batteries #6631 and #6632 to the service arm control switches located in Holddown Arms 2 and 4, figure 3-24. This action provides 28 volts dc to two identical and redundant service arm control switch circuits.

When the vehicle begins rising, the holddown arm Primary Service Arm Control Switch closes, allowing 28 volts de from the battery supply to be applied through the service arm control switches to the primary buses in the MI Firing Distributor #9083. Energizing these buses applies power to the primary buses in each of the five Service Arm MI Firing Distributors (paragraph 3-27) on the tower. Protected latching relays are also energized through the service arm control switches, allowing a parallel path around the switches to insure a maintained circuit during peak heat and vibration conditions.

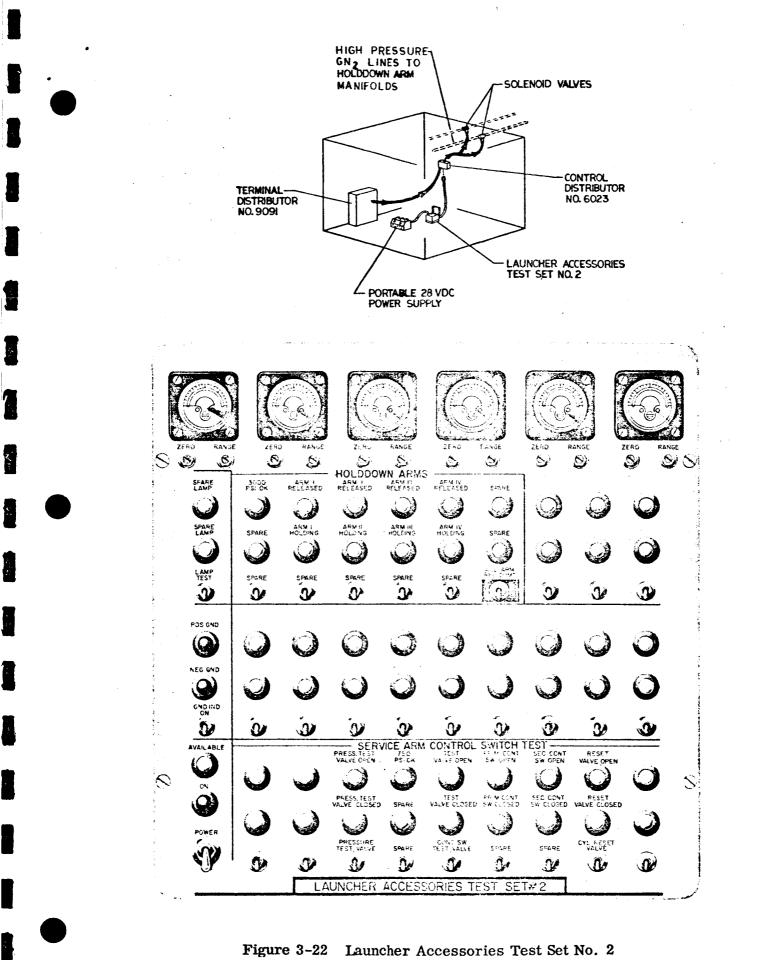


Figure 3-22 Launcher Accessories Test Set No. 2

*. `

Each MI Firing Distributor Bus supplies a signal which initiates umbilical carrier kickoff for the inflight arms. As the umbilical carrier separates, a release switch simultaneously initiates carrier withdrawal and arm retraction. Voltage for this last function is also supplied through the Service Arm MI Firing Distributor.

If the entire primary firing circuit functions properly, then carrier kickoff, carrier withdrawal, and arm retractions all occur before the secondary firing circuit is closed. Note, however, that the secondary firing circuit will be activated even if it is not needed.

The Secondary Service Arm Control Switch closes when the vehicle rises to 18 inches, providing excitation of a secondary firing circuit. This secondary circuit initiates carrier withdrawal and arm retraction directly, since carrier kickoff has already been effected mechanically. In all, two primary and two secondary circuits provide four possible actuations of the umbilical carrier and service arm at launch.

Functions relating to the five inflight arms not discussed here are functions of the Service Arm Subsystem, paragraph 3-2.

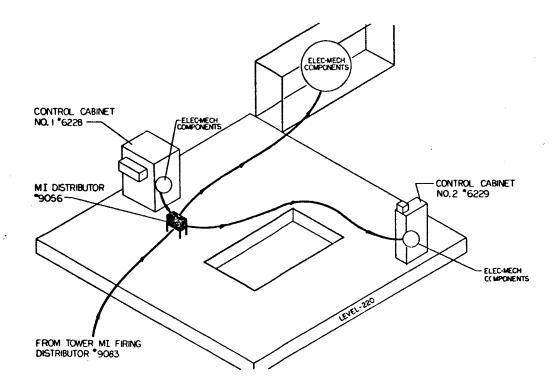
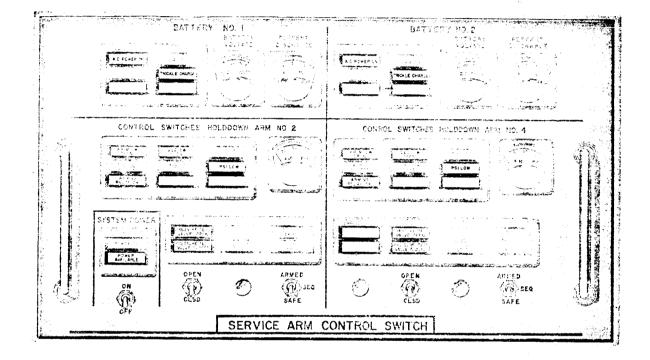
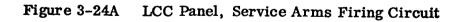
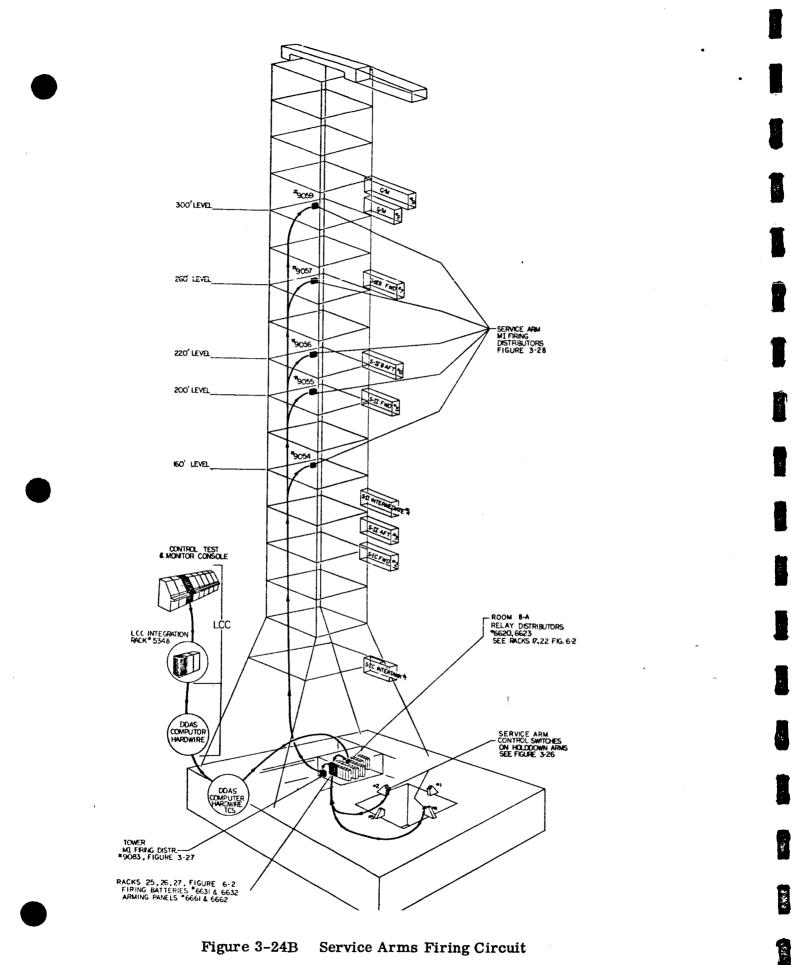


Figure 3-23 Typical Level, Service Arms Firing Circuit



1





Service Arms Firing Circuit Figure 3-24B

1 Strates and

3-26. <u>Local Test of the Service Arm Firing Circuit</u>. Within the service arm firing circuit, the only local test is that of the service arm control switches, figure 3-25. A pneumatic test mechanism is provided in the service arm control switch assembly which physically closes the switch for test. Pressure for the mechanism is supplied through a Pneumatic Box Assembly (figure 3-26), which is controlled through a standard eighteen-connector electrical distributor.

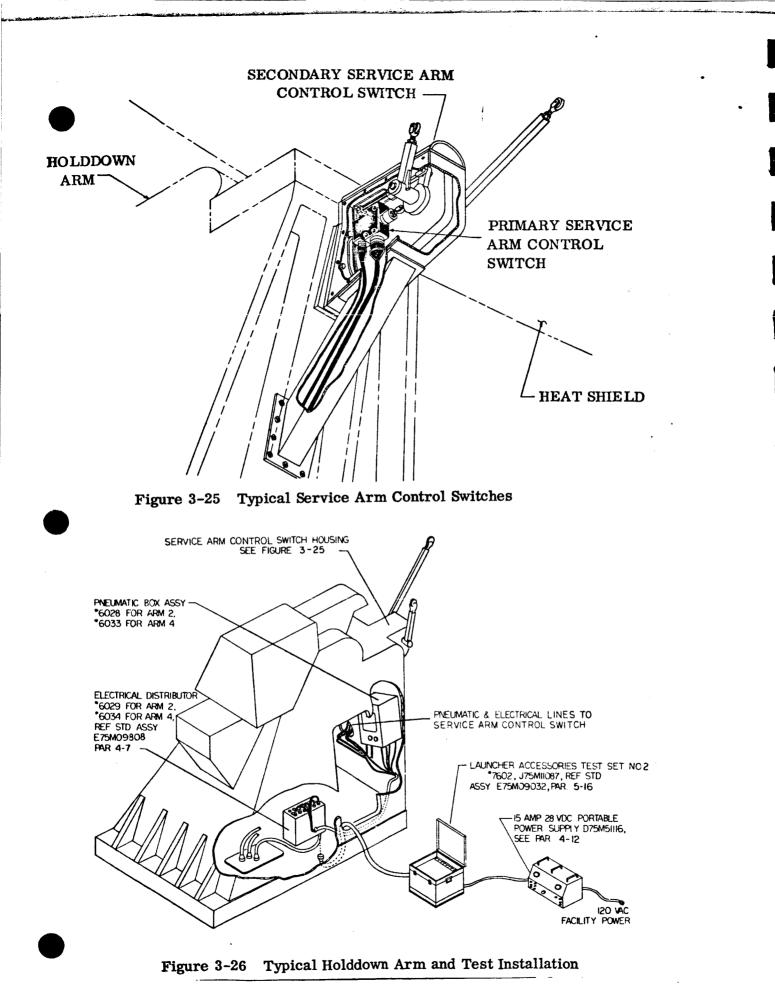
Test functions are applied to this distributor by the Launcher Pneumatics Test Set No. 2, a programming of the Universal Patchboard Test Set. This test set actuates the liftoff switch test mechanism and monitors pneumatic pressures and switch positions. Its test functions are outlined in figure 3-22.

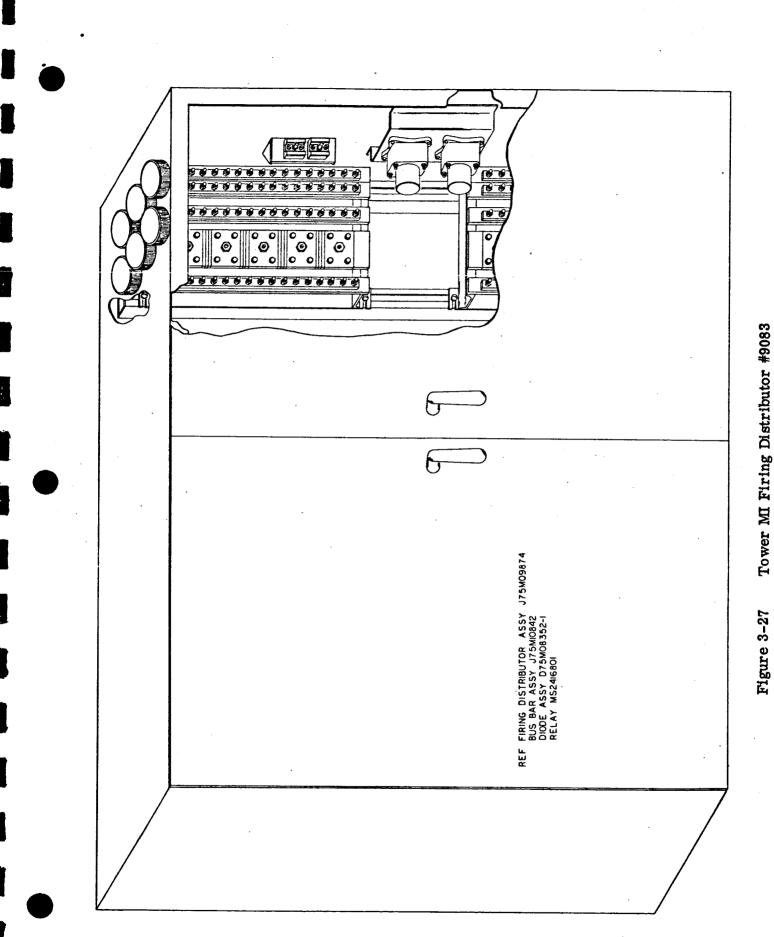
3-27. <u>MI Distributors and Cabling.</u> The service arms firing circuit carries the umbilical release command through Tower MI Firing Distributor #9083 to the five service arm MI Firing Distributors. These are facility enclosures with distribution modules provided by the Launch Equipment Branch. The modules are shown in figures 3-27 and 3-28.

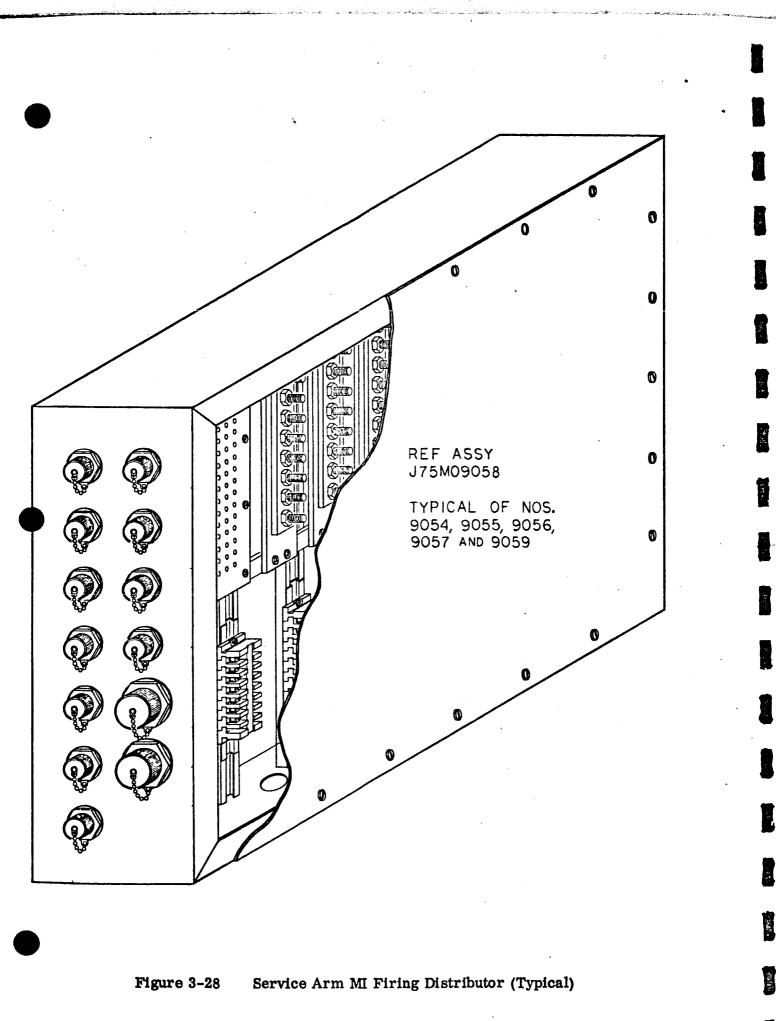
Owing to the critical nature of the service arm firing circuit, the firing distributors are interconnected by Mineral Insulated (MI) Cable. This is a copperencased cable carrying one to seven conductors within compacted magnesium oxide insulation. It is highly resistant to heat, mechanical disturbances and RF radiation. Special bulkhead fittings are available for sealing its conductors against moisture and atmosphere.

Figure 3-29 shows a typical MI cable installation for the service arm firing circuit. MI cable also carries power to certain lighting J boxes and emergency J boxes aboard the service arms.

A typical termination of MI cable is shown in figure 3-30 and is described fully in procedure drawing A75M07450.



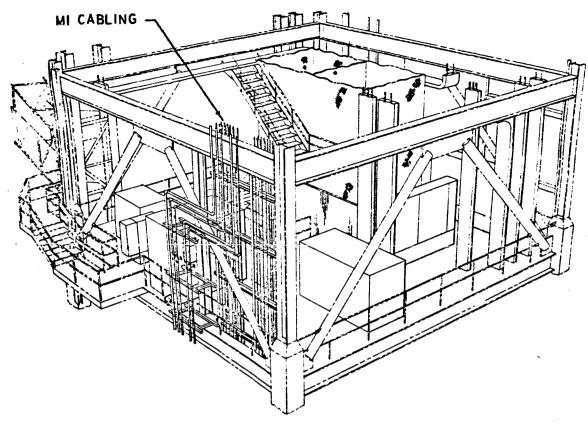




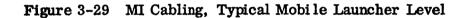
3-40

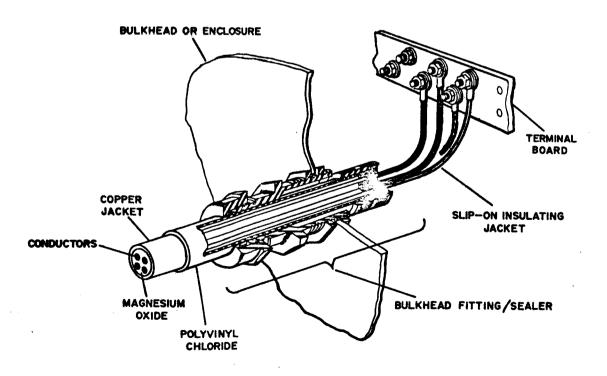
134292

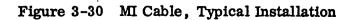
14.5











3-28. TAIL SERVICE MASTS SUBSYSTEM.

3-29. <u>Purpose of the Subsystem</u>. This electrical subsystem provides system control, test, and monitor for tail service mast operation. The Tail Service Masts provide support for fuel, electrical and air conditioning lines which must enter the aft section of the S-IC vehicle.

3-30. Equipment for the Subsystem. The Tail Service Masts Subsystem uses the following equipment:

- a. TSM 1-2 #6006, 3-2 #6007, and 3-4 #6005, Level 0.
- b. TSM Control Distributor #6009, TSM 1-2
- c. TSM Control Distributor #6010, TSM 3-2
- d. TSM Control Distributor #6008, TSM 3-4
- e. Control Test and Monitor Consoles, LCC (TSM 1-2 #5420, TSM 3-2 #5421, TSM 3-4 #5422)
- f. TSM Portable Test Set #7601, Level 0
- g. Relay Racks, room 8-A. See Racks 18 and 19, figure 6-2
- h. Power Distribution Rack #6422, room 8-A. See Rack 29, figure 6-2
- i. Power Supply Rack #6421, room 8-A. See Rack 6, figure 6-2
- j. Standby Battery Rack #6408, room 8-A. See Rack 5, figure 6-2
- k. Terminal Distributor #9091, room 13-A
- 1. Terminal Distributor #9092, room 5-A
- m. Terminal Distributor #9082, room 7-A
- n. Portable DC Power Supply, Level 0

3-31. <u>Description of the Subsystem.</u> The Tail Service Masts (TSM) are operated by pneumatically pressurized hydraulic fluid and are electrically controlled. Each mast has 31 electrical and electro-mechanical components and an electrical control distributor located within its structure. Component control and monitoring signals are transferred into 60-conductor cables via these control distributors and routed through an interface plate and facility distributors to two Launcher Accessory Relay Distributors (6650 and 6651) located in Room 8-A of the Mobile Launcher.

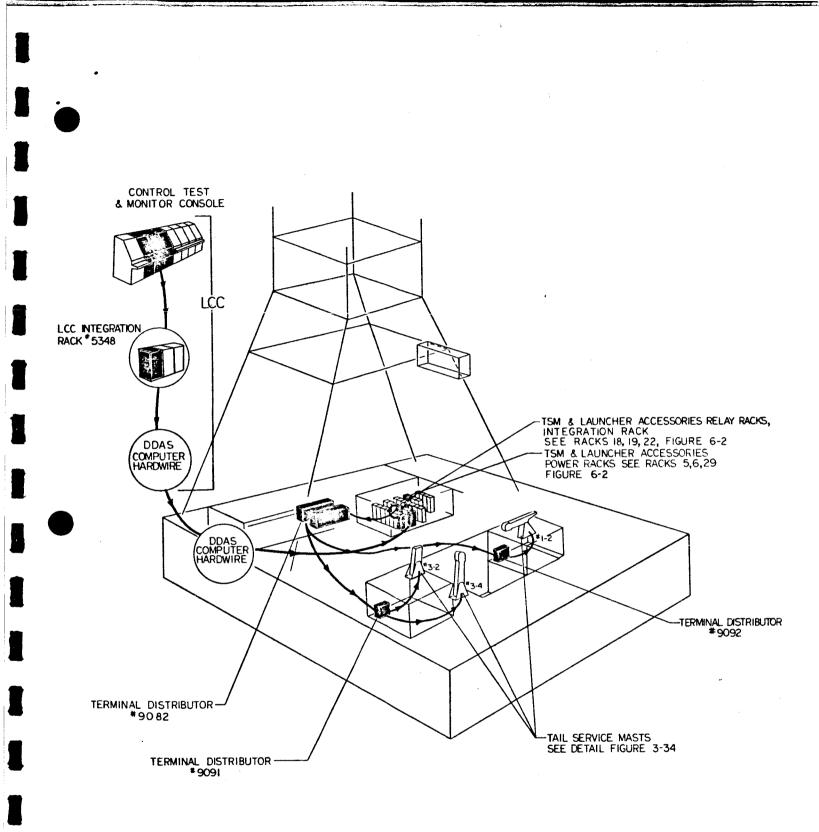


Figure 3-31 Tail Service Masts Subsystem

Relay closures in the relay distributors provide control logic and permit the transfer of 28 volts dc via the terminal and control distributors to limit and pressure switches and solenoid-operated valves within each TSM. Transducer excitation voltage (50 vdc) is transferred from the MSFC-Astrionics integration distributor through the relay distributor and the terminal and control distributors to the transducers. See figure 3-31.

The three control panels provide visual displays of all critical parameters for the TSM subsystem. These panels also provide controls for pneumatic and hydraulic charging and mast retract tests. The masts can be extended locally only by a control on the side of each mast or by a test set (a typical panel is illustrated in figure 3-32). Test and control functions may be programmed into the LCC Mobile Launcher computer, but all signals so programmed can be manually overridden from the control panel.

3-32. Local Testing of the Tail Service Masts. For local control and testing, a portable TSM Test Set (see figure 3-34) is employed. This is a special programming of the Universal Patchboard Test Set. The face panel of this test set is illustrated in figure 3-33. Two cables from the TSM Test Set replace cables from the base of the TSM Control Distributor. Power is supplied to the test set by a portable 28 volt dc supply, paragraph 4-12. By use of the test set, the masts may be individually retracted or extended and each electrical component within the masts may be controlled and monitored.

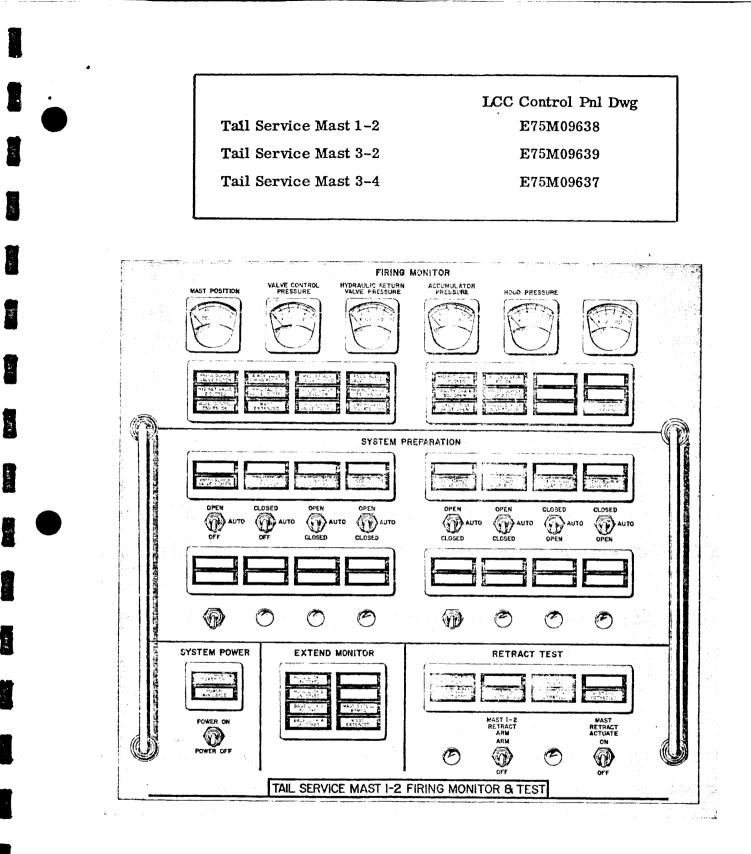


Figure 3-32 LCC Control Panel, Tail Service Mast

MAST POSITION VALVE CONT PRESS, HYD RET V PRESS, ACCUM PRESS, HOOD PRESS, FS+100 PSI F5+100 PSI F5+										
					Contraction of the second		1 STATE	90		
			ELL C	<u> <u>B</u></u>		EH /				1. 1. 1. 1. 1. 1. 1. 1. 1. 1. 1. 1. 1. 1. 1
ZERO RAN		RANGE	ZERO	RANGE	and the second s	RANGE	ZERO RAN			and the second
0 6 6	-	S	226	S	S	3	PRESS. SUP	GRAV DR	ACCUM	8
SPARE	SUP V OPEN	LEVEL OK	PRESS. SUP	PRESS.SUP		HYD RET	V3 OPEN	SOV OPEN	PSIOK	
SPAFE	SURV	SPARE	PRESS. SUP	PRESS.SUP V2 CLOSED	HYD RET V PSI OK	HYD RET	PRESS. SUP	GRAV DR SOV CLOSED	SPARE	
	CLOSED		VI VENTED	\bigcirc	\bigcirc	\bigcirc	\bigcirc	\bigcirc		
LAMP TEST	HYD SUP VALVE	PRESS.SUP	FRESS. SUP VI OPEN	SYSTEM PRESSURIZE	SPARE	SPARE	SPARE	SPARE	SPARE	
<u>•</u>	ARMING	<u>.</u>	<u>A</u>	<u> </u>		XTEND-	<u></u>	<u></u>	<u>ð</u>	
POS GRO	UMB REL SIM	LOCK REL V	UNLOCKED	LOCK 2 UNLOCKED	LOCK 3	LOCK 4	4-WAY V EXT POS	CYL SUR		
G	UMB REL SIN	LOCK REL V	LOCKI	LOCK 2	LOCK 3	LOCK 4	A-WAY V	CYL SUP.	LUCK SWITCHES	
NEG GRD	V CLOSED		LOCKED				FIRE POS	V CLOSED	BYRISSED	
GRD IND ON	UMB REL SIM	LOCK REL VALVE	SPARE	SPARE	SPARE	SPARE	MAST	SPARE	LOCK STATCHES BYPASSED	5
<u>(</u>)	SYSTEM	TEST -	N PROP. VA	IVE TEST	<u></u>	<u>.</u>	MAST TES	-A		
AVAILABLE	HYD BYP V OPEN	BLEED V OPEN	PROP.V SOV OPEN		UMB DISC.		HYD RET	HYD SET		
O ON	HYD BYP	BLEED V	PROP. V	PROP.V		UNB LS	HYD RET	HYD RET	MAST	0
	V CLOSED		SOV CLOSED	CLOSED	V CLOSED			V2 CLOSED	RETRACTED	
POWER	HYD BYP VALVE	PNEU BLEED VALVE	FROP.V SOV	SPARE	UMB DISC.	SPARE	SPARE	SPARE	LIFT-OFF SIMULATE	
	1	N.	TAN O	N.		ST SET	A -	•1-		•
			TAIL SI	ERVICE	MAST TE	EST SET		·····		فعندر

1989 1987

Î

Color and

19.00

A days

2542 V

Ar Carl

1.4.562

SLAME.

State of

Figure 3-33 Face Panel, TSM Test Set E75M07658

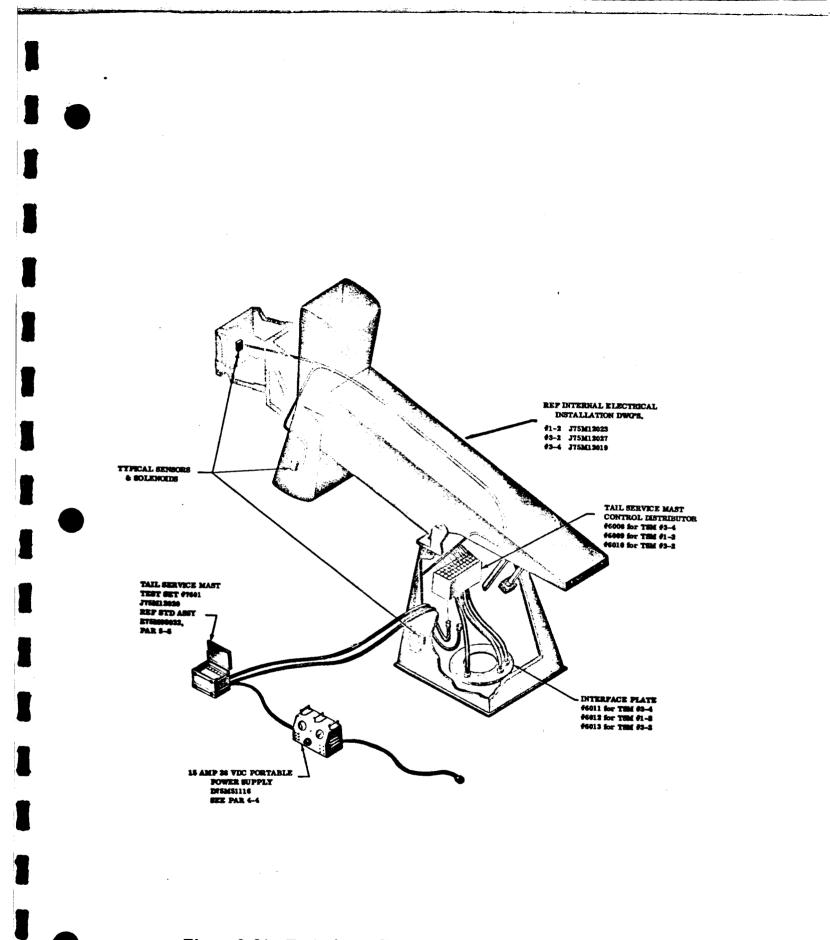


Figure 3-34 Typical Installation, TSM Test Set E75M12020

3-33. PNEUMATICS CONTROL AND DISTRIBUTION SUBSYSTEM.

2-34. <u>Purpose of the Subsystem</u>. The Mobile Launcher pneumatic electrical system provides electrical controls for operating valves and monitoring pressure and valve positions for the GN₂ and helium high pressure valving and panels, Valve Panel No.11 and Valve Panel No.12, and also the purge valves. These panels and valves supply pneumatic pressure for the holddown arms, tail service masts, service arms, hydraulic checkout console, Q-ball, engine servicing and purge applications.

3-35. Equipment for the Subsystem. The following equipment is used by the Pneumatics Control and Distribution Subsystem:

- a. Helium Control Panel #6404, room 1-B
- b. GN₂ Control Panel #6405, room 1-B
- c. High Pressure Pneumatics Control Distributor #3407, room -1B
- d. Valve Panel No.11 #6605, room 12-A
- e. Valve Panel No.11 Control Distributor #6667, room 12-A
- f. Holddown Arm Control Distributor #6023, room 12-A
- g. Valve Panel No.12 #6167, level 160
- h. Valve Panel No.12 Control Distributor #6165, level 160
- i. Relay Distributor #6651, Room 8-A. See rack 19, figure 6-2.
- j. UB Instrumentation and Control Distributor #9016, level 160.
- k. Terminal Distributor #9082, room 7-A
- 1. Terminal Distributor #9091, room 12-A
- m. Terminal Distributor #9092, room 5-A
- n. Control, Test and Firing Monitor Console, LCC

3-36. Description of the Subsystem. GN_2 and helium values, value positions, and pressures aboard the Mobile Launcher are controlled and monitored from the Pneumatic Distribution System Panel (figure 3-36). This panel interfaces with the Launcher Accessories Relay Distributor #6651 via the DDAS-Computer-Hardwire complex, figure 3-35. The relay rack provides the required logic for control and monitoring functions, and acts as an interface transfer unit. Terminal and control distributors connect this relay rack to the value panels and values for each of the following areas:

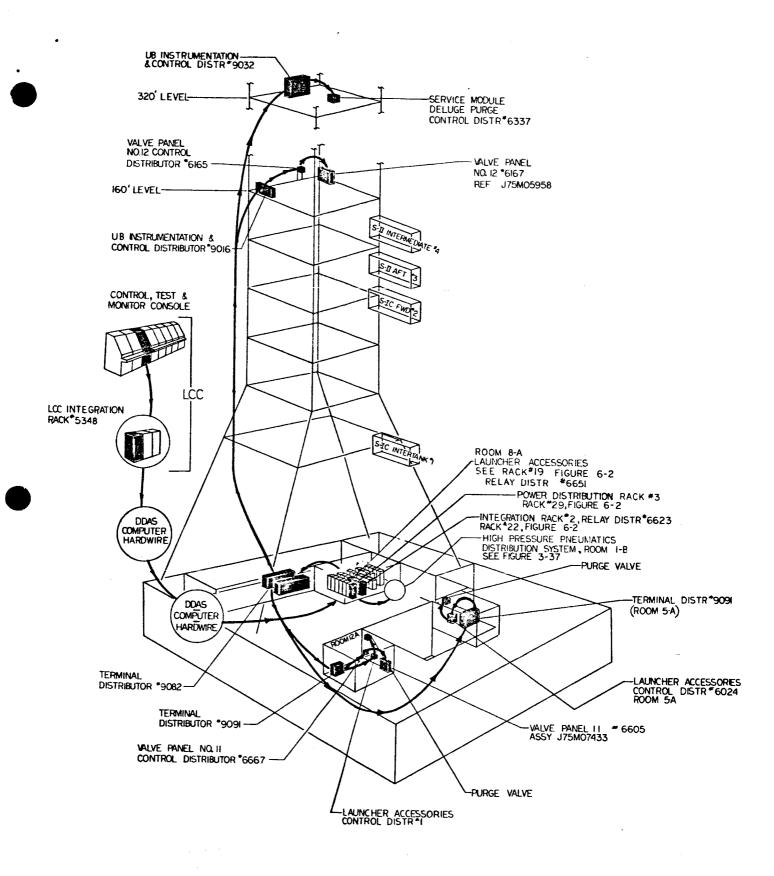


Figure 3-35 Pneumatics Control and Distribution Subsystem

- a. High Pressure Pneumatics System
- b. Valve Panel No. 11, J75M07433
- c. Valve Panel No. 12, J75M05958
- d. Deluge Purge Panel
- e. Launcher Purge Valves

The High Pressure Pneumatics System, figure 3-37, consists of a manifold and valving system for distributing GN_2 and helium to valve panels and consoles throughout the Mobile Launcher.

Valve Panel No.11 provides:

- a. 3000 psi and 750 psi GN_2 for the Tail Service Masts and for checkout of the ground support hydraulics
- b. 1000 psi GN_2 for engine servicing and cleaning
- c. 1500 psi helium for the holddown arms pneumatics

d. 50 psi GN, for hazardous purge requirements

Valve Panel No.12 provides:

a. 750 psi and 125 psi GN₂ for all service arm control pneumatics

b. 50 psi GN, for hazardous purge

3-37. <u>Tests for the Subsystem.</u> Two different test units are used within the Pneumatics Control and Distribution Subsystem for local testing:

a. Launcher Accessories Test Set No.1 is a programming of the Universal Patchboard Test Set, figure 5-8. It measures pneumatic pressures, provides actuating test signals and monitors switch and valve positions for the High Pressure Pneumatics system (figure 3-37) and for Valve Panel No.11 (figure 3-38). These functions are shown on the test set face panel, figure 3-39.

By plug swapping a test cable at the control distributor (#6407) and utilizing a portable power supply, the GN_2 and helium panels may be used for local control and monitoring.

b. Tower Test Set E75M14137 is a programming of the Miniaturized Universal Test Set, figure 5-6. It measures pneumatic pressures, provides actuating signals, and monitors switch and valve positions for Valve Panel No. 12, figures 3-40 and 3-41. It also provides control and monitor functions for the Q-Ball Cover Removal Subsystem, figure 3-43.

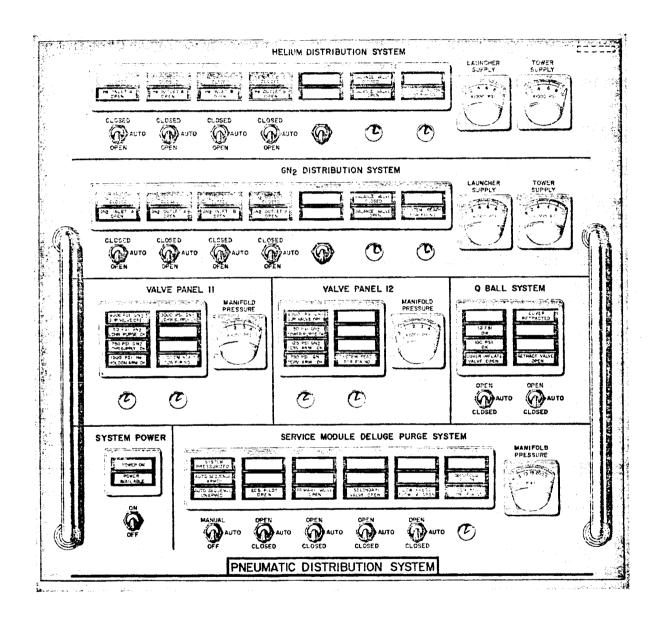
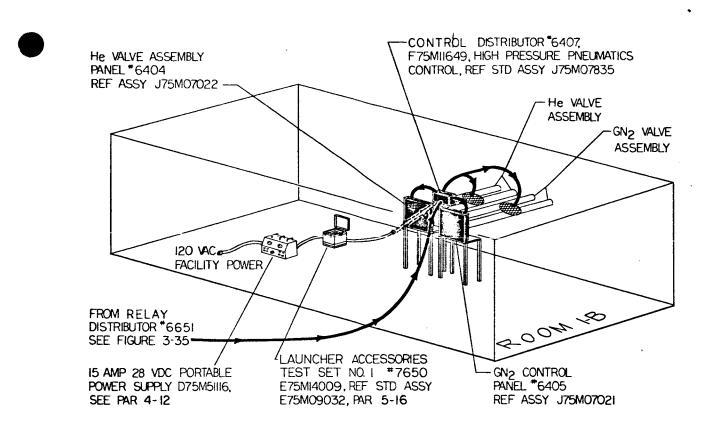


Figure 3-36 LCC Panel, Pneumatic Distribution System





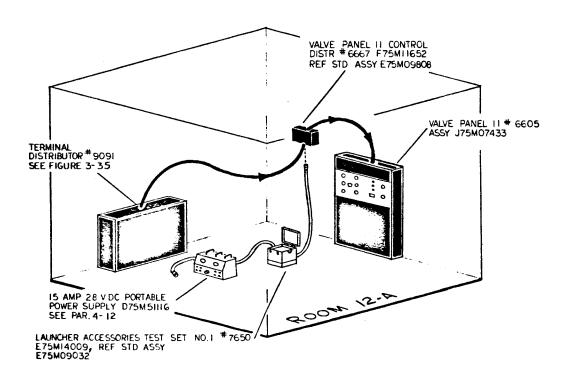
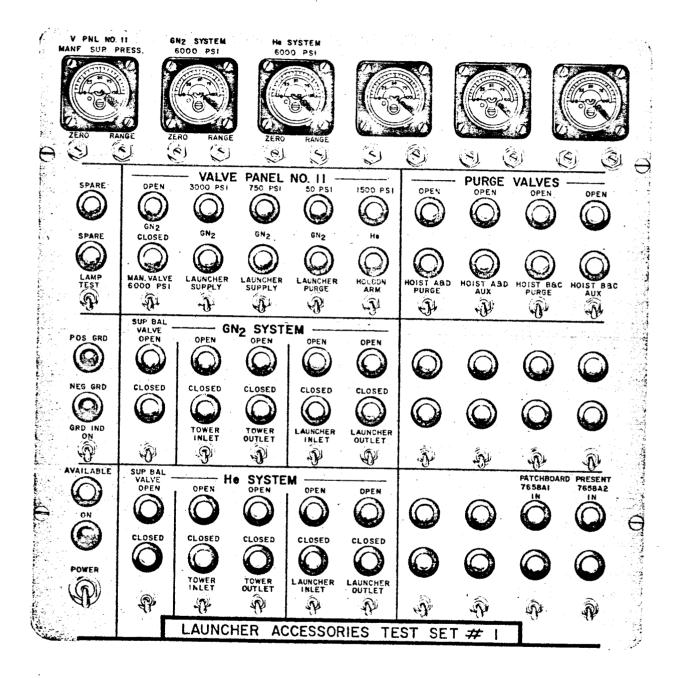


Figure 3-38 Typical Test Installation, Valve Panel 11



304

Action of

345.00 M

a lighter

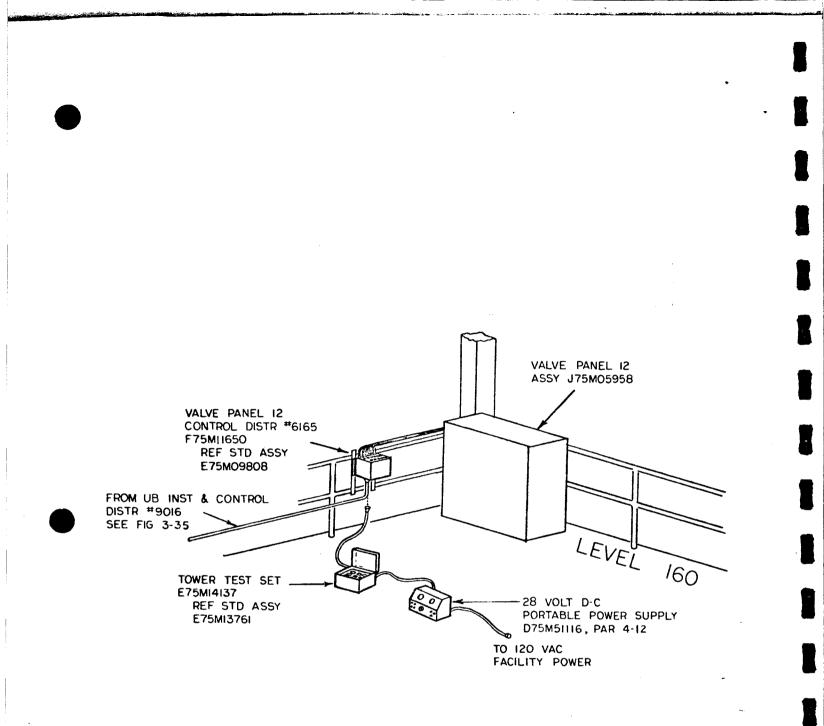
A W. P.

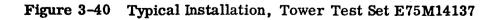
Will.

Sec. 14

1.0

Figure 3-39 Face Panel, Launcher Accessories Test Set No. 1, E75M14009





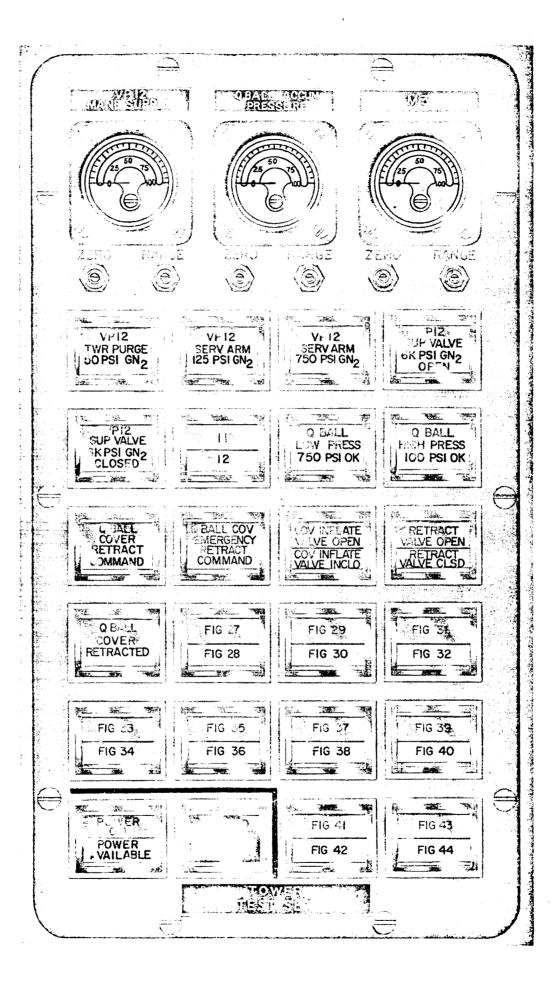


Figure 3-41 Face Panel, Tower Test Set E75M14137

3-38. Q-BALL COVER REMOVAL SUBSYSTEM

3-39. <u>Purpose of the Subsystem</u>. The Q-Ball Cover Removal Subsystem provides electrical control and monitoring of the Q-Ball cover mechanism.

3-40. Equipment for the Subsystem. The Q-Ball Cover Removal Subsystem employs the following equipment:

a. Hoist Assembly (J75M15383)

- b. Pneumatic Assembly #6360 (J75M15388)
- c. Q-Ball Cover Control Distributor #6361 (F75M13155)
- d. UB Instrumentation and Control Distributor #9032
- e. Terminal Distributor #9082A32
- f. Relay Distributor #6651. See rack 19, figure 6-2.
- g. Panel Assembly Pneumatic Distribution System, LCC Panel, (D75M09674)

3-41. Description of the Subsystem. The Q-Ball Cover is a fiberglass cover for vehicle guidance components in the tip of the Saturn V Launch Escape System rocket. Prior to launch, this cover is ejected by pressurized GN_2 (10 psi). It then is retracted from the vehicle by a Hoist Assembly consisting of a lanyard, a pneumatic motor, and a weight and pulley system (see figure 3-43).

Pneumatic pressure for cover removal is supplied by the pneumatic assembly (#6360) located beside the electrical control distributor (#6361) on the 360-foot tower level. This pneumatic assembly is controlled remotely from the LCC Pneumatic Distribution System Panel, as in figure 3-42.

Several minutes before vehicle liftoff, a manual cover inflation switch on the Pneumatic Distribution System Panel (figure 3-36) is closed, providing a signal routed through the DDAS-Computer-Hardwire Complex to Relay Distributor #6651 in room 8-A. Relay closure provides a 28 volt dc signal through distributor #9082A32

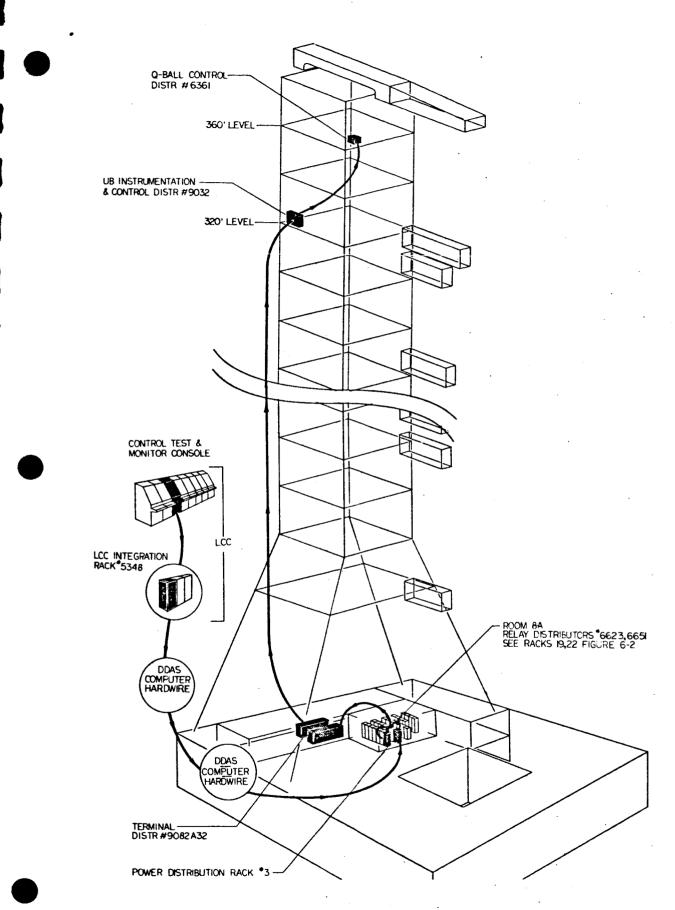


Figure 3-42 Q-Ball Cover Removal Subsystem

and #9032 to a cover inflation solenoid value in the pneumatic assembly. The GN 2 (10 psi) is thus supplied through pneumatic hose to a nylon bladder in the Q-Ball Cover. The bladder inflates and raises the Q-Ball Cover.

As the Q-Ball Cover falls, a steel weight in the gravity retract system actuates a motor-start limit switch and energizes a relay. One contact closure energizes another relay in series with the motor-stop limit switch and another seals in the motor-start limit switch. A contact from the relay in series with the motorstop limit switch energizes the second solenoid valve. This retract valve opens and allows nitrogen at 125 psi to actuate the pneumatic motor and wind up the lanyard. When a mechanical stop on the lanyard reaches the hoist assembly, the motor-stop limit switch is actuated, closing the retract valve. As the cover retracts fully, a limit switch returns a signal to the LCC along a path parallel to the command signals. After the lower limit switch is activated, a Q-Ball Cover retract lamp on the LCC control panel indicates full retraction.

Panel indicator lamps are illuminated whenever the retract value is open, the cover inflate value is open, the low pressure reaches 10 psi in the Q-Ball pneumatic assembly GN_2 accumulator, or the high pressure reaches 100 psi.

In case of emergency, a manual retract switch on the LCC pneumatic distribution system panel may be closed. This switch closure bypasses the relay logic and energizes the retract valve solenoid. The retract valve then opens, allowing 125 psi nitrogen to actuate the pneumatic motor, which in turn pulls a mechanical disconnect pin. This allows the Q-Ball Cover to open and be retracted by the counterweight.

3-42. Local Test. The Q-Ball Cover mechanism can be locally tested by use of Tower Test Set E75M14137. This test set is connected to Q-Ball Cover Control Distributor #6361 in place of the system cable, and provides control and monitor functions shown in figure 3-41.

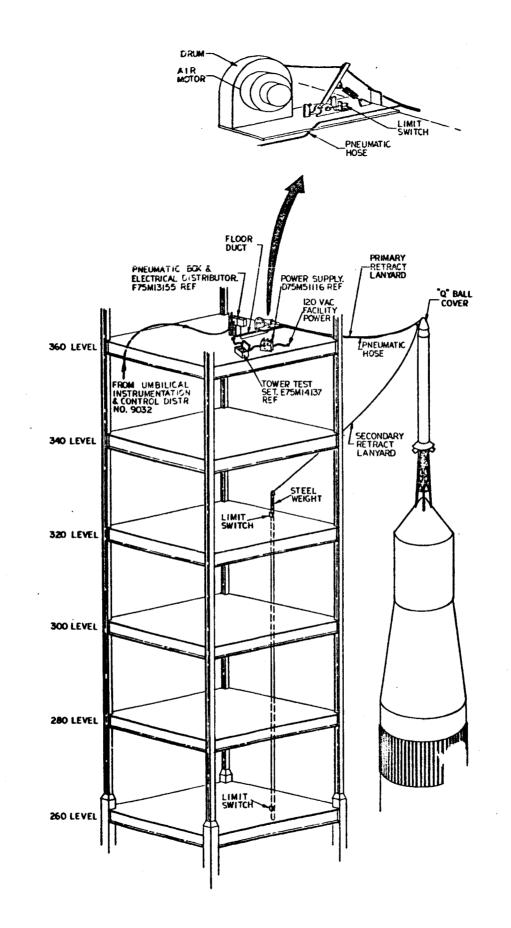


Figure 3-43 Q-Ball Cover Mechanism, Level 360

3-43. ESE POWER SUBSYSTEM.

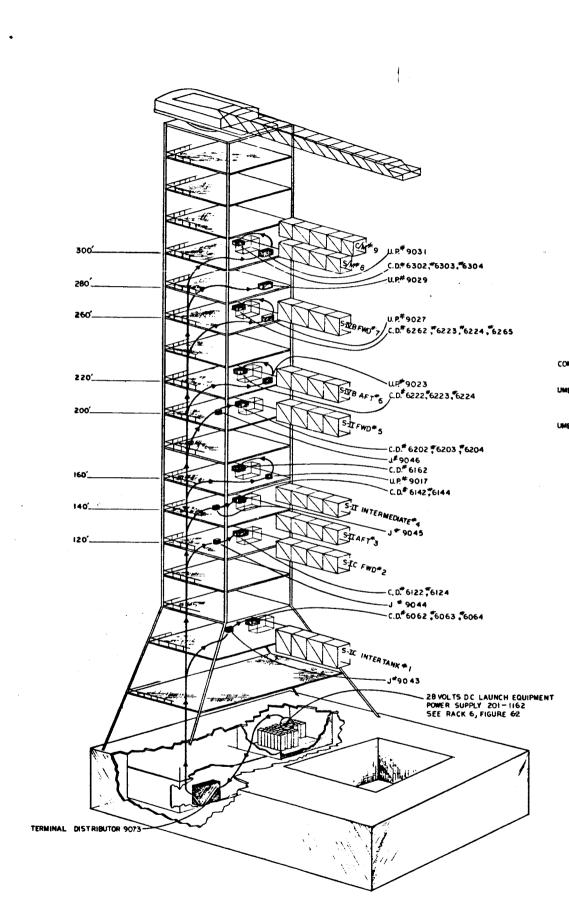
3-44. Purpose of the Subsystem. This subsystem provides dc power to the ESE.

3-45. Equipment for the Subsystem. The ESE Power Subsystem uses the following equipment:

- a. Relay Distributor #3620 (power distribution only).
- b. Relay Distributors #6613 thru #6619, #6621, #6622, #6650, #6651 (Power distribution and other subsystems), figure 3-53.
- c. Battery Rack #1172, figure 3-52.
- d. Power Supply Rack #1162, figure 3-50.
- e. Power Distribution Racks, figure 3-51. See Racks 7, 8, 29, figure 6-2.
- f. UB Power Terminal Distributors #9042, #9044, #9054, #9017, #9023, #9027, #9029, #9046, and #9031 on various launcher levels as shown in figure 3-44.
- g. Terminal Distributor #9082, Room 7-A.
- h. Terminal Distributor #9073, Room 7-B.
- Control Distributors #6008, #6009, #6010, #6023, #6024, #6063, #6064, #6122, #6124, #6142, #6144, #6162, #6202, #6203, #6204, #6222, #6223, #6224, #6262, #6263, #6264, #6265, #6302 #6303, #6304, #6667 and #6407 on various launcher levels, figure 3-44.
- j. Power Control Console (Astrionics)

3-46. <u>Description of the Subsystem.</u> Power for ESE electrical equipment consists of (a) 28 volt monitor power, (b) 28 volt control power, and (c) power for DDAS and hardwire transmission. These three types of power are discussed separately below:

(a) 28 volt monitor power is applied to limit switches, pressure switches, and other components which monitor the state of the ESE subsystems. It is also applied to the ESE patch distributors where relay contacts monitor the state of the subsystems. Power in this category is generated in power supply 201-1162, and is distributed directly to the monitor switches and relay contacts without any command switching from a control panel. Monitor power for the service arms is shown in figure 3-44 and that for the service arms patch distributors in figure 3-45. Monitor power for the launcher networks and corresponding patch distributors is shown in figure 3-46.



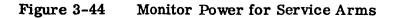
家

¢

64.6

C.D. CONTROL DISTRIBUTOR NO.1 U.P. UMBILICAL POWER TERMINAL DISTRIBUTOR

J UMBILICAL POWER J BOX



(b) 28 volt control power is that power required for relays, solenoidoperated valves, and other components which effect actuation of the service arms, tail service masts, and launcher accessories. It is applied to buses in the patch distributors through power modules (contactors) which are controlled from the LCC control panels, as shown in figure 3-47. A similar control power is supplied to a bus in the MSFC-Astrionics signal conditioning equipment (figure 3-48) which returns the power through relay contact closures.

c. Power for DDAS and hardwire transmission as well as for the LCC control panels is supplied by Astrionics to buses in the integration racks and patch racks. See figure 3-49.

The source power for the system described above is supplied by MSFC and consists of a basic 500 amp power supply rack (#1162) and a standby battery rack (#1172). Operation of these racks is controlled from an LCC power control console also furnished by MSFC. See figure 3-55.

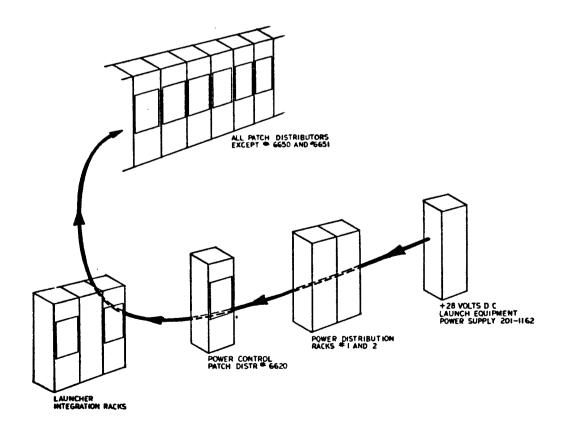
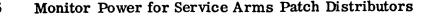


Figure 3-45



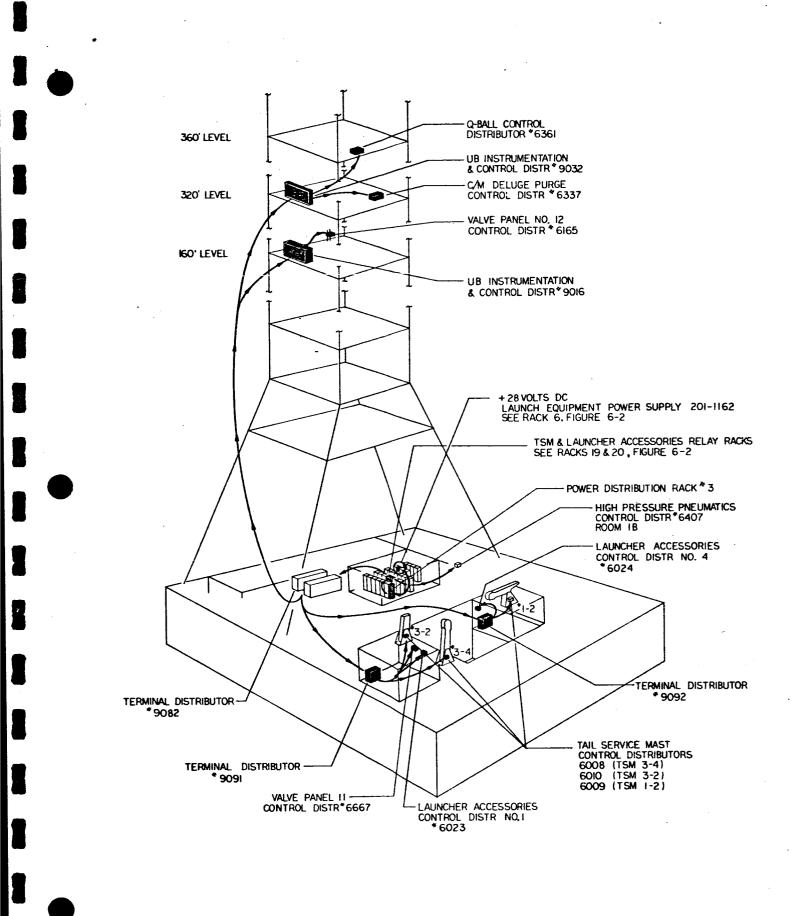
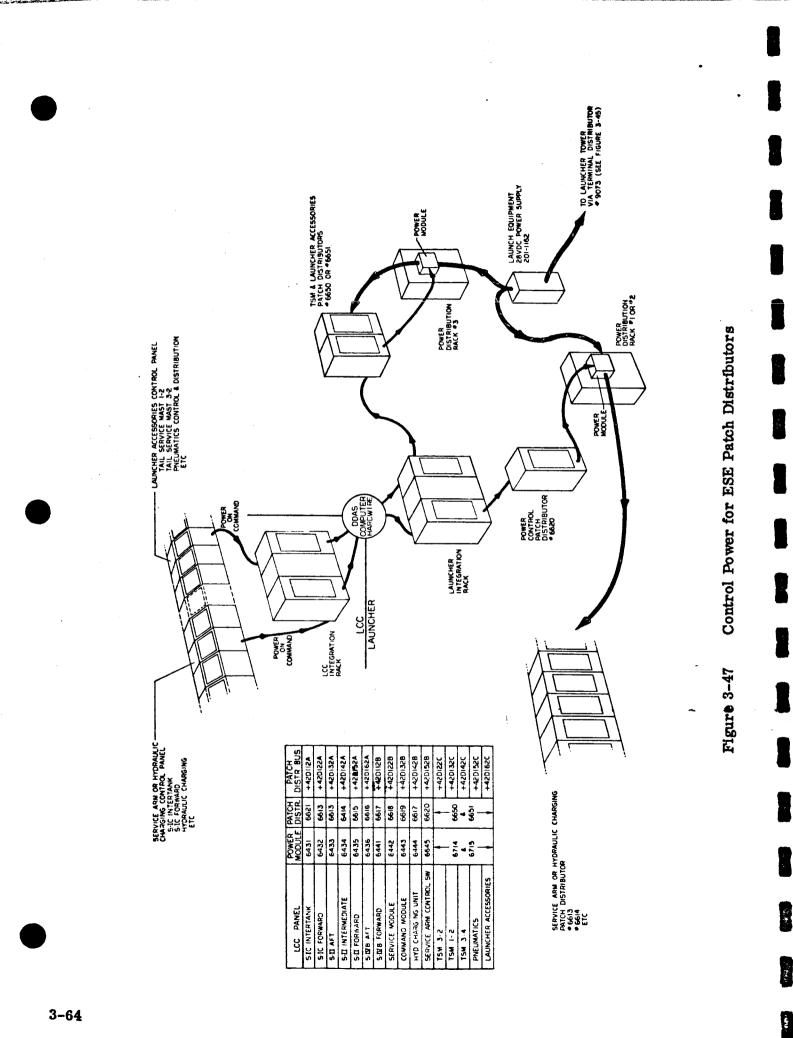
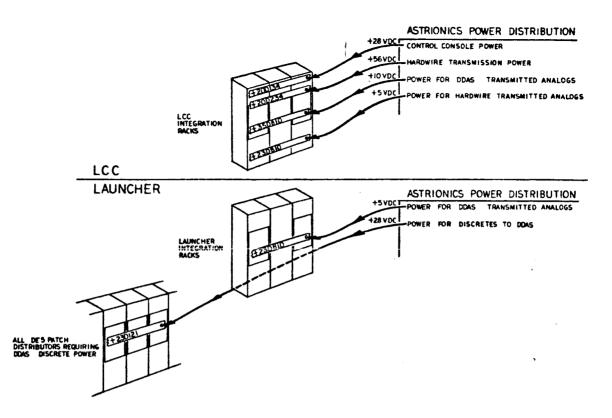
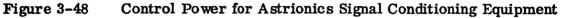


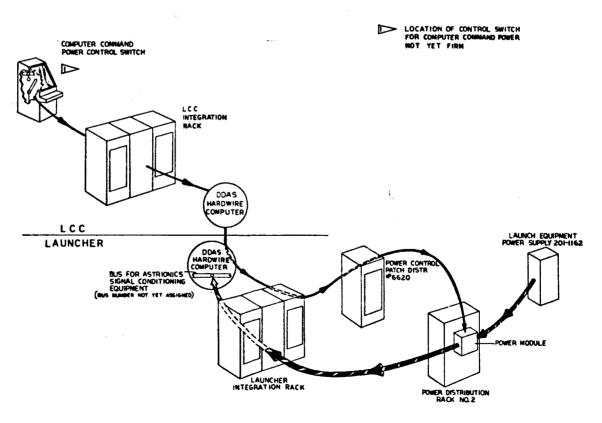
Figure 3-46Monitor Power for Launcher Networks and
Patch Distributors #6650 and #6651

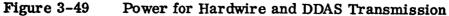






,





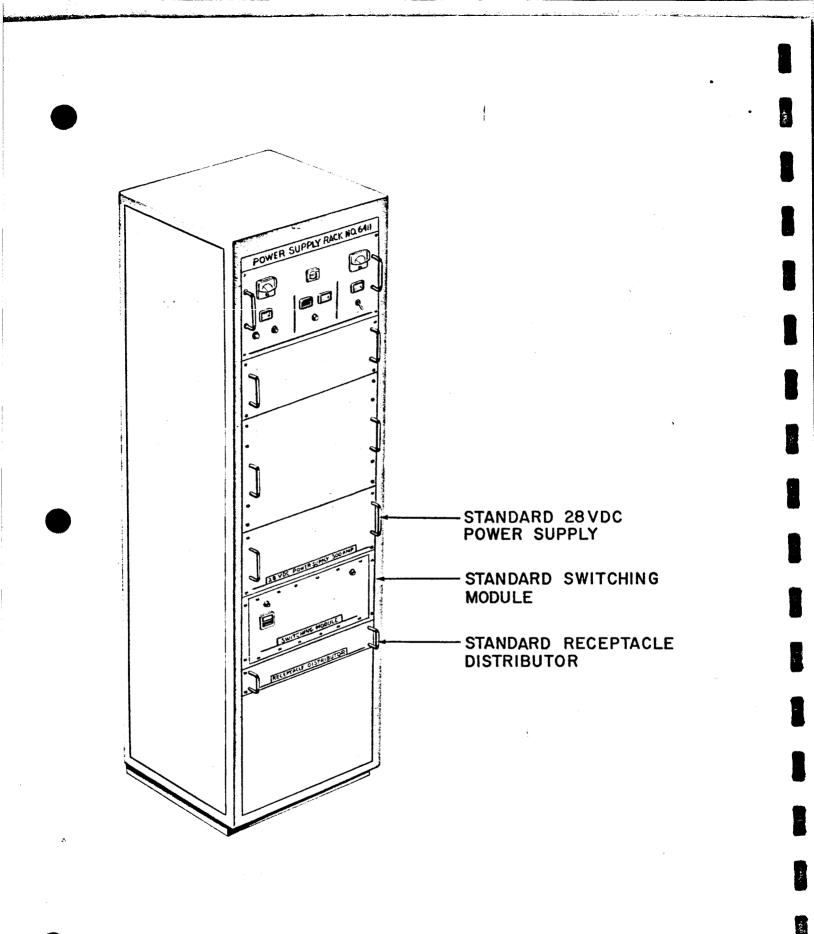
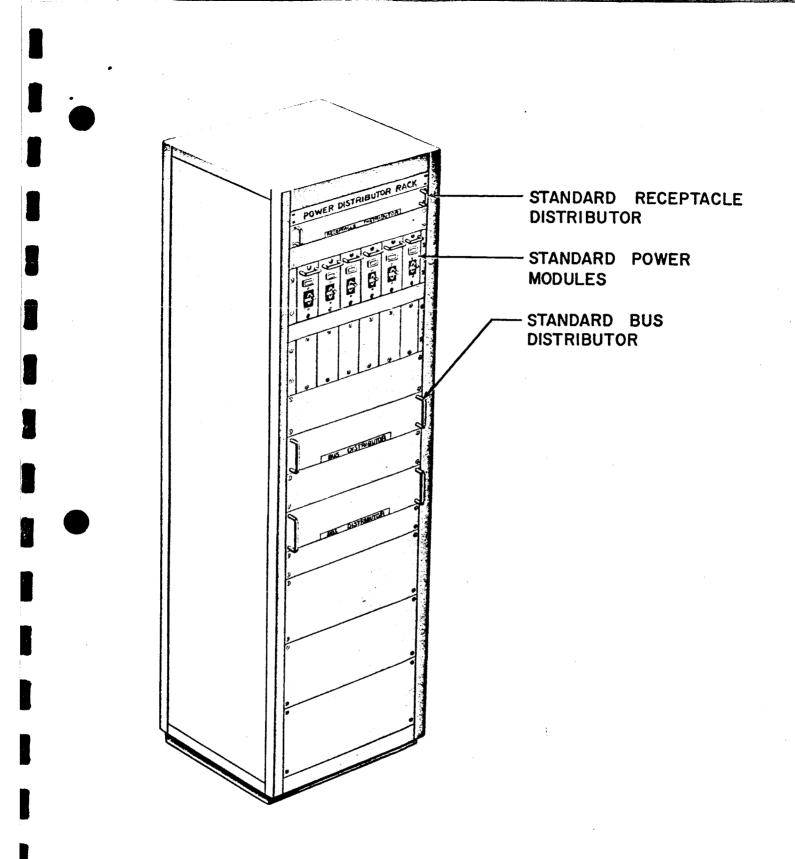
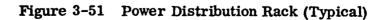


Figure 3-50 Power Supply Rack (Typical)





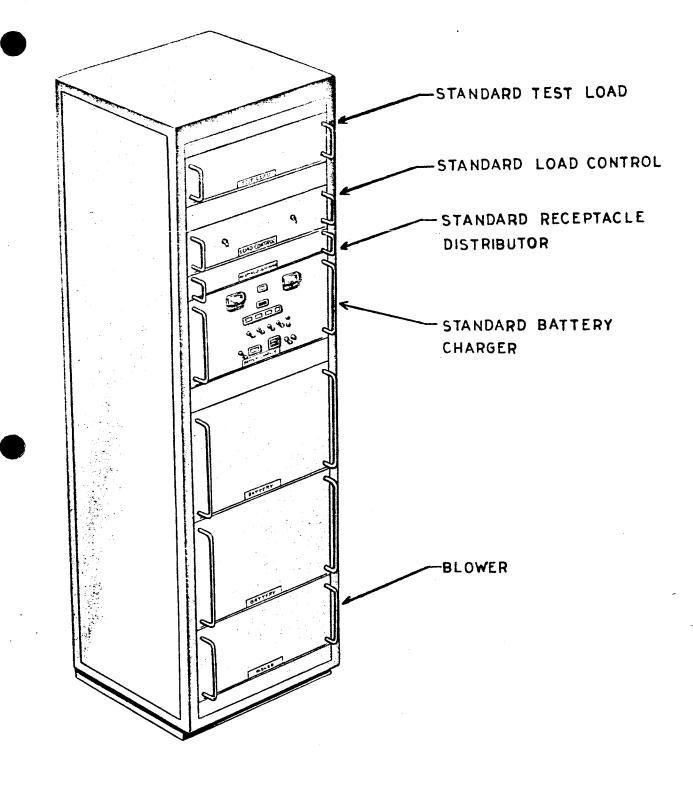
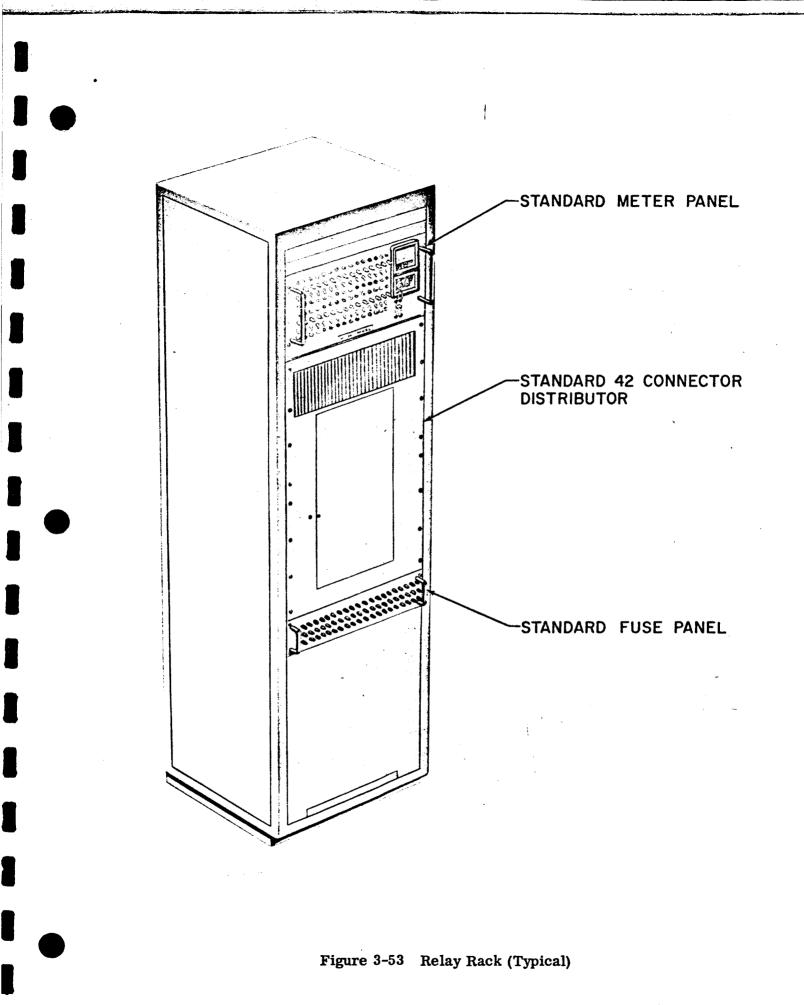
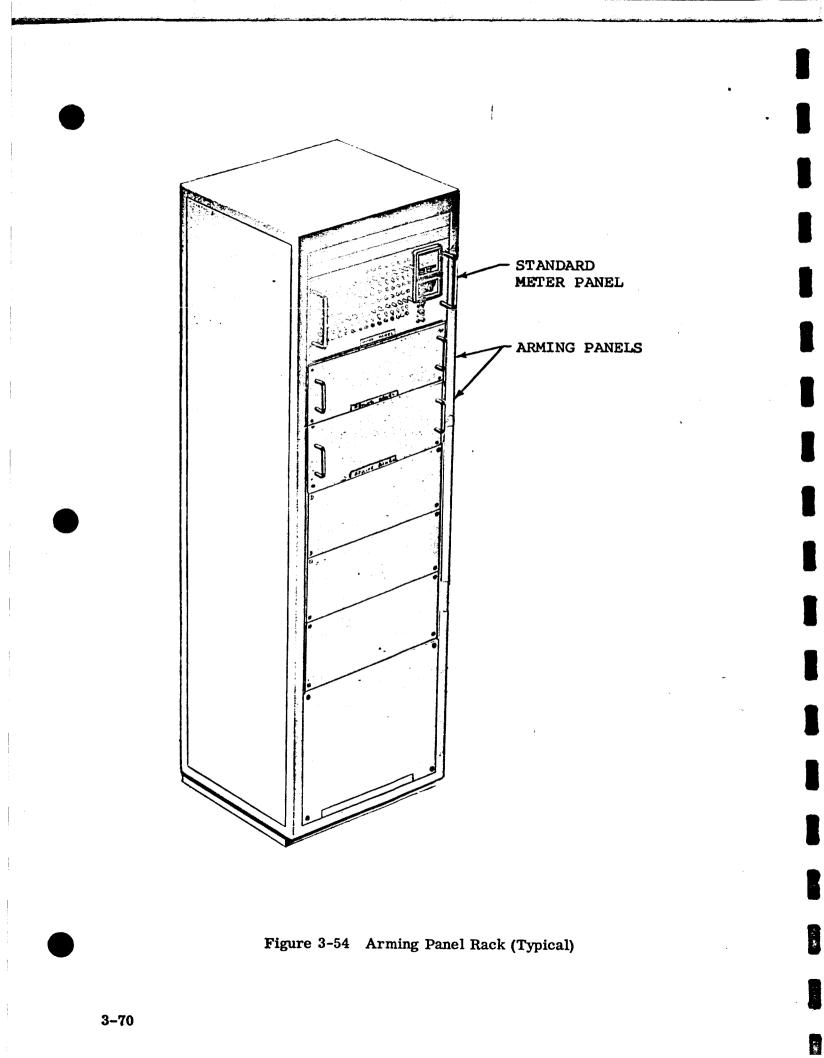
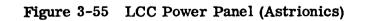


Figure 3-52 Battery Rack (Typical)





TO BE PROVIDED



3-47. S-IC ENGINE SERVICING PLATFORMS SUBSYSTEM.

3-48. <u>Purpose of the Subsystem</u>. The S-IC Engine Servicing Platforms Subsystem controls the hoist assembly required to position any of the S-IC engine servicing platforms.

3-49. <u>Electrical Equipment for the Subsystem</u>. The S-IC Engine Servicing Platforms Subsystem uses the following equipment:

- a. Relay Distributor #6606, room 2-A
- b. Deck Distributor #6000, Level 0
- c. Platform Distributor #7201 and #5089, Servicing Platform
- d. Platform Controller #6000A1, Level 0
- e. Level Sensor #7200A3 and 5088A3 Servicing Platform
- f. Motor/Brake/Winch Assemblies #7853, #7852, #7851 and #7850, level 0
- g. Warning Buzzer #7200A4 and #5088A4
- h. Top Limit Switche #7200A8, A9, A12, A13 and #5088A8, A9, A12, A13
- i. Load Detector #7200A10, A11 and #5088A10, A11
- j. Interlock Distributor #7200A1, A2 and #5088A1, A2

3-50. <u>Description of the Subsystem</u>. Three separate platforms are required at the VAB and launch pad for servicing the Saturn S-IC stage:

a. The Mobile Launcher Level Servicing Platform is a passive platform which provides servicing access to the S-IC stage at deck level 0.

b. The S-IC Engine Servicing Platform (Pad) services the engine area while the Mobile Launcher is at the launch pad. This platform has level sensors, interlocks, and other control equipment not contained in the Mobile Launcher Level Servicing Platform, and offers greater load capacity.

c. The S-IC Engine Servicing Platform (VAB) is used to service the engine area, install engine skirts or change an F-1 engine while the Mobile Launcher is in the VAB. It has the same control equipment as the S-IC Engine Servicing Platform (Pad), and still greater load capacity. See figure 3-57.

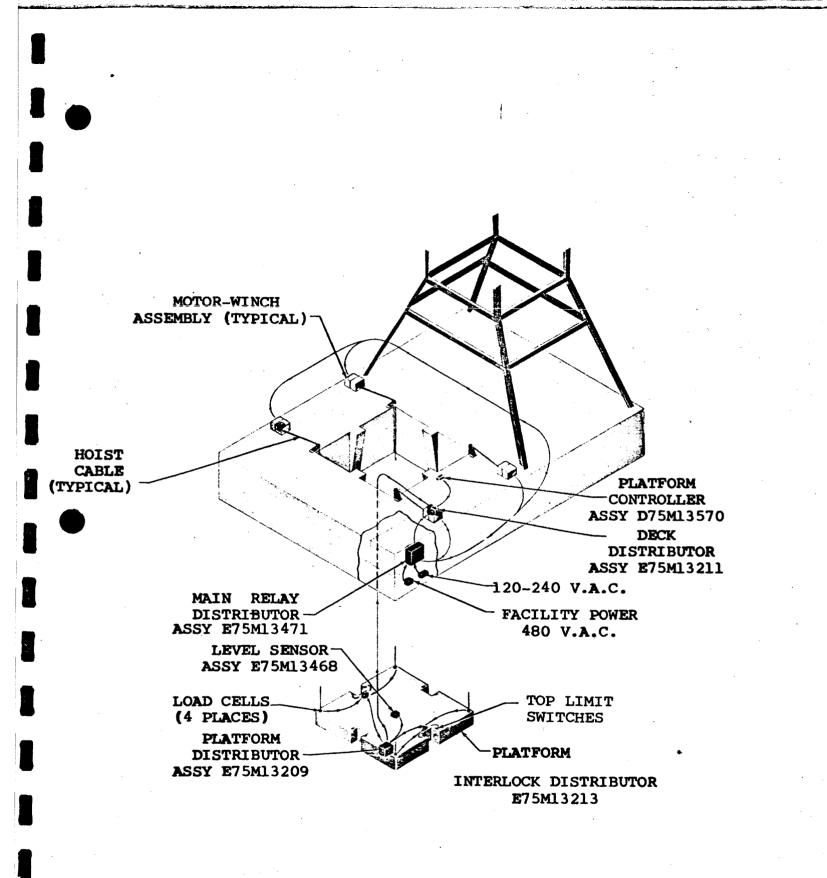


Figure 3-56 S-IC Engine Servicing Platforms Subsystem

[]

Π

These three platforms, represented by a single generalized platform in figure 3-56, are raised and lowered by a cable-hoist mechanism which is locally controlled by an operator on the Mobile Launcher deck. The electrical control system for this hoist is called the S-IC Engine Servicing Platforms Subsystem. See figure 3-58.

This subsystem normally employs four motor/winch assemblies operating simultaneously, giving it the capacity to lift a balanced platform loaded with one inboard F-1 engine. Step-control is achieved with two-speed motors.

Electrical power is also provided for a removable fifth motor/winch assembly. This assembly will provide additional hoist capacity for any platform corner raising an outboard F-l engine.

The motor/winch assemblies are supported by several special equipments. These equipments and their functions are as follows:

a. The Relay Distributor (room 2-A) provides control logic and power to the motors and control system. This distributor uses direct cabling from the distributor to the motors and control elements. It is shown in figure 3-59.

b. The Deck Distributor (figure 3-60) located inside motor/winch housing "D" provides the means for connecting the platform controller to the system.

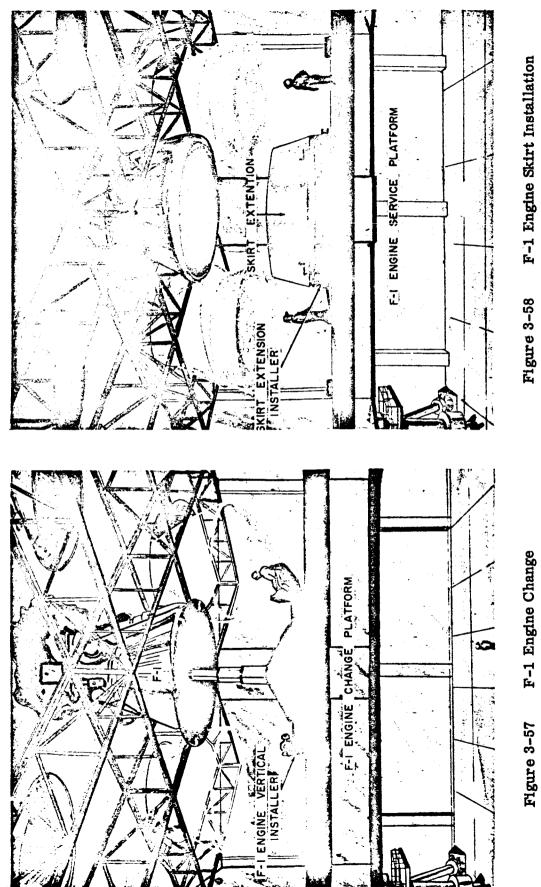
c. The Platform Controller, which is connected to the system via the deck distributor, provides on-off, speed, and direction control of the motors. The motors may be controlled either individually or all together. See figures 3-61 and 3-62.

d. The Platform Distributor routes interlock signals and distributes power to various platform locations.

e. The Level Sensor (figure 3-60) detects tilt of the platform and sends this information to the platform controller via the platform and deck distributors. The information is monitored by the operator for manual corrections.

f. A maximum-height interlock and cable-loading indicators are incorporated into the overall system. All exposed equipment is designed for protection against explosions and RFI.

Finally, it should be noted that the Platform Distributor, the Level Sensor, the maximum-height interlock, and the cable-loading indicators are not used on the Mobile Launcher Level Servicing Platform. This platform is positioned by an operator using the platform controller without signals from the platform itself.



· · · ·

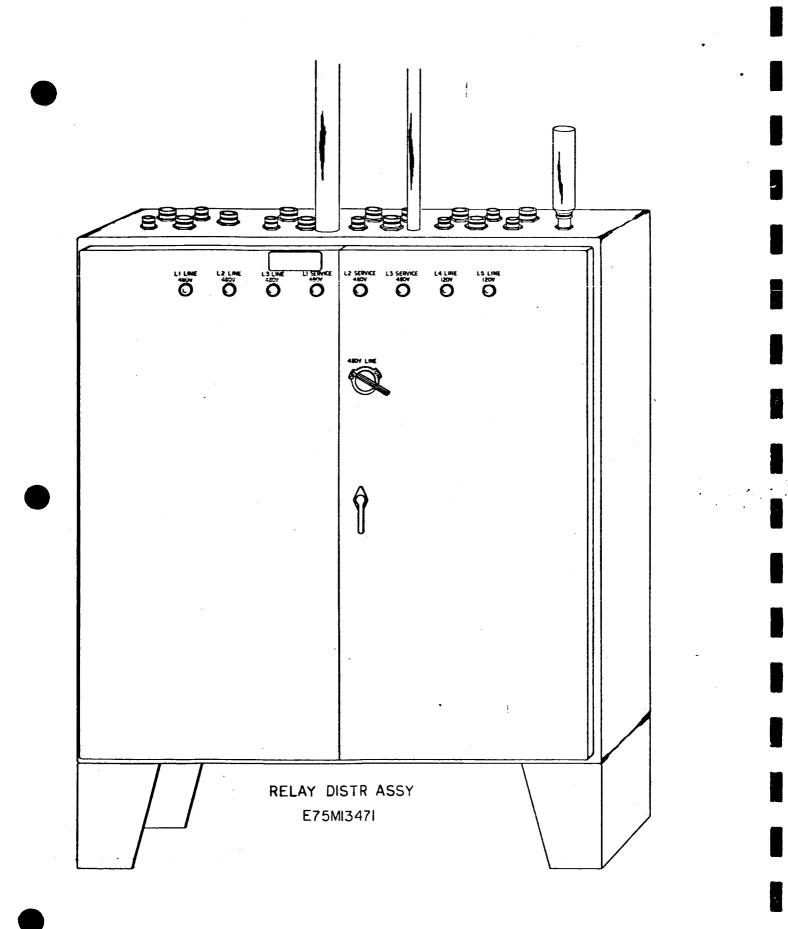
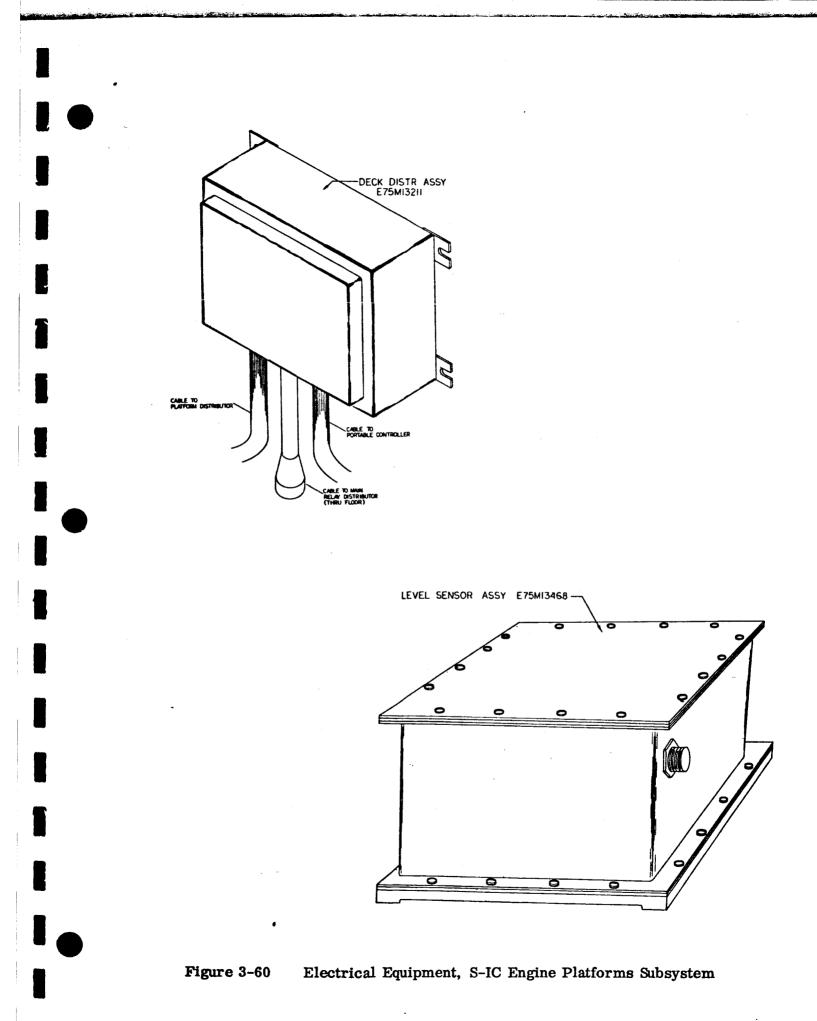


Figure 3-59 Relay Distributor, S-IC Engine Platforms Subsystem



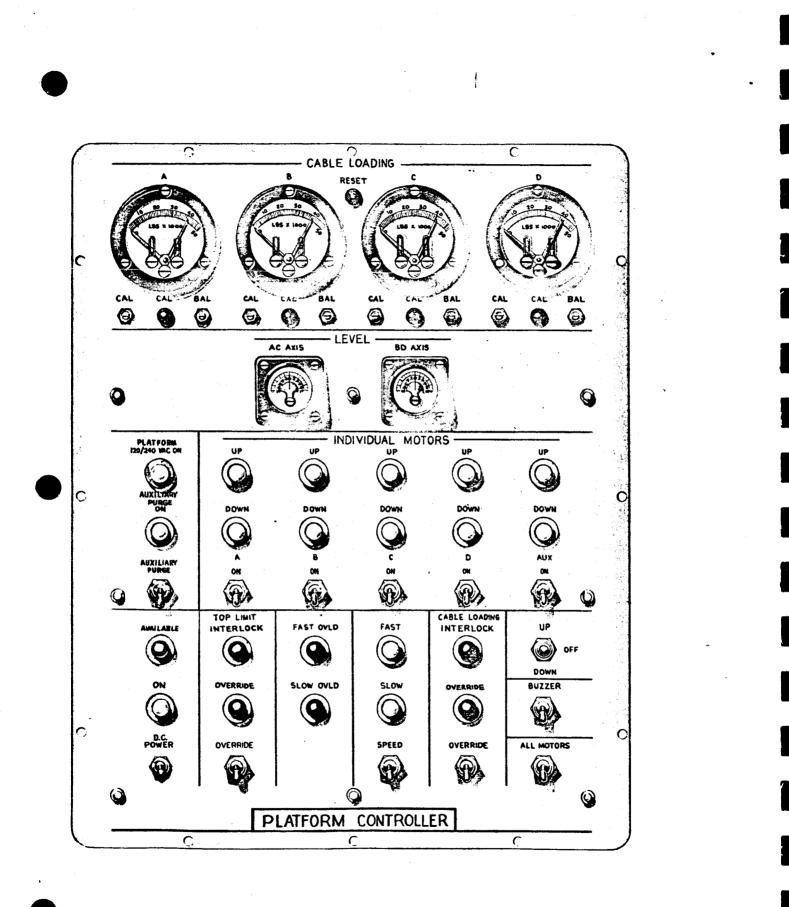


Figure 3-61 S-IC Engine Servicing Platform Controller E75M13570

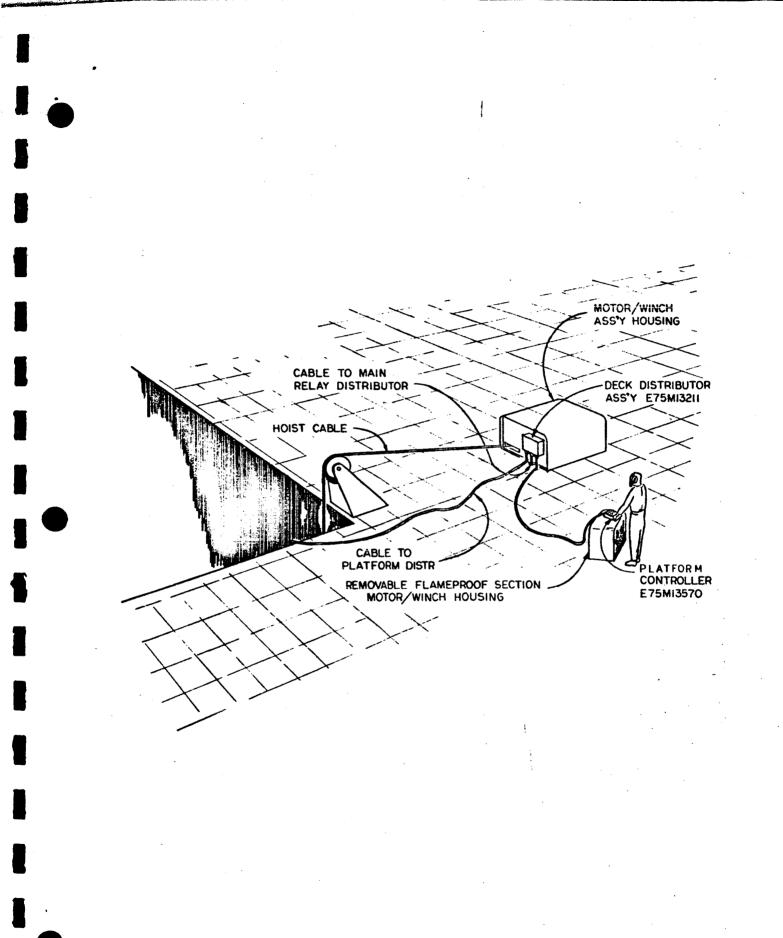


Figure 3-62 Typical Installation, Platform Controller

3-51. PLATFORM TRANSPORTER SUBSYSTEM.

3-52. Purpose of the Platform Transporter Subsystem. This equipment controls movement of the F-1 engine servicing platform transporter on the launch pad.

3-53. <u>Electrical Equipment Employed</u>. The following electrical equipment is included in the transporter subsystem:

a. Electric Motors (4)

b. Distribution Boxes (4)

c. Portable Controllers (2)

3-54. <u>Description of the Subsystem</u>. The Platform Transporter is a rail-borne carriage which positions the F-1 Engine Servicing Platform under the Mobile Launcher. It rides just above the launch pad flame trench.

The transporter is pulled underneath the Mobile Launcher by a pair of winch assemblies (figure 3-63) at the north end of the platform transporter rails. These two winches normally operate simultaneously, with a balanced load. Load balance is achieved by use of a single cable which is strung continuously between the winches, through pulleys on the transporter. Both winches are controlled from a single portable controller.

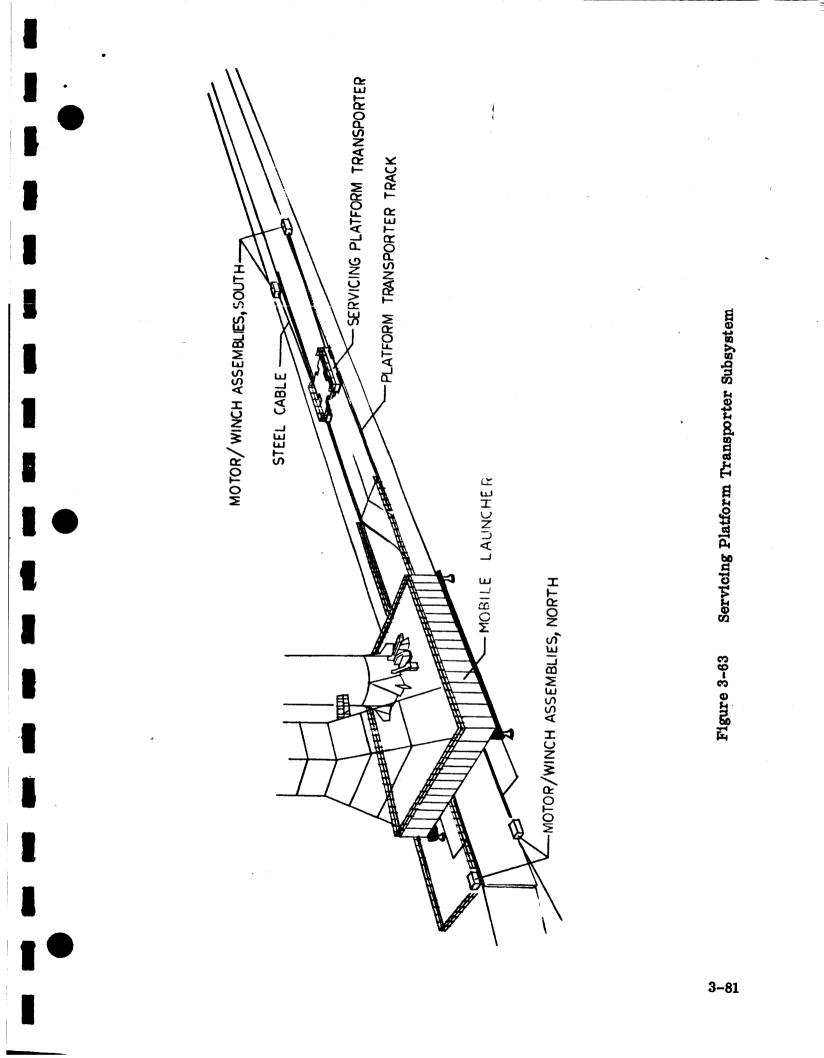
A similar pair of winches at the south end of the rails withdraws the transporter from the Mobile Launcher.

A typical winch assembly is shown in figure 3-64. It is driven by a 3 phase, 480 volt, induction motor which is totally enclosed and fan-cooled.

The motor is controlled through a Nema type 4 distribution box equipped with a motor circuit switch, a starter contactor, a transformer for the contactor, and three overload elements.

The motor circuit switch is for quick make and break operation. It is manually operated by a handle on the front of the distributor.

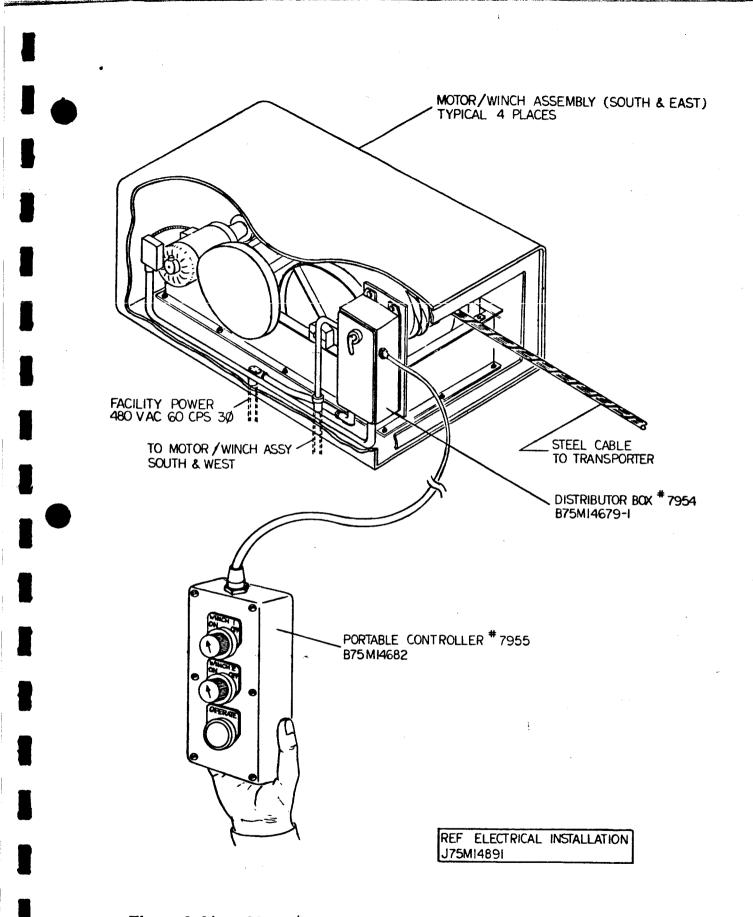
The starter contactor is a Nema type 2 contactor for carrying 480 volt, 60 cycle power. Its magnetic coil is operated by 120 volts, 60 cycles and is readily accessible for quick change. Power for the coil is provided by a stepdown transformer mounted inside the contactor enclosure.

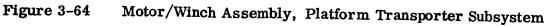


Three overload elements in the distributor provide protection for the 3-phase motor. An overload on any line to the motor opens the starter contactor, thereby stopping the motor.

Control for the distributor box is provided by a portable controller, figure 3-64, which is capable of operating either of two winches individually or both simultaneously. One controller is used in conjunction with the north winches and another with the south winches. Each controller is encased in heavy-duty watertight stainless steel and weighs about ten pounds. It has magnetic reed-type pushbutton switch and two hand-operated selector switches equipped with watertight caps. Each controller operates under 120 volt, 60 cps power with a minimum contact rating of 2.0 amperes.

On launch pad A, the control cable interconnecting the two north-end motor winch assemblies is suspended over the flame trench by a steel rope. On future launch pads this control cable is carried in conduit through the concrete pad structure. All other control cables and all power lines are permanently installed in embedded conduit.





3-55. S-II AND S-IVB ENGINE SERVICING PLATFORM SUBSYSTEM.

3-56. <u>Purpose of the Subsystem</u>. This subsystem provides electrical control for raising and lowering the S-II and S-IVB engine servicing platform.

3-57. Electrical Equipment for the Subsystem. The S-II and S-IVB Engine Servicing Platform Subsystem includes the following equipment:

a. Motor/winch assemblies 7204A1, A2, A3 and A4.

b. Electrical junction boxes 7205, 7206, 7207, and 7208.

c. Distribution box 7204.

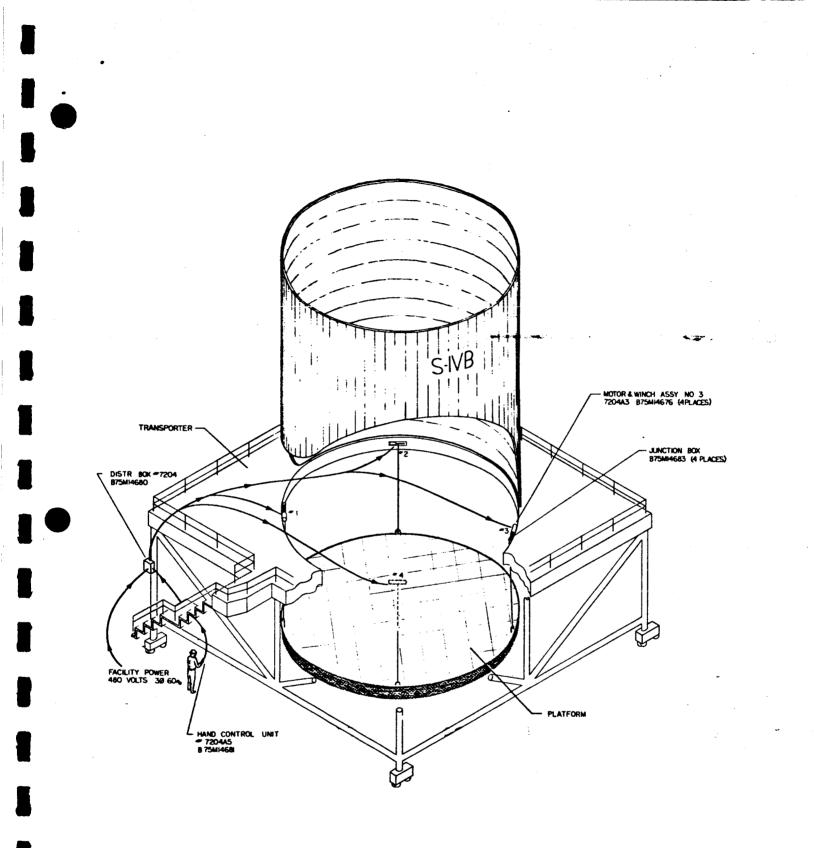
ૼ

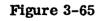
d. Hand control unit 7204A5.

3-58. <u>Description of the Subsystem</u>. In the Vehicle Assembly Building, a special transporter (figure 3-65) is provided for Saturn V stages S-II and S-IVB. The vehicle stage is positioned on this transporter over a circular cutout. Personnel then gain access to the stage engines by means of a vertically movable platform, which is controlled by the electrical system herein called the S-II and S-IVB Engine Servicing Platform Subsystem.

The platform is raised and lowered by four winches, figure 3-66. Each winch is driven by one squirrel-cage induction motor which is totally enclosed and fan-cooled. This 3-horsepower motor is operated by 3 phase, 480 volt, 60 cps power. Full load current at 3500 rpm will not exceed 4.5 amperes. It is equipped with one 480 volt single-phase space heater that is capable of maintaining motor temperature at 5 to 8 degrees above ambient. Limit switches are provided to override motor operation when reel-in or reel-out limits are reached.

The electrical junction box attached to each motor/winch assembly has a Nema type 4 watertight enclosure. It is equipped with three overload elements (one for each phase) and a non-fused ac reversing starter. The starter is a commercial Nema size 0, open-type, with contacts for carrying 400 volts, 60 cycles. Its 120 volt, 60-cycle magnet coil is easily accessible for quick change.





S-II and S-IVB Engine Servicing Platform Subsystem

Voltage for this coil is supplied by a step-down transformer in Distribution Box #7204. The 0.5 kva, single phase transformer receives 480 volt facility power and has two 115 volt outputs for the motor heaters and motor-starter coils. Distribution Box #7204, a Nema type 4 watertight enclosure, also has terminal blocks for distributing power to the transformer, motors, and hand control unit.

Operation of the four winches is actuated and controlled by means of hand control unit No. 7204A5. This unit has a separate select switch for each winch so that any combination of switches can be operated simultaneously. REEL-IN and REEL-OUT commands are actuated by pushbutton switches. All switches are magnetic reed type. The unit weighs approximately 10 pounds. See figure 3-66.

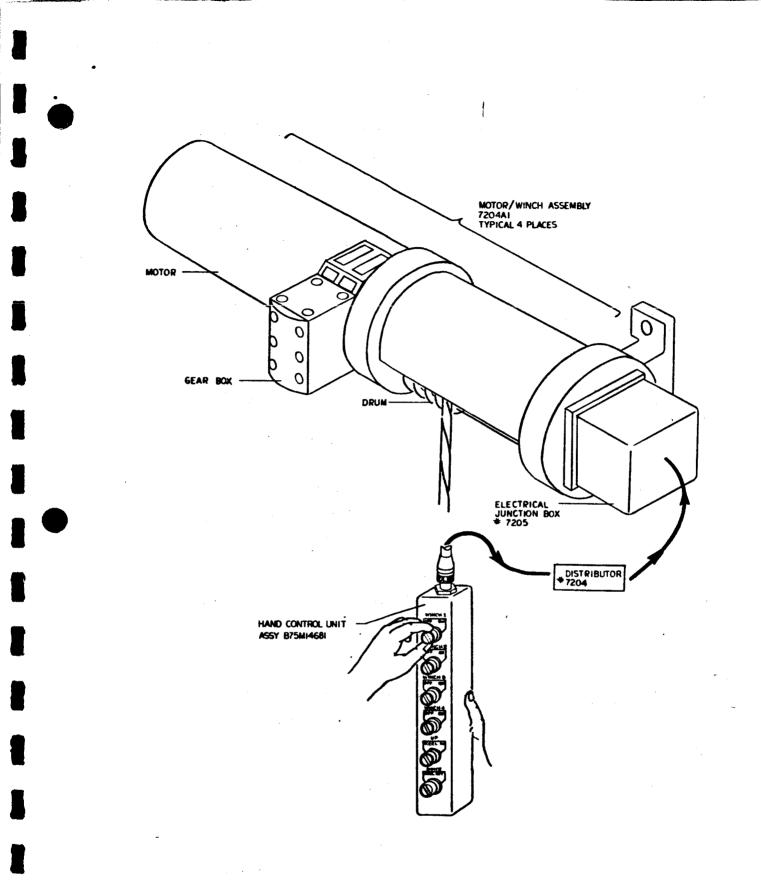


Figure 3-66 Motor/Winch Assembly B75M14676

3-59. LAUNCH EQUIPMENT CONTROL AREA, LCC

In the Launch Control Center sixteen consoles (figure 3-67) are assigned for control, test, and monitor of eight of the twelve ESE subsystems. These eight subsystems and their corresponding control panels are:

> a. Service Arms Subsystem S-IC Intertank thru Service Module (eight panels). b. Apollo Access Arm Subsystem **Command Module** c. Pneumatic Control and Q-Ball and Pneumatic **Distribution** Subsystem Distribution d. Q-Ball Cover Removal. **Q-Ball and Pneumatic** Subsystem Distribution e. Hydraulic Charging Unit Hydraulic Charging Unit Subsystem

f. Tail Service Masts Subsystem

g. Launch Equipment Firing Circuits Subsystem TSM 1-2, TSM 3-2, TSM 3-4

Holddown Arms & Purge Valves and Service Arm Control Switches

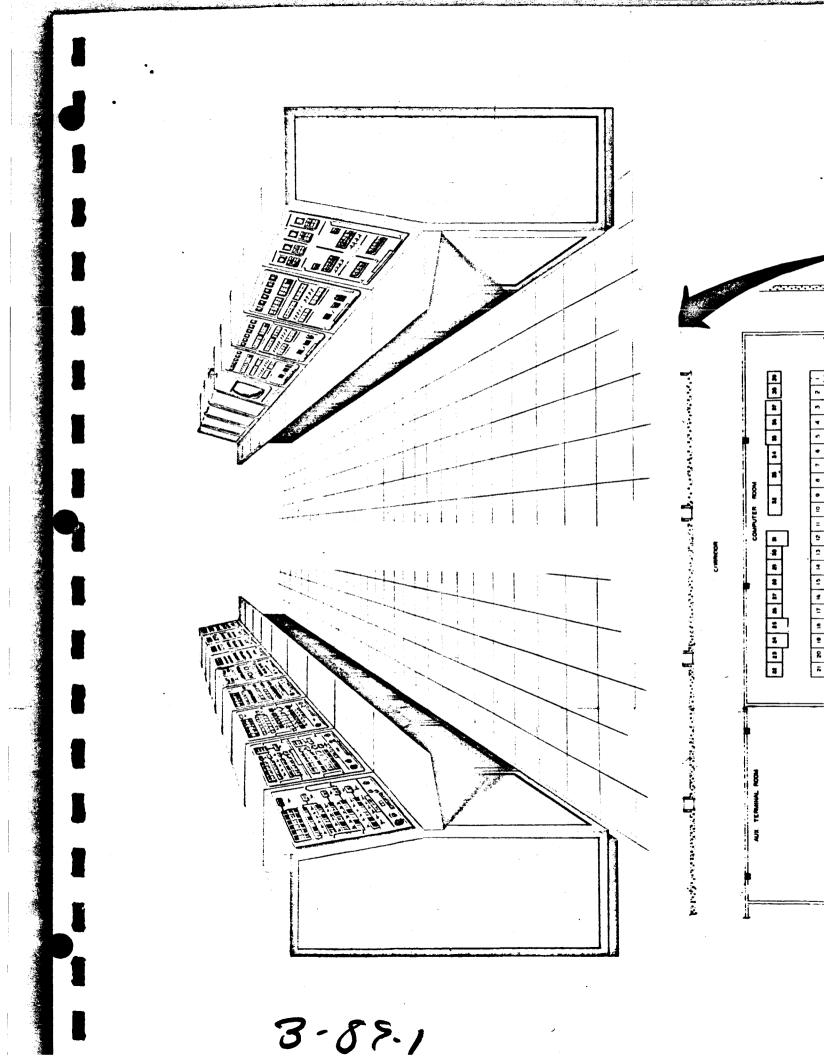
h. ESE Power Subsystem

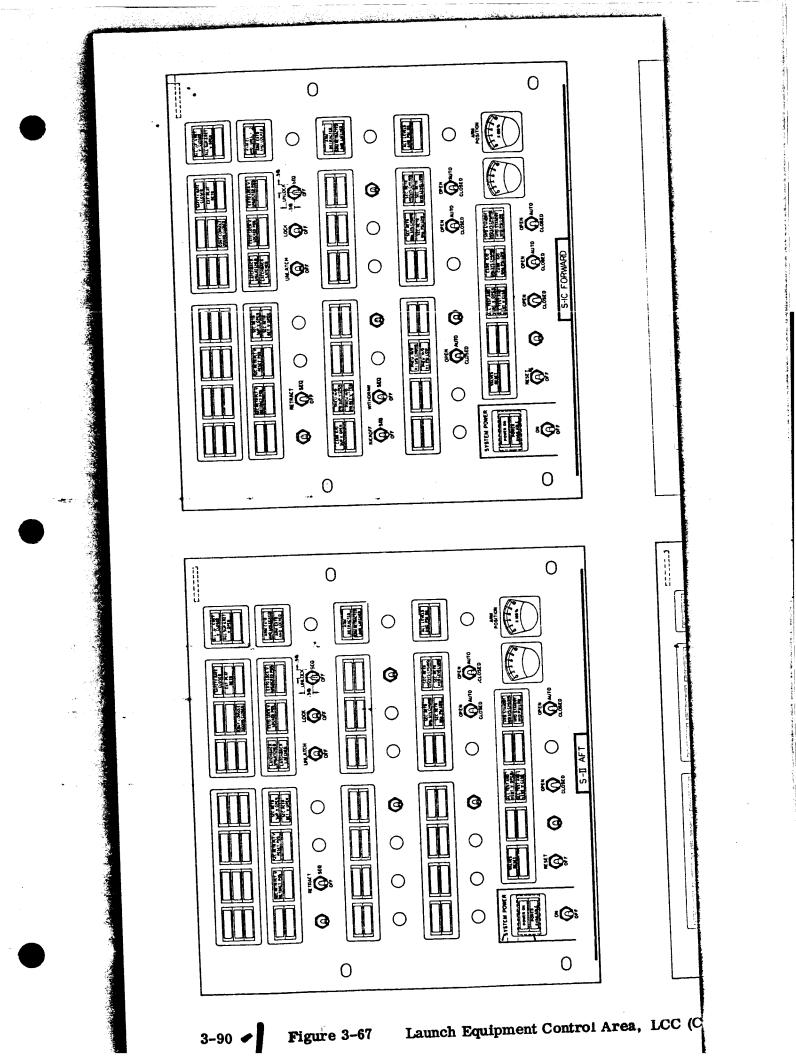
Launch Equipment Power Panel (Astrionics)

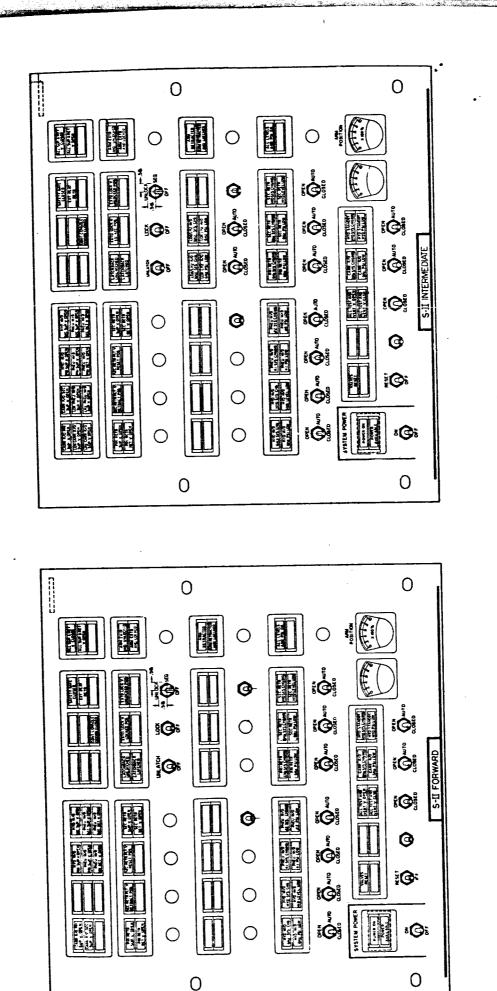
The Status Panel provides monitoring for the eight Service Arms, the Command Module, Pneumatic Distribution, the Hydraulic Charging Unit, the three Tail Service Masts, the Holddown Arms and the Service Arm Control Switches.

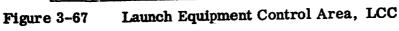
Control, test, and monitor signals for these subsystems are transmitted between the LCC and Mobile Launcher via the DDAS, Mobile Launcher/LCC computers, and hardwire installations.

The basic enclosure for these LCC consoles is described in paragraph 4-102.

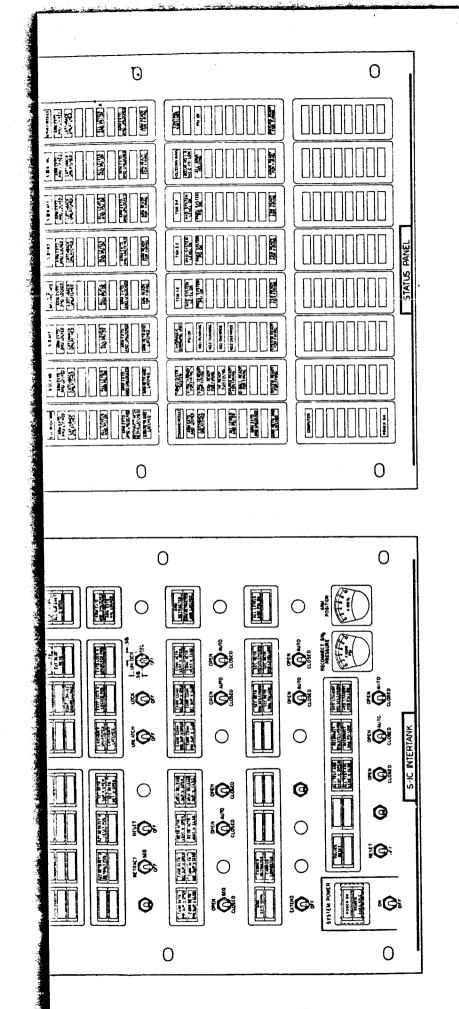








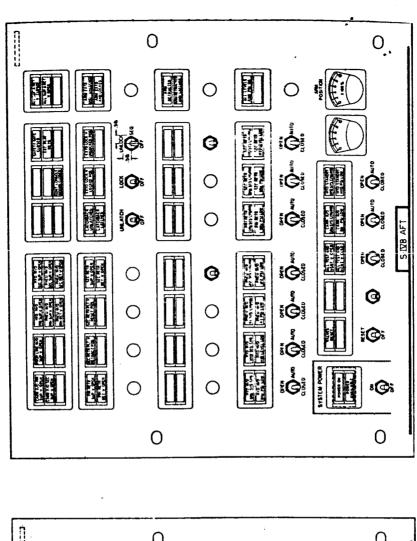


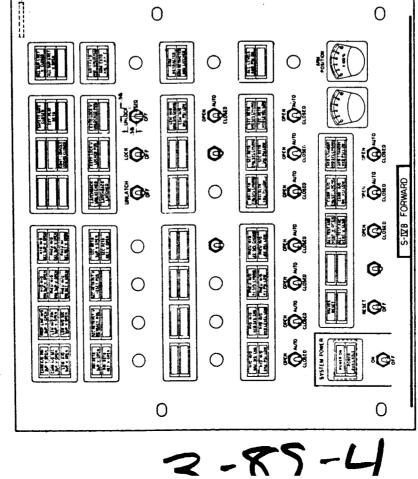


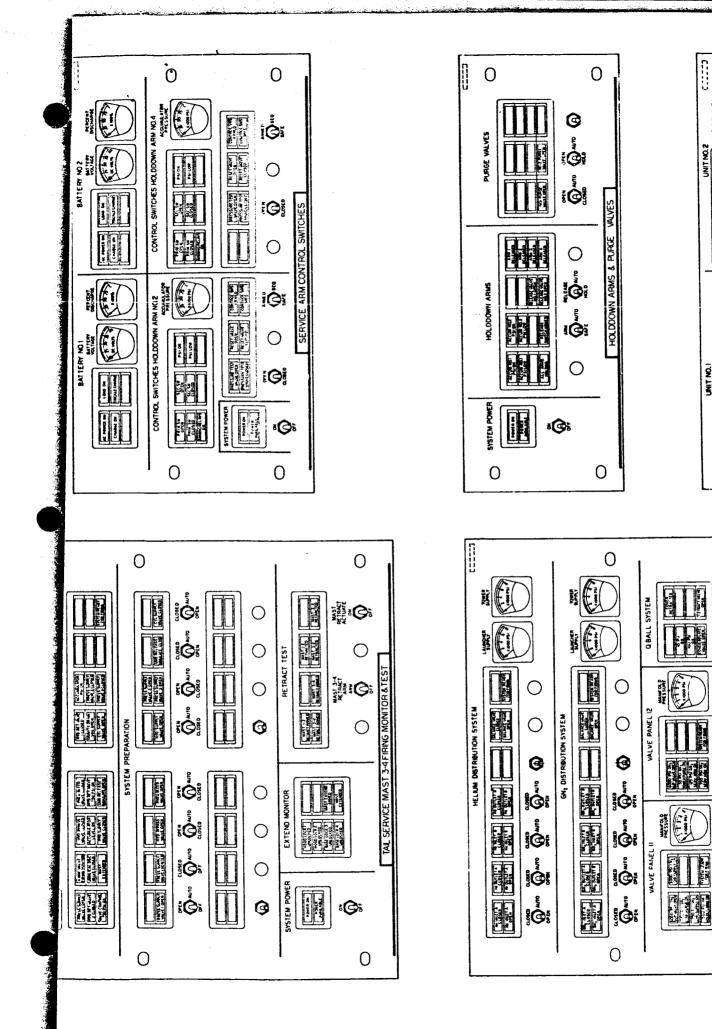
FIRMS MONTOR MAY FORME WAY FORME MAY FORM

ontinued)

3.90 - 2







UNIT NO.2

C

INNE OF

THE PACEMENT

(

ŝ

ŝ

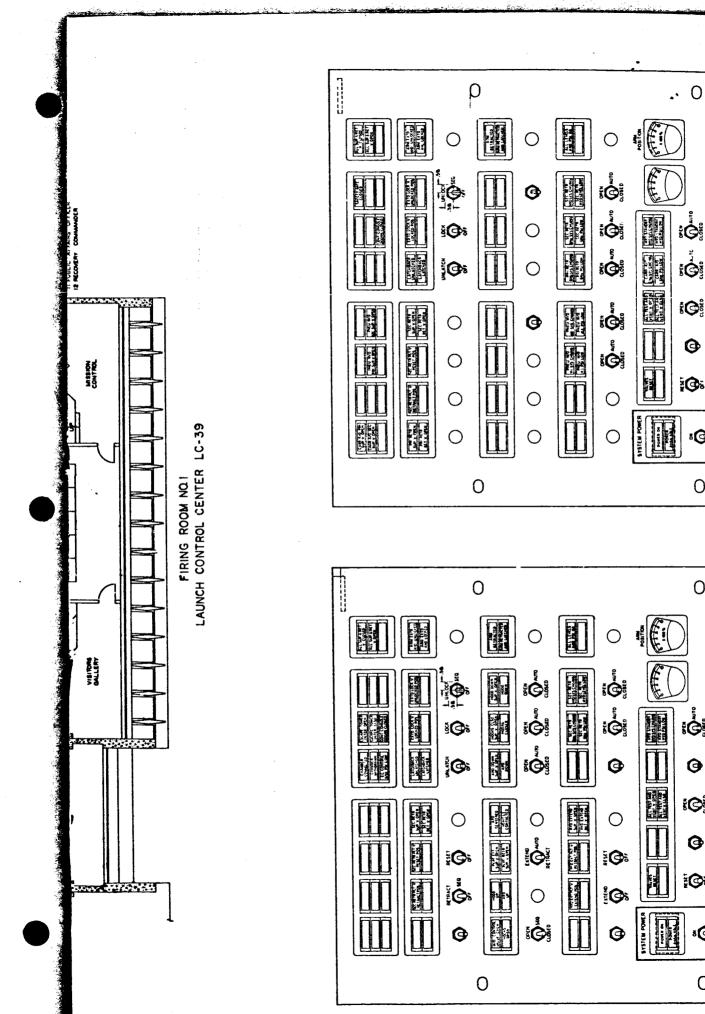
С

 \bigcirc

Ο

Ο

UNIT NO.



0

io:

i©i

0

2**0**%

5**0**5

0

0

io!

0

(C)

0

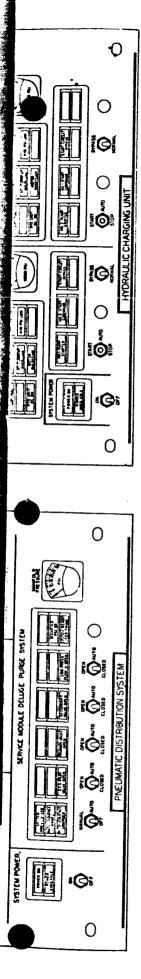
: ©:

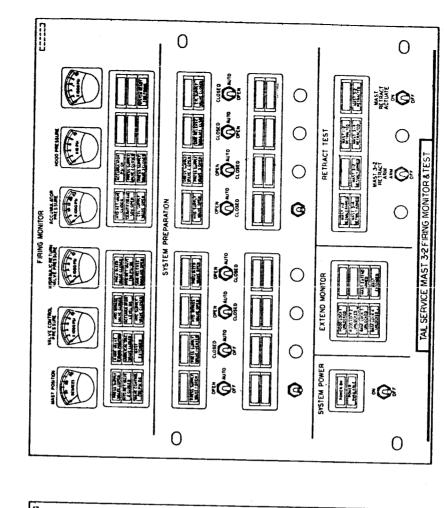
***@**`

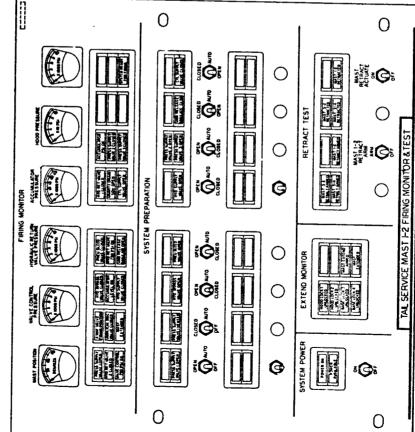
0

MODULI

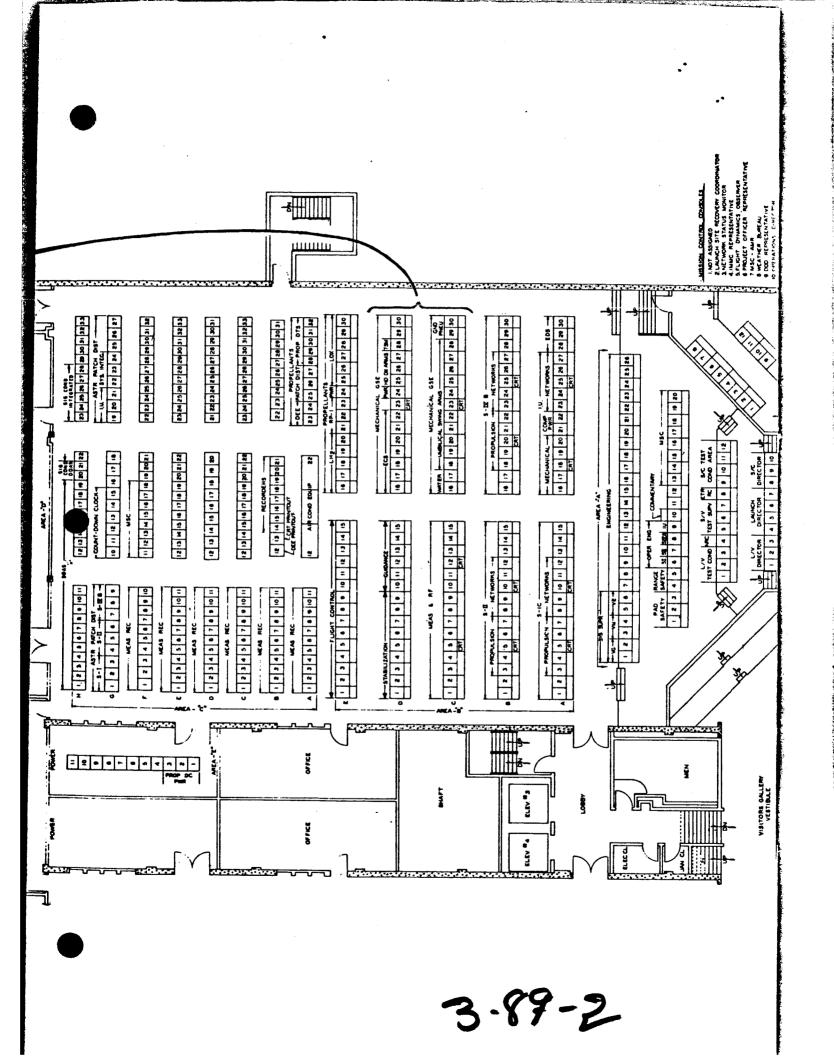
SERVICI







3-50-4



SECTION IV

ESE STANDARD COMPONENTS

4-1. GENERAL.

The standard Launch Equipment Branch assemblies shown in figure 4-1 are individually delineated in paragraphs 4-2 through 4-116 following.

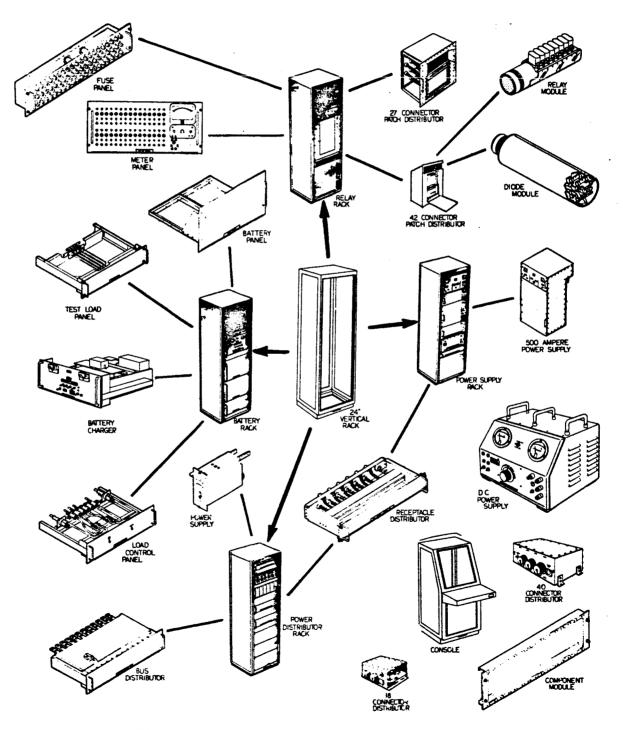


Figure 4-1 Standard Assemblies for ESE

4-2. STANDARD ELECTRICAL DISTRIBUTOR E75M07835.

4-3. <u>Description of Distributor</u>. Standard Electrical Distributor E75M07835 (figure 4-2) is a pressurizable enclosure 7.5 inches high, 18 inches wide, and 12 inches deep. It contains forty 7-pin connectors (Cannon BFR16S-1S or equal), one 5-pin connector (Bendix 71-323940-75P or equal) and two 61-pin connectors (Bendix 71-323940-73P or equal). All connectors are wired to terminal boards using standard solder terminals (Cambion X1782-D or equal), and patching between terminal boards is required to complete the interface between input and output connectors.

This basic distributor is normally mounted with angle brackets, but is also available for mounting in a standard rack panel.

4-4. <u>Use of Distributor</u>. Distributor J75M07835 is used for electrical control of the Mobile Launcher service arms, tail service masts, high pressure pneumatics system, and holddown arms. At each location it gathers electrical lines from individual components and consolidates them into two cables for further transmission. The distributor also contains positive and negative power buses for supplying power to relays, solenoid-operated valves, and other dc components within the service arms and masts.

4-5. <u>Locations of Distributor</u>. Distributor J75M07835 appears in approximately 28 places in the Launch Equipment Branch ESE. The distributors, although not precisely located in the illustration, are:

a. Contr	ol Distributors 1, 2, 3	Service Arms Subsystem See figures 3-2 and 3-3
b. Tail S	Service Mast Distributor	Tail Service Masts Subsystem, See figure 3–31
c. Holdd	lown Arm Control Distributor	Launch Equipment Firing Circuits, Room 12–A See figure 3–22
-	Pressure Pneumatics ol Distributor	Pneumatics Control and Distribution Subsystem See figure 3-37

4-6. <u>Design Organization</u>. Distributor J75M07835 was designed by the Electrical Section, Launch Equipment Branch, KSC.

1.8

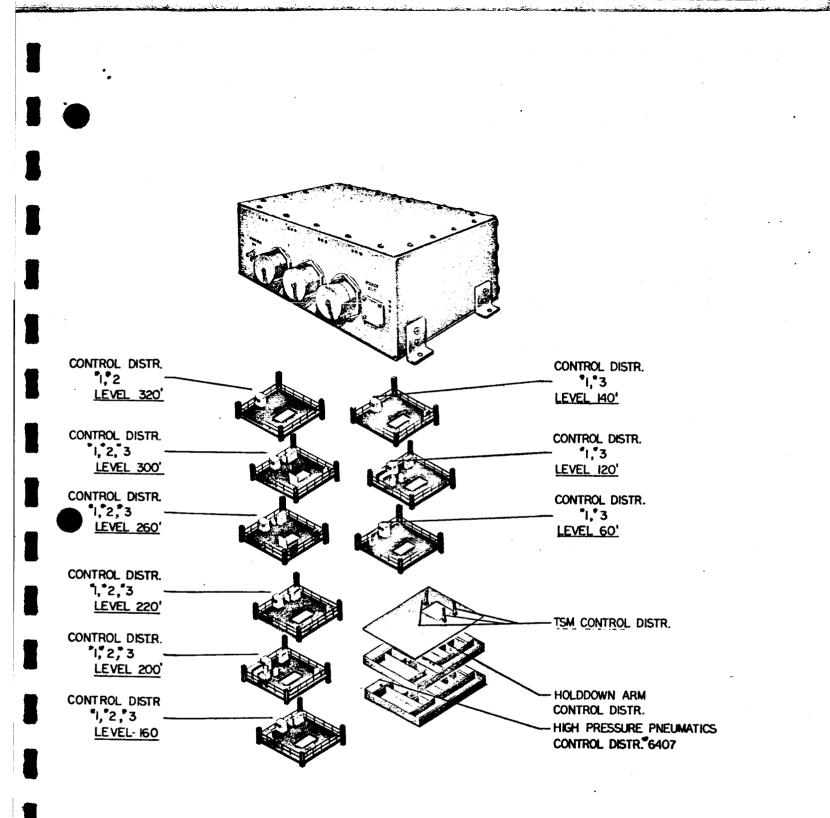


Figure 4-2 Standard Electrical Distributor J75M07835

4-7. STANDARD ELECTRICAL DISTRIBUTOR E75M09808.

4-8. <u>Description of Distributor</u>. Standard Electrical Distributor E75M09808 (figure 4-3) is a pressurizable stainless steel unit 7.5 inches high, 11.5 inches wide, and 12.75 inches deep. On the back are eighteen 7-pin connectors (Cannon BFR16S-1S or equal). On the front are a 5-pin connector (Bendix 71-323940-75P or equal) and a 61-pin connector (Bendix 71-323940-73P or equal). All connectors are wired to terminal boards using standard solder terminals (Cambion X1782-P or equal), and patching between terminal boards is required to complete the interface between input connectors.

This distributor is outfitted with anchor nuts and angle brackets for cabinet or panel mounting.

4-9. <u>Use of Distributor</u>. Distributor E75M09808 is used in the Water Glycol Control System, Service Arm Control Switch Test System, Q-Ball and Deluge Purge, Valve Panel No.11 and Valve Panel No. 12. At each location it gathers electrical lines from individual components and consolidates them into one cable for further transmission. The distributor also contains positive and negative power buses for supplying power to relays, solenoid-operated valves, and other dc components within the systems it serves.

4-10. Locations of Distributor. Distributor E75M09808 appears in approximately six places in the Launch Equipment Branch ESE. See figure 4-3.

	a.	Valve Panel 11 Distributor	Pneumatics Control and Distribution. See figure 3-38
10 10 10 10 10 10	b.	Valve Panel 12 Distribitor	Pneumatics Control and Distribution. See figure 3-40.
	c.	Service Arm Control Switch Test Distributor	Firing Circuits. See figure 3-26.
	d.	Water Clycol Control Distributor	
	e.	Q-Ball Purge Control Distributor	Q-Ball Cover Removal Subsystem. See figure 3-42.
***	f.	Deluxe Purge Control Distributor	Pneumatics Control and Distribution. See figure 3-35.
4-11.	. Design Organization. Distributor E75M09808 was designed by the Electrical		

Section, Launch Equipment Branch, KSC

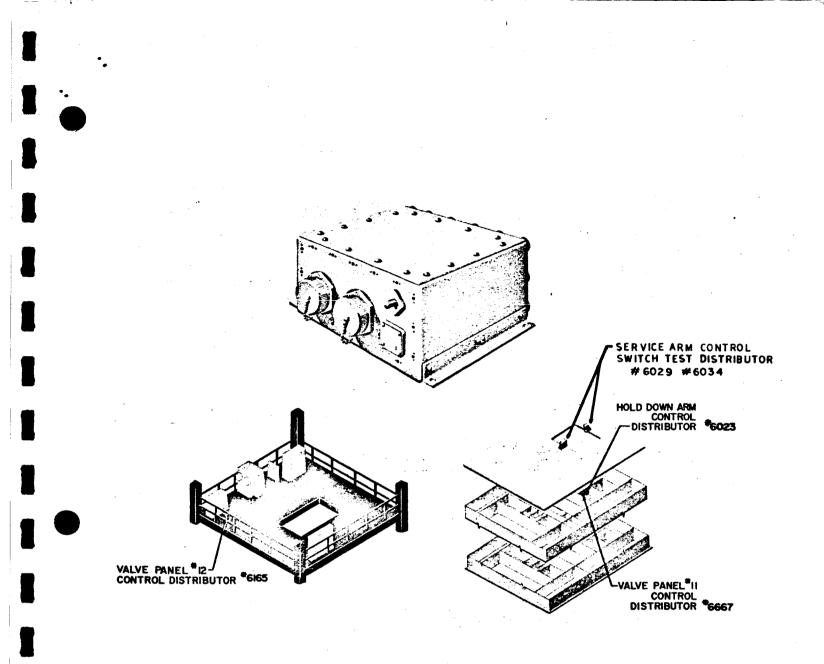


Figure 4-3 Standard Electrical Distributor J75M09808

.

4-5

4-12. PORTABLE DC POWER SUPPLY D75M51116.

4-13. Description of Supply. Portable Power Supply D75M51116 (figure 4-4) is packaged in a hand-carry case 13 inches high, 14 inches wide, and 13 inches deep. It provides unregulated dc in the ranges 0-18 volts or 18-36 volts, controlled by a range select switch and rheostat. Maximum output load is 15 amperes. Both voltage and load are displayed on front-panel meters.

Output power is supplied from front and rear binding posts, or from a rearmounted cable connector (MS3102E-22-2S). Input power, 120 volts, 60 cps, is supplied through a rear-mounted standard 3-pin ac panel connector.

A positive and negative ground-detector circuit is actuated by a switch on the front panel.

4-14. <u>Use of Supply.</u> Power Supply D75M51116 can be easily hand carried about the Mobile Launcher to provide dc power where requirements do not exceed 36 volts, 15 amperes and 1 volt ripple/noise. It is used to provide power to the Tail Service Mast Test Set, Launcher Accessories Test Set No. 1, Launcher Accessories Test Set No. 2, and Tower Test Set.

4-15. Locations of Supply. See figures 3-22, 3-26, 3-34, 3-37, 3-38, 3-40.

4-16. <u>Design Organization</u>. Power Supply D75M51116 was designed by the Electrical Section, Launch Equipment Branch, KSC.

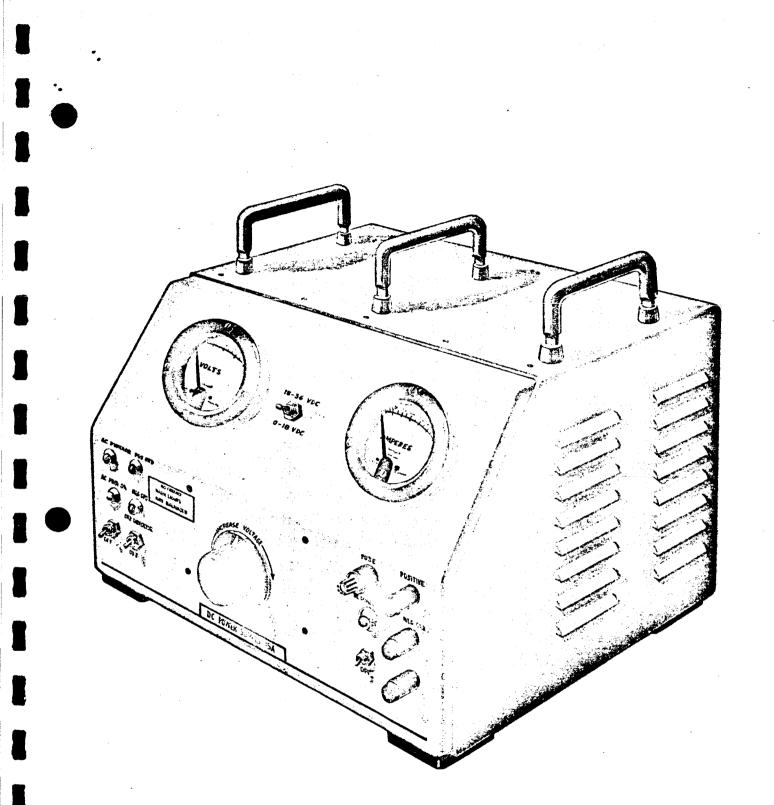


Figure 4-4 Portable DC Power Supply D75M51116

_

4-17. STANDARD 24 INCH VERTICAL RACK E64-KN-F-260-10.

4-18. <u>Description of the Vertical Rack</u>. Standard 24 Inch Vertical Rack E64-KN-F-260-10 is an R.F.I. shielded enclosure 86.25 inches high, 26.06 inches wide and 30.56 inches deep. It is designed to house standard 24 inch component panels. See figure 4-5.

Panels may be mounted either front or rear, and may be individually shielded. The rack base, side panels, and cable entry (at the enclosure bottom) are removable.

4-19. <u>Use of the Vertical Rack</u>. Standard 24 inch racks for the ESE house seven types of assemblies:

a. Nine service arm relay racks, figure 3-53.

b. Two TSM & launcher accessories relay racks containing an internal fuse panel in lieu of Fuse Panel J75M10163.

c. Four standby battery racks and firing batteries, figure 3-52.

d. Three power distribution racks, figure 3-51.

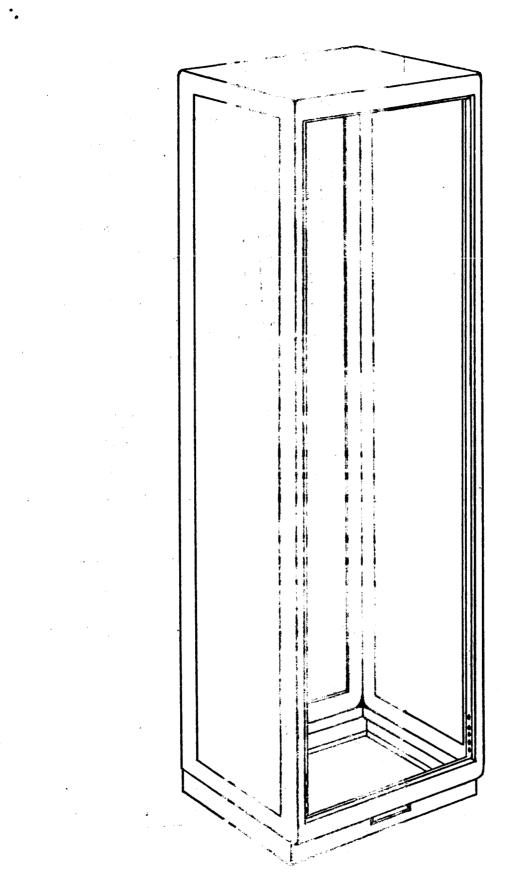
e. Two power supply racks (one spare), figure 3-50.

f. One arming panel rack, figure 3-54.

g. Four integration racks and two cable entrance racks. Spare racks will be assigned as required.

4-20. Locations of Vertical Rack. The ESE employs approximately twenty-four racks in Mobile Launcher room 8-A and three in the LCC. See figure 6-2.

4-21. <u>Design Organization</u>. The Standard 24 Inch Vertical Rack is commercial equipment purchased to KSC/MSFC specifications.



R

Figure 4-5 Statement 24 Juch Vertical Rack E64-KN-F-260-10

4-22. STANDARD PATCH DISTRIBUTOR F75M04681 (42 CONNECTOR).

4-23. <u>Description of the Distributor</u>. Standard Patch Distributor F75M04681 is an enclosure 12.5 inches deep, 36.73 inches high and 24 inches wide. It may be mounted in a standard 24 inch rack. On the back of the enclosure, mounted on two hinged plates, are forty-two 61 pin connectors (Bendix PT0SE-24-61P or equal).

Patching between these connectors is provided by the following patchboard assembly, which is accessible through the front panel door:

- a. Frame & Spring Assembly (Anderson Electric Co. BT 2560 or equal)
- b. Removable 2560-pin patchboard (Anderson Electric Co. IPB2560S or equal)

Above the front panel door is a row of 84 taper-pin blocks. These are two types, mounted alternately; 42 blocks (AMP 480107-6 or equal) and 42 blocks (AMP 480107-9 or equal). Each taper-pin block has 30 female test jacks wired to the patchboard.

4-24. <u>Use of the Patch Distributor</u>. The standard patch distributor is employed in various 24-inch relay racks used in the ESE. In these locations, the distributor provides monitoring and flexible distribution of signals. It also accommodates standard relay modules (paragraph 4-37) and standard diode modules (paragraph 4-42), which may be patched into the ESE systems as required.

4-25. Locations of Patch Distributor. The ESE employs approximately eleven standard patch distributors located in the relay racks, room 8-A. See figure 4-6.

4-26. <u>Design Organization</u>. Patch Distributor F75M04681 was designed by MSFC-Astrionics and adopted by KSC-Launch Equipment Branch.

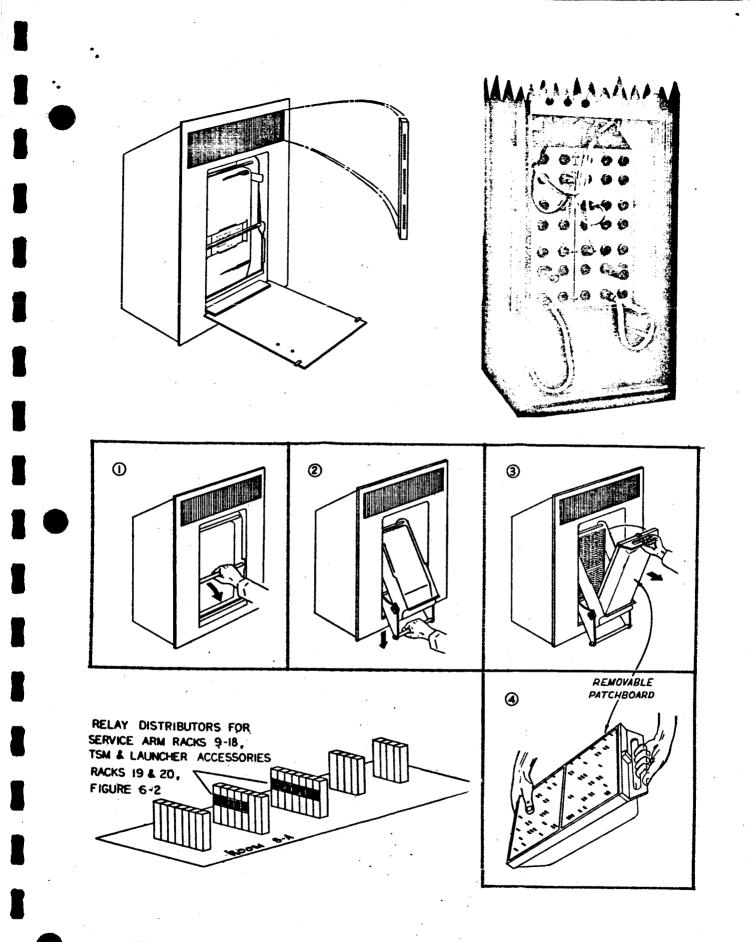


Figure 4-6

Standard Patch Distributor (42 Connector) F75M04681

4-27. STANDARD PATCH DISTRIBUTOR F75M04681 (54 CONNECTOR).

4-28. <u>Description of the Distributor</u>. Standard Patch Distributor F75M04681 is an enclosure 15.12 inches deep, 48.47 inches high and 24 inches wide. It may be mounted in a standard 24 inch rack. On the back of the enclosure, mounted on two hinged plates, are fifty-four 61 pin connectors (Bendix PT07SE-24-61P or equal).

Patching between these connectors is provided by the following patchboard assembly, which is accessible through the front panel door:

a. Frame & Spring Assembly (AMP Inc. 420050-2 or equal)

b. Removable 3254-pin patchboard (AMP Inc. 420051-2 or equal)

Above the front panel dcor are 2 rows of 54 taper-pin blocks each. These are two types, mounted alternately; 27 blocks (AMP 480107-6 or equal) and 27 blocks (AMP 480107-9 or equal) in each row. Each taper-pin block has 30 female test jacks wired to the patchboard. See figure 4-7.

4-29. <u>Use of the Patch Distributor</u>. The standard patch distributor is employed in various 24-inch relay racks used in the ESE. In these locations, the distributor provides monitoring and flexible distribution of signals. It also accommodates standard relay modules (paragraph 4-37) and standard diode modules (paragraph 4-42), which may be patched into the ESE systems as required.

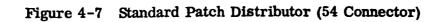
4-30. Locations of Patch Distributors. The ESE employs approximately four standard patch distributors; two are located in the Integration Racks, room 8-A and one each is located in the Integration Racks in the LCC.

4-31. <u>Design Organization</u>. Patch Distributor F75M04681 was designed by MSFC-Astrionics and adopted by KSC-Launch Equipment Branch.

TO BE PROVIDED

1.00 **1**0

18 18 19



4-32. STANDARD PATCH DISTRIBUTOR D75M09313 (27 CONNECTOR).

4-33. <u>Description of the Distributor</u>. Standard Patch Distributor D75M09313 (figure 4-8) has an enclosure 22.5 inches high, 21.0 inches wide, and 13.06 inches deep. On the back of this enclosure are twenty-seven 61-pin connectors (Bendix PTO7SE-24-61P or equal). Patching between these connectors is provided by the following patchboard assembly:

a. Frame & Spring Assembly (AMP, Inc. 420048-4 or equal)

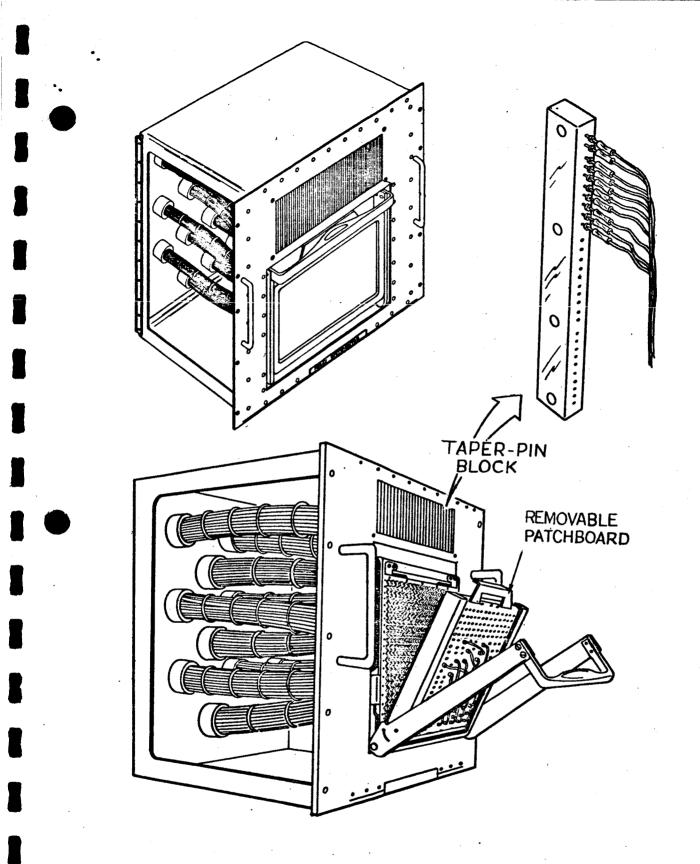
b. Patchboard (AMP, Inc. 420049-3 or equal)

Above the front opening is a row of 54 taper-pin blocks, mounted alternately: 27 blocks (AMP 480107-6 or equal) and 27 blocks (AMP 480107-9 or equal). Each taper-pin block has 30 female test jacks wired to the patchboard.

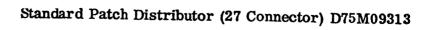
4-34. <u>Use of the Patch Distributor</u>. This patch distributor provides monitoring and flexible distribution of signals. It also accommodates standard relay modules (paragraph 4-37) and standard diode modules (paragraph 4-42), which may be patched into any system using the distributor. It is presently used in the Portable Arm Control Console.

4-35. <u>Locations of Patch Distributor</u>. See Portable Arm Control Console, figure 5-12.

4-36. Design Organization. Patch Distributor D75M09313 was designed by the Electrical Section, Launch Equipment Branch, KSC.







4-37. STANDARD RELAY MODULE D75M50348.

4-38. Description of the Relay Module. Standard Relay Module D75M50348
(figure 4-9) is an assembly 2.43 inches high and 5.87 inches long, and weighs
14 ounches. It contains:

a. 8 relays Per MSFC-SPEC-339/53 (DPDT 28 volts dc relay, coil resistance 600 ohms, contact rating 2.0 amperes at 28 volts dc resistive load)

b. 8 bases Burndy Engineering Co. MT9R-2 or equal 72 base contacts (Burndy RC16Y-1-F59 or equal)

Each relay contact is wired to one pin in the 61-pin connector (Bendix PTO6P-24-61S or equal). For four of the relays, each side of the dc coil is independently wired to a connector pin. In the remaining four relays, one side of each dc coil is wired to a common bus.

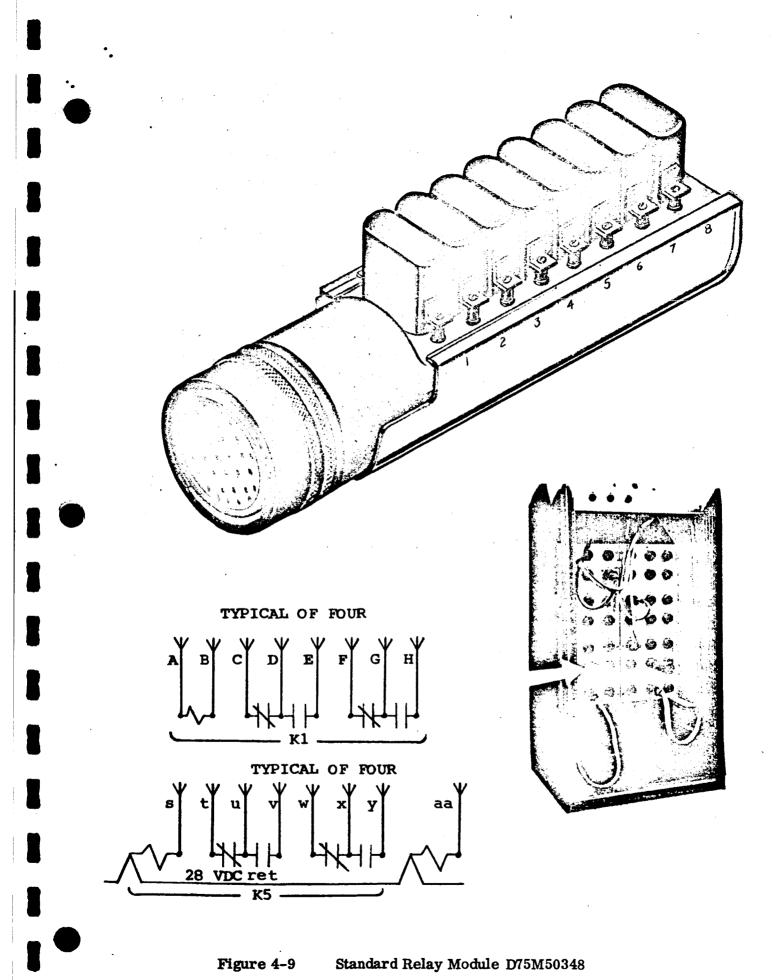
4-39. Use of the Relay Module. The 2-ampere relay contacts in this module are adequate both for logic circuits and for control, since no solenoid valve in the ESE control systems required more than 1.5 amperes.

In the ESE, these relays are used as logic and control elements in the control systems for the service arms, tail service masts, firing circuits and others. They are also used as transfer elements for interface separation so that no control subsystem uses dc power from another.

In the Launcher Ground Equipment Test Set (Launcher GETS), standard relay modules simulate switches and relays in the ESE. They also close indicator lamp circuits within the GETS.

4-40. <u>Locations of Relay Module</u>. In the ESE, about 150 relay modules are connected to various patch distributors in the relay racks in room 8-A and in the Launcher GETS.

4-41. <u>Design Organization</u>. Relay Module D75M50348 was designed by MSFC-Astrionics and adopted by KSC-Launch Equipment Branch.



4-42. STANDARD DIODE MODULE D75M50364.

4-43. <u>Description of the Diode Module</u>. Standard Diode Module D75M50364 is 2.75 inches in diameter and 10.25 inches long, including connector. It weighs 2.25 pounds.

Inside are thirty diode-fuse pairs, each series-wired to two pins in the 61-pin connector (Bendix PT06P-24-61S or equal). The diodes are type 1N540 (Texas Instruments or equal); the fuses are Bussman GLD-1 or equal.

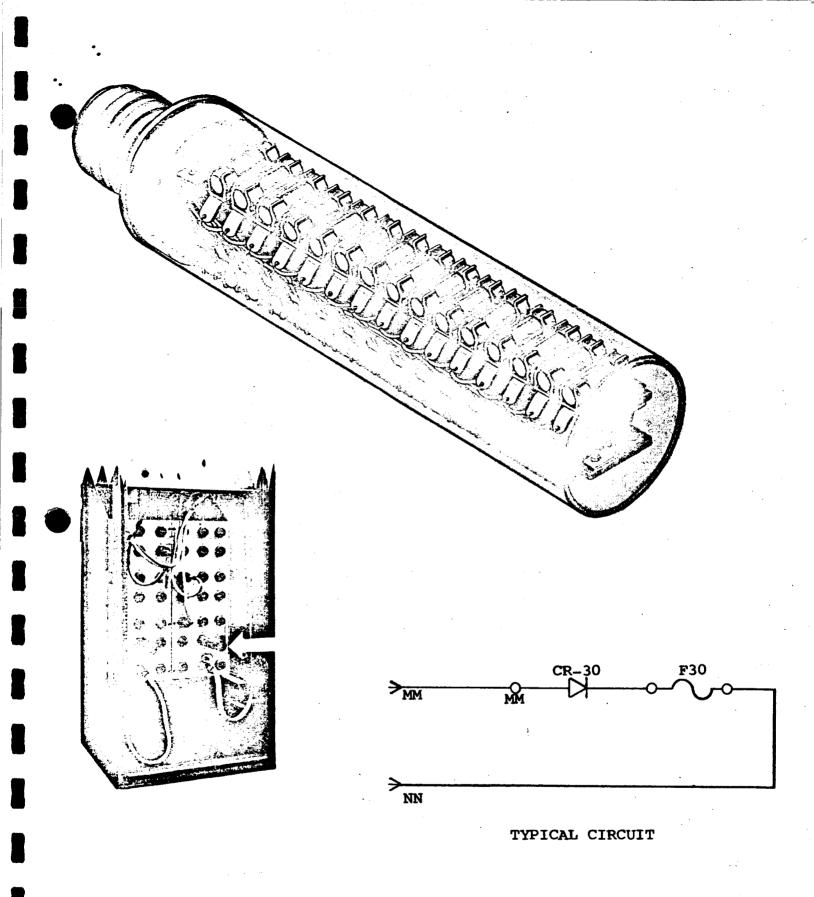
All fuse clips are mounted together on two fuse boards, and all diodes are mounted on a single board in between. All wiring at the adapter end is potted.

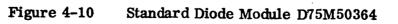
4-44. <u>Use of the Diode Module</u>. Standard Diode Module D75M50364 can be used with any electrical distributor mating with the connectors named above, and is presently used with the Standard Patch Distributors. See figure 4-10.

In the ESE, the diodes in this module are used to suppress arcing between relay contacts.

4-45. <u>Locations of Diode Module</u>. About 50 diode modules are used in Electrical Support Equipment, including the Launcher GETS and the relay racks in room 8-A.

4-46. <u>Design Organization</u>. Diode Module D75M50364 was designed by the Electrical Section, Launch Equipment Branch.





4-47. STANDARD METER PANEL J75M10165.

4-48. <u>Description of the Meter Panel</u>. Standard Meter Panel J75M10165 (figure 4-11) is designed for placement in the standard 24-inch rack. It is 10.5 inches high and approximately 3 inches deep. The panel has:

a. 30 fuses	Bussmann GLD-1 or equal Bussman Fuseholder HLD or equal
b. 30 switches	Toggle T-2150, Control Co. of America or equal
c. 60 jacks	E. F. Johnson Co. or equal 108–901 thru 108–913 (assorted colors)
d. 1 multimeter	Simpson Model 270 or equal

Interconnections of these components are shown on the front panel silkscreen. Note that a shorting switch is added to the multimeter input for zeroing the ohmmeter scale.

4-49. <u>Use of the Meter Panel</u>. In the ESE, this meter panel is employed in the relay and power racks in room 8-A. In each case, test leads will normally be connected to test pin blocks in a patch distributor. See figure 4-8.

Standard Meter Panel J75M10165 is normally used to measure resistance and dc voltage.

4-50. Locations of Meter Panel. Approximately seven standard meter panels are used in the Mobile Launcher relay racks and arming panel racks, room 8-A.

4-51. <u>Design Organization</u>. Standard Meter Panel J75M10165 was designed by the Electrical Section, Launch Equipment Branch.

i 0 \odot Ø 0 Ć (Õ) Ó (0) Ø. ٢ 0 0 6 ٢ Ó Ó 6 0 0 6 ۲ G) $\mathbf{\hat{o}}$ 0 ٢ Ó 0 0 0 0 0 0 Ô 9 (Ĉ) 0 ٥ 0 O 00 1 0 Ó Ó **(Ô**) Ó 0 0 0 0 0 Ó 6 0 0 METER PANEL DARD METER PANELS J75MIDI65 SEE RELAY RACKS 10, 12, 14, 17, 24, FIGURE 6-2 621 / 6626 6628 6625 ***66**27

Figure 4-11 Star day & Meter Manel J75M10165

4-52. STANDARD FUSE PANEL J75M10163.

4-53. <u>Description of the Fuse Panel</u>. Standard Fuse Panel J75M10163 (figure 4-12) is designed for placement in the standard 24 inch rack. It is 5.25 inches high and 6 inches deep.

Sixty 3-ampere indicator fuses (Bussman GBA-3 or equal) are connected to two 61-pin connectors (Bendix PTO7SE-24-61P or equal). All signals enter one connector and leave through the other.

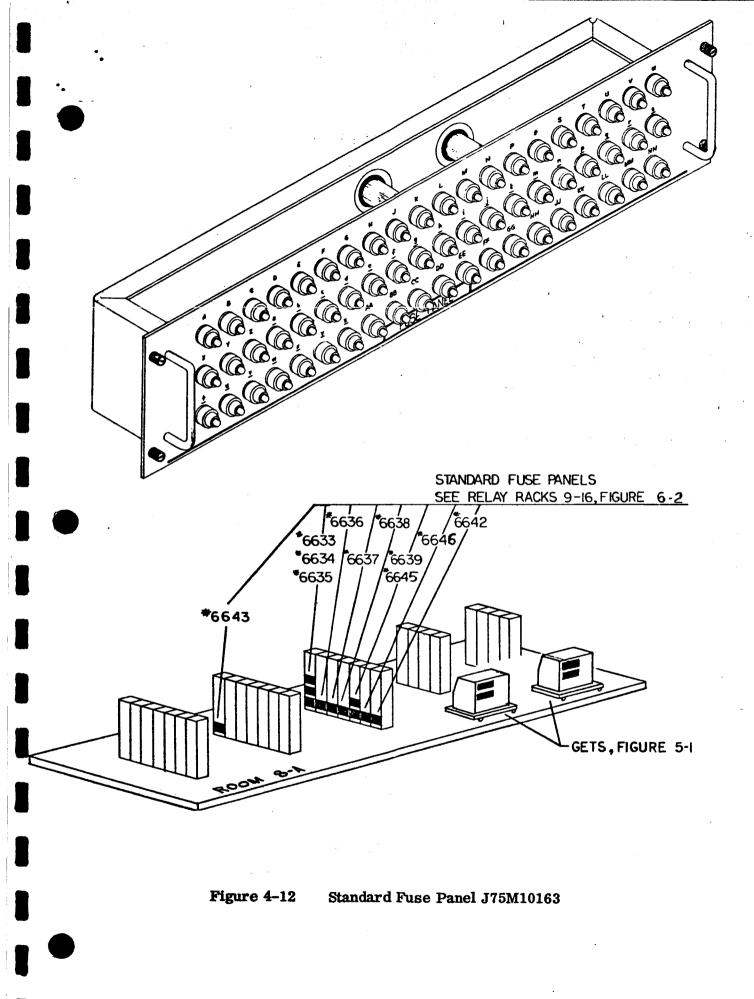
4-54. <u>Use of the Fuse Panel</u>. From one to three fuse panels are used in each relay rack in room 8-A. Signals passing through the relay racks are routed through this fuse panel, so that cabling is protected during tests and exercises.

Prior to launch, the fuse panel is bypassed by disconnecting its two cable connectors and mating them to each other.

4-55. Locations of Fuse Panel. Approximately 15 fuse panels are used in the relay racks and GETS, as in figure 4-12.

4-56. <u>Design Organization</u>. Standard Fuse Panel J75M10163 was designed by the Electrical Section, Launch Equipment Branch.

4-22



4-57. STANDARD MODULE FRAME J75M10432.

4-24

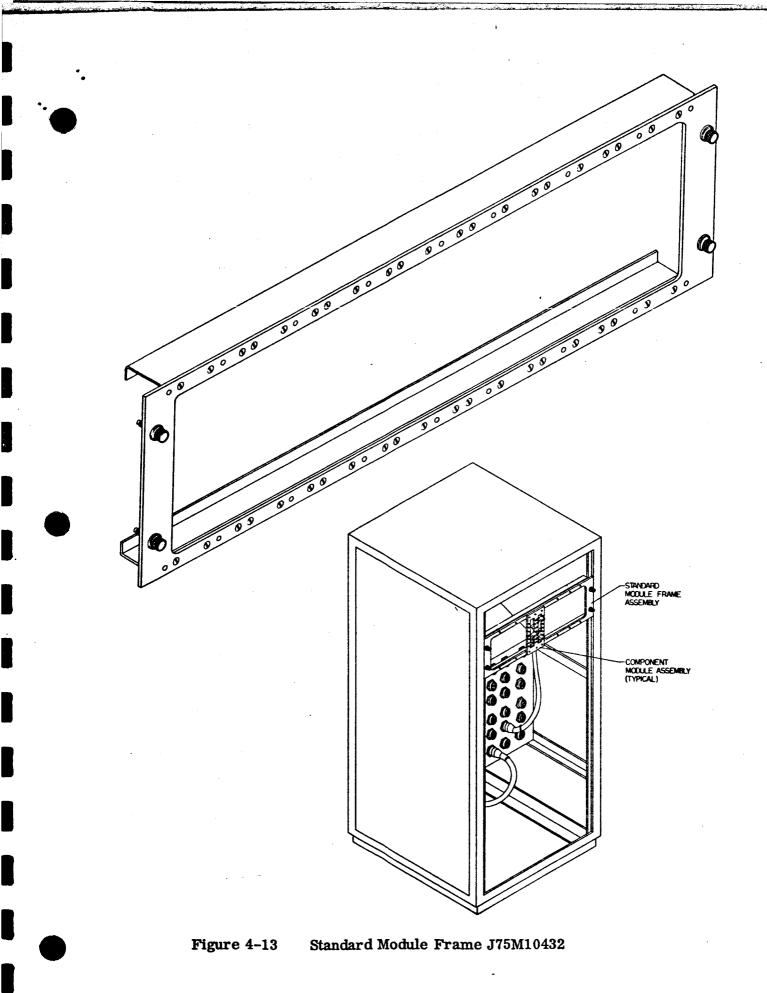
4-58. <u>Description of the Module Frame</u>. Standard Module Frame J75M10432 (figure 4-13) is designed for placement in the back of the standard 24-inch vertical rack. It is approximately 7 inches high and 2 inches deep.

4-59. <u>Use of the Module Frame</u>. The module frame accommodates a variety of components to be connected to the standard patch distributor, as illustrated in figure 4-13. These are normally components other than diodes or relays, since standard relay and diode modules are already available.

A typical use of the frame is shown in figure 4-13, where three electronic timers are mounted in a component module assembly D75M10433, which is mounted to the Standard Module Frame.

4-60. <u>Locations of Module Frame</u>. At present the module frame is not used in the ESE.

4-61. Design Organization. The Standard Module Frame J75M10432 was designed by MSFC-Astrionics and adopted by KSC-Launch Equipment Branch.



⁴⁻²⁵

4-62. STANDARD 500 AMPERE POWER SUPPLY.

4-63. <u>Description of the Power Supply</u>. The 500 Ampere Power Supply (figure 4-14) is designed for placement in a standard 24-inch rack. It is 47.25 inches high and 24 inches deep. Electrical connections to this supply are provided as follows:

a.	AC Input Connector	Bendix 323340-57P or equal
b.	Remote Sensing Connector	Bendix PT07SE-20-39S or equal
c.	DC Output	Bus bars extending from rear of unit

The front panel of the supply contains:

a. Panel meters for dc output voltage and current

b. Switches POWER ON and START-STOP

c. Control for voltage output level

d. Meter indicating elapsed time

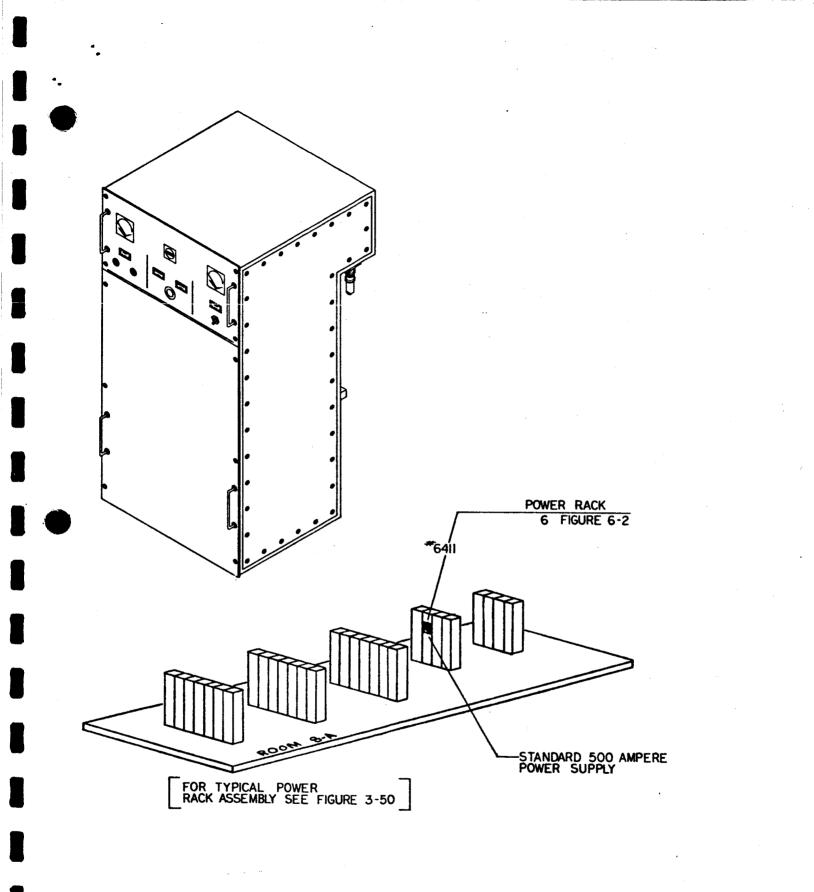
This supply is designed for a maximum output current of 500 amperes and a nominal output voltage of 28 volts dc, adjustable from 24 volts to 36 volts. Output ripple is less than 0.1 volts peak to peak.

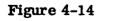
The input power required is 220/440 volts, 3-phase, 60 cycle. A stepdown transformer with a three-phase, full-wave semiconductor bridge is used for the ac to dc conversion. The output voltage level is controlled by silicon control rectifiers.

4-64. <u>Use of the Power Supply</u>. The standard 500-ampere power supply is the prime source of 28 volt dc power for all electrical functions on the service arms other than battery-operated devices. It is also used to provide 28 volt power for the Tail Service Masts, Launcher Accessories, and other ESE Subsystems.

4-65. <u>Location of Power Supply</u>. Two 500 ampere power supplies are located in room 8-A of the Mobile Launcher. One is currently a spare.

4-66. <u>Design Organization</u>. The standard 500 ampere power supply was designed by MSFC-Astrionics. For LC-39, it is furnished by MSFC and used by KSC-Launch Equipment Branch.







4-67. STANDARD TEST LOAD E40M03971.

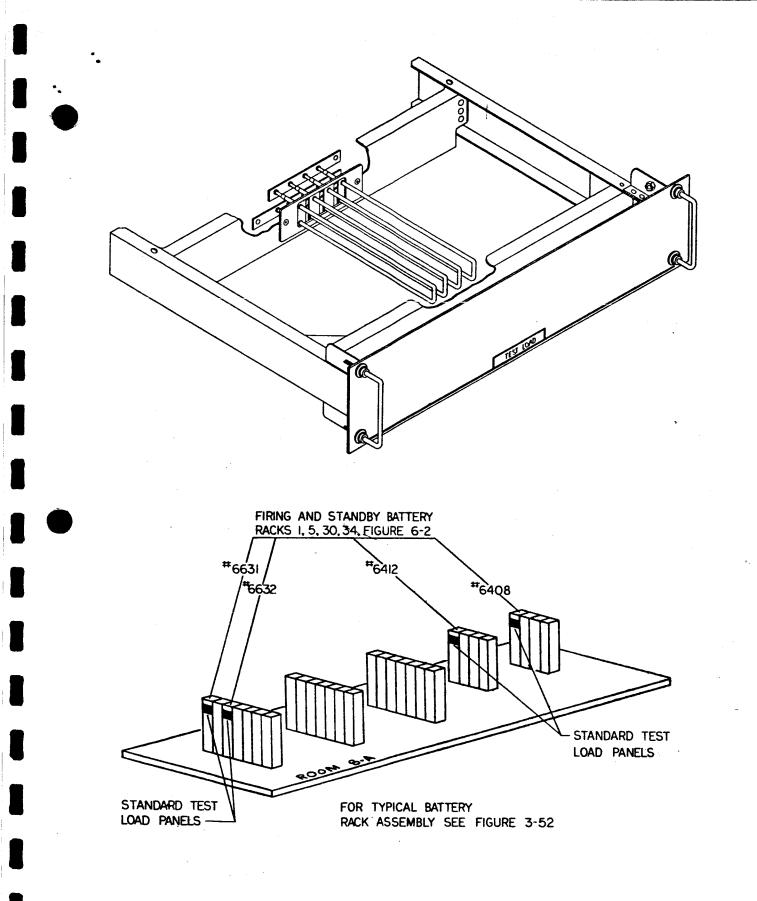
4-68. <u>Description of the Test Load.</u> Test Load Assembly E40M03971 is designed for placement in the standard 24-inch rack. It is 5.15 inches high and 18 inches deep.

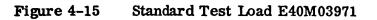
It contains four resistance elements (Chromalox 4-43209 or equal, 30 volts, 300 watts) connected in parallel by bus bars at the rear of the chassis. The buses are connected to input cables by terminal lugs.

4-69. <u>Use of the Test Load</u>. Test Load Assembly E40M03971 is a part of the standby battery racks and firing battery racks. It serves as a test load for the batteries and may be connected to the batteries by the Load Control Panel, figure 4-16.

4-70. Locations of Test Load. See figure 4-15.

4-71. <u>Design Organization</u>. Standard Test Load E40M03971 is designed by MSFC-Astrionics. For LC-39, it is furnished by MSFC and used by KSC-Launch Equipment Branch.





4-72. STANDARD LOAD CONTROL PANEL E40M03535.

4-73. <u>Description of the Load Control Panel</u>. Load Control Panel E40M03535 is designed for placement in the standard 24 inch rack. It is 7 inches high and 20.25 inches deep.

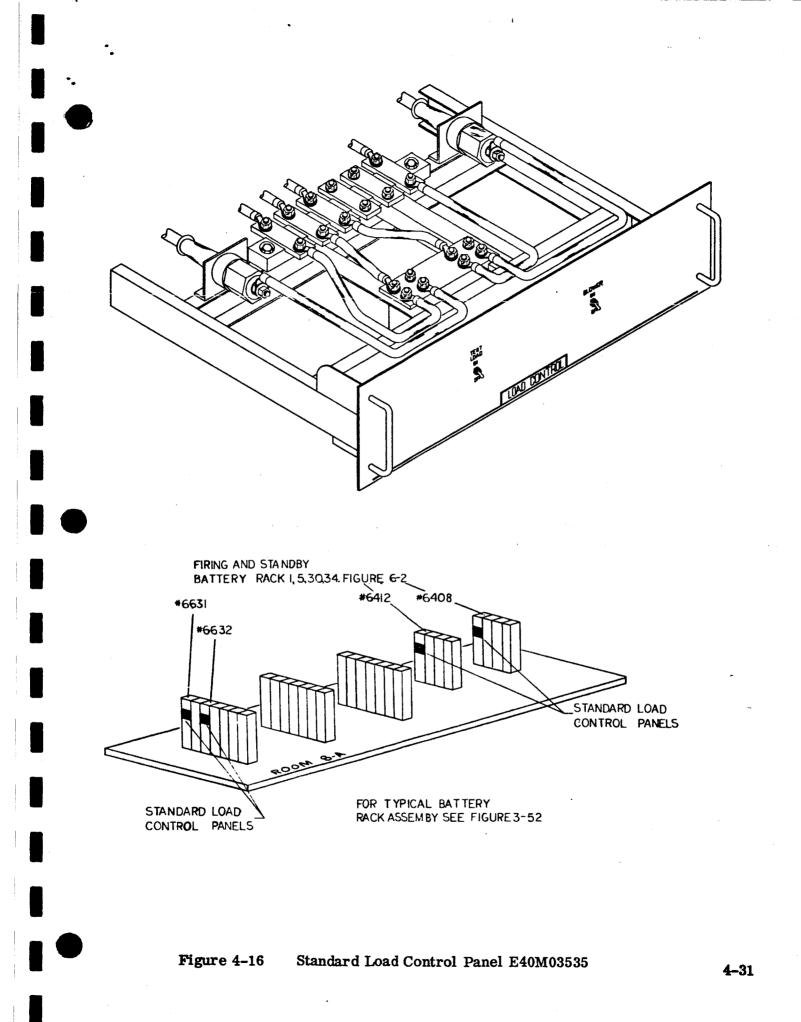
It contains a DPDT contactor (Hartman A-848K or equal) controlled by a front-panel switch. The contactor is actuated by a 28-volt dc coil. The normally open contacts are rated at 200 amperes, and the normally closed contacts at 300 amperes.

The dc power controlled by the contactor is supplied through two singlepin connectors (Superior RS250GB, RS250GR or equal) and is routed through contacts to a group of bus bars.

4-74. Use of the Load Control Panel. In the ESE, a Load Control Panel is located in each standby battery rack and firing battery rack. It is used to disconnect the normal circuits from the batteries in these racks and to connect the standard test load, figure 4-15.

4-75. Locations of Load Control Panels. Four Load Control Panels are used in room 8-A as shown in figure 4-16.

4-76. <u>Design Organization</u>. The Standard Load Control Panel E40M03535 is designed by MSFC-Astrionics. It is furnished by MSFC and used by KSC-Launch Equipment Branch.



4-77. STANDARD BATTERY CHARGER.

4-78. <u>Description of the Battery Charger</u>. The Standard Battery Charger (figure 4-17) is designed for placement in a standard 24-inch rack. It is 12.25 inches high and 24 inches deep.

Input power to the charger is 220/440 volts, 3-phase 60-cycle. A terminal strip is provided for this input. A connector is provided for remote-readout and control-function cabling. DC output cables to the battery load are connected by terminal lugs fixed to bus bars.

The front panel of the charger contains:

- a. Switch-lights which indicate Power ON-OFF, Load Charge, Reset, Temperature Alarm, and Cells Charging
- b. Panel meters showing charger dc voltage and current
- c. Digital readouts for percent discharge, peak discharge, and charge cycle.

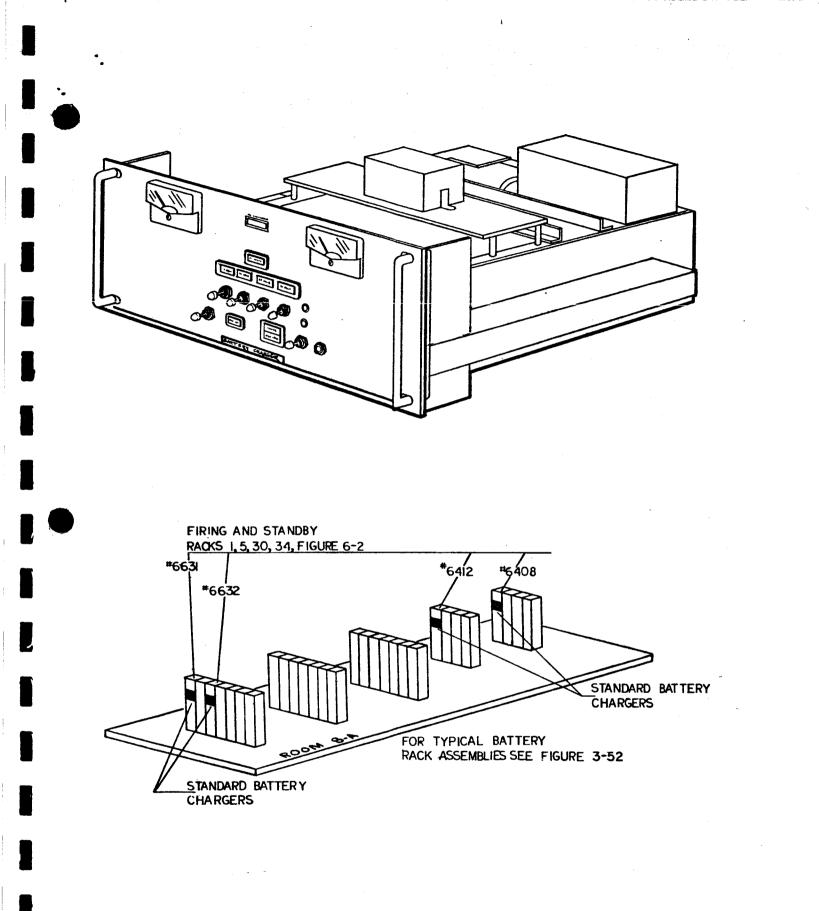
The dc power for battery charging is obtained by transforming and rectifying the three-phase line voltage. Semiconductors are used for rectifying, some of which are Silicon Control Rectifiers (SCR) which also control load current. The SCR's are gated by magnetic amplifiers whose input is a comparison of battery voltage to a Zener reference.

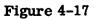
4-79. Use of the Battery Charger. To maintain battery charge, one battery charger is used in each ESE battery rack. These are:

- a. Two firing battery racks used in the service arms firing circuit, paragraph 3-25.
- b. Two standby battery racks which provide power supply backup.

4-80. <u>Locations of Battery Chargers</u>. The ESE uses four Standard Battery Chargers as in figure 4-17.

4-81. <u>Design Organization</u>. The Standard Battery Charger was designed by MSFC-Astrionics. It is furnished by MSFC and used by KSC-Launch Equipment Branch.





Standard Battery Charger

4-82. STANDARD BATTERY PANEL J40M03605.

4-34

4-83. Description of the Battery Panel. Standard Battery Panel J40M03605 (figure 4-18) is designed for placement in the standard 24-inch rack. It is 15.72 inches high and approximately 19 inches deep, excluding front handles.

The panel is designed for a maximum load of 240 pounds (120 pounds per battery module, 2 modules maximum).

4-84. <u>Use of the Battery Panel</u>. Standard Battery Panel J40M03605 accommodates batteries for the Firing Batteries and Standby Battery Racks.

4-85. <u>Locations of Battery Panels</u>. Eight battery panels are used for the ESE in room 8-A.

4-86. <u>Design Organization</u>. Standard Battery Panel J40M03605 was designed by the Electrical Section, Launch Equipment Branch.

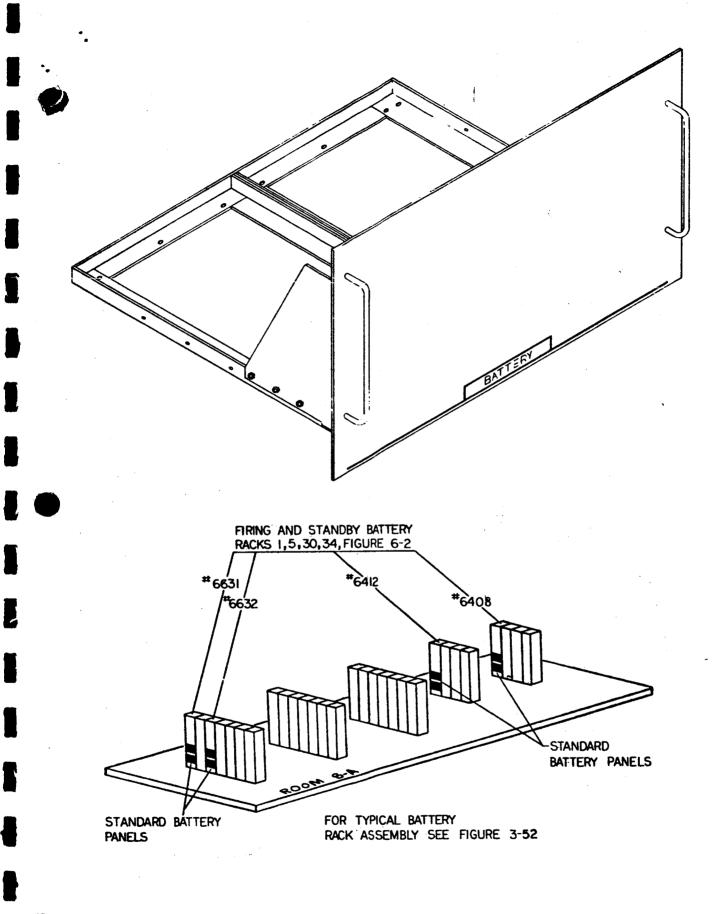


Figure 4-18 Standard Battery Panel J40M03605

4-87. STANDARD POWER MODULE TYPE IIA J75M12295.

4-88. <u>Description of the Power Module</u>. Standard Power Module J75M12295 (figure 4-19) is 8.5 inches high, 4 inches wide, and 14 inches deep. These modules are mounted in a Standard DC Power Frame Type II E75M12301, which in turn mounts in the standard 24-inch rack. Each power module contains:

a. 1 4-pin connector	Bendix HUS-KEY 10-323332-17S or equal
b. 1 10-pin connector	Bendix PT07SE-12-10S or equal
c. 1 circuit breaker	Heinemann Electric Co. AM1510-70 Single-pole, 70 amperes or equal
d. 1 contactor	Cutler-Hammer 6042-H119 or equal
e. 1 shunt	Empro Mfg. Co. 1242-11-50 75 amperes or equal

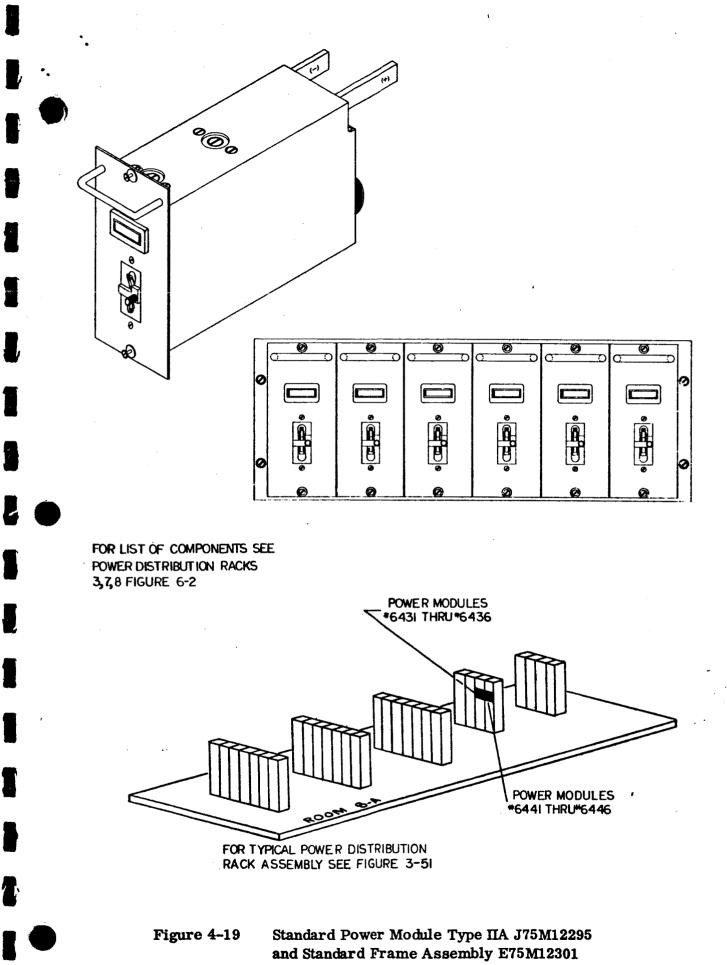
f. 1 indicating light

A circuit breaker and contactor are series-wired to provide both overload protection and remote control capability. A shunt is provided for current . measurement.

4-89. <u>Use of the Power Module.</u> All power modules are mounted in the ESE power distributor racks. A power module is provided for each of the nine Service Arms, the Hydraulic Charging Unit, Computer Command Functions and the Service Arm Control Switch Assembly (liftoff switch pneumatics). One module is used for the three Tail Service Masts and another for the Pneumatics Control and Distribution Subsystem.

4-90. Locations of Power Modules. There are 18 power modules used in the ESE. Power Distribution Racks #6413, #6414 and #6719 contain six each. All are in room 8-A of the Mobile Launcher base.

4-91. <u>Design Organization</u>. Standard Power Module Type IIA was designed by MSFC-Astrionics and is used by KSC-Launch Equipment Branch.



4-92. STANDARD RECEPTACLE DISTRIBUTOR TYPE I E75M12316.

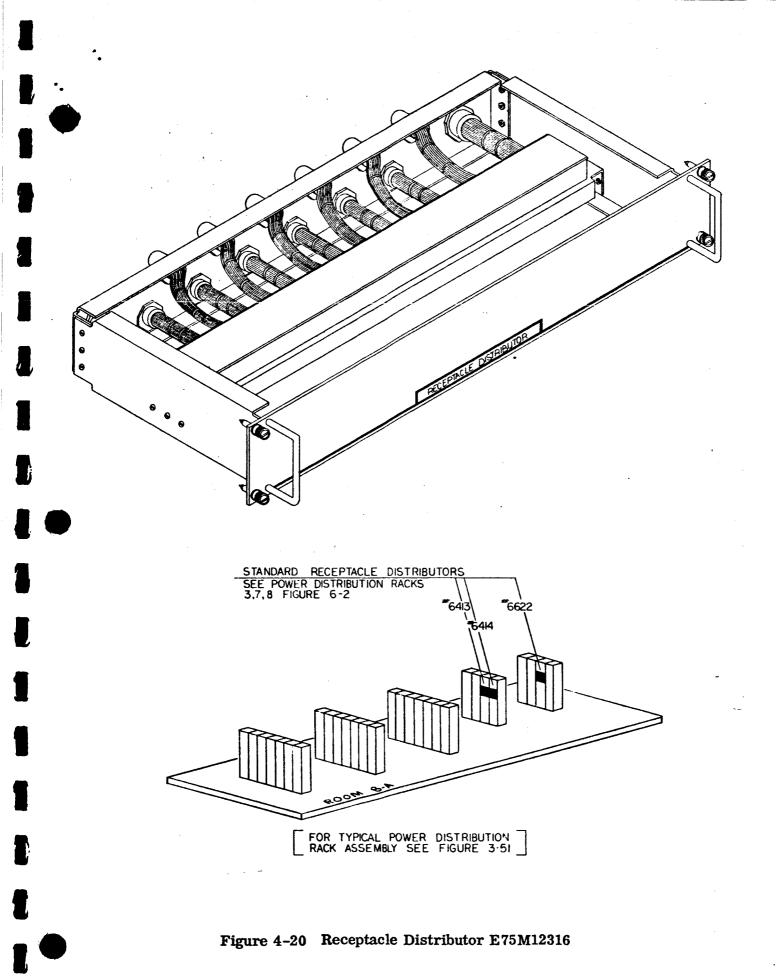
1-93. <u>Description of the Receptacle Distributor</u>. Standard Receptacle Distributor
Type I E75M12316 is designed for placement in a standard 24-inch rack. It is
3.6 inches high and 12 inches deep (see figure 4-20).

It contains twelve 10-pin connectors (Bendix PT07SE-12-10P or equal) and two 61-pin connectors (Bendix PT07SE-24-61S or equal). One of the 61pin connectors is wired to six of the 10-pin connectors, and the other 61-pin connector is wired to the remaining 10-pin connectors.

4-94. Use of the Receptacle Distributor. The Receptacle Distributor Type I is used in the Power Distribution Racks (figure 3-51) to route the 28 volt control and monitoring signals from individual power modules to the relay racks. It combines the twelve 10-conductor cables from the power modules into two 61 conductor cables.

4-95. <u>Locations of Receptacle Distributors</u>. Three type I receptacle distributors are used in the ESE: one each in Power Distribution Racks #6413, #6414, and #6719. All are in room 8-A of the Mobile Launcher base.

4-96. <u>Design Organization</u>. Standard Receptacle Distributor Type I E75M12316 was designed by MSFC-Astrionics and used by KSC-Launch Equipment Branch.



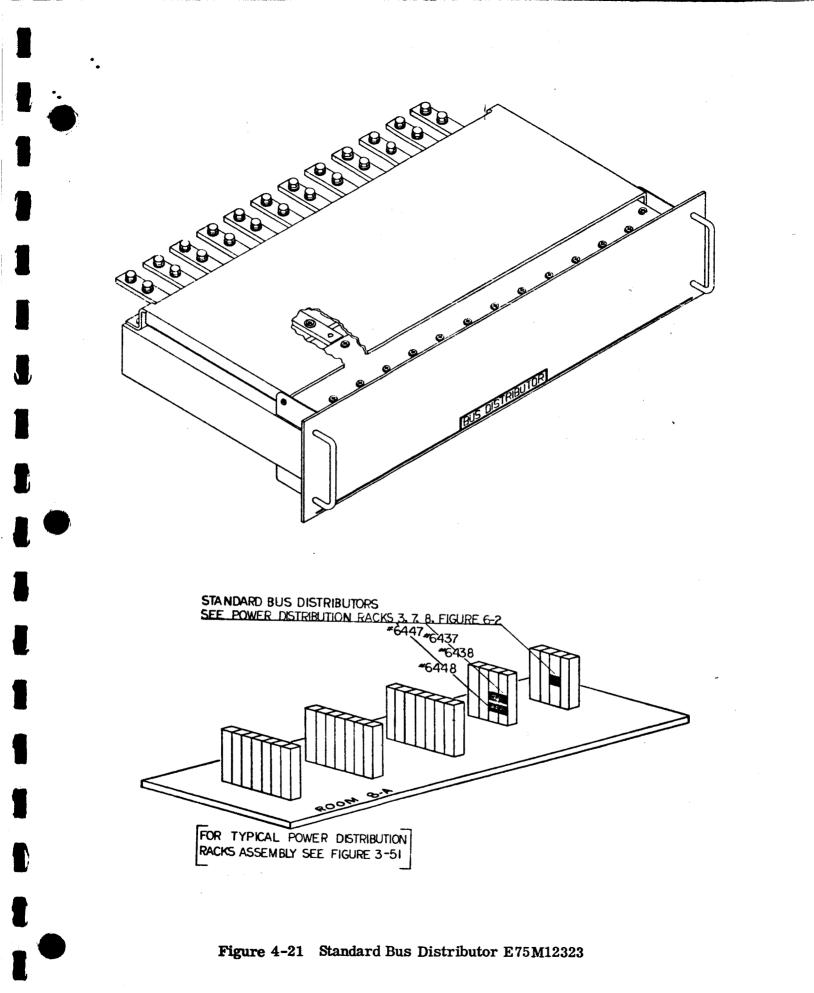
4-97. STANDARD BUS DISTRIBUTOR E75M12323.

4-98. Description of the Bus Distributor. Standard Bus Distributor E75M12323 (figure 4-21) is designed for placement in the standard 24 inch rack. It is 5:25 inches high and 14 inches deep. This distributor contains twelve 0.25 inch x 0.75 inch bus bars, each designed for terminating one #4 conductor. The twelve bus bars are divided into six adjacent pairs, with each pair connected to one 61-pin connector (Bendix PT07SE-24-61S or equal).

4-99 <u>Use of the Bus Distributor</u>. The Bus Distributor is used in power distribution racks to distribute dc power from individual power modules to individual 61-pin connectors. For ESE use both bus bars of a given pair have the same polarity, so that all pins of any one 61-pin connector are of the same polarity. It is through these connectors that power is distributed to the ESE systems.

4-100. <u>Locations of Bus Distributors</u>. The Bus Distributors are in Power Distribution Racks #6413, #6414, and #6719, located in room 8-A, (see figure 4-21).

4-101. Design Organization. Bus Distributor E75M12323 was designed by MSFC-Astrionics and is used by KSC-Launch Equipment Branch.



4-102. STANDARD CONSOLE E64-KN-F-261-4.

4-42

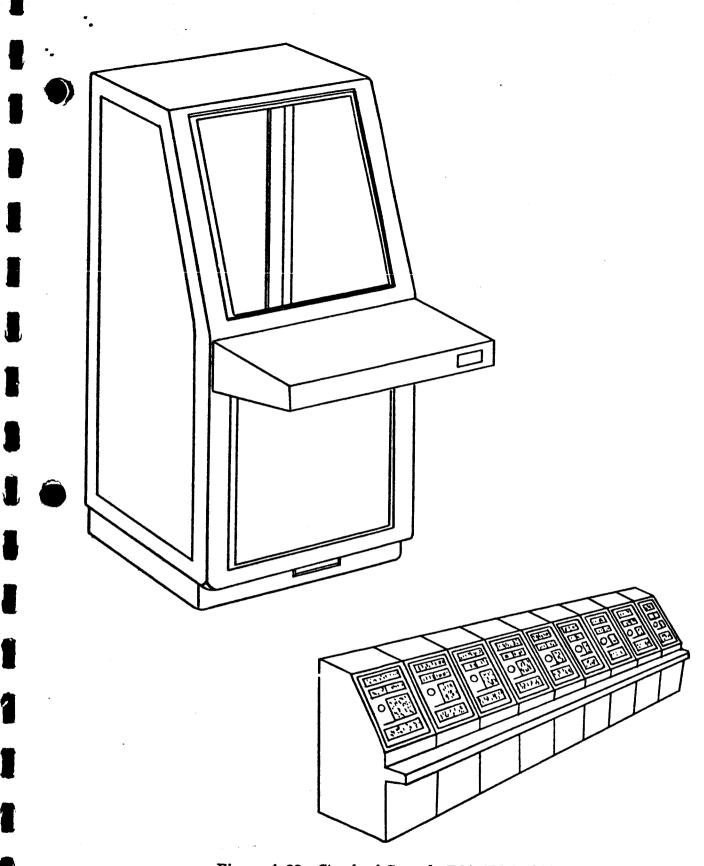
4-103. Description of the Console. Standard Console E64-KN-F-261-4 (figure 4-22) is a radio frequency interference (R. F. I.) shielded enclosure 53 inches high, 26.06 inches wide, and 30.56 inches deep, with a 19 degree sloping front. The console is designed to house 19-inch component panels on its upper front.

The enclosure base, side panels, and cable entry (enclosure bottom) are removable.

4-104. Use of the Console. These consoles house the LCC control panels which provide remote control, test, and monitor functions for the ESE.

4-105. <u>Location of Consoles</u>. The consoles are in the firing rooms, third floor, Launch Control Center.

4-106. Design Organization. Standard Console E64-KN-F-261-4 is commercial equipment purchased to KSC/MSFC specifications.





ļ

4-107. INTEGRATION RACK.

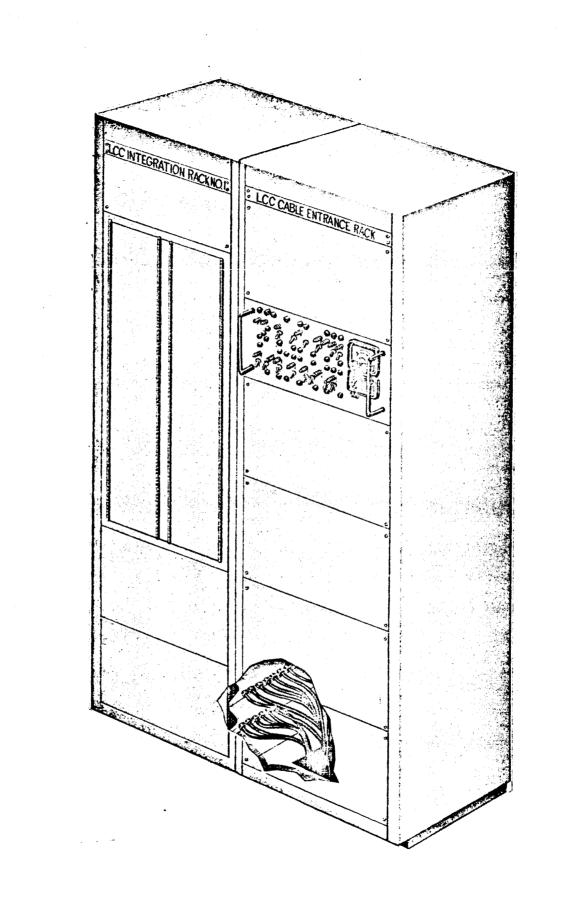
4-108. <u>Description of the Integration Rack</u>. The integration rack basically consists of a standard 24-inch vertical rack (Para. 4-17) installed with a standard 54-connector patch distributor (Para. 4-27). Since the base entrance plate to the standard rack can accommodate a maximum of 25 connector entrances, it is necessary to incorporate a side entrance plate to meet the total capacity of the 54-connector distributor.

To provide side cable access, an entrance rack is placed between two integration racks. It will also be used to mount ancillary equipment such as meter panels and communication panels. The entrance rack is a standard 24-inch vertical rack with the side panels removed. (See figure 4-23).

4-109. <u>Use of the Integration Rack</u>. The integration rack is used as an interface between various relay distributors, communication links and control consoles. Its purpose is to gather related signal functions (i.e., commands, analogs, discrete indications, etc.) which emanate from various sources and, by means of patching, route these functions to specific cables which interface with communications hardware. In this way, the total number of interface cables can be reduced by maximum use of available circuits in each cable.

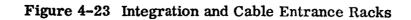
4-110. <u>Locations of Integration Racks</u>. There is an integration rack set located both in room 8-A of the mobile launcher and in each firing room of the launch control center. A set consists of two basic integration distributor racks and one entrance rack.

4-111. <u>Design Organization</u>. The integration rack was designed by the Electrical Section, Launch Equipment Branch.



••

10



4-112. STANDARD LATCHING RELAY MODULE D75M05895.

4-113. <u>Description of the Latching Relay Module</u>. Standard Latching Relay Module (Figure 4-24) is an assembly 2.43 inches high and 5.87 inches long, and weighs approximately 13 ounces. It contains:

a. 7 Relays Per MSFC-SPEC-339/63 (DPDT 28 vdc relay)

Coil resistance - 800 ohms Contact rating - 2.0 amperes at 28 volts dc resistive load

b. 7 Bases Burndy Engineering Co. MT9R-2 or equal, including 72 base contacts (Burndy RC 16Y-1-F59 or equal)

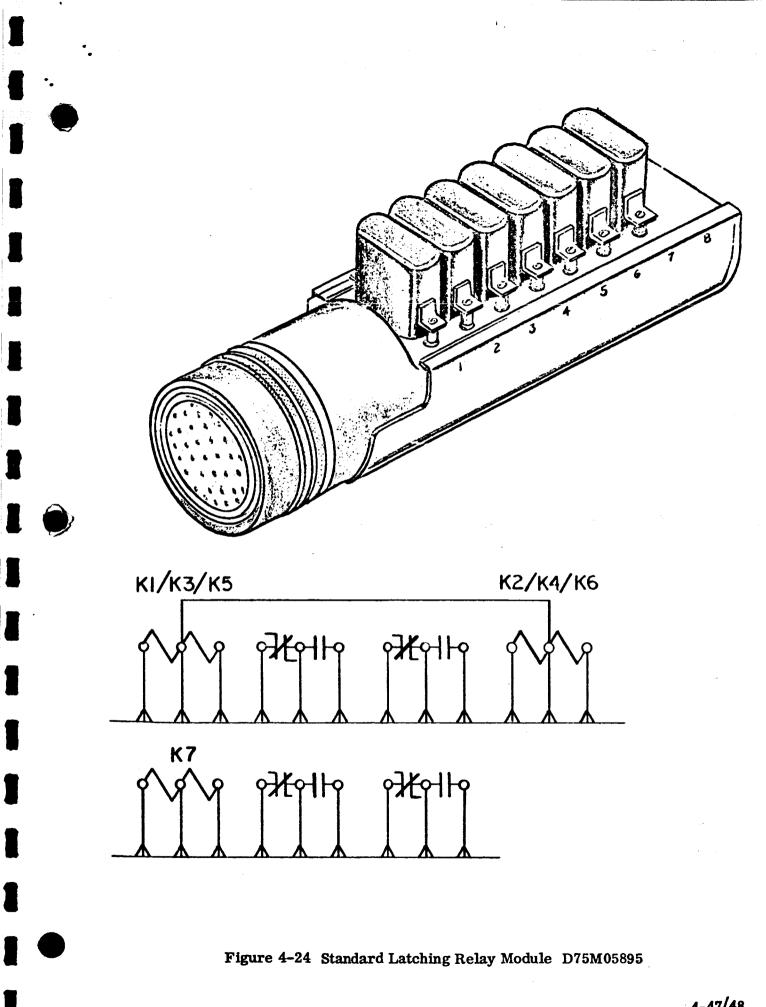
Each relay contact is wired to one pin in the 61-pin connector (Bendix PT06P-24-61S or equal). In all cases, the two coils associated with a specific relay are tied to a common return. For six of the seven relays, two adjacent relays also share a common return for their four coils.

The contacts of the relay are magnetically locked into a specific position when the coil is activated. They remain in this position, even after voltage dropout. The opposite coil must be activated to switch the contacts.

4-114. <u>Use of the Latching Relay Module</u>. In the ESE, these relays are used as logic and control elements in the control systems when the source signal is of a momentary nature or when the system logic calls for locking out certain functions which are no longer required in series with system operations.

4-115. Location of Latching Relay Modules. In the ESE, these relay modules are connected as required to various patch distributors in the relay racks in room 8-A.

4-116. <u>Design Organization</u>. Standard Latching Relay Module D75M05895 was designed by the Electrical Section, Launch Equipment Branch.



4-47/48

SECTION V

TEST EQUIPMENT

ELECTRICAL SUPPORT EQUIPMENT

5-1. GENERAL.

Also included in this section is a description of $||||^{\mu}$ swing arm and tail service mast qualification test site at MSFC, where $cer^{(\mu)}$ of the test equipments will be first employed.

5-2. LAUNCHER GROUND EQUIPMENT TEST SET E75M13286.

-3. <u>Description of the Test Set</u>. The Launcher Ground Equipment Test Set (GETS) (figure 5-1) is a roll-around portable test console approximately 57 inches high, 53 inches wide, and 28 inches deep. It contains:

- a. 1 standard 60-connector patch distributor
- b. 1 standard 40-connector patch distributor
- c. 250 indicator lamp modules (JAY-EL P/N10182 or equal)
- d. 90 meters Triplett model . 5E or equal
- e. 90 analog potentiometers
- f. 300 switches (Cutler Hammer 8867K1 or equal
- g. 2 standard fuse panels
- h. 1 standard communication panel

i. 1 power monitor panel

This test set has provision for 20 function-cable connections and four powercable connections. These cables enter the console through a hinged door in the console base. Two doors on the front of the console provide access for connecting cables, diode modules, and relay modules to the patch distributor.

Two drawers with writing surfaces are provided for the operator.

5-4. Use of the Launcher GETS. For the purpose of checkout and test of the ESE subsystems, two Launcher GETS are employed as in figure 5-3. One is programmed as a control and monitor unit which generates all signals normally provided to the ESE subsystems from the LCC. At the same time, it simultaneously displays all discrete and analog signals arising in the particular subsystem under test. See figures 5-4 and 5-5. As a control and monitor unit, the Launcher GETS uses up to 10 diode modules (paragraph 4-42) in conjunction with the components listed above.

The other Launcher GETS is connected to system cables coming from the LCC. It is thus a simulator, and is programmed to simulate the electricalmechanical control and monitor components within a given ESE subsystem. These are mainly solenoids, pressure transducers, and pressure switches in the service arms, tail service masts, and Mobile Launcher base. Programmed as a LEFT BAY FRONT PANEL

RIGHT BAY FRONT PANEL

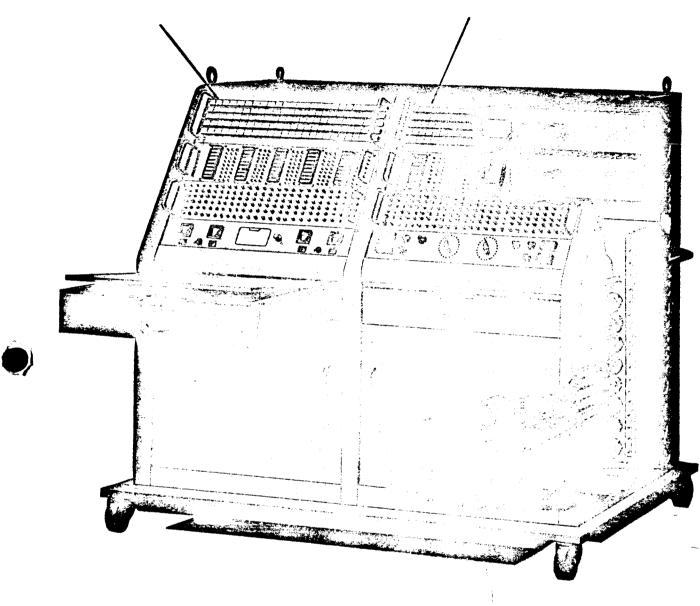


Figure 5-1 Ground Equipment Test Set E75M13286

simulator the Launcher GETS uses up to 36 relay modules (paragraph 4-37) along with its other test components.

Both the GETS control unit and the simulator unit interface with the ESE subsystems at the relay racks in room 8-A. The control unit is connected to the relay racks in place of systems cables from the LCC; the monitor unit is connected to the LCC cables themselves.

For these purposes, pre-installed test cables from the relay racks are routed beneath the deck to a deck plate in room 8-A. During tests the GETS is placed near the deck plate and connected to appropriate test cables.

5-5. Location of the Launcher GETS. The Launcher GETS is used in room 8-A.
5-6. Design Organization. The Launcher GETS was designed by the Electrical Section, Launch Equipment Branch.

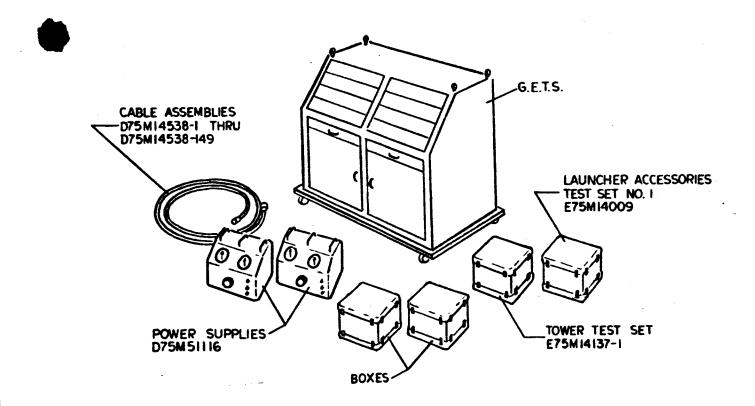


Figure 5-2 GETS Auxiliary Equipment

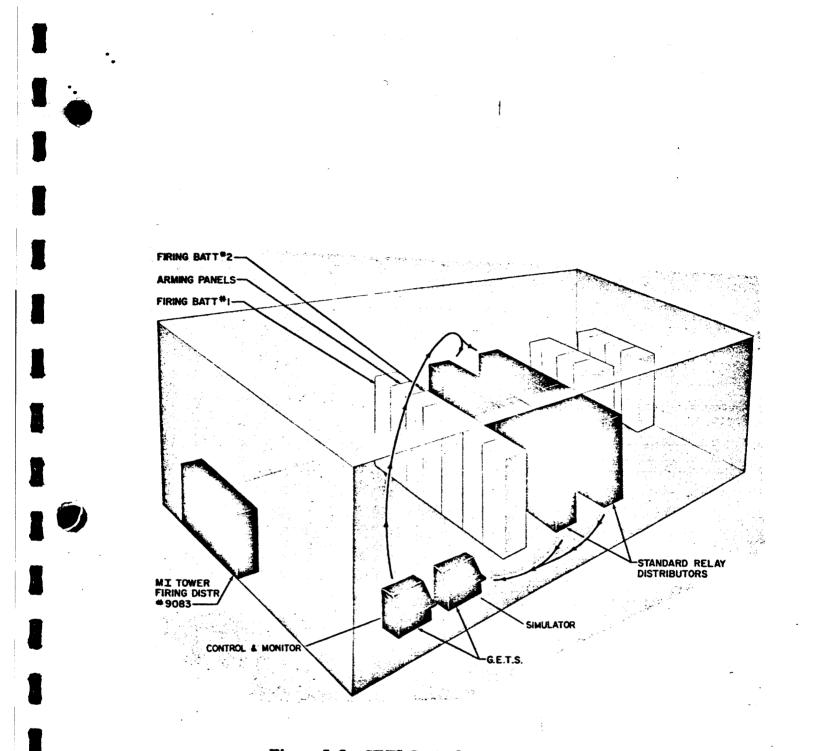


Figure 5-3 GETS Control, Monitor and Simulation

. ١. ø le le . Ø 0:0:0;0;0;0:0:0 Θ \odot G \odot Θ Õ 57.0 A LOUGH DIG THE REAL 1 100 Õ **Ö** 10-67 8C

\triangleright	LIGHT PANEL ASSY.	E75MI3292
	ANALOG PANEL ASSY.	E75MI3294
B	SWITCH PANEL ASSY.	E75MI3296
B	POWER MONITOR & CONTROL PANEL ASSY.	E75MI3298

ļ

Figure 5-4 Left Bay Front Panel, Launcher GETS

2.5 11 - 1 1 - 1 ••• •••• 1 - 1 ł . 4. . : 0 • • • • 0 ** ** ** 0 5 5 5 ** 5 5 5 ** READOUT 25 4 304 : - -8 9 V 9 V V ź -5 5 5 5 5 5 655 <u>299</u> 999 \$ 5 5 655 999 -Is S S Ð. Ô ĕ O 3 9 9 • \odot Ú € • C 0 0 • C C _9008_

2>

LIGHT PANEL ASSY.	E75MI3293
ANALOG PANEL ASSY.	E75MI3295
3 SWITCH PANEL ASSY.	
COMMUNICATIONS PANEL ASSY	E75MI3299

Figure 5-5 Right Bay Front Panel, Launcher GETS

MINIATURIZED UNIVERSAL TEST SET (MUTS) E75M13761.

5-8. <u>Description of the MUTS</u>. The Miniaturized Universal Test Set (figure 5-6) is a portable test unit 8 inches high x 13.5 inches wide x 11.5 inches deep, and weighs approximately 35 pounds. It contains these components:

a. 3 meters	Weston Model 3911, 50 microamperes or equal
b. 24 Twist-Lites	Series 10/E; horizontally split display DPDT switch combinations
c. 16 relay sockets	Burndy MTR-92 or equal
d. 19 taper pin blocks	AMP 53 series, 30 pins or equal
e. 16 relays	MSFC-SPEC-339/53

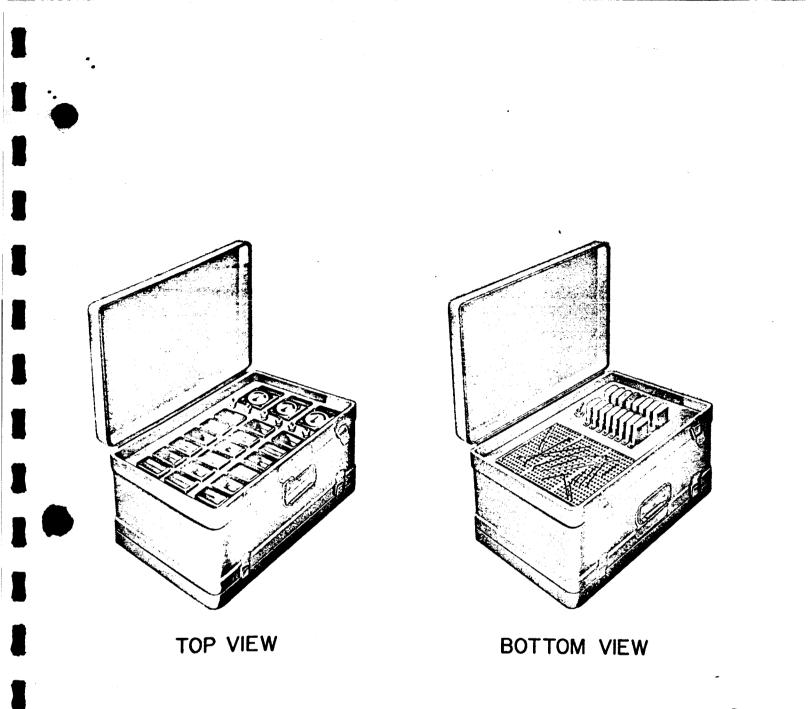
The program of the MUTS is carried to the controlled unit by one or two 61-conductor cables using a pigmy connector (Bendix PT07P-24-61S or equal). Input 28 volts dc is supplied through one 3-pin connector (Cannon BFR22-2P or equal). For self-checks, the MUTS contains a transistorized ground indicator circuit, a POWER AVAILABLE indicator, and a POWER ON indicator.

5-9. <u>Use of the MUTS</u>. The MUTS can be patched to check pneumatic pressures and the operation or position status of valves and relays, and in addition can be hand carried about the Mobile Launcher by test personnel. It is presently used as the Tower Test Set for the Q-Ball Cover Removal Subsystem and for Valve Panel #12.

5-10. Locations of MUTS. The MUTS tests the Q-Ball cover mechanism on level 360 (figure 3-43), and Valve Panel 12 on level 160 (figure 3-40).

5-11. <u>Design Organization</u>. The MUTS was designed by the Electrical Section, Launch Equipment Branch.

5-7.





į

5-12. PORTABLE ARM CONTROL CONSOLE J75M07542.

5-13. <u>Description of the Console</u>. The Portable Arm Control Console (figure 5-7) is a roll-around portable test console approximately 50 inches high, 45 inches wide, and 28 inches deep. It contains its own dc power supply, and so can use either 28 volts dc or 115 volts ac input. The console's arm control panel has the following components:

a.	103 indicator lamps	Lamps Control of America (Series L3000) or equal
b.	1 meter	Minneapolis-Honeywell HS2F, 0-50 ua. or equal
c.	39 switches	Cutler Hammer 8833-K4 & 8821-K6 or equal
Its charging control panel has these components:		

a.	108 indicator lamps	Lamps Control of America (series L3000) or equal
b.	54 switches	Cutler Hammer 8825-K6 or equal

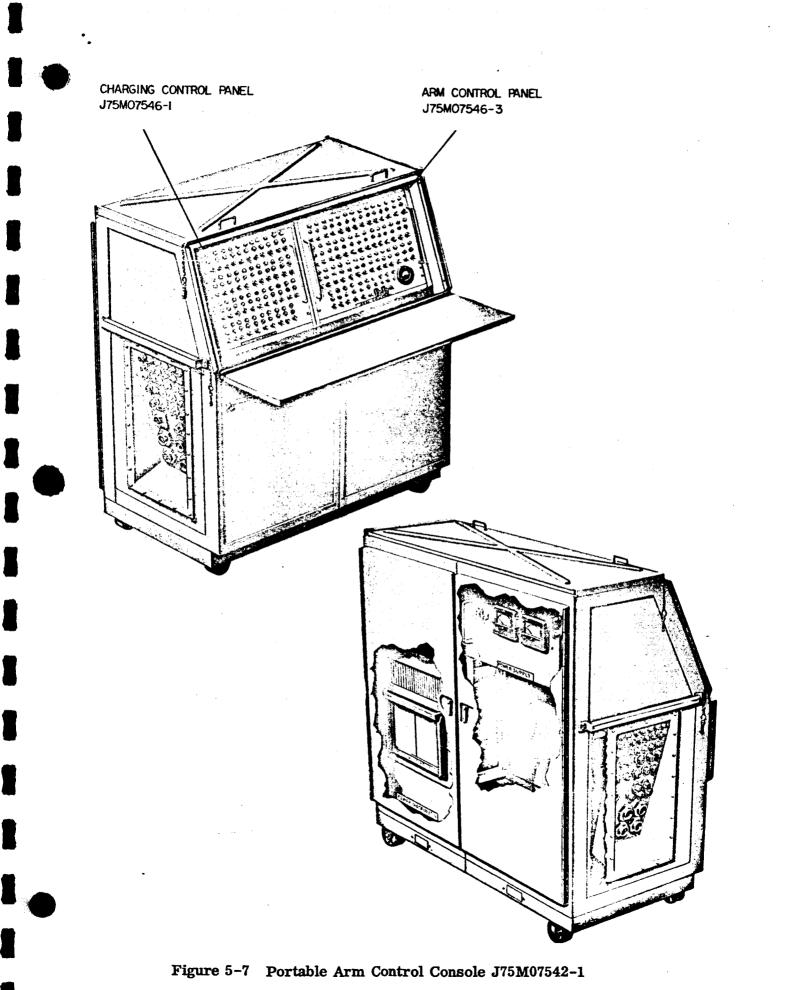
The console also contains a 27-connector patch distributor (paragraph 4-32), plus the following connectors for power and control-function connections:

a.	24 Cannon Connectors	TBFR 16S-1PS or equal
b.	7 Cannon Connectors	TBFR 24-10PS or equal
c.	1 Cannon Connector	TBFR 22-2PS or equal
d.	5 Cannon Connectors	TBFR 22-22PS or equal
е.	1 Bendix Connector	10-248454-75PS or equal
f.	6 Bendix Connectors	10-248454-73PS or equal

5-14. <u>Use of the Console.</u> The Portable Arm Control Console may be programmed to control, test, and monitor the Apollo access arm and any of the eight service arms. Test programs are altered by changing patchboards in its patch distributor. The console is positioned on the tower level associated with the arm under test, and its test cables are connected to distributors and electro-mechanical components as required.

5-15. Locations of Console. The Portable Arm Control Console is used on tower levels 60, 120, 140, 160, 200, 220, 260, and 300.

5-16. <u>Design Organization</u>. The Portable Arm Control Console was designed by the Electrical Section, Launch Equipment Branch.



5-17. UNIVERSAL PATCHBOARD TEST SET E75M09032.

5-18. Description of the Test Set. The Universal Patchboard Test Set (figure 5-8) E75M09032 is a portable test unit approximately 16 inches x 16 inches x 14.9 inches high, weighing about 50 pounds. It contains these test components:

a.	6 Meters	Weston Model 3911, 50 microampere dc or equal
b.	54 Lights	Dialco Socket 101-8430W-975; lamps ASA 327 or equal
c.	27 Switches	Cutler-Hammer 8867K6 or equal
d.	28 Relay Sockets	Burndy MTR-92 or equal
e.	1 Patchboard Assy.	AMP P816D, 695081-3 Frame & Spring or equal AMP 595005-2 Removable Board or equal

The program from the Universal Patchboard Test Set is carried to the controlled unit by one or two 61-conductor cables using pygmy connectors (Bendix PT07P-24-61S or equal). Input 28 volts dc is supplied through one 3-pin connector (Cannon BFR22-2P or equal). For self-checks, the test set contains a ground indicator circuit, a lamp filament test switch, POWER ON and POWER AVAILABLE indicators, and a test socket and switch for individual lamp tests.

5-19. <u>Use of the Test Set.</u> The Universal Patchboard Test Set can be programmed to measure pneumatic pressures and check the operation or position status of various valves and relays. It can be hand carried about the Mobile Launcher by test personnel. At present, the test set has three separate programmings:

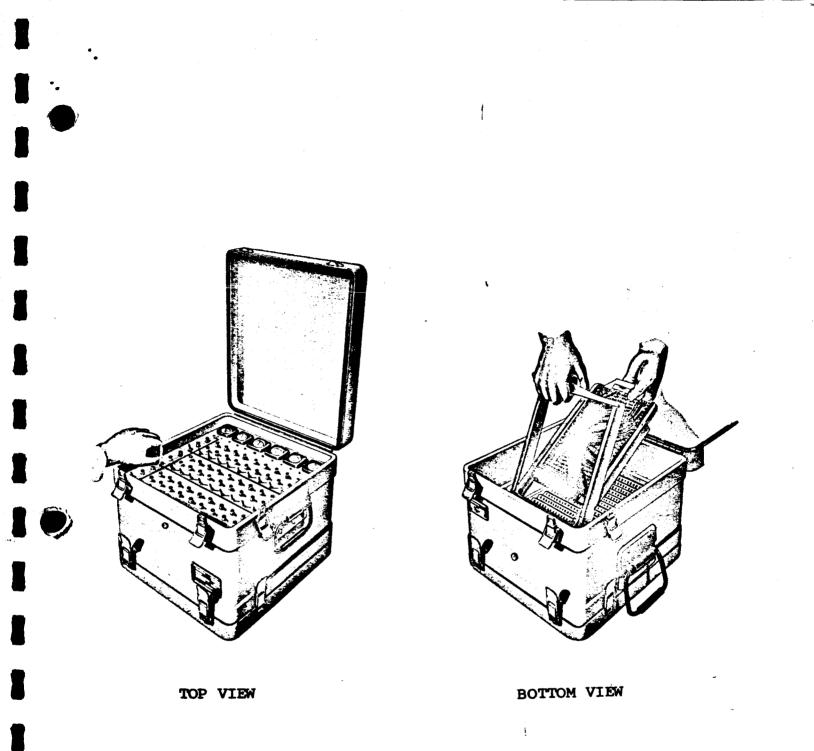
a. Tail Service Mast Test Set, figures 3-33 and 3-34

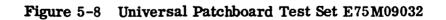
b. Launcher Accessories Test Set No.1, figures 3-37 and 3-38

c. Launcher Accessories Test Set No. 2, figures 3-22 and 3-26

5-20. Locations of Test Set. The Universal Patchboard Test Set is used on levels 0, A, and B.

5-21. <u>Design Organization</u>. The Universal Patchboard Test Set was designed by the Electrical Section, Launch Equipment Branch.





5-22. MSFC SERVICE ARM/TAIL SERVICE MAST TEST SITE.

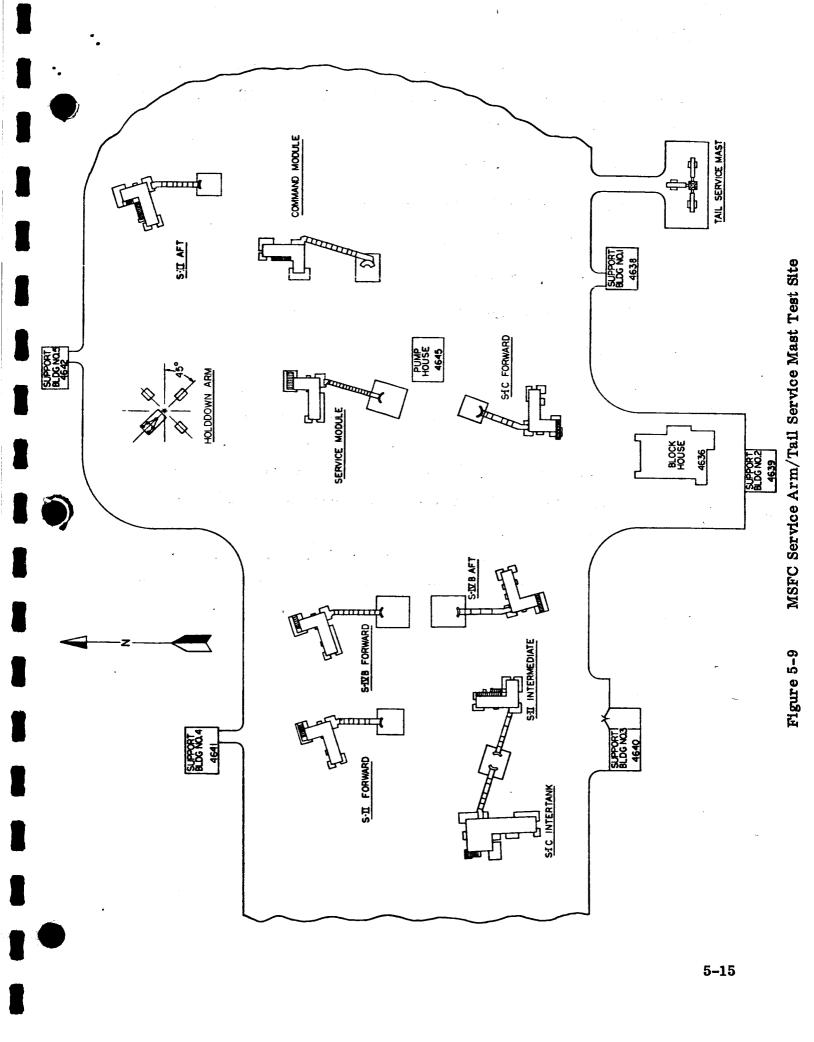
5-23. <u>Purpose of the Test Site.</u> The MSFC Test Site is used to functionally test and qualify for operational use each service arm and tail service mast in conjunction with its associated control equipment. The tests simulate launch conditions and are performed on the equipment prior to delivery to KSC.

5-24. Equipment for the Test Site. The Test Site at Marshall Space Flight Center (figure 5-9) is a test facility comprised of ten test stands, a tail service mast pad, and a block house.

5-25. <u>Service Arm Test Stands</u>. There are nine swing arm test stands, four of which contain preflight vehicle simulators and five of which contain vehicle liftoff simulators. All swing arm test stands contain power and signal distributors, mounting facilities for equipment tested and closed-circuit television for visual monitoring of tests. Four of the test stands also share cryogenic handling equipment and nine stands share an engine exhaust pressure simulator.

On each swing arm test stand there is one service arm which is to be functionally tested before it is installed on a Mobile Launcher at LC-39. Also available for test are the swing arm related control cabinets No.1 and No.2, and hydraulic/pneumatic actuators. This equipment will be provided with the swing arm on delivery to LC-39.

Certain electrical distributors used on the Mobile Launcher for service arm control are facility end items; on the test stands they are replaced by special permanently installed distributors. Hence, each Umbilical Instrumentation and Control Distributor is replaced by a test site Tower Terminal Box. Each Mineral Insulated Firing Distributor is replaced by a special test site Firing Distributor. In addition, a test site Vehicle Receptacle Box on each inflight test stand connects the liftoff switches to the arm control system.



Some of the electrical equipment needed for control and monitoring of the service arm is shown in figure 5-11. Equipment supplied by the Electrical Section, Launch Equipment Branch, includes:

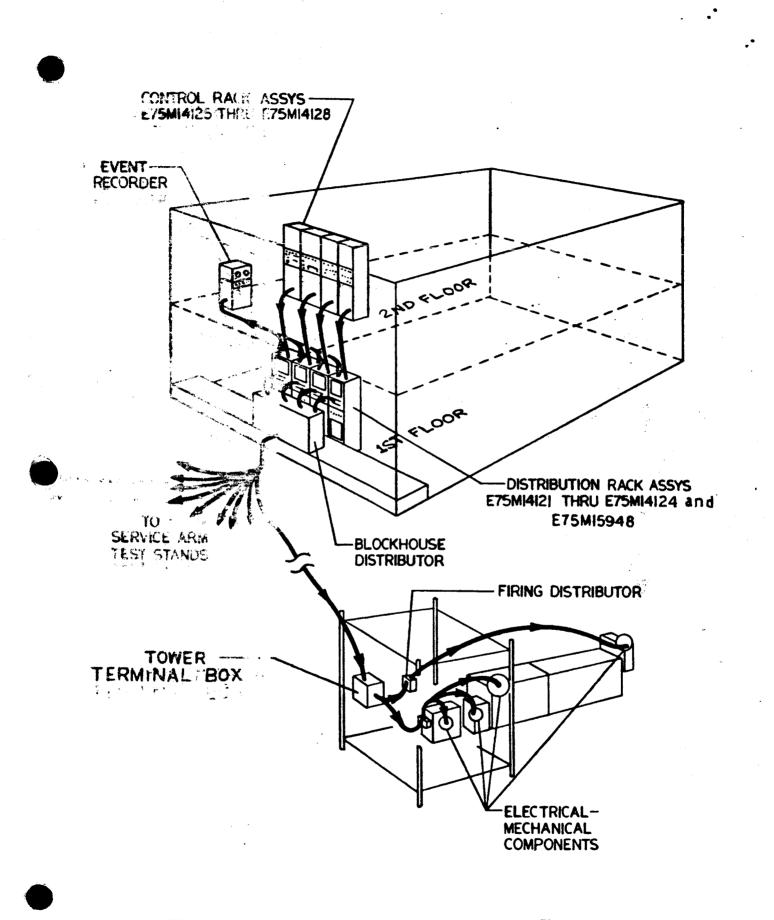
- a. Blockhouse control rack assemblies (4)
- b. Blockhouse distribution rack assemblies (5)
- c. Firing distributors (5)
- d. Tower terminal boxes (9)
- e. Vehicle receptacle boxes (5)
- f. Integration rack (1)

5-26. Electrical Control for the Service Arms. Primary control and monitoring of the service arm under test is conducted from the second floor of the blockhouse. Four control racks (figures 5-11 and 5-12) provide test functions shown on the typical control rack overlay (figure 5-10). There is one overlay for each of the nine arms, one for the liftoff switches, and one for the hydraulic charging unit and holddown arm control.

The control panel overlay shows that the electrical control system can be programmed to allow the swing arm to perform as a complete, automatic system. Alternately, many discrete mechanical functions can be tested separately.

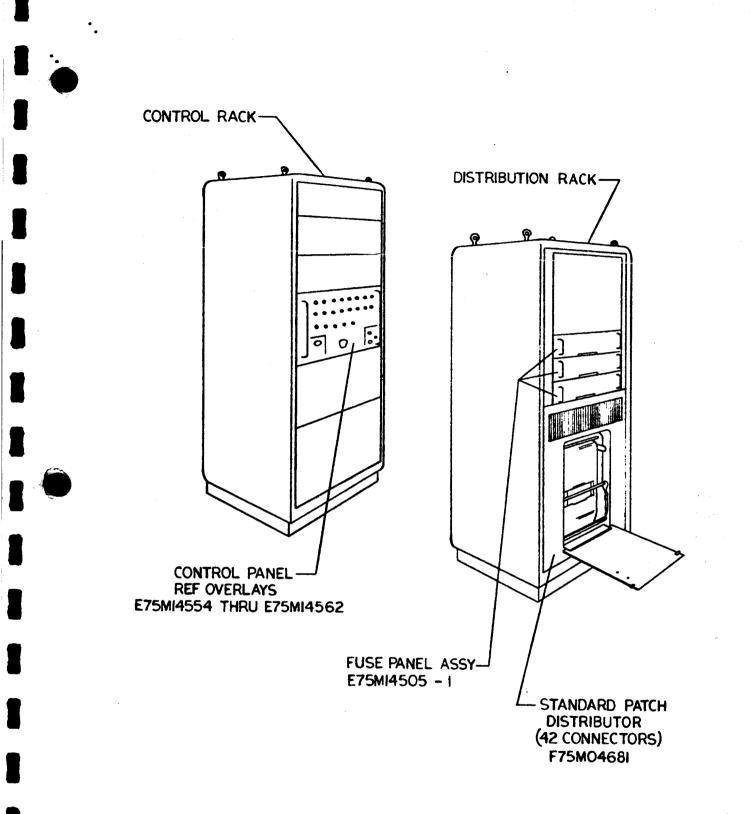
				PRI RET OPEN	SLOT OFEN		S		HY SEC SUP OPEN	D WD PRI SUP OPEN	SE SE		RETR PLT
C i			6	C		6		•	69		0	0	NETR POS
		0	0	6		6			6	0		0	0
RETR PRI SUP OPEN	E Ha	HAND V SEA	GATE GATE CLOSED			8 8 8 8 8 8 8 8 8 8 9				AND PSI	ALL SUP AND RET V G SD	0	6
	6		0	6	UNE NO POS	C	6		C	0	C	6	Q.
ί.			6	6		G	6			6	6	6	S.
	UMB CARR K/O PRI STS CHI PRI LOR	HYD SYSS OM LEWEL LOW	WP SHOW	Phel #1 5:::::::::::::::::::::::::::::::::::	W/D #72 SYS CMS P3N LOW	PRI RETR GM2 STOCK PC PC C	SEC HYD SYS DHA LEVEL LOW	PSI LOT	S S S S S S S S	A C C C C C C C C C C C C C C C C C C C	STANDBY HYD ST3 CHS PS1 LON PS1 LON	ALL TEST	
£	97						CHEVENEDE TRD COFF	OVENINGE TRO GET		DIVERSING AU		OPEN	
1997 - 1991 1997 - 1997 - 1997 1997 - 1997 - 1997 - 1997 - 1997 - 1997 - 1997 - 1997 - 1997 - 1997 - 1997 - 1997 - 1997 - 1997 - 1997 - 1997 -		3	5							2000 6	ARM PC	SITION	
			6	6	6	6		6		RANGE			
	ē	ē	0	•	•	G .,	()	•	6		PERCENT OF		
				Г Г			RMEDIAT	E			PERCENT OF I		

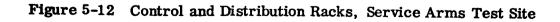
Figure 5-10 Typical Control Panel, Service Arms Test Site



.

Figure 5-11 ESE Equipment, Service Arms Test Site





5-27. <u>Tail Service Masts Test Pad.</u> The tail service masts (TSM) test pad (figure 5-13) has one to three tail service masts, hydraulic and pneumatic systems for the masts, a GN_2 storage tank, a vehicle lifoff simulator, and a control room separate from the swing arm control blockhouse.

Testing of the tail service mast is controlled primarily from the Firing Monitor and Test Panels in the control room. These panels are functionally similar to the TSM control panels in the Launch Control Center at LC-39. They provide control and monitor functions for that portion of the tail service mast system which is furnished by the Electrical Section, Launch Equipment Branch. Other equipment provided by the Electrical Section includes:

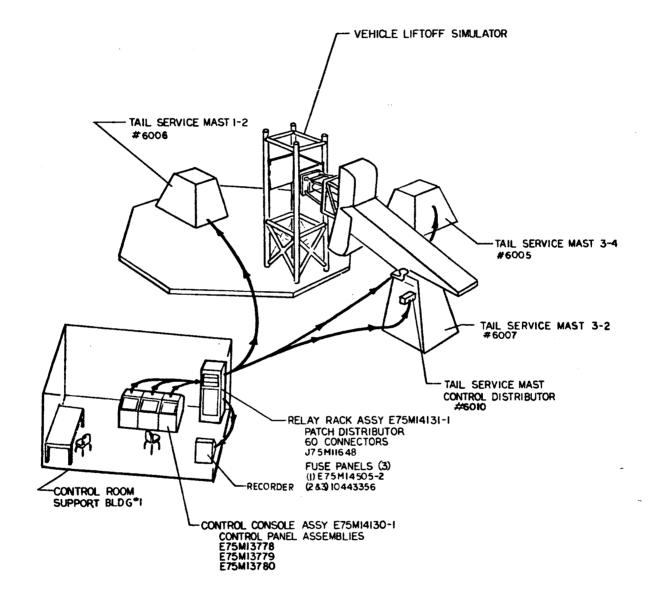
- a. Tail Service Mast 1-2 #6006 (Internal Cables)
- b. TSM 1-2 Control Distributor #6009
- c. Tail Service Mast 3-2 #6007 (Internal Cables)
- d. TSM 3-2 Control Distributor #6010
- e. Tail Service Mast 3-4 #6005 (Internal Cables)
- f. TSM 3-4 Control Distributor #6008
- g. Tail Service Mast Distribution Rack #T6650

h. Firing Monitor and Test Panel 1-2

i. Firing Monitor and Test Panel 3-2

j. Firing Monitor and Test Panel 3-4

Control power (28 volts dc) and an event recorder are furnished by the MSFC test facility.



••

Figure 5-13 Tail Service Masts Test Site, MSFC

5-28. Electrical Control for the Tail Service Masts. Before a tail service mast is tested, pneumatic and hydraulic accumulators for its actuators are charged and vented, and appropriate lines are bled. Some of these preparations require adjusting various pneumatic and hydraulic valves by hand. Other valves in the system are actuated by solenoids, and are therefore electrically controlled and monitored from the Firing Monitor and Test Panel, figure 5-14. When the monitor lamps and meters on this panel indicate that all conditions are ready, the mast extension operation can be started.

By the use of the MAST LOCK RELEASE and MAST EXTEND switches on the tail service mast, an operator now gradually extends the mast toward the vehicle liftoff simulator. He is in communication with a control room technician who monitors lamps and meters on the Firing Monitor and Test Panel. Final connection of the mast's umbilical carrier to the vehicle liftoff simulator is completed manually.

The mast retract test is then conducted from the Firing Monitor and Test Panel. Closure of the MAST RETRACT PRESET and MAST RETRACT ACTUATE switches retracts any one mast or any combination of the three, as required. Switch closure provides a 28 volt dc signal to relay rack #T6650, which in turn performs all necessary relay logic and transmits control signals to appropriate components in the mast. All such events are recorded by instrumentation equipment.

Alternately, the mast can be armed for retract and actuated by motion of the vehicle liftoff simulator. The simulator itself is MSFC test facility equipment, and is not controlled by the KSC Firing Monitor and Test Panel.

As the mast retracts, all major movements of the mast and its actuators are detected by transducers, limit switches, and potentiometers. Signals from these components return to relay distributor #T6650 and subsequently to lamps on the control panel. All such events are also recorded by instrumentation equipment.

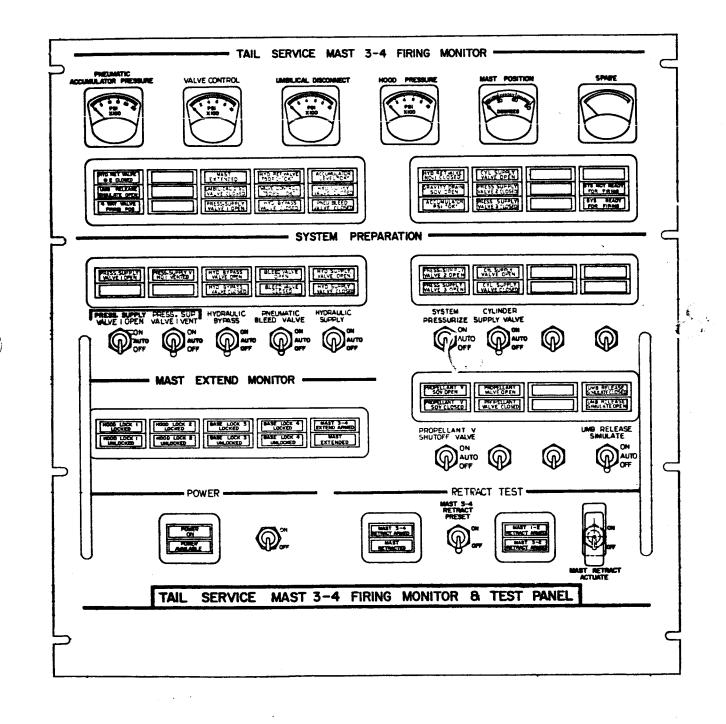


Figure 5-14 TSM Test Site Control Panel E75M13780

5-23/24

SECTION VI

REFERENCE INFORMATION ELECTRICAL SUPPORT EQUIPMENT

6-1. ESE FAMILY TREE OF DRAWINGS.

The ESE Family Tree of Drawings illustrates, by generation breakdown, that Electrical Launch Support Equipment (ELSE) which is the responsibility of the Launch Equipment Branch, EDV-15. The tree contains the following branches:

LC-39 Ground Support Equipment	J75M05970
Off LUT Installation LUT-1	J75M05971
Launcher Installation LUT-1	J75M05972
Launcher Deck Installation LUT-1	J75M05973
Tower Supply Installation LUT-1	J75M05974
Tower Platforms Installation LUT-1	J75M05975
LC-39 GSE Test Equipment	
F-1 Engine Servicing Equipment	J75M13482

The ESE Family Tree of Drawings may be found in Volume II of this ESE Electrical Reference Handbook.

6-2. **REFERENCE** SPECIFICATIONS.

The documents listed in Table 6-1 will be used by contractors for Launch Equipment Branch fabrication and installation. Table 6-2 contains substitutions and deletions of specifications and procedures. Abbreviations for cable specifications are listed in Table 6-3.

1

DOGIN	
DOCUMENT	DESCRIPTION
NUMBER	DESCRIPTION
	KENNEDY SPACE CENTER DOCUMENTS
KSC-S-101	Semiconductor Devices, General Specification
KSC-S-101/1	Diode, Zener, Silicon, Power, Type K1N2970B
KSC-S-101/2	Transistor, NPN Switching, Type K2N914
KSC-S-101/3	Transistor, NPN, General Purpose, Type K2N2192A
KSC-S-101/4	Transistor, PNP Switching, Type K2N2905A
KSC-S-101/5	Transistor, Power, NPN Silicon, Type K2N1724
KSC-S-101/6	Transistor, NPN, Dual, Silicon, K2N2060
KSC-S-101/7	Diode, Silicon, High Speed Switching, Type K1N4153
KSC-S-101/8	Diode, Silicon, General Purpose, Type K1N645
KSC-S-101/9	Diode, Zener, Silicon, Type K1N967B
KSC-S-101/10	Diode, Silicon, Power Rectifier, Type K1N250C
KSC-S-101/11	Transistor, NPN, Switching, Type K2N2846
KSC-S-101/12	Transistor, NPN, Silicon, Power, Type ZN2811
KSC-S-101/13	Diode, Zener, Silicon, Power, Type K1N2814A,
	K1N2827A, K1N2828A
KSC-S-101/14	Transistor, NPN, Silicon, Power, Type K2N3265
KSC-S-101/15	Diode, Rectifier, Silicon, Power, Fast Recovery,
	Type K1N3902
KSC-S-101/16	Transistor, PNP Silicon, Power, Type K2N3026
KSC-S-101/17	Transistor, PNP, Silicon, Power, Type K2N3026
KSC-S-101/18	Transistor, PNP, Silicon, Type K3516
KSC-S-101/19	Transistor, NPN, Silicon, Power, Type K2N2658
KSC-S-101/20	Transistor, PN Silicon, Unijunction, Type K2N493
KSC-S-101/21	Transistor, PNP, Silicon, Type K2N2276
KSC-S-101/22	Transistor, NPN, Silicon, High Power, Type K2N2746
KSC-S-101/23	Transistor, NPN, Silicon, Type K2N2432
KSC-S-101/24	Diode, Temperature Compensated Reference, Sub-
	miniature, Type K1N825A
KSC-R-102	Resistors, Fixed, Film (High Stability) (Styles
	KRN 55/60/65/70C)

Table 6-1 REFERENCE SPECIFICATIONS

4

DOCUMENT NUMBER	DESCRIPTION
	KENNEDY SPACE CENTER DOCUMENTS (Continued)
KSC-R-103	Resistors, Fixed, Film (High Stability) (Styles KRL 07/20/32/42)
KSC-R-104	Resistor, Fixed, Power, Wirewound (Chassis Mount) (Style KRE 65/70/75)
KSC-R-105	Resistors, Fixed, Wirewound (Insulated) (Style KRW 67/68/69)
KSC-C-106	Capacitor, Fixed, Glass Dielectric, High Reliability (Styles CYER 10/15/20/30)
KSC-C-107	Capacitors, Fixed, Tantalum (Polarized, Etched Foil) (Style KCL 21)
KSC-C-108	Capacitors, Fixed, Tantalum Non-Polar Etched Foil (Style KCL 23)
KSC-C-109	Capacitors, Fixed, Tantalum (Polarized, Plain Foil) (Style KCL 31)
KSC-C-110	Capacitors, Fixed, Tantalum (Nonpolarized, Plain Foil) (Style KCL 33)
KSC-C-111	Capacitors, Fixed, Solid, Tantalum (Style KCS13)
KSC-C-112	Capacitors, Fixed, Plastic Dielectric, (Nonmetallic Case) (Style KCTM)
KSC-C-113	Capacitors, Fixed, Polarized, Tantalum Foil (Style KCL 51 and KCL 53)
KSC-C-114	Capacitors, Fixed, Paper or Paper-Plastic (Style KCVP09)
KSC-P-116	Packaging and Marking for Cables and Harnesses, Procedure for
KSC-STD-132	Potting and Molding Cable Assembly Termination
KSC-W-151	Solderless Wrap Process, Electrical Connections, Specification for
KSC-STD-152-1	Graphical Symbols for Drawings, Part 1 – Electrical and Logic Symbols
KSC-STD-152-2	Graphical Symbols for Drawings, Parts, Mechanical Symbols
KSC-E-153	Enclosures, Modular, Radio Frequency Interference Shielded, Specification for
KSC-STD-164	Environmental Test Methods for Ground Support Equipment Installations at John F. Kennedy Space Center
KSC-E-165	Electrical Ground Support Equipment Fabrication
KSC-W-167	Wiring Programming System Patchboards, Procedure for
KSC-STD-169	Marking of Ground Support Equipment
SP-4-28-D	Design Test Data, Mechanical and Electro-Mechanical Components

DOCUMENT NUMBER	DESCRIPTION
	KENNEDY SPACE CENTER DOCUMENTS (Continued)
SP-4-38-D	Shock and Vibration Environments and Test Specification Levels, Ground Support Equipment, Launch Complex 39
SP-80-D	Guide for Environmental Protection When Using Electrical Ground Equipment Within the Areas of
SP-82-D	Saturn Complexes Where Hazardous Areas Exist Electrical and Electro-Mechanical Components for Saturn Launch Complexes
10M01671	Cleanliness Levels, Cleaning, Protection and Inspection Procedures for Parts, Assys., Sub-systems, and Systems for Pneumatic Use in Support Equipment, Specification for
A10430105	Sealing Compound
A75M00100	Pneumatic and Mechanical Components, Electrical Spec for
B75M012383	Ring Nut (for Cannon 40 Shell Size Connectors)
B75M02122	Cable Plug Assy (2-3/8 Thd)
B75M02123	Plug Hull (for Cannon 40 Shell Size Connectors)
B75M02124	Nut (for Cannon and Bendix 40 Shell Size Connectors)
B75M02177	Boot (for Banana Plug)
B75M02183	Ring Nut (for Bendix 40 Shell Size Connectors)
C75M02196	Adapter
C75M02197	Protection Cap (Umbilical)
C75M02197-1	Boot
B75M02943	Cable Plug Assy (2-5/16 Thd)
B75M02944	Plug Hull (for Bendix 40 Shell Size Connectors)
A75M03589	Application of Heat and Blast Resistant Coating of Cables, Proc for
B75M04097-X	Adapter (Connector to Conduit) (Refer to dwg. for dash no.)
F75M05218	Cast Chassis, Complex 39
A75M05668	Solderless Electrical Connections, Proc for
A75M05875	Fabrication and Installation of Tube Assemblies and Installation of Fitting and Fitting Assemblies, Spec for
A75M07450	MI Cable Terminations, Proc for
B75M07830	Adapter
A75M09397	Electrical Cable Assembly, Spec for
A75M09465	Splicing and Serving Lifting Cable, Procedure for
A75M09467	Hydraulic System Components and Hydraulic Fluids for Ground Support Equipment (GSE) Cleaning, Testing, and Handling Procedure
A75M09468	Sealing of Electrical Components, Spec for

DOCUMENT								
NUMBER	DESCRIPTION							
	KENNEDY	SPACE C	ENTER	DOCUMENTS (Continued)				
A75M09470	Welding Stainless Steel and Invar Pipe, Specification for							
A75M09946	Welding of A	Welding of Aluminum Alloy Pipe, Specification for						
A75M11300	Tape, Marke	er (Green -	-1, Red	-2)				
A75M12110	Heat and Bla	st Protecti	ve Coat	ting				
A75M13302	Connector In	spection Sp	pecifica	tions				
C75M13303				ose Connector				
C75M13304		•	-	ose Connector				
C75M13676	Shield Rings		-					
A75M13965		ration Met	hod of I	Determining Non-Volatile				
	• -			Trichloroethylene Solvents,				
	Procedure for							
B75M50703	Adapter (Fle	xible Armo	ored Ca	ble)				
A75M51074				ort Equipment Installation				
A75M10180	6 Cond.	16 AWG	os	6 UNSH				
A75M10181	60 Cond.	16 AWG	OS	60 UNSH				
A75M10182	60 Cond.	16 AWG						
A75M10183	60 Cond.	16 AWG	OS	30 PTSI				
A75M10184	60 Cond.	16 AWG	OS	20 TTSI				
A75M10185	60 Cond.	16 AWG	OS	15 QTSI				
A75M10186	60 Cond.	16 AWG	OS	60 SSI				
A75M10187	60 Cond.	14 AWG	OS	2TTSI, 3PTSI, 48UNSH				
A75M10188	60 Cond.	14 AWG	OS	3TTSI, 3PTSI, 45SSI				
A75M10189	7 Cond.	8 AWG	OS	7 UNSH				
A75M10190	60 Cond.	20 AWG	OS	60 UNSH				
A75M10191	60 Cond.	14 AWG	OS	60 UNSH				
A75M10192	60 Cond.	14 AWG	OS	60SS				
A75M10193	60 Cond.	14 AWG	OS	30 PTSI				
A75M10194	60 Cond.	14 AWG	OS	20 TTSI				
A75M10195	60 Cond.	14 AWG	OS	15 QTSI				
A75M10196	10 Cond.	16 AWG	OS	10 UNSH				
A75M10197	6 Cond.	8 AWG	OS	6 UNSH				
A75M10198	4 Cond.		OS	3#8, 1#16				
A75M10199	5 Cond.	12 AWG		5 UNSH				
A75M10200	4 Cond.	0	OS	4 UNSH				
A75M10201	61 Cond.	20 AWG		61 UNSH				
A75M10202	32 Cond.	20 AWG	OS	32 UNSH				
A75M10203	3 Cond.	8 AWG		3 UNSH				
A75M10204	6 Cond.		OS	4RG214/U, 1RG62A/U, 1#12				

ŧ

.

. . :

• • • •

DOCUMENT NUMBER	DESCRIPTION							
	KENNEDY SPACE CENTER DOCUMENTS (Continued)							
A75M11003	40 Cond.		OS	5 QTSI#10, 20UNSH#14				
A75M11004	52 Cond.		OS	4PTSI#7, (11TTSI, 5PTSI,	1SSI#14			
A75M11005	60 Cond.	14 AWG	OS	5PTSI, 50SSI				
A75M11006	33 Cond.		OS	3#0, 3#4, 13PTSI#14, ISSI#	14			
A75M11007	29 Cond.		OS	4 #2, 9# 6, 16#1 4				
A75M11008	40 Cond.	14 AWG	OS	20 PTSI				
A75M11009	60 Cond.	14 AWG	OS	4TTSI, 12PTSI, 24SSI,				
				ABLATIVE SHEATH				
A75M11010	60 Cond.	14 AWG	OS	5PTSI, 50 UNSH	AB			
A75M11011	60 Cond.	14 AWG	OS	2TTSI, 3PTSI, 48 UNSH	AB			
A75M11012	60 Cond.	14 AWG	OS	6TTSI, 17PTSI, 8SSI	AB			
A75M11013	60 Cond.	14 AWG	OS	15 OTSI	AB			
A75M11014	60 Cond.	14 AWG	OS	20 TTSI	AB			
A75M11015	10 Cond.	16 AWG	OS	10 UNSH	AB			
A75M11016	60 Cond.	16 AWG	OS	60 UNSH	AB			
A75M11017	6 Cond.	16 AWG	OS	6 UNSH	AB			
A75M11018	4 Cond.	0	OS	4 UNSH	AB			
A75M11019	29 Cond.		OS	4#2, 9#6, 16#14	AB			
A75M11020	40 Cond.		OS	5QTSI#10, 20 UNSH#14	AB			
A75M11021	52 Cond.		OS	4PTSI#7, (11TTSI,				
				5PTSI, 1SSI#14)	AB			
A75M11022	60 Cond.	14 AWG	OS	5PTSI, 50SSI	AB			
A75M11023	33 Cond.		OS .	3#0, 3#4, 13PTSI#14,				
				1SSI#14	AB			
A75M11024	60 Cond.	14 AWG	OS.	60 SSI	AB			
A75M11025	60 Cond.	14 AWG	OS	30 PTSI	AB			
A75M11026	60 Cond.	14 AWG	OS	60 UNSH	AB			
A75M11027	6 Cond.		OS	4RG214/U, 1RG62A/U,				
				1#12	AB			
A75M14073	60 Cond.	14 AWG	OS	6TTSI, 17PTSI, 8SSI				
A75M14074	7 Cond.		OS	7RG180B/U				
A75M14075	60 Cond.	16 AWG		60 UNSH				
A75M14076	60 Cond.	14 AWG	OS	5PTSI, 50UNSH				
A75M14077	21 Cond.	8 AWG	OS	21 UNSH				
A75M14078	1							
A75M14079								
A75M14080								
A75M14081								

NUMBER	DESCRIPTION								
;	KENNEDY SPACE CENTER DOCUMENTS (Continued)								
A75M14082									
A75M14083	• •		<i>t</i> .		• •				
A75M14084			•		• .				
A75M14085	,			·	•				
A75M14086					•*				
A75M14087				;			• · · _		
A75M14088					st. • .				
A75M14089									
A75M14090	1	•			•				
A75M14091							,		
A75M14092	6 Cond.	8 AWG	OS	6 UNSH		AB			
							•		
	MAR	SHALL SP	ACE FL	IGHT CEN	TER DOCU	MENTS			
MSFC-STD-105	Synthetic	Rubber, A	Age Cont	rol of. Std	for		٠,		
MSFC-SPEC-106		Compatibil							
MSFC-SPEC-119		rs, Recep							
MSFC-SPEC-130					ators and	Welders.			
	Spec for								
3	-	Diama J m							
MSFC-SPEC-143	Fittings.	riared It	ıbe (Prei	nium Qual	ity) Press	ire			
MSFC-SPEC-143			•	nium Qual	ity) Pressu	ire			
	Connectio	ons, Spec	for	-	•7	ire			
MSFC-SPEC-143 MSFC-STD-154 MSFC-STD-156	Connection Printed C	ons, Spec Circuit Des	for sign & Co	nstructior	h, Std for	ıre	•		
MSFC-STD-154 MSFC-STD-156	Connection Printed C Riveting,	ons, Spec Circuit Des Fabricati	for sign & Co ion and Ir	onstruction spection,	n, Std for Std for				
MSFC-STD-154	Connection Printed C Riveting, Cleanline	ons, Spec Circuit Des Fabricati ess of Com	for sign & Co on and In ponents	onstruction aspection, for use in	h, Std for				
MSFC-STD-154 MSFC-STD-156	Connection Printed C Riveting, Cleanline and Pneu	ons, Spec Circuit Des Fabricati ess of Com matic Syst	for sign & Co on and In ponents cems, Sp	onstruction aspection, for use in ec for	n, Std for Std for Oxygen, F	uel,			
MSFC-STD-154 MSFC-STD-156 MSFC-SPEC-164	Connection Printed C Riveting, Cleanline and Pneu Cleanline	ons, Spec Circuit Des Fabricati ess of Com matic Syst ess Level 1	for sign & Co on and In ponents cems, Sp Requiren	onstruction spection, for use in ec for uents and I	n, Std for Std for	uel, Methods	•		
MSFC-STD-154 MSFC-STD-156 MSFC-SPEC-164	Connection Printed C Riveting, Cleanline and Pneu Cleanline for Deter	ons, Spec Circuit Des Fabricati ess of Com matic Syst ess Level 1	for sign & Co on and In ponents tems, Sp Requirem eanliness	onstruction aspection, for use in ec for aents and I s Level of	n, Std for Std for Oxygen, F nspection I Gas Bearin	uel, Methods			
MSFC-STD-154 MSFC-STD-156 MSFC-SPEC-164 MSFC-PROC-195	Connection Printed C Riveting, Cleanline and Pneu Cleanline for Deter Supply an	ons, Spec Circuit Des Fabricati ess of Com matic Syst ess Level 1 mining Cl ad Slash M	for sign & Co on and In ponents cems, Sp Requirem eanliness easuring	onstruction aspection, for use in ec for aents and I s Level of Systems,	n, Std for Std for Oxygen, F Inspection I Gas Bearin Proc for	uel, Methods ng Gas			
MSFC-STD-154 MSFC-STD-156 MSFC-SPEC-164 MSFC-PROC-195 MSFC-SPEC-202	Connection Printed C Riveting, Cleanline and Pneu Cleanline for Deter Supply an Compoun	ons, Spec Circuit Des Fabricati ess of Com matic Syst ess Level 1 mining Cl ad Slash M d, Potting	for sign & Co on and In ponents tems, Sp Requirem eanliness easuring and Mol	onstruction spection, for use in ec for ents and I s Level of Systems, ding, Elas	n, Std for Std for Oxygen, F nspection I Gas Bearin Proc for tomeric, S	uel, Methods ng Gas Spec for			
MSFC-STD-154 MSFC-STD-156 MSFC-SPEC-164 MSFC-PROC-195 MSFC-SPEC-202	Connection Printed C Riveting, Cleanline and Pneu Cleanline for Deter Supply an Compoun	ons, Spec Circuit Des Fabricati ess of Com matic Syst ess Level 1 mining Cl d Slash M d, Potting ystems, E	for sign & Co on and In ponents tems, Sp Requirem eanliness easuring and Mol	onstruction spection, for use in ec for ents and I s Level of Systems, ding, Elas	n, Std for Std for Oxygen, F Inspection I Gas Bearin Proc for	uel, Methods ng Gas Spec for			
MSFC-STD-154 MSFC-STD-156 MSFC-SPEC-164 MSFC-PROC-195 MSFC-SPEC-202 MSFC-SPEC-222	Connection Printed C Riveting, Cleanline and Pneu Cleanline for Deter Supply an Compoun Resins Sy Epoxy, S	ons, Spec Circuit Des Fabricati ess of Com matic Syst ess Level I mining Cl ad Slash M d, Potting ystems, E pec for	for sign & Co on and In ponents tems, Sp Requirem eanliness easuring and Mol lec and F	onstruction aspection, for use in ec for aents and I s Level of Systems, ding, Elas anvironme	n, Std for Std for Oxygen, F nspection I Gas Bearin Proc for tomeric, S	uel, Methods ng Gas pec for ution,			
MSFC-STD-154 MSFC-STD-156 MSFC-SPEC-164 MSFC-PROC-195 MSFC-SPEC-202 MSFC-SPEC-222	Connection Printed C Riveting, Cleanline and Pneu Cleanline for Deter Supply an Compoun Resins Sy Epoxy, S Fittings,	ons, Spec Circuit Des Fabricati ess of Com matic Syst ess Level I mining Cl ad Slash M d, Potting ystems, E pec for	for sign & Co on and In ponents tems, Sp Requirem eanliness easuring and Mol lec and F hreads (onstruction aspection, for use in ec for aents and I s Level of Systems, ding, Elas anvironme	n, Std for Std for Oxygen, F Inspection I Gas Bearin Proc for tomeric, S Intal, Insula	uel, Methods ng Gas pec for ution,			
MSFC-STD-154 MSFC-STD-156 MSFC-SPEC-164 MSFC-PROC-195 MSFC-SPEC-202 MSFC-SPEC-222 MSFC-PROC-238	Connection Printed C Riveting, Cleanline and Pneu Cleanline for Deter Supply an Compoun Resins Sy Epoxy, S Fittings, and Lubr	ons, Spec Circuit Des Fabricati ess of Com matic Syst ess Level 1 mining Cl ad Slash M d, Potting ystems, E pec for Female T	for sign & Co on and In ponents cems, Sp Requirem eanliness easuring and Mol- lec and H hreads (roc for	onstruction aspection, for use in ec for aents and I s Level of Systems, ding, Elas Invironmen Prem. Qua	a, Std for Std for Oxygen, F nspection I Gas Bearin Proc for tomeric, S ntal, Insula al.), Prepa	uel, Methods ng Gas pec for ution,			
MSFC-STD-154 MSFC-STD-156 MSFC-SPEC-164	Connection Printed C Riveting, Cleanline and Pneu Cleanline for Deter Supply an Compoun Resins Sy Epoxy, S Fittings, and Lubr Bonding a	ons, Spec Circuit Des Fabricati ess of Com matic Systess Level 1 mining Cl ad Slash M d, Potting ystems, E pec for Female T ication, P and Ground	for sign & Co on and In ponents cems, Sp Requirem eanliness easuring and Mol lec and H hreads (roc for ting, Ele	onstruction aspection, for use in ec for aents and I s Level of Systems, ding, Elas anvironmen Prem. Qua ctrical, Sp	a, Std for Std for Oxygen, F Inspection I Gas Bearin Proc for tomeric, S Intal, Insula al.), Prepa	uel, Methods ng Gas Spec for ation, aration			
MSFC-STD-154 MSFC-STD-156 MSFC-SPEC-164 MSFC-PROC-195 MSFC-SPEC-202 MSFC-SPEC-222 MSFC-PROC-238 MSFC-PROC-238	Connection Printed C Riveting, Cleanline and Pneu Cleanline for Deter Supply an Compoun Resins Sy Epoxy, S Fittings, and Lubr Bonding a Conforma	ons, Spec Circuit Des Fabricati ess of Com matic Systess Level 1 mining Cl ad Slash M d, Potting ystems, E pec for Female T ication, P and Ground	for sign & Co on and In ponents tems, Sp Requirem eanliness easuring and Mol- lec and H hreads (roc for ting, Ele of Printe	onstruction aspection, for use in ec for ents and I s Level of Systems, ding, Elas Invironmen Prem. Qua ctrical, Spectorical	a, Std for Std for Oxygen, F Oxygen, F Inspection I Gas Bearin Proc for tomeric, S Intal, Insula al.), Prepa pec for Assys, Pr	uel, Methods ng Gas Spec for ation, aration			
MSFC-STD-154 MSFC-STD-156 MSFC-SPEC-164 MSFC-PROC-195 MSFC-SPEC-202 MSFC-SPEC-222 MSFC-PROC-238 MSFC-PROC-238 MSFC-SPEC-249 MSFC-PROC-257	Connection Printed C Riveting, Cleanline and Pneu Cleanline for Deter Supply an Compoun Resins Sy Epoxy, S Fittings, and Lubr Bonding a Conforma	ons, Spec Circuit Des Fabricati ess of Com matic Syst ess Level 1 mining Cl d Slash M d, Potting ystems, E pec for Female T ication, P and Ground al Coating Heat React	for sign & Co on and In ponents cems, Sp Requirem eanliness easuring and Mol- lec and H hreads (roc for ling, Ele of Printe	onstruction aspection, for use in ec for aents and I s Level of Systems, ding, Elas invironmen Prem. Qua ctrical, Sp ed Circuit 1. of, Pro	a, Std for Std for Oxygen, F Oxygen, F Inspection I Gas Bearin Proc for tomeric, S Intal, Insula al.), Prepa pec for Assys, Pr	uel, Methods ng Gas Spec for ation, aration			
MSFC-STD-154 MSFC-STD-156 MSFC-SPEC-164 MSFC-PROC-195 MSFC-SPEC-202 MSFC-SPEC-222 MSFC-PROC-238 MSFC-PROC-238 MSFC-PROC-257 MSFC-PROC-273	Connection Printed C Riveting, Cleanline and Pneu Cleanline for Deter Supply an Compoun Resins Sy Epoxy, S Fittings, and Lubr Bonding a Conforma Tubing, F	ons, Spec Circuit Des Fabricati ess of Com matic Syste ess Level 1 mining Cl d Slash M d, Potting vstems, E pec for Female T ication, P and Ground al Coating	for sign & Co on and In ponents cems, Sp Requirem eanliness easuring and Mol- lec and H hreads (roc for ting, Ele of Printe ive, Inst of, Proc	onstruction aspection, for use in ec for aents and I s Level of Systems, ding, Elas invironmen Prem. Qua ctrical, Sp ed Circuit 1. of, Pro for	a, Std for Std for Oxygen, F Oxygen, F Inspection I Gas Bearin Proc for tomeric, S Intal, Insula al.), Prepa pec for Assys, Pr	uel, Methods ng Gas Spec for ation, aration			

••

s às**∳**-

	DOCUMENT NUMBER	DESCRIPTION
		MARSHALL SPACE FLIGHT CENTER DOCUMENTS (Continued)
	MSFC-SPEC-278 (Continued)	Terminals, Bifurcated and Turret, Standoff, Insulated Screw Type, Spec for
	MSFC-PROC-310	Potting of Electrical Distributors, Proc for
	MSFC-SPEC-331	Enclosures, Modular Shielded, RFI, Spec for
	MSFC-SPEC-332	Cables, Electrical General Spec for (Slash Sheets not applicable)
	MSFC-QPL-332	Qualified Products List for MSFC-SPEC-332
	MSFC-SPEC-338	Semiconductor Devices, General Spec for
		(Slash Sheets/1 through/119 may apply)
	MSFC-SPEC-339	Relays, DC, Hermetically Sealed, for Space Vehicles
		and GSE, General Spec for
	MSFC-SPEC-	
	339/1 thru/94	Individual Relay Specs
	MSFC-STD-350	Abbreviations for use on Drawings, Std for
	NPC 200-2	Quality Program Provisions for Space System Contractors
	NPC 200-3	Inspection System Provisions for Supplies of Space Materials,
	•	Parts, Components, and Services
	NPC 200-4	Quality Requirements for Hand Soldering of Electrical Connections
	D10507600	Flared Tube Fittings Concentricity Gage
	A10509301	Riveting, Fabrication, and Inspection, Standard for
	A10509308	Welding, Carbon, Low Alloy and Stainless Steel, (Manual or Auto)
		Spec for
	A10M01671	Cleanliness Levels, Cleaning, Protection and Inspection
	•	Procedures for Parts, Assemblies, Subsystems, and Systems,
		for Pneumatic use in Support Equipment, Spec for
	A50M01161	Aluminum GSE (Except Equipment Racks) Paint Finishing of,
		Proc for
	A50M01162	Aluminum Equipment Racks, GSE, Paint Finishing of, Proc for
I	A50M01165	Aluminum Electrically Conductive Chromate Coating of, Proc for
	ī	MILITARY STANDARDS AND SPECIFICATIONS
	MIL-STD-28	Drawing Titles, Approved Method For Assignment of
	MIL-STD-108	Definitions of and Basic Requirements for Enclosures For
		Electric and Electronic Equipment
	MIL-STD-129	Marking for Shipment and Storage
	MIL-STD-163	Steel Mill Products Preparation for Shipment and Storage
	MIL-STD-171	Finishing of Metal and Wood Surfaces

••

•

DOCUMENT NUMBER	DESCRIPTION
	MILITARY STANDARDS AND SPECIFICATIONS (Continued)
MIL-STD-183	Canceled (See FED-STD-183)
MIL-STD-429	Printed Circuit Terms & Definitions
MIL-STD-810	Environmental Test Methods for Aerospace and Ground
	Equipment
MIL-A-1154	Adhesive, Bonding, Vulcanized Synthetic Rubber to Steel
MIL-A-6091	Alcohol, Ethyl, Specially Denatured, Aircraft
MIL-A-8625	Anodic Coatings, for Aluminum and Aluminum Alloys
MIL-A-25994	Aluminum Alloy Angles, Channels I and Z Beams,
	Extruded or Rolled Structural Shapes
MIL-B-117	Bags, Interior Packaging
MIL-B-5087	Bonding, Electrical (For Aircraft) ASG#
MIL-B-7883	Brazing of Steels, Copper, Copper Alloys, and Nickel Alloys
MIL-B-22205	Bags, Transparent, Flexible, Heat Sealable for Packaging
	Applications
MIL-B-43014	Boxes, Water Resistant Paper Board, Folding, Set-up and
	Metal-Stayed
MIL-C-17	Cables, Radio Frequency, Coaxial, Dual Coaxial, Twin
	Conductor, Twin Lead
MIL-C-104	Crates, Wood, Lumber and Plywood Sheathed, Nailed and
	Bolted
MIL-C-5015	Connectors, Electric, AN, Type
MIL-C-7769	Cushioning Material, Bound Fiber
MIL-C-12000	Cable, Cord, and Wire, Electric Packages of Connectors.
	Electrical, Waterproof, Quick Disconnect, Heavy Duty
	Туре
MIL-C-26482	Connectors, Electric, Circular, Miniature, Quick
	Disconnect (Navy)
MIL-C-26636	Contacts, Crimp Type, For Electrical Connectors
MIL-C-40091	Crimping Tool, Terminal, Hand
MIL-D-3464	Desiccants, Activated, Bagged, Packaging use and Static
	Dehumidification
MIL-E-463	Ethyl Alcohol (for Ordnance Use)
MIL-E-5272	Environmental Testing, Aeronautical and Associated
	Equipment, General Spec.
MIL-E-15090	Enamel, Equipment, Light Gray, Formula No. (11)
MIL-F-21608	Ferrule, Shield Grounding, Insulated, Crimpt Style,
	Brass (ASG)
MIL-G-613	Grapnels, Marine, Trip Wire, and Crash Truck

DOCUMENT NUMBER	DESCRIPTION
	MILITARY STANDARDS AND SPECIFICATIONS (Continued)
MIL-G-5510	Packing, Preformed, Straight, Thread Tube Fitting Boss
MIL-G-45204	Gold Plating - Electrodeposited
MIL-I-631	Insulation, Electrical, Synthetic - Resin Composition; Nonrigid
MIL-I-3190	Insulation Sleeving, Electrical, Flexible Treated
MIL-I-6181	Interference Control Requirements, Aircraft Equipment
MIL-I-7444	Insulation Sleeving, Electrical, Flexible
MIL-I-8660	Insulating and Sealing Compound, Electrical
MIL-I-10428	Isopropyl Alcohol, Technical
MIL-I-22129	Insulation Tubing, Electrical Polytetrafluorethylene Resin
MIT D 7 9	Nonrigid Diagtic Material Laminated Thermosotting For Designation
MIL-P-78	Plastic-Material-Laminated, Thermosetting For Designation Plates
MIL-P-116	Preservation, Methods of
MIL-P-130	Paper, Wrapping, Laminated and Creped
MIL-P-8585	Primer Coating, Zinc Chromate, Low-Moisture-Sensitivity
MIL-P-11414	Primer, Lacquer, Rust-inhibiting
MIL-P-13949	Plastic Sheet, Laminated, Copper Clad
MIL-P-18177	Plastic Sheet, Laminated, Thermosetting, Glass Fiber
	Base, Epoxy-Resin
MIL-R-26	Resistors, Fixed, Wirewound (Power Type)
MIL-R-93	Resistors, Fixed, Wirewound (Accurate) General Spec for
MIL-R-94	Resistors, Variable, Composition, General Spec for
MIL-R-6855	Rubber, Synthetic, Sheet, Molded, and Extruded for
	Aircraft Applications
MIL-R-10509	Resistors, Fixed Film, High Stability General Spec for
MIL-R-55182	Resistors, Fixed, Film, Established Reliability,
	General Spec for
MIL-S-4461	Sealing Machines, Heat, Bench, and Portable (Temperature,
	Pressure and Time Controlled).
MIL-S-5002	Surface Treatments (Except Priming and Painting) for Metal
	and Metal Parts in Aircraft
MIL-S-7952	Steel - Sheet and Strip, Uncoated, Carbon 1020 and 1025
	(Aircraft Quality)
MIL-S-18729	Steel-Plate, Sheet and Strip, Alloy, 4130 Aircraft Quality
MIL-S-19500	Semiconductor Devices, General Spec for
MIL-T-713	Twine and Tape, Lacing and Typing (for use in Electrical
	and Electronic Equipment)

••••

DOCUMENT NUMBER	DESCRIPTION
	MILITARY STANDARDS AND SPECIFICATIONS (Continued)
MIL-T-5021	Tests, Aircraft and Missile Welding Operations Qualifications ASG#
MIL-T-7928	Terminals, Lug and Splice, Crimp Style, Copper (ASG)
MIL-T-10727	Tin Plating, Electrode Deposited or Hot-Dipped, For
	Ferrous and Non-Ferrous Metals
MIL-T-21595	Tape, Pressure-Sensitive Adhesive, Paper, Masking- Non-Staining
MIL-V-173	Varnish, Moisture-and-Fungus-Resistant (For the
	Treatment of Communications, Electronic, and
2 - 1 2	Associated Electrical Equip)
MIL-W-5086	Wire, Electrical, 600 Volt, Copper, Aircraft
MIL-W-5088	Wiring, Aircraft, Installation of (ASG) Three-Phase
	For the Production of Aircraft Quality Welds
MIL-W-7973	Welding Machines, Electrical Resistance, Spot, Press Type
MIL-W-8604	Welding of Aluminum Alloys - Process of
MIL-W-8611	Welding, Metal Arc and Cas, Steels, and Corrosion and Heat Resistant Alloys, Process for
MIL-W-16878	Wire, Electrical, Insulated, High Temperature (Navy)
MIL-Y-1140	Yarn, Cord, Sleeving, Cloth, and Type-Glass
	FEDERAL STANDARDS AND SPECIFICATIONS
FED-STD-102	Preservation, Packaging, and Packing Levels
FED-STD-183	Identification Markings for Iron and Steel Proc
FED-STD-595	Colors
BB-N-411	Nitrogen
J-C-98	Cable and Wire, Insulated, Methods of Sampling and Testing
L-P-391	Plastic-Methacrylate, Sheets, Rods and Tubes-Cast
L-P-590	Plastic Compounds, Molding and Extrusion, Polyethylene
О-Е-760	Ethyl Alcohol Ethanol, Denatured Alcohol, and Proprietary
0 70 094	Solvent Michlandsthelene Mechanical
O-T-634	Trichlordethylene, Technical
P-S-661 PPP-T-60	Solvent-Dry-Cleaning Tape, Pressure Sensitive Adhesive, Water Proof for
FFF-1-00	Packaging and Sealing
PPP-B-601	Boxes, Wood, Cleated, Plywood
PPP-B-621	Boxes, Wood, Nailed and Look Corner
PPP-B-636	Box Fiber Board

, + ·

DOCUMENT NUMBER	DESCRIPTION
	FEDERAL STANDARDS AND SPECIFICATIONS (Continued)
PPP-B-640	Boxes, Fiber Board, Corrugated Triple Wall
QQ-A-200/9	Aluminum Alloy Bar, Rod, and Structural and Special
	Shaped Sections Extruded 6063
QQ-A-250/8	Aluminum Alloy 5052, Plate & Sheet
QQ-A-250/11	Aluminum Alloy 6061, Plate & Sheet
QQ-A-274	Aluminum Alloy Bars, Rods, and Shapes Extruded 6063
QQ-A-282	Aluminum Alloy Bars, Rods, Wire, and Special Shapes,
	Rolled, Drawn, or Cold Finished 7075
QQ-A-362	Aluminum Alloy Plate and Sheet, Alclad 2024
QQ-B-613	Brass, Leaded and Non-Leaded, Plate Rolled, Bar Sheet and Strip
QQ-P-416	Plating, Cadmium, Electrodeposited
QQ-S-571	Solder, Lead Alloy, Tin Lead Alloy, and Tin Alloy, Flux
	Cored Ribbon and Wire and Solid Form
QQ-S-633	Steel Bars, Carbon, Cold Finished and Hot Rolled
	(General Purpose)
QQ-W-343	Wire, Electrical and Non-Electrical, Copper, (Uninsulated)
TT-C-490	Cleaning Methods and Pretreatment of Ferrous Surfaces for
	Organic Coatings
TT-E-489	Enamel, Alkyd, Gloss, (For exterior and interior surfaces)
TT-E-529	Enamel, Alkyd, Lustreless
TT-I-558	Ink, Marking Stencil, Opaque, for nonporous Surfaces, Metals, Glass, etc.
TT-M-261	Methyl-Ethyl-Ketone for use in Organic Coatings
TT-P-636	Primer Coatings, Synthetic, Wood and Ferrous Metal
TT-P-662	Primer Surfacer, Sanding, Lacquer and Enamel Type
TT-P-666	Primer Coating, Zinc Yellow, for Aluminum and Magnesium
	Surfaces
TT-X-9-6	Xylene for use in Organic Coatings
UU-T-81 WW-T-799	Tags, Shipping and Stock
MM-T-(33	Tubing, Copper, Seamless for use with Solder-Joint or Flared to be Fitting
Z-O-358	Oil-Peanut
2000	
	MILITARY STANDARDS
MS 20426	Rivet, Solid, Countersunk 100 Degrees, Precision Head Aluminum
MS 20470	Rivet, Solid-Universal Head, Aluminum and Aluminum Alloy

DOCUMENT NUMBER	DESCRIPTION
	MILITARY STANDARDS (Continued)
MS 20600	Rivet, Blind, Structural Pull Stem Self Plugging, Protruding Head, Type II, Class I
MS 20601	Rivet, Blind, Self Plugging, 100 deg flush Head, Type II, Class II
MS 20602	Rivet, Blind, Chemically Expanded, Protruding Head, Type I, Class I, Styles A and B
MS 20603	Rivet, Blind, Chemically Expanded, 100 degree flush head, Type I, Class II, Styles A and B
MS 20659	Terminal, Lug, Crimp Style, Copper, Uninsulated, Class I (ASG)
MS 25036	Terminal, Lug, Crimp Style, Copper Insulated, Class I (ASG)
MS 25037	Crimping Tool, Hand, For Copper Insulated Terminal
MS 25042	Cap, Electrical Connector Plug, Dust
MS 25043	Cap, Electrical Connector, Receptacle, Dust
MS 25083	Jumper - Assemblies Bonding and Current Return (ASG)
MS 25274	Cap, Wire End (Class I)
MS 25311	Ferrule, Shield Grounding, one piece, Insulated, Class I, for Coaxial and Shielded Cable
MS 25312 MS 33584	Tool – Crimping, Hand, For Insulated Shield, Ground (ASG) Tubing End, Standard Dimensions for Flared
	MISCELLANEOUS SPECIFICATIONS
AN 735	Clamp, Loop Type Bonding
AN 742	Clamp, Plain, Support, Loop Type, Aircraft
AND 10387	Drill Sizes and Drilled Hole Tolerances, Twist
NAS 523	Code, Rivet

Table 6-2DOCUMENTS DISCONTINUED FROM USE BY LAUNCH
EQUIPMENT BRANCH, ELECTRICAL SECTION

i

•

.

F

DISCONTINUED DOCUMENTS SUPERCEDING DOCUMENTS ABMA-PD-W-45 ABMA-PD-R-197 MSFC-SFEC-339 ABMA-STD-428 MSFC-STD-154 A10338552 AL0423785 KSC-STD-169 A10423785 KSC-STD-169 A10401071 A75M01260 KSC-STD-132 (In Process) A75M01260 KSC-STD-132 (In Process) A75M01260 KSC-STD-169 A10401071 SP-80-D A75M01260 KSC-STD-132 (In Process) A75M50223 H1027, 14073 thru 14092, and MSFC-SPEC 332 A75M50741 KSC-E-165 MIL-STD-12 MSFC-STD-165 MIL-STD-130 KSC-STD-169 A75M09468 QQ-A-320 QQ-A-320 QQ-A-320 QQ-A-320 MSFC-STD-163 MSFC-STD-163 MSFC-PROC-256 MSFC-PROC-196 MSFC-PROC-256 MSFC-PROC-196 MSFC-PROC-256 MSFC-PROC-256		
ABMA-PD-W-45 MIL-W-8604 ABMA-PD-R-187 MSFC-SPEC-339 ABMA-STD-428 MSFC-STD-154 A10338552 KSC-E-165 A10419905 KSC-STD-169 A10423785 KSC-STD-169 A10001071 SP-80-D A75M01260 KSC-STD-132 (In Process) A75M50222 KSC Dwgs. A75M10180 thru 10204, 11003 thru A75M50223 11027, 14073 thru 14092, and MSFC-SPEC 332 A75M50497 KSC-E-165 A75M51073 KSC-E-165 MIL-STD-12 MSFC-STD-350 MIL-STD-130 KSC-STD-169 and KSC-P-116 MIL-STD-130 KSC-STD-276 MIL-S-8484 A75M9468 QQ-A-318 QQ-A-250/8 (5052) QQ-A-327 QQ-A-200/9 Extrusions MSFC-STD-163 Discontinued MSFC-STD-163 Discontinued MSFC-PROC-186 MSFC-PROC-256 MSFC-PROC-196 MSFC-PROC-256	DISCONTINUED	
ABMA-PD-R-197 MSFC-SPEC-339 ABMA-STD-428 MSFC-STD-154 A10338552 KSC-STD-169 A10419905 KSC-STD-169 A10423785 KSC-STD-169 A1001071 SP-80-D A75M01260 KSC-STD-132 (In Process) A75M50222 KSC Dwgs. A75M10180 thru 10204, 11003 thru A75M50223 11027, 14073 thru 14092, and MSFC-SPEC 332 A75M50497 KSC-E-165 A75M50741 KSC-STD-350 MIL-STD-130 KSC-STD-169 and KSC-P-116 MIL-C-13777 MSFC-SPEC 332 MIL-STD-130 KSC-STD-169 and KSC-P-116 MIL-ST-144 MSFC-SPEC 332 MIL-STD-150 KSC-STD-169 and KSC-P-116 MIL-STD-130 KSC-STD-100 Vgc-A-318 QQ-A-250/8 (5052) QQ-A-250/8 (5052) QQ-A-250/11 (6061) QZ-A-327 QQ-A-200/9 Extrusions	DOCUMENTS	SUPERCEDING DOCUMENTS
	ABMA-PD-W-45 ABMA-PD-R-197 ABMA-STD-428 A10338552 A10419905 A10423785 A10401071 A75M01260 A75M50222 A75M50223 A75M50497 A75M50497 A75M50741 A75M51073 MIL-STD-12 MIL-STD-12 MIL-STD-130 MIL-C-13777 MIL-I-7444 MIL-S-8484 QQ-A-318 QQ-A-327 MSFC-STD-110 MSFC-STD-158 MSFC-STD-163 MSFC-PROC-186 MSFC-PROC-196	MIL-W-8604 MSFC-SPEC-339 MSFC-STD-154 KSC-E-165 KSC-STD-169 SP-80-D KSC-STD-132 (In Process) KSC Dwgs. A75M10180 thru 10204, 11003 thru 11027, 14073 thru 14092, and MSFC-SPEC 332 KSC-E-165 KSC-E-165 KSC-E-165 MSFC-STD-350 KSC-STD-169 and KSC-P-116 MSFC-SPEC 332 MSFC-SPEC-276 A75M09468 QQ-A-250/8 (5052) QQ-A-250/11 (6061) QQ-A-200/9 Extrusions Use only for guideline without use of applicable documents MSFC-PROC-256 Discontinued MSFC-PROC-256

Table 6-3 ABBREVIATIONS, CABLE SPECIFICATIONS

1

ſ	ABBREVIATION	DESCRIPTION
-	С	Conductors
	P	Pairs
	S	Single
	Т	Triplets
	Q	Quads
	(Q)	Quints
	SH	Shield
	SS	Single conductor shielded
	SSI	Single conductor shielded and insulated
	PTS	Pairs twisted and shielded
	PTSI	Pairs twisted, shielded, and insulated
	TTS	Triplets twisted and shielded
	TTSI	Triplets twisted, shielded, and insulated
	QTS	Quads twisted and shielded
	QTSI	Quads twisted, shielded, and insulated
	(Q)TS	Quints twisted and shielded
	UNSH	Unshielded
	OS	Overall shield
	OSDB	Overall shield double braid
		÷
	· · ·	
L		

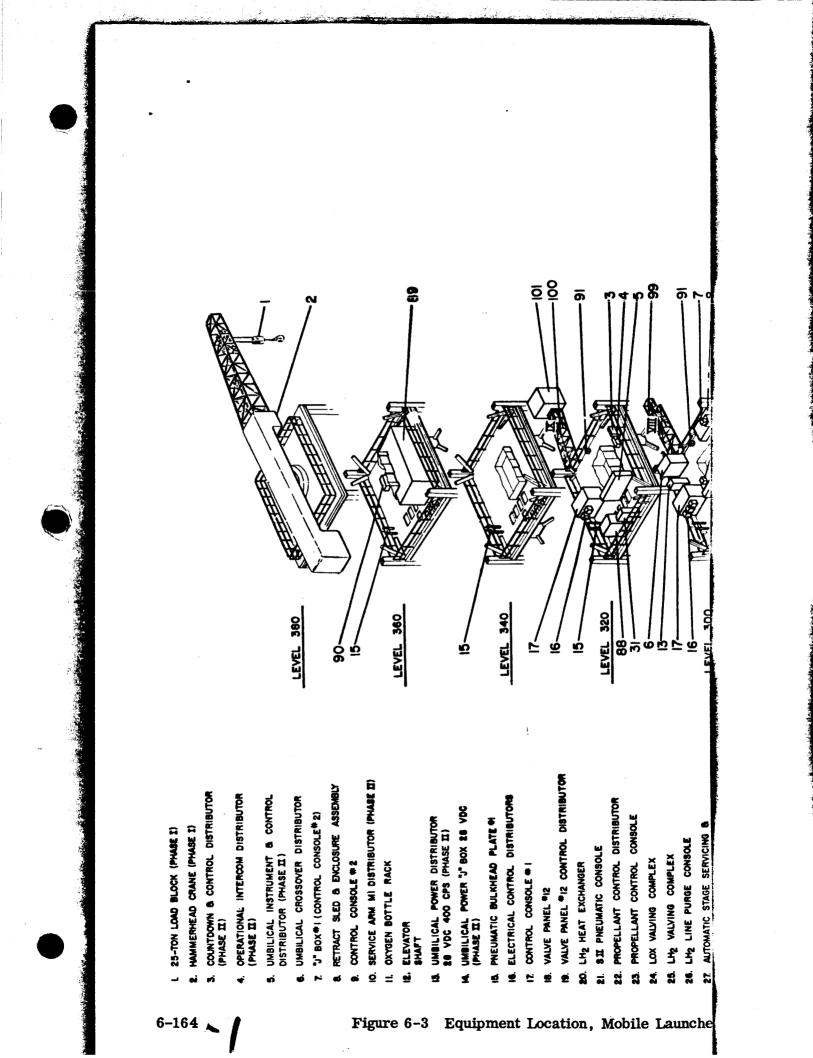
6-3. ESE EQUIPMENT LOCATIONS.

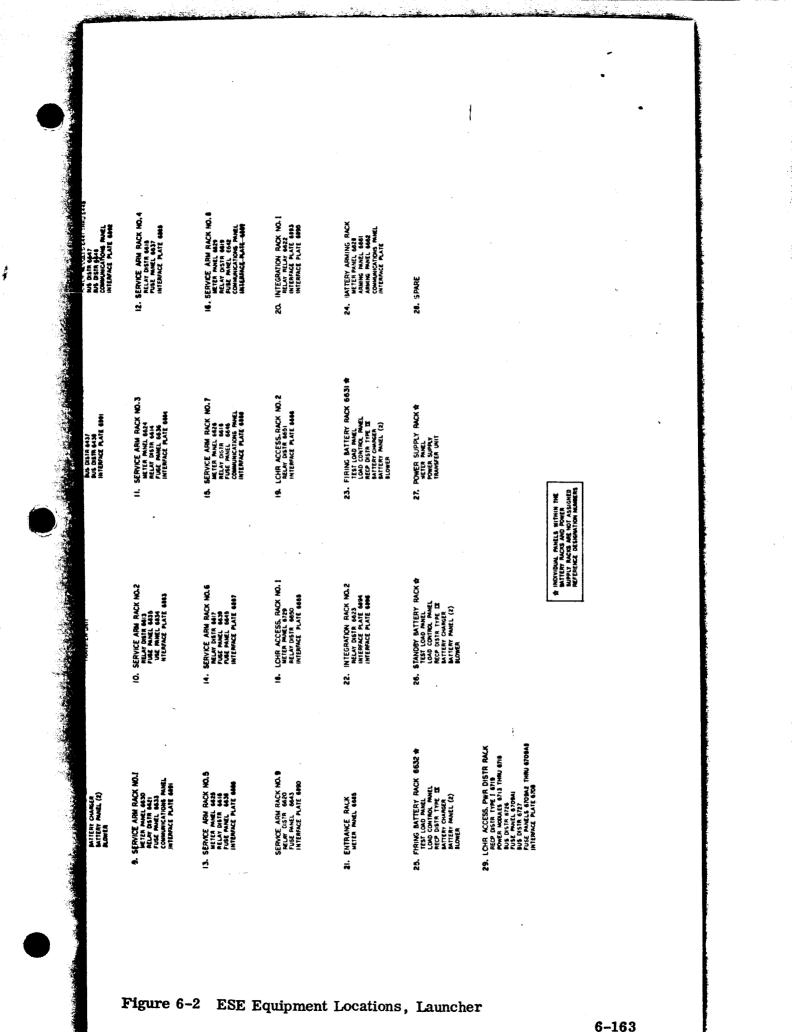
ESE electrical enclosures on the Mobile Launcher are shown in at least one of the figures 6-2, 6-3, or 6-4. Enclosures for the following subsystems are largely absent:

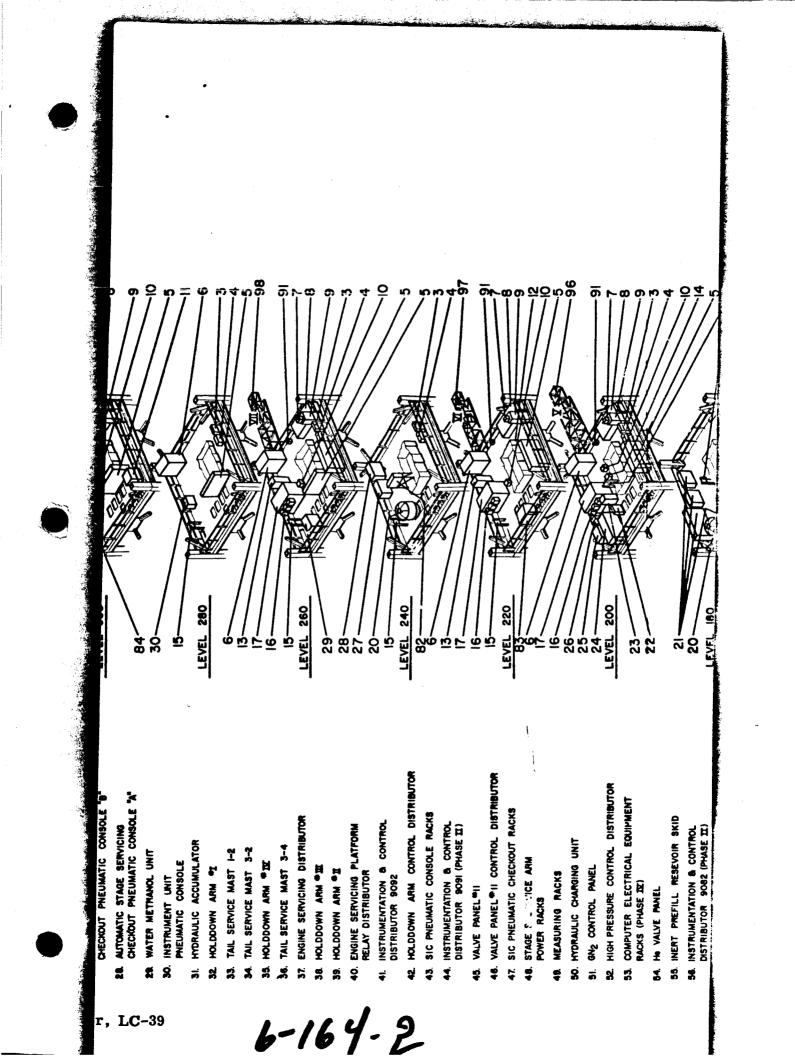
a. S-IC Engine Servicing Platforms Subsystem

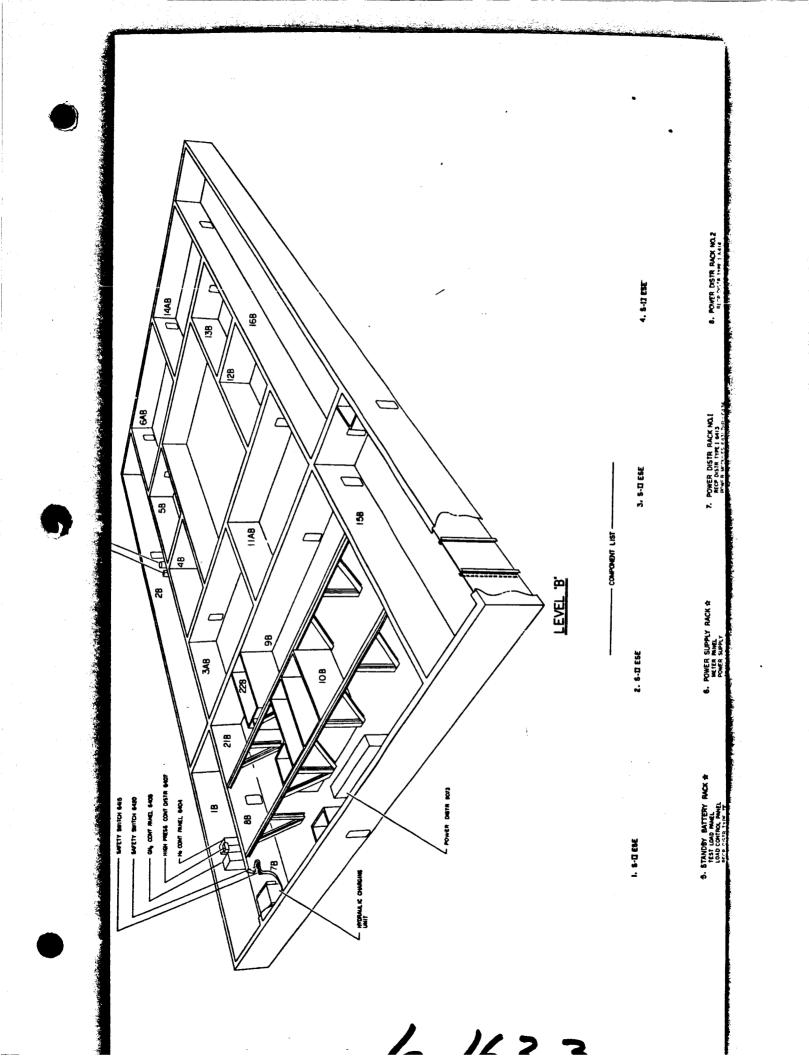
b. Platform Transporter Subsystem

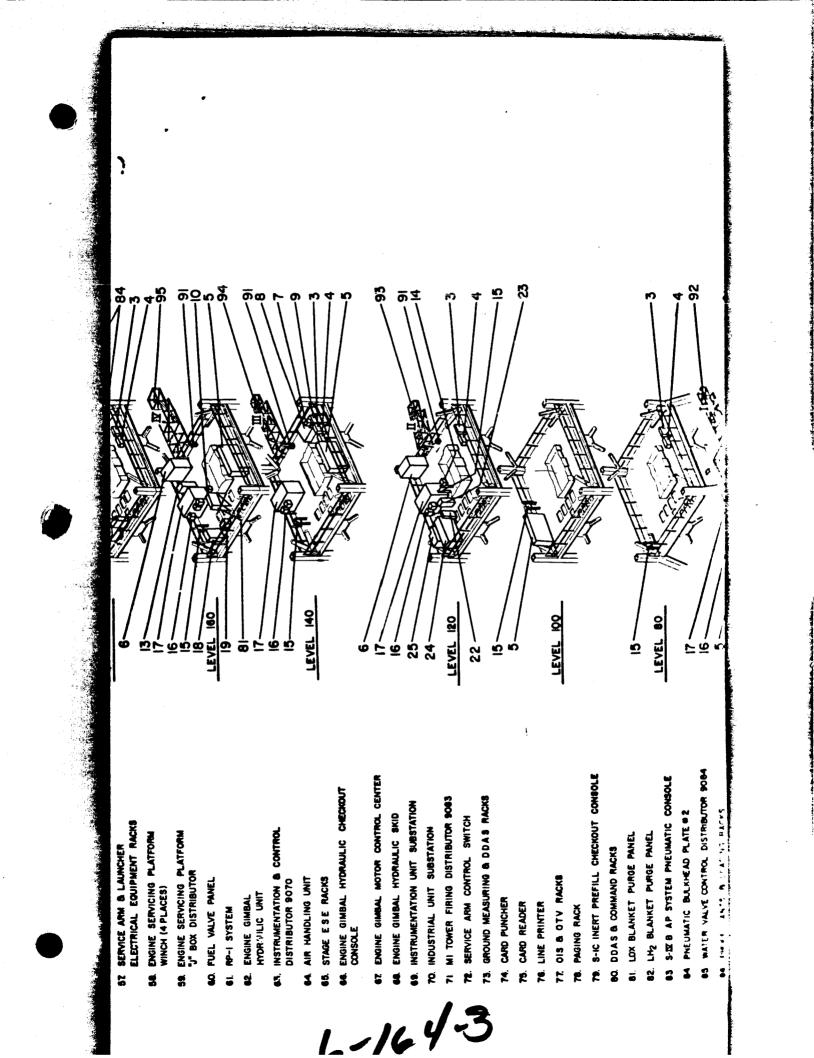
c. S-II and S-IVB Engine Servicing Platform Subsystem

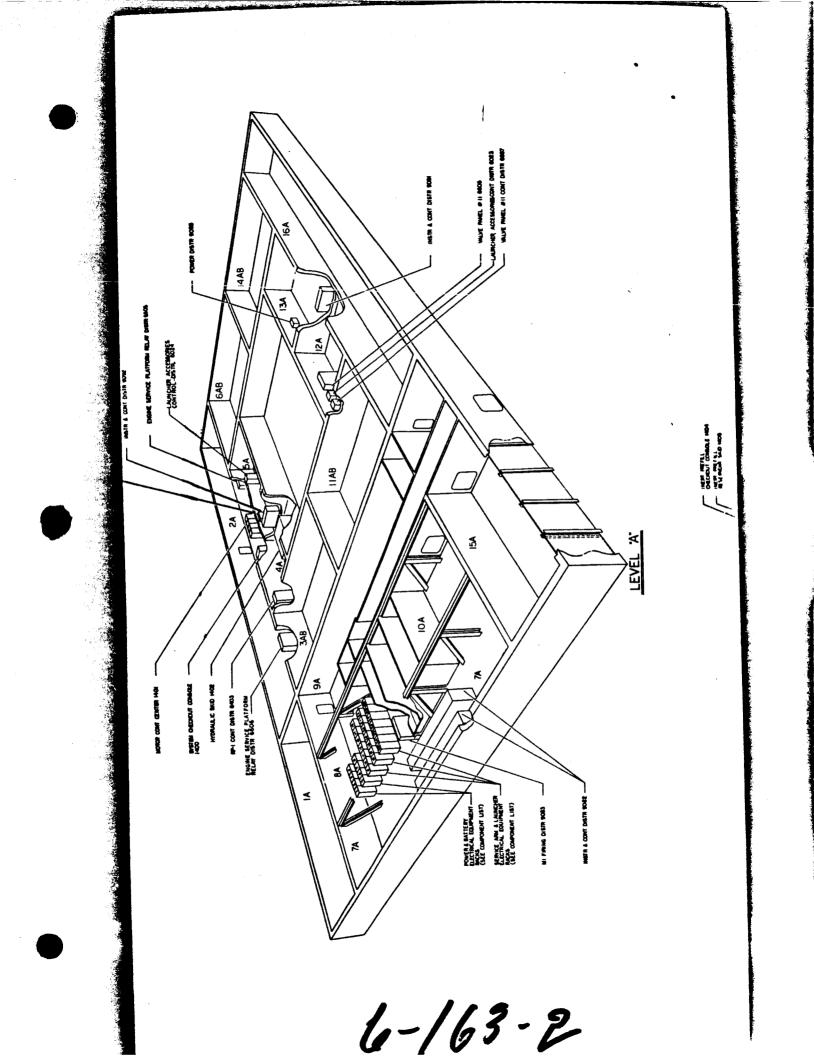


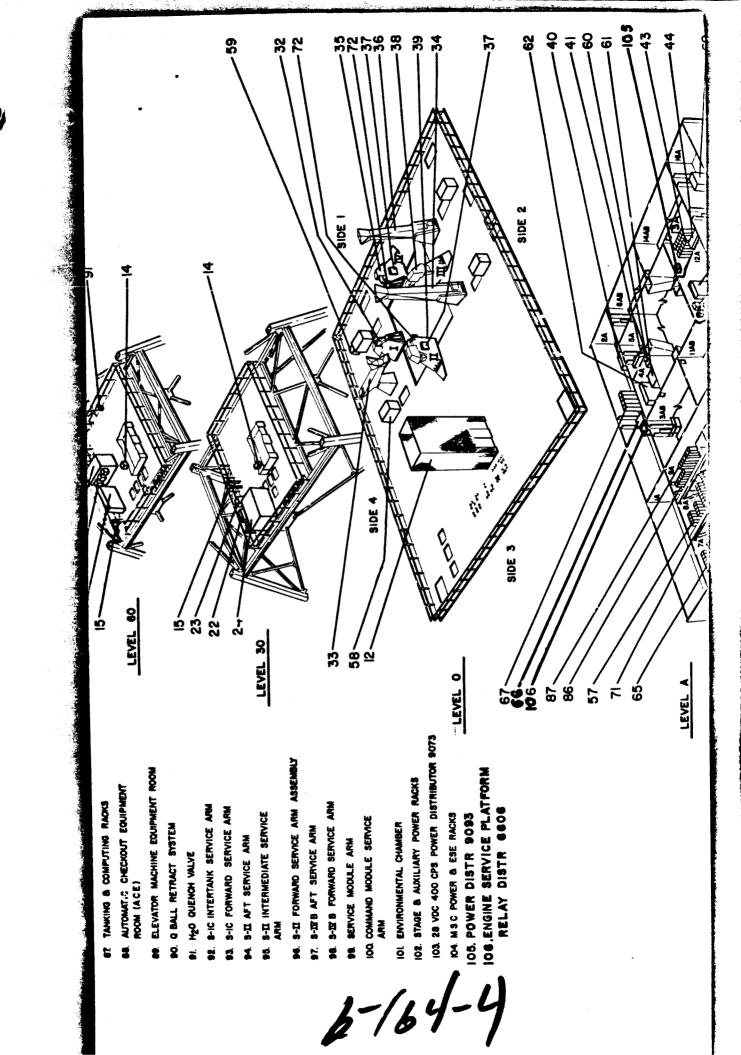




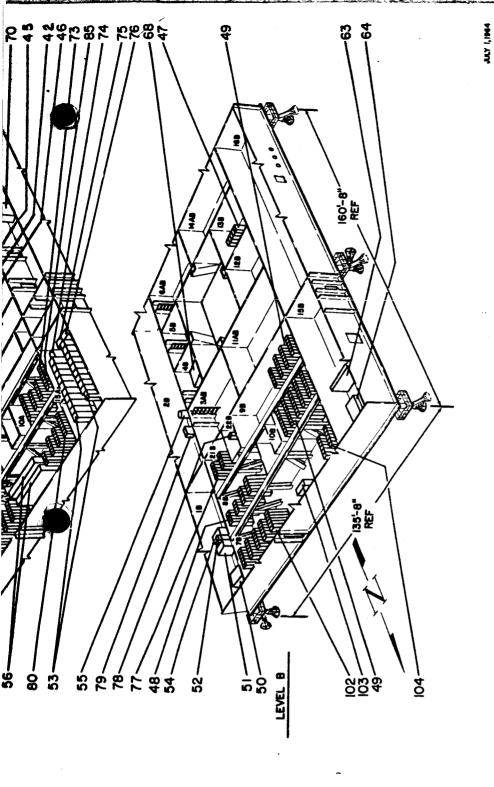




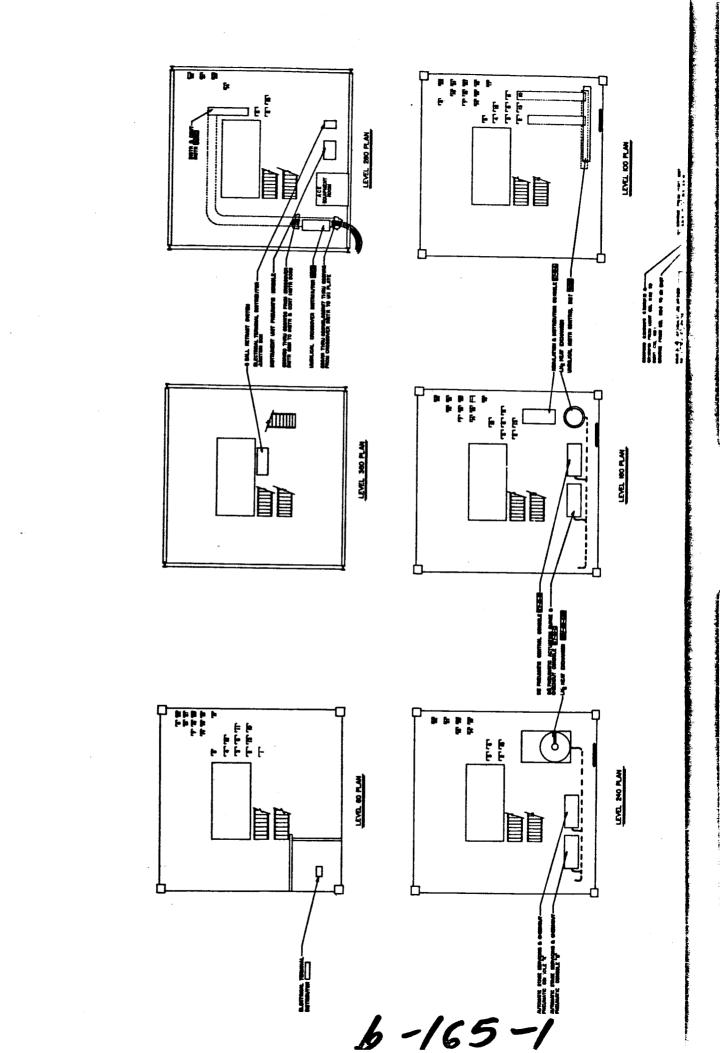


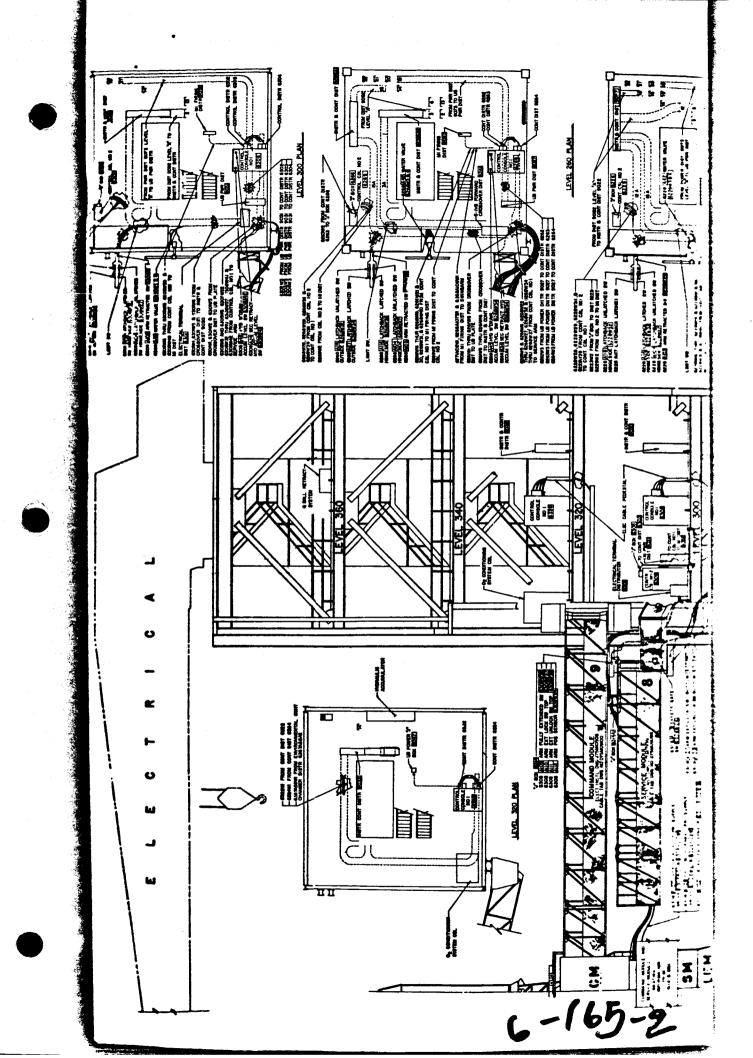


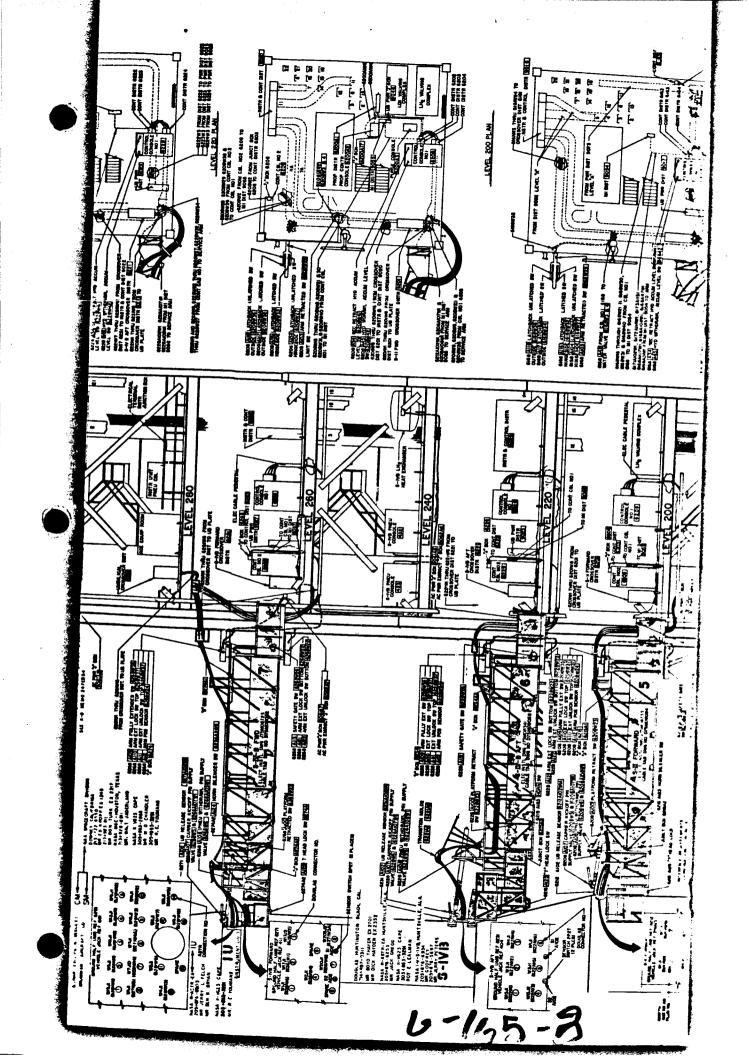
ş TAIL SERVICE MAST 3-4 DISTR 8008 TAIL SERVICE MAST 3-2 DISTR 8010 SERVICE PLATEORN WINCH MOTON HOUSING "E 7004 ADIVICE ANN CONTROL, BUITCH DISTR 6029 MENNICE PLATFORM WINCH MOTOR HOLEING 'C' PARE REAVED AND CONTROL BUILDO DEAT SCHA Kang Mi Corta Prio Ø NUM BILL DATE NON Ф¢ LEVEL 0 R 6-163-1



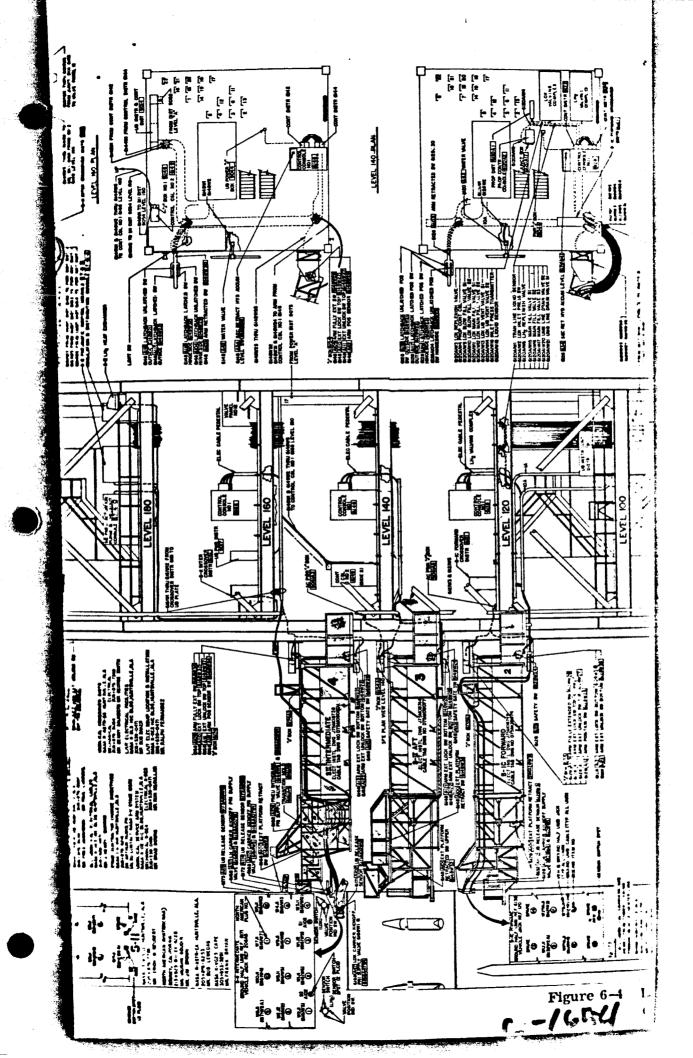
6-164-5

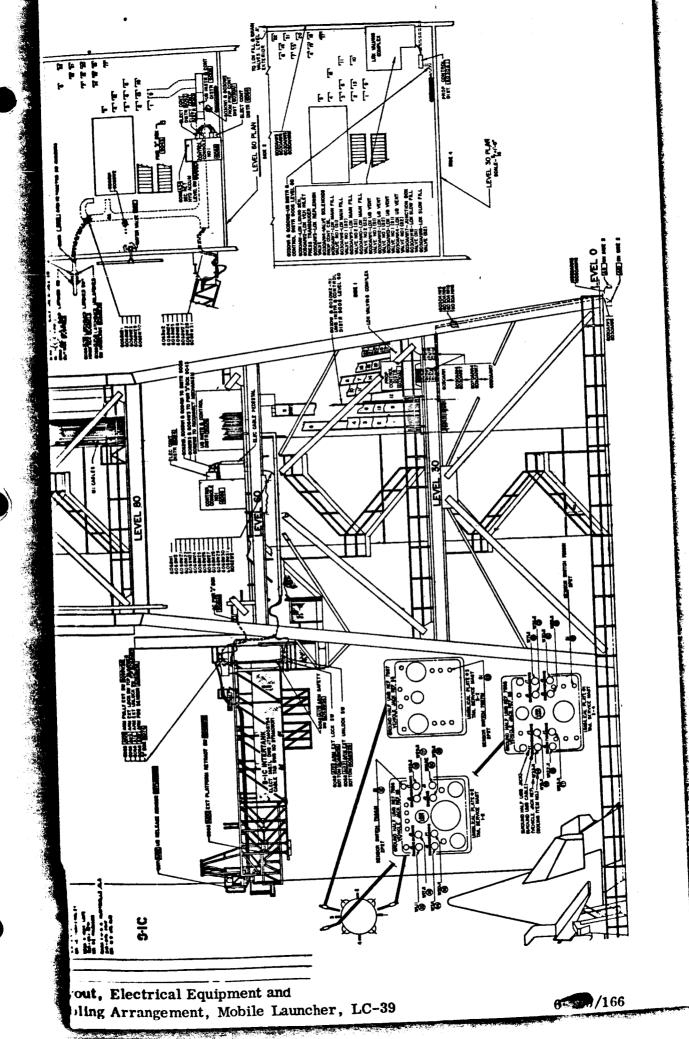




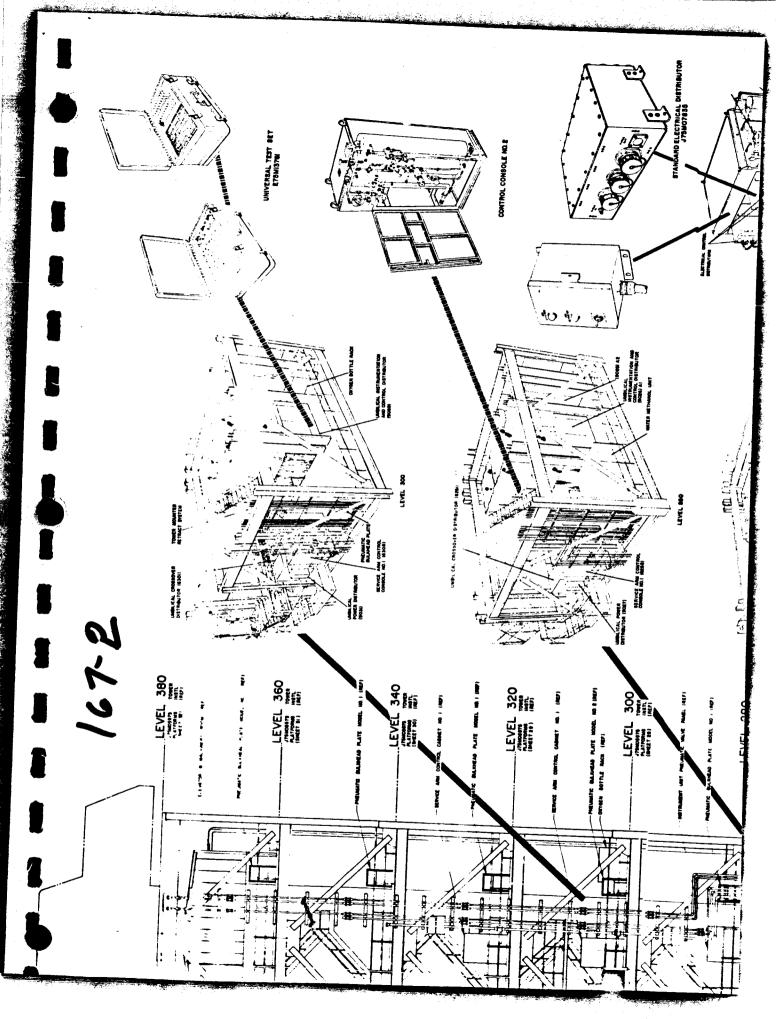


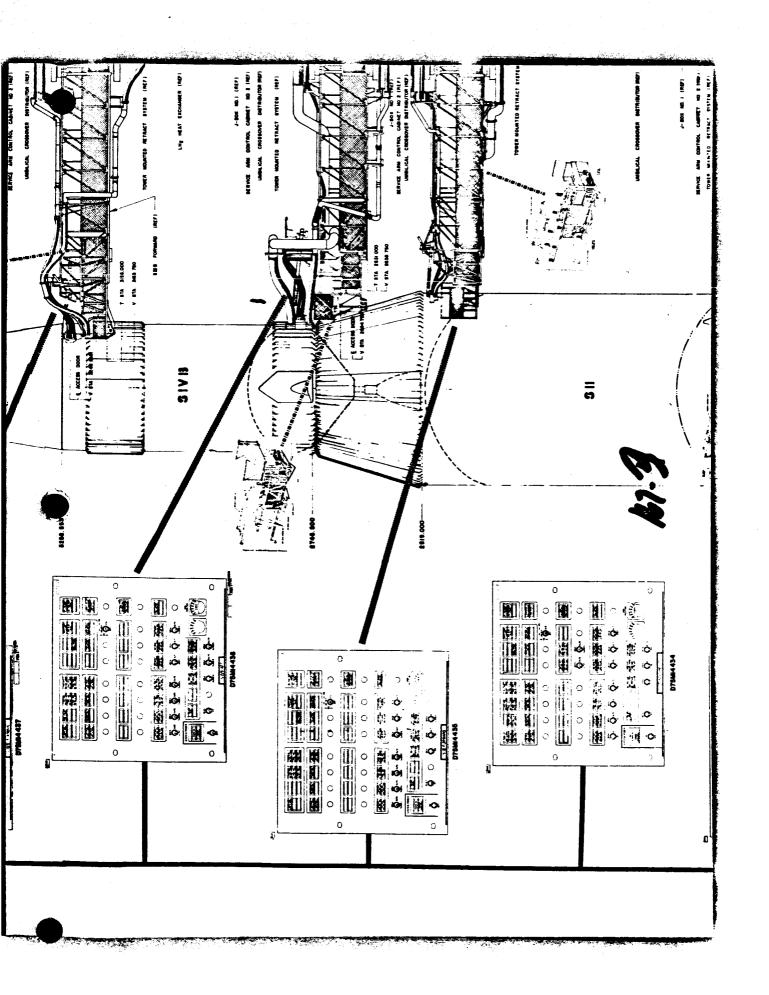
بليستعب كالجلف والسطية والمحد ومعاليتهم والمستعمل والمستعمل ومراعدتهم الأراد المتشر والمستع والمستعم وال

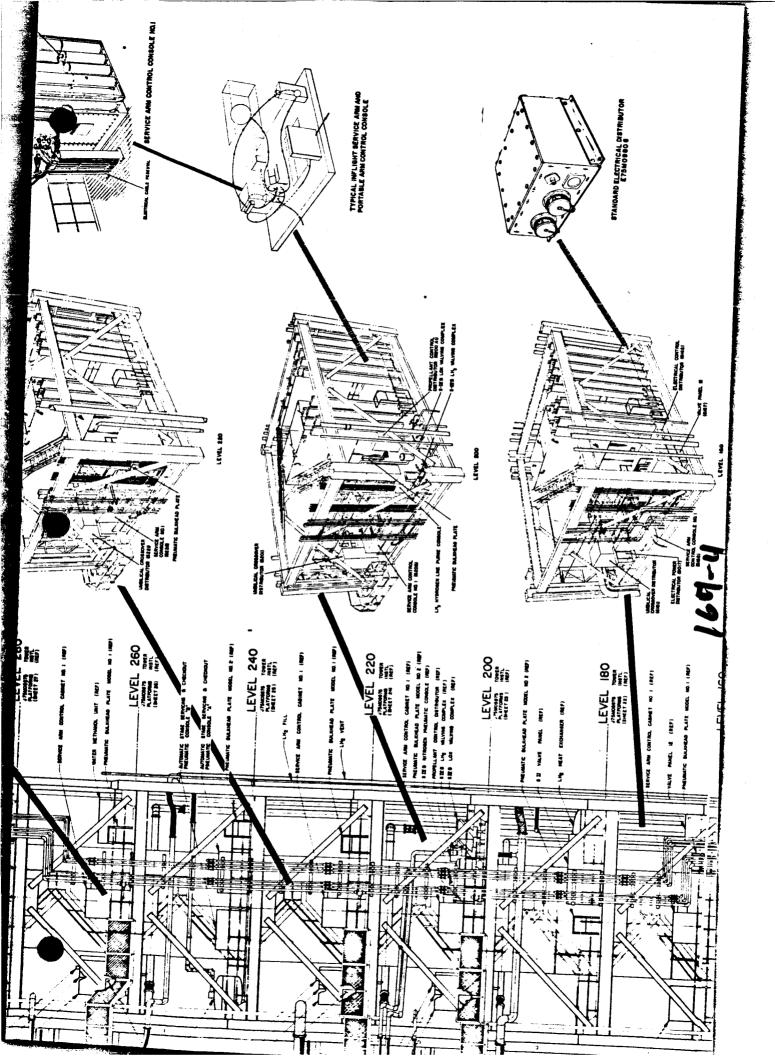


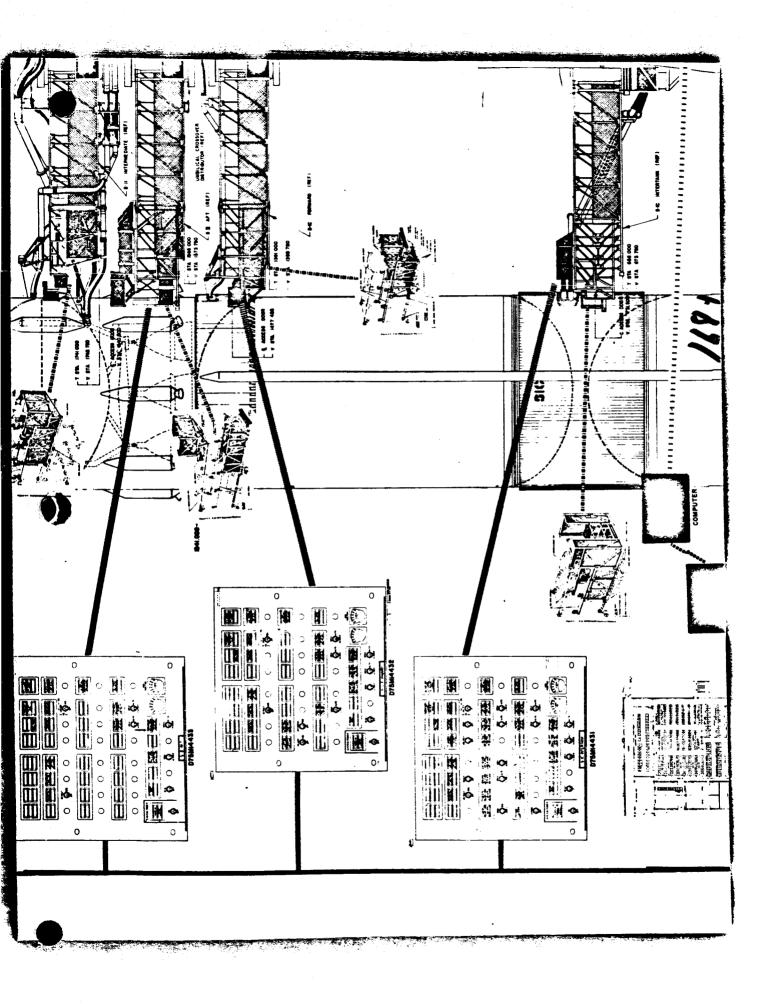


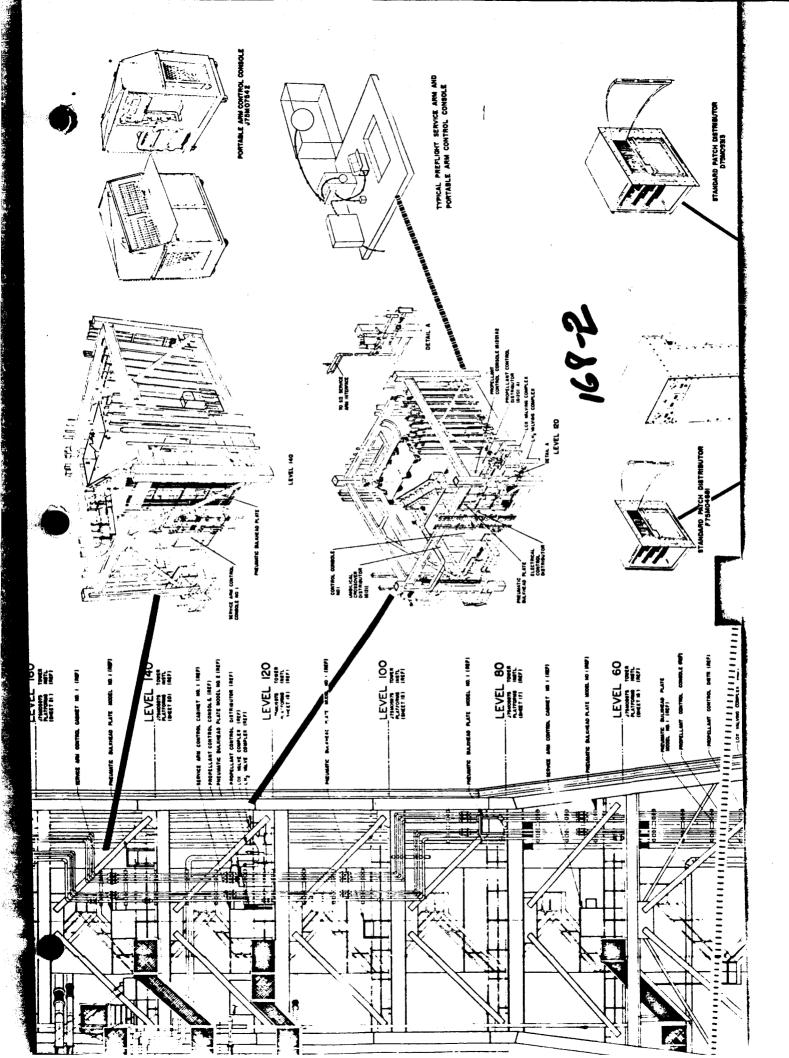
1= -THE AND CONTROL DETAILS THIS OF A THE CROMMAR DISTRUCTION (NOT 1-BOX NO - [100 ENVIRONMENTAL CHAMBER HODILE INCO ×. CINCE NOBULE T. STA. MAA.000 V STA. MITATO LIEM **₹** 3044.555 3748, **5**86 0 M ¥] ⊂ 10 N 0 075M4439 0 0 0 0 0 0 ų

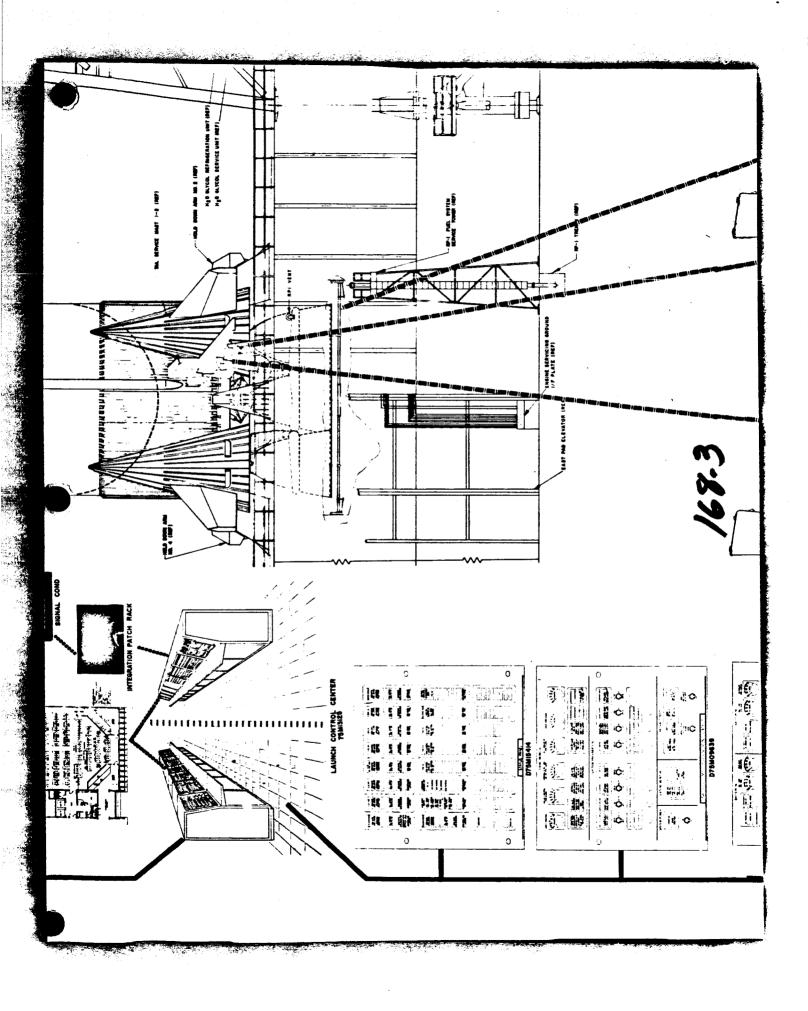


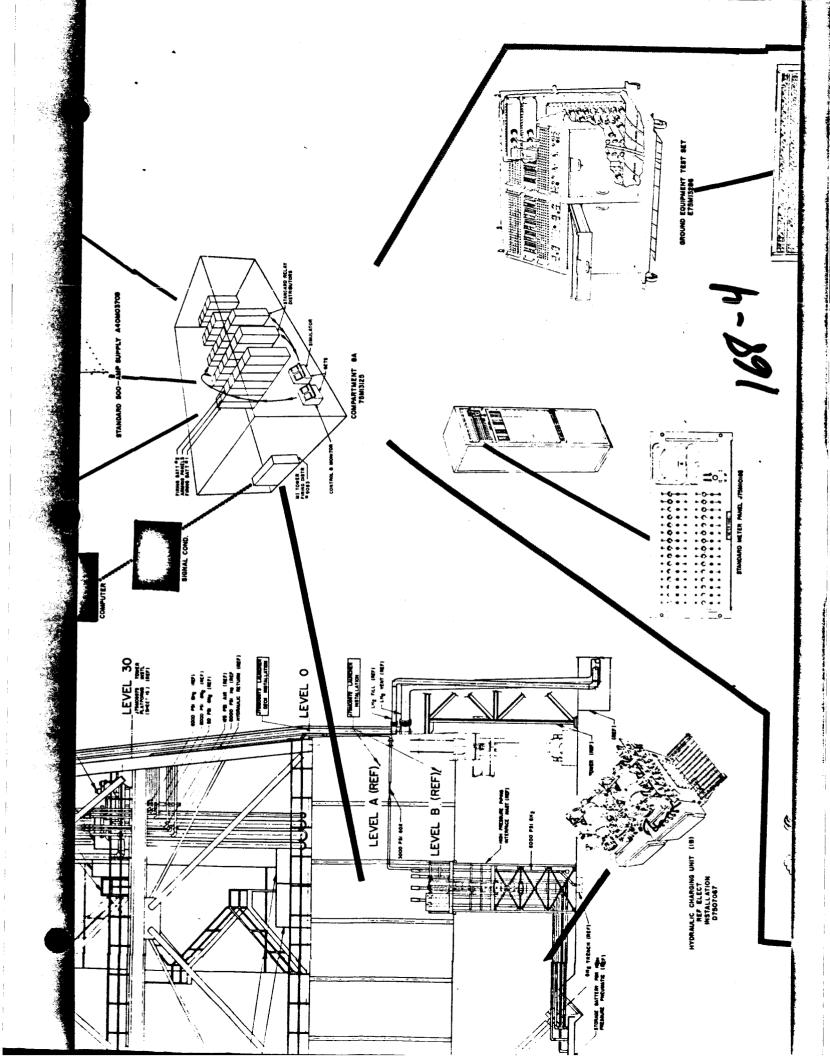


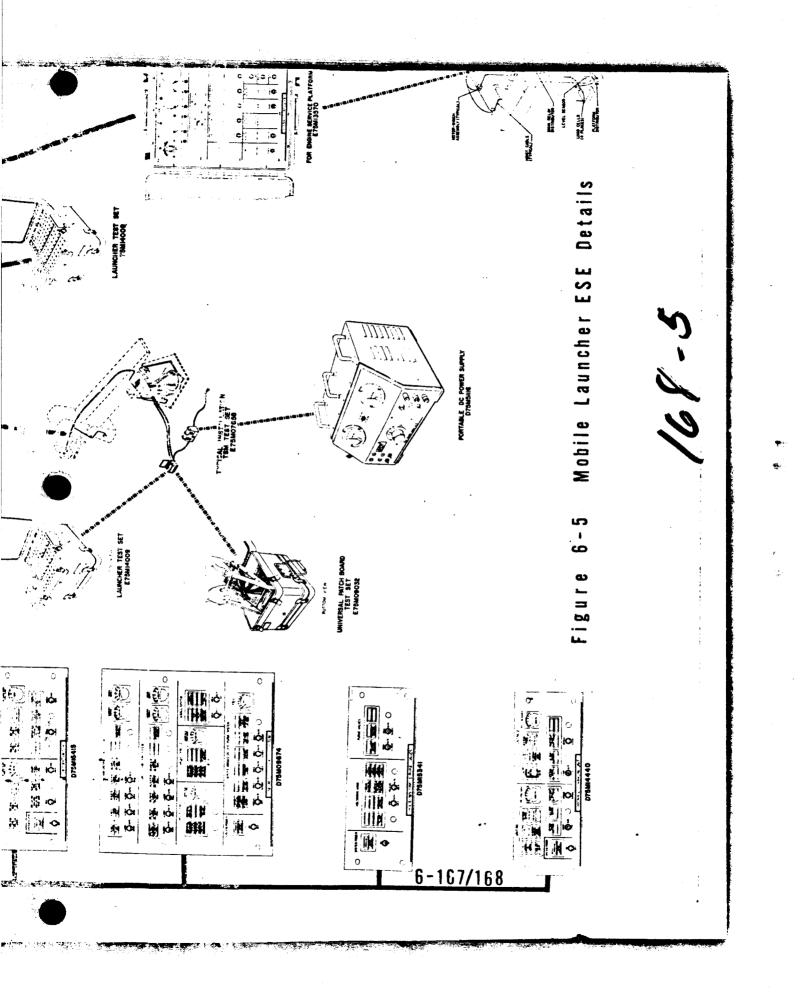


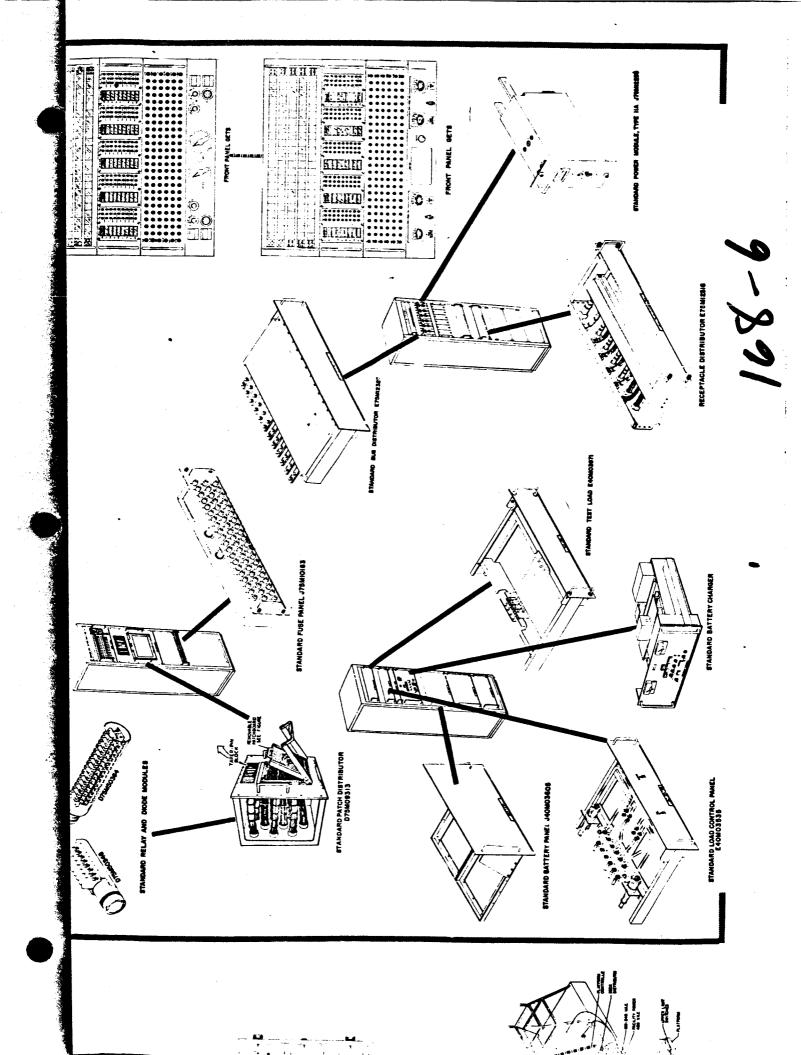








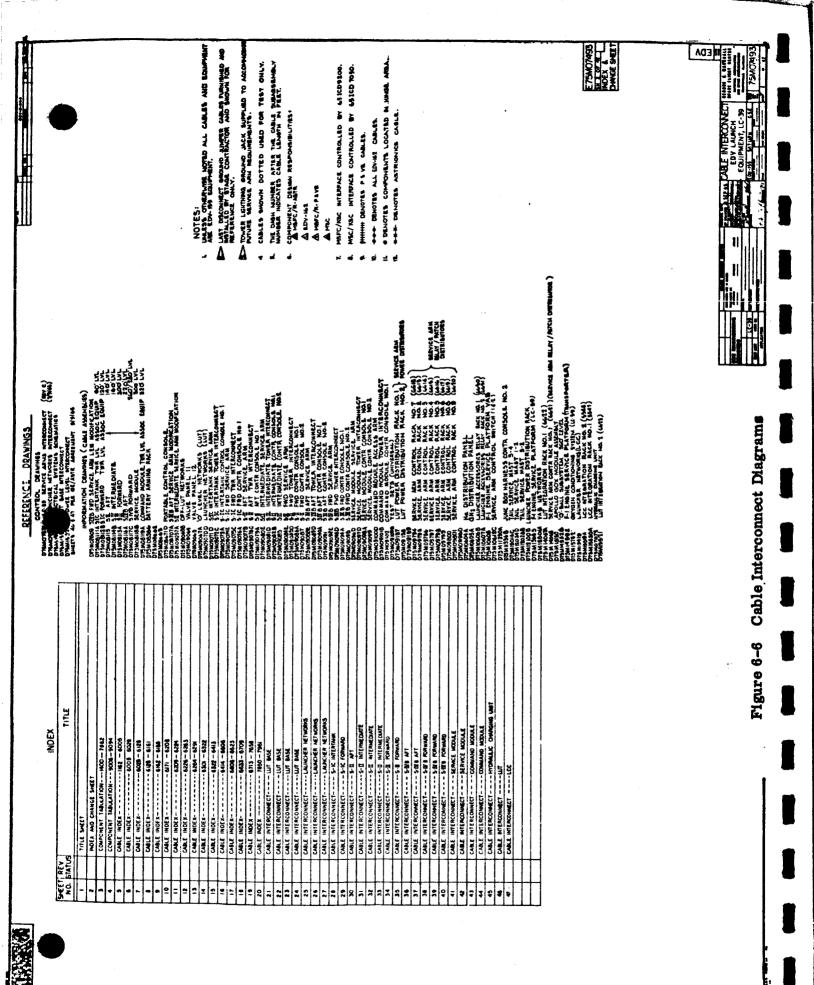




6-4. ESE CABLE INTERCONNECT DIAGRAMS.

Cable interconnect diagrams for Electrical Support Equipment are on pages 6-170 through 6-215 following. These diagrams were recently revised andare scheduled for further revision before the next issue of this document.

Also included here are four pages of Cable Interconnect and Electrical Schematic Criteria (6-216 through 6-219).



		APPLICABLE ANTER AMERICAME		
		€¥		CARE NTERCON EDV LAUNCH EDV LAUNCH EQUEMENTLC33
ASSY PMG. NO PMG. NO PMG. NO PMMCR201 P		11200000000000000000000000000000000000	174 120%	
	PPER MOOLE FUNE FINISH FUNE FINISH FOR MOOLE FINISH FUNE FUNE FUNE FUNE FUNE FUNE FUNE FUNE	PLASE PANEL PLASE PANEL PLASE PANEL LAUNCHER ACCESS RELAY DR TR LAUNCHER ACCESS RELAY DR TR LAUNCH		
REF DREF DREF DREF DREF DREF DREF DREF D		94556 94557 94557 94557 94557 94547 94547 94547 94547 94547 94549 94549 94549 94549 94549 94549 94549 94549 95739 957595		11
RE WORKS SH NO				aas
			THER UP of No. JPARADOO - 40 DISTRIBUTION JPARADOO - 40 Namine of No. JPARADOO - 40 <	Final to a
COMPONENT LICCA. CONTROL BOW USCA. CONTROL BOW USCA. CONTROL BOW LICCA. CONTROL BOW LICCA. CONTROL BOW THE CONTROL BOW CONTROL LOST FINIBUTION CONTROL LOST FINIBUTION CONTROL LOST FINIBUTION CONTROL LOST FINIBUTION CONTROL LOST FINIBUTION CONTROL CONTROL BOW CONTROL CONTROL BOW CONTROL CONTROL BOW CONTROL CON	V. BOX. V. BOX. V. BOX. V. BOX. V. BOX. CONTROL. DISTIBILIDIA IND. CONTROL. DISTIBILIDIA IND. CONTROL. DISTIBILIDIA IND. V. BOX. V. CONTROL. V. CONTRO	 P. BOR HING: UNKER SERVICE MODULE HING: UNKER SERVICE MODULE HING: UNKER SERVICE MODULE UNKER DESTRIBUTION UNKER DESTRIBUTION LOWING, DESTRIBUTION CONTROL DESTRIBUTION LOWING, DESTRIBUTION	LUTINAL DE FIREUREN NO I CONTRAL DE FIREUREN NO I LUNCTON BOX JUNCTON BOX LUNCTON BOX ACONTRAL ARMAN MARKEN A	LING 2 STABILER ASSY
NO REWUNDS SHI NO		මෙන් දී දීමාවමාදුදෙකකිම් පිරිමාන	ේ දුමුළුමුදුළුලුදුළුදුළුවන්න නිතිවිට මා	Tionine AA
S COMPORE NT DWG NO Systellar Component DWG NO NO Mail Preprint Component DWG NO D Mail Preprint Component DS D D D D D Mail Preprint Component D	1 LUNI -/- -/ -/ -/ -/ -/ -/ -/ -/	Законскот склужани Сонтост вытек () тужи (25) Массилист склужани Сонтост вытек () тужи (25) Панациска Гужи (

i

6-171

																											t X															<u></u>			1					168:	FOR APPLICABLE NOTES BEE			-			E25MO743		TABULATION								T	N	03		CONTRACTOR NERCONACT Still the Party			to some / 11 Lat. Lan.	
	NUMBS PL. NO			I	ļ					T					I	T	Π		Ţ	T		T			Ţ				Ţ			Ţ			T				Ţ								Ţ			ş		ā							Π	T			T	T					T		IN THE NU			ALL	
Ш									+																												+							-		-+																			7										
	22	8																																																										ĺ													(panu)	•	
	22	Ţ	Π	Ι	Π	T	Π	T	T	Π	T	Т	Π		I	Π	Π	T	İ		1	Π		T	Т	П	Ι	T	Π	Π	Т	Г	Π	T	Γ		Π	Т	Т	Π	T		T	Π	Π	T	Π	Π	Т	Π	Π	T	Π		T	Π	Т	Π	T		Π	T	Γ	Π	·	Т	Π	T	Π]					
	FEMMIS *	t						T	t	·		T			t			╎	╉						T						┥	┢		+							╉								+			ł														╞		+							ſ
															ł													+										1																								╉						╎					JIAKIA		
	COMPORING	LOC INTEGRATION RACH NO!	LOC INTICATION AND AD	COMECTOR PLATE	COMMECTOR PLATE	CONTROL PAND LAD &W	T Status Much	S-IC MIGRAM	1 de 13-5	Q3MLNI D-9	S-II FWO		197	CN NAT CASE LANT	2-1 mg	154 3-2	154 14	CONTROL PAINT HOLDOWN AME		DONNECTOR PLATE-LUT	DAME CADA PLATE - LUT		INTEGRAPHION MATCH MACK LUT																																																	lo Intoncompet 1	cause miter counter magratus (continued)		
						Т	Π						i i		2420	Ĩ.	242	T		Hill Hill	1 1 1	1.1																																					T																
		Π	T	Γ		Ţ	Π	Т	Π	Ĩ	1	Π	Π	T	Γ	Π	T	T	Π		5 T	, П	ŧ	T	Π	T	Ι	Τ	Π	T	T	Π	Π	T	Π	Т	Π	Т	ſ	Т	Π	T	Т	Π	Т	Т	П	Π	Т	Π	Π	Т	Π	Т	Π	Π	T	Π	Т	!	ר	: Г	Π	Π	T	Π	Π	Т	Π			Dimma R_R			
	RINNES P		╉				H	+		+	╁				$\left \right $		+		$\left \right $		╀		t	$\frac{1}{1}$		\dagger	Ì			+	╁			Ť		╀		╀	$\left \right $				╎						1				╎		H				ł		┥		$\left \right $	$\left \right $	╁				H			lann.	79er		ſ
	00 NO 1		ł							174141113	A NOT CHIC	JPSIMONSIA MA	PICES SIESA	Susidis			CINICIPAL STATE	100011	5 860002	J65860005	- Sizanta		21660HC/P						J'PEMBHAI3	+															╎						1000	M0130	J75406724	61260	J75409086							J'ISHORATA			$\frac{1}{1}$			175407562	164104				4		(
	woo									T		SAC ENLAGE SERVICE TOP PLATE				<u> </u>				UB PLATE 1588-3-2 (165		H		1	AUX NOTOR TO	NOTOR LIPAKE T	LO DAR MAN	T	Π		UN MALA CONTROL TERMINAL DISTR		LE META CONTROL TEMMINAL DISTA.	UB INSTACONTROL TERMINAL DISTR	THE PARTY PARTY PARTY	US POWER TERMINAL DISTR			UR HIGT. & CONTROL TERMINAL DISTR	UB POWER TERNINAL DISTR	LIB INSTACONTROL TERMINIAL DISTR	JE POWER TERNIMAL DISTR		A INSTA CONTROL TERMINAL DISTR	UR HOWER TEMBINAL DISTR LA INST A CONTROL TEMANAL DISTR		UB POWERYPOX			us rough? acr	44 FRIME CS18		1640		1734 July 1		KENCE AN EXCEPTION OF AN			POWER USTR 28 VDC			WATER WILL CONTROL DISTR					LA POURTAIN AND							
			-1			1001	0 0 0 0					2	102	100	1607	1152		1 SEE	N.C.	142	55	3860	100	7963	7165	998	1	191				_		100	Π	104	Π	Т	1205	Т	1026	Т	Т	П		П	T	NON T		П		-	20	-		1	1 705	T	Т	4406	and a	104	101	1	Т	Ĩ									

								BUCK CABLE INDEX	2001/07/07/07 28	O/ 01 and 00		14/03, 60%54	NOS CURSA	05,48,446,21479,1866,2479,1=12 05,48,476,21479,1866,2479,1=12	VOS, TUNSH	AVOS, POUNSH SVOS, AB, POUNSH	NOS 21UN5H /05.4-2.9-6.16 = 14	105.48.4-8.9-4.16-14- 20/UKSH	/05, Jeo, Jee, 1367,34614,1351,614 /05, /48,340,354,13675614,1351,614	205, 50151=10, 20 UNSAP-14 205, 50151=10, 20 UNSAP-14	/+/05.137751 05.455144,1177514 5751414, 551414	06 /8, 455/a4, //TT5/a/4, \$PT5/a.4, \$5/a/4	14/ UNSH 14/ 03, 60 UNGH	14705, 60531 14705, 60531	6/02.30/151 (4/201751)	ALOS AR GOUNSM	14/05 6055	-44/05 5075 50 551	-14/05, 27751, 37751, 48 U45N -14/05, 37751, 37731, 45 551 -14/05, 47761, 37671, 45 551	14/05 20115	14/05/48.60UNSN	14/05 A B. 5PTSI \$0 UNEK 14/05 A B. 5PTSI \$00551	14/05.48 30PT51 4/03.48.27T51, 3PT51, 40 UNSH	4/05/8 61151,127151, 24951 4/05/8 61151,177151, 2581			010 HIELD	H LARIH LLOED, INSULATED STED GAIRS DE USULATED	NSTF 0, SMELDED, NGULATED N.E.S.P.N.S.T.E.DED, NGULATED		 -			ICABLE NOTES BEE BACKTE.			E /2W0/493								Ţ	c	-	ALCARE INTERCONNECT	EDV LAUNCH	-
	┲┰Ŧ	.		, ,	Π		' 		A75MI4089-1C	APBNI0200-4C		APSINOIDO- 6C		-2-Capinory	A75MIOI89-PC	A754091-100	A75MH077-21C	A75MID19-29C		A75MII020-400	AP-400IIMETA	A75MI021-90	A75M0181- 60C	A75MIOIBE-60C		A75MIDI91-60C	A75MICI92-60C	ATENICOS 600	109-BEIDINSZY	A75W0194-600	A75MI026-L0C	A75MIQIO-600	A75MIO1-60C	A75MI009-600			DISH - UNSHIEL	SSI-SINGLE, SHI	TTSI-TRIM 6.		T	r T	FT	POR APPL	r r	-	-			гт	11	- T -	- 1		тт	, 	-			
NAN NO VE CHANNE	1								•																			44		12			-																										1]	
A CONNERTOR PU																																																							┽╆ ╽╽ ┥╋					
BULF COMPETER		084	3			9	8	00																						041	061		193	1204	3 2	0.91	395		100																		0140	لل		
ALCANE DU		0754/07755-56775440400	075407753-31	D75H 07768-24 A75H0H0	075M07799-16	D75M07755-38 A75MIC	D75MD776 52 A75MD	975M07757-4 479M080	_																					i.	107547766 A75410190			0794020 A7940204	1		D75MISIAI A75MIDIS5		075M7757 A75MI0160		075M7753	D75W7755	075M7755	075M7755	075M7753	D75M7755	075M7762	07547754	075M7753	075M7755	C577M270	07547755	075M7751	1677MC10	075M7753	1511 MELO			ļ	
TION A SALL	12-11	SOBBAINT DTSHISEES-I	5- 10 5MI	11115 -0			÷ 3	1 - ES 3 CIMOL		7		N5 1	\$7455W5 \$7455W7	9M	6.41	NII -	EIM	W14	61.4	W20	1117	455AIW4	AIWS	LIN'T	7955AIWI 7855.Aliw2	VINZ		STASSAUM6		075409047-1	- 49050H540		075MI2016-				9	V12 31 52		301	2 M	6.VI 7.LI	ewi e	10mi - 22	1)WI 24 · · · 23	13WI 25	14N1 - 30	5K 145				20		22W2	11.		ANT ALL ALL ALL ALL ALL ALL ALL ALL ALL AL			
ALL REALIZION		506	8905	5085	2008		5048	100	T		SV/3	5//5		574	5745	SV2		SWS	51/22		5115	51455	55V25	SANTE L	57455		57455	51455		9003	2 M0009		6005W1		\$002M	\$007	50	15001	15003		10031	1001			15009	(002)	V5009	1001	15009	5003	1003	15009	6005A21M	15009	1003	6003				
PANNED IN AL AN	<u>.</u>	×	•	-		8							1		-	58	17-53	4	4	8					8	4							4	9 /				- 4	2		2	2 9		8					+										1 1 1	•
COMETOR																																																									+			
AUTA . COMECTOR		VIIDI92-	A75MILD 05- A75MILD05-	SPEC 33284	ATSMDIBE	MAY SPEC 332/44	A	LIPEW	10 = 3 W/WE	3/0	W 20-8	W 20-0	SPIC XVA		A75MIOI60-	A75MIR40	26101	50011	W785 3345	10192	TO MAK SPEC 372AM	10138	SPEC 332/M	1600 I	26101	5PEC 372/NO	10193	MERT SPEC 332/79A	10193	INC 334/754	C6101W52V	161019	MI0193	26101454		AT5 MICHAE	76101	10131		1015	1601M	NICISE		A75 MIG192	16101		:					RIOIDE	A /5MiOl92				Ŧ		i	
a Ciercary		75M15164-78	75M16069- 48	75M4047 40	79411281- 54	075M112M- 10	D75413994- 101 1177	711 28 -945019510 542-	D75MI1285- 67 3CO	0.754 06 -265 111527	TAYT OB -HETEINET	3471 25 PP900004 255		La Laconcia	D75M09070-259 075M16080- 17 A784	D75412012- 25	D75M16046- (38) A75M	175405070-254 DTSMI1203 9 A7541005	DISMORIT- 44 DISMI4477- 60 MSFC SPEC 324	018 MISINE - 110 4751	FFEMISIA3- 30 MSK	075MISHE 2112MISH	DZWISIĘS WOLC	MSTA 1 SUSINISTO	HELP DE - BIISINSLD	075M16043+ 15 1455	075MIS112 1 4754	D75416067 MSR	D75MIS117 . A75H	075M16063 075M	Ted 875 MISIN 210	075M5162 A75	D75MISH7 IS A75	10 075415115- 140 APS		DISMONIT TIDIMENDIA NO ATS	ISLY COL SIIS INSLO	D75N0 0117 74 075M1545 445 A75M10791		SHOWER IN SILENELD IN THE WELD	LEV IO- SIISING	DTSMPerT -se DTSMISHS -50 ATSMIDISE		-12 D79MISILE -47 A751	VSLV OIL- SIISINSLU	DF- CHENCLO	71 DTSMISIE4 140	DTSMISHS 145	075MISI64	Sti Cilcincto	ELV MILE EILEIMEIN	DISMINIS - 64 ATSMICIDE				DTIMOTTS 24 APA				
CARLE		D758099070-384			015M0900242	UTSAUGIO	-		+			202 202	152 · · · · · · · · · · · · · · · · · · ·	CANNEN IN		/ D75M12024- 46	ļ		1		075N09070-343 875N15163- 3					/ DISMDBOID-340	32		59C.			2		075MOBUT-		Π	Т		Π	Π	T	Τ	Τ	D79MOBIN	DT SMO	+-		<u> </u>		-		D75M0E0146-				879MC 043 - 2				
NEF NEF		INOCH!	100m	14004.7	14004/	NIGION	HOPING	LA PROPI	140/081	ALD ON	140/041	140-06 M	MOLENI	(40.44	PIMBORI .	140311	IN POPI	341011	IND ANI	INSON	INIT INI	MININ	MITTEN	141144 M	MIT IN	141121121	14124141	141242161	1412A2#	1412A3W	EMETZIAI	141244W	1412441	ISIAWI	141412	1414W4	INSIDI	1415142		141611	1416W3	1418MI		1420MI	ALT'S AL	M271211	1421A2W2	WEVIZVI	1421434	1421434	1421241	1429WI	5 462 PI		1001	SCEEWI				

										-																-					• • •											SEMONELE		111LLX										A Jos I have		
	-T :	11	••••		TT		- -			- 1-	TT	11			• •	71			T		.			ŤŦ	гт	ŦŦ	ŦŦ	TT		.	-		· -T	Note:		TT				ŤŦ	T 1				, • • •				ŦŦ			CALL NEW CALL NEW CONCU	2			
1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1		2-	- 2	35	5.8 	*							25								22	• •	22	2				2	1		2	ä	28		12																					
2	τ ι																249-01-1828-01 518-07-18-0																																		CALLER DISTOLATION					
									14- 16-	-				·		ć.	I HSAE SAKTEMIONS OF	ġ š					ŻÉ									di i		è			ġ		33			żż	*	÷ 3			k a			1.1.				intinued)		
CANUT CANUT		A-11 ONNOTAB-	HI ON- HIMANIA	PEMILOR- ST DTHING THE	LONGLASS - BORNELD	NEWINELO I - GLOWELO	162n - C-	Ţ.	-1001- -1 1001-	÷ •	1.4	USI WHAT IS ROOMELO	TTONER N-OFORMERS				4M 12-				Prevention - Englished	LINIMELODON-OLDSONILO	D'RMORD MULTIMOTTIN.	ILOWED & OLOGONALD	DISMONORACOTMOTIRS-		LIDWSIG 25 GLOSOWLIG	FTOMETO AN UTOROMATO	- HELLOWELD HE - CLOSONALD	DISMONDIA MA DISMONTI	10 11- 1	-IGTTOMPTORE-UTOPOTATO	PRACEOCI-2 DIMONTRA	DISMOSOEI - DISMIAS	DISHOROGI I DISHOTAR		99000	• •	9₹			- CTICHARDEN 1020 - CTICA	110 KOT	10	00 1 00		DEMONSIC L-2000441	DISMICHT & DISMOTISI	FLOWER 1-21-04-210	COLBAZVI - DISMOTIBI-	TONATO A-SHORAT			cconnect Diagrams (Continued)		
- 14 VIL	IN BOOM	T THEODY							COILVE				11		П	Т	CORVE	CORVO	1015		CONTRACT		4014WE	IT	4015V2		Π	H	COLENS	11		SWFIOL		livelos	6021 MI		11ME 209	SIME202	POSSWH			6024VI	EIN VION	COZAWIA EOZAWIS	6024W16		1020M	17100	LOZB AWI	TT	-			connect Dis		
																																																						Cable Interc		
N																																	-							11						1 1 1								Figure 6-6 Ca		
. Pute		1				++					 						żż								1				÷*					1			•											· 1		· ·				Figur	• •	
CAL CAL	÷	20		Drantona I InteroTita			10 + 6	5 1 5 1 5 <u>1</u>	110 8.	01- 1-		10	10 10	110 13.	110 88-	10 62.	10 N-	5	E.	10 14	5 5 7 7	-40 0TF	49 40	10	10	3			HEI HELDER VEORINE	75M12028 41 DISHING	1410 1- 1-	-6410 4-		ATTO 01-	2410 EI-	Setto SI-	2410 · 11-	122- 14110 ES-	-24 07154	1410	Mu 10	110 of .	×	110 L	1410		6114 6114				CTMILLI IL DIMILL					
	1.	11	1		600ev 8	177000	400xm2	COOLUIN	POLATUL	LOOLANI	LUCON MIL	1004ATWI	1 TOOLA VI	POSA IOUI	GOGARWI	POCKAIAWI	COUNTENT	LOOLANSWE	1744 PO0	EDOCALSW!	POCKATOWI	LOOLATVA	LUCALLY C	Indepoor	600654WI	100000	CODES BULL	1000 S 1001	10 1/18187000	601W8	1VZATOOJ	COTASHI	600745WI	1414000	1 MGATOON	LWINTOD	COUTAISMI	IN NALA TOON	COTAISWI	CC O ANNI	1001 × 20 MI	COUTA 21WE	6007A22W1	10035101	CUC 7 31.1	000	600157WI		6 (c 15 m)							14 V

AN AN

à

												_																								_																												
				•																																														CONCL3	CABLE	INDEX						T	0.	·			Thursday 1	A CONTRACTOR OF A CONTRACTOR OF A CONTRACTOR A CONTRACTOR A CONTRACTOR A CONTRACTOR A CONTRACTOR A CONTRACTOR A
201V P																																																														EQUIPMENT, LC39		the second se
																		_																							Norm:																							14
]	8 X 8	ŀ	┥	\prod	┼┝	-	-	┥┤	╢	\parallel	+	2		H	+	\prod	H		$\left \right $	+	+	╟	+	$\left \right $	μ	╀	Н	+	+	$\left \right $	+	+	\prod	╀	$\left \right $	+	$\left \right $		+	╢	+	H	+	Η	╀		Н	╀	\prod		+	\prod	\parallel	+	+	╢	\parallel	╉		2	1			
			1		╢	\dagger	$\dagger \dagger$	\dagger	11	\dagger	\uparrow	Ħ	Ħ	T	T	Ħ	t		11	T	T		t	t	Ħ	t	Π	T	T	Ħ	Ħ	T	Ħ	T		T	T	Ħ	╈	Ħ	T	Ħ	T	Ħ	t	Ħ	Ħ	╋			Ť	Ħ	\dagger	T	Ħ	Ħ	Π	+			. 1			
ļ																																																																
			\dagger	╟╋	╢	$\dagger \dagger$	Ħ		11		+		╞┼	Π		╫	Ħ			t	t	Ħ	t	t	╟	t	Ħ	T	T		T	İ	ŤŤ	İ	İİ	Ħ	ŀ	İİ		İΪ	Ť	Π	Ť			╏╏		T					Ħ	T		İİ				HL	n n Th			ĥ
	anner a																																																														Ħ	
ļ	ě		T	Π	Ħ	Π	T	\prod	T	Π	T	Π	Ħ	Π	T	Ħ	Π	i				Π	Ţ	T	Π	Ť	Π		T	Π		T	Π	T	Π	Π	T		T	Π		Π	T	Π	T	Π	Π	T		Π			Π	Π			Π	T		Πl				Ľ
	COMPCTON CON																																																															
		T	Τ	Π	2	Π	Π	970	010		2610	211	66	0070	0000	1410	0810	Π				Π		Π	Π	T					T	T	Π	Τ	Π	Π	8	1610	000		Τ	Π	Ι	Π	T	Π	Π		Π	0010	0010	0	11	017 180	180	(10	22	000	0010				ho	3
	Canton Contraction	INSCR			ATSMIOIBO			IIWS47	IIMS A	A75MIC	ATSMIC	AZEMIO	6101M514	A 75MIC	A 75Mic	A75MI0191	ATSMIG				Ì		T	Ī		Ī				Π					Π		A75MIOIAO	DIMSCV	A75M10200		T			II	İ					A 75 MI0180	0 1454 V	A75MID	A75MII017	A75MIQ	A75MID180	A 75MIL	A 7 FMIO	A 75MIO	A75MIOIBO A75MIOIBO	WHEELY I			H nu	
				5566	D75M0773-1 075M07723	0%6	1466	6411	862	D75M08112-1 D75M13967 A75M10191 D75M08112-2 D75M13967 A75M10191	3942	3942-60		1 1	17782	DTSMOBILE-9 DTSMIA457	775	1511	15/1	7755	1753	1154	7751	154	1321	2524	154	7753	154	154	1.151	756	1	15/1	7.55	7758	1754	767	20%	1753	1811	1344	1511	137	1.5	D75M09076-48 D75M07755	15/1	735	1)56	5511	154	151	104	101	7544	196	1.22	121	D75M09075- 1 D79M07756 A	947			connect Diagrams (Continued)	
	2	D76MG	114540	04440	DISMO	09660₩520	015MI 3464	D75MO	D75MI	D75MI	D75MI	IWS/Q	019HI 3968-	DISMO	P75M0	DISMI	D1540	OM210	DH2/1	D15M0	OM240	OWSID	DM220	07540	DNSID	OM270	DISMO.	DISMO	DYSMO	OWSLO	D75M0	DISMO	104510	015MO	D75MD	D75MO	DISMO	P75M13	D1540	DWSLO	DM270	OWSIG	D73MO	011510	DWSLD	D75M0	04510	DTSM0	DNS10	DW5/D	01540	DWSIC	DPSMI	DTSMO	04510	D75MII	OWSID	015M0	DIAMO	1075-01)) a	/ 0
	CALLY ALCONG	1.110	10		1-044	1-09660#522	075M04014-8	7-9-204	P074-10	B112-2	0112-4	8-119	D75M08112-13	51-2110	91-21100	0112-9	90164	1 9/06	F-92060	1-94060	9-9/040	1-92040	8-92060	01-74044	1-92060	1-94040	9076-14	51-9206	11-7406	91-9406-	9076-19	12-9406	PALLOWSIG EZ-WOLOWSIG	9076-25	97-9406	12-3404	9076-29	01-2110	11-110	9076-32	11-9404	1076-35	1076- JC	64-9206	<u>9076-45</u>	9076-48	1016 43	9076-4	39-9204	10-10	01-5-10	11-54.04	5-1/04	9075-1	1-5105	1-9-06	9-5206	1-5106	5-5/06	1-1406			mo	
	ŝ,		-	0.00	075M	P75MQ	5W520	DMSLO	075M0	D75M0	D75MC	D75MG	DY5M0	D75mc	D75MC	DISMO	DISMO	DWSID	D75MC	D75MC	DISMO	D75MC	DY5WC	075MC	wsta	D75M0	DTSMO	DNSIG	D75MG	DTSM0	D75M0	ONSIG	OH510	DWS10	DYSMO	DM270	DISMO	DT5MO	ON210	0115110	D75M0	DT5MG	DTSMC	D75M0	DN810	D75MO	DTSMC	075MC	DTSM0	DM510	DNS10	D15MO	DMELO	D75M0	01240	015MO	DWS2 1	D75M0	015M0	01940			500	5
	ALT.	14 40	6071 W I B	1	AOT DW 2	CTOW 1	IMGIID	3		6121W 20	W 28	W 24	90 A	W 28		1 1			6122W14		11 11	9 1	2 :	EL MI	12M	16 M 10	1 me	87. N. 18		CE NZ	-	N 31	612W 35 0	77 7	2 M 4 1	4 + A #	2 W 48	1.11	6 M O	1.1	EIM I	¥1M	8 A A	11.1	UZ.M	6/24W21	2 3	18.1	¥ 46	6124W43									PIE BWZ9				Ĕ	1
		Ĩ	5		2	209	Ī	3	55	13	613	5	5		3	519	5	2	3	5		3	2017	3	3	3	012	10		5			7	3	5		1	618		19	1	5				Ę.	6124	613	÷.		23	8			1.1				33				U O U)
			-	-		H		П	П	Π		Π	П	Π	Π	Π		Π		Π	Γ	Π	Ţ	I	Π	T	Π	Π	T	П	Π	Π		T	Π		Π		36			Π		П	Τ		Ī		Π				Ì	Τ	Π	Π		Τ					iu o o	;
	9					╫		+	+			$\left \right $	╂╋	+	$\left \right $		ſ	H			+	╫	+	ł	$\left \right $	+	H	$\left \right $	+	H	+	+	+	╉	H	+	╟	+	-	Ĥ	+	Н		H	╉				╟	+		$\left \right $		+	-	┼	+	╀	╟	H			tari	
		ľ	i																																																												Cahla Inter	
1	8	5	SF-13	14-74C	8 .	1	i.		2 2	55-53	5-51	54-35	51-32	51-35	St - 75	2 5	3		+	$\left \right $	╉	╫	╉	╀	╂┨	╉	Η	$\left \right $	╀	┼┼	+	+	+			╋	╟╋	+		$\left\{ \right\}$	+	Η	+	┼┼	╉	╢	H	+	╟	+	+-	╢	4	+-	+	╫	+	+	$\left \right $	H			ahli	
	CONTERENT	CINALICE	P. HOLEY	SUCK RN	IN DAR	10-223640-015	23-98 JAON 54	NO-	MS 2006 8145- 25	18 JOLE	55-51/290/1 SM	01090183	19106R10	DIALOUT	01090165	190165	10-123640-845																							1																							C)
1	No.	14.75	11 11 11 11 11	1-53	16-5+14 90H SH 41-59189018 SH	110-0	4 451-0		12 -15P	¥ 45-6	H 45-5	41-1	-1 BFR 1051-3P MS 3106R1051-3E	M 41-59	W d1-59	W 41-59	1919-		Η	Ħ	╈	┼┤	╉	┢		╈	H		t	Ħ		H	\dagger		Ħ	╈			ł		+	i		Ħ	╋				Ħ				+			<u>+</u> +			1	H			8 - 8	>
	DUNKTON	51 11 11	NUCCES -	190164	190165	418-047828-0V	10-323640-75P	147018 54	10-329640-75P	185814	1 BFR 14	-16- ROI - 10	BFR 105	1290165	1990165	1004	10-33540-81P																																				-	1		:							d r	>
ļ	1	а.			8	ľ	11	11	: I 1							101	T	Π	- 8	2	\dagger			t		+		╟╋	\dagger	╢	t		\parallel			\dagger		Ì	8		29		28	12	+					T		0		20	2.9	3 5 8	20	5,5	20				Fioura 6-6	•
	PULL CADLE		criomice		ATFMIOLOO	A75MICI91	A75MI0200	1014514	A 75MI0100			-		A 75MI0180	 	A TSMIDIBO	121010191		A75MI0191	10IWSL	╉	┼		╀		╉	+		╉		+	┝┼	+		$\left \right $	+	╟	+	475MI0180	10IWELL	A75MI1026	PO JEE JE	101WS4	12MIQ1	+	075407751	$\frac{1}{1}$			1	Ļ.	A 75MIOLOG	175MIDI	12MINE	101W521	DIIMST	101W54	17HAULT	ATSALOIRO	onietes			μ.	•
	1	5	554	2		42	11	752	75.2	787	787		11	1	252	TT		11	82	756	751		53			15	25	154	122	1	751	755	756	111	1	3 5	55	† ;;;	11	11	. 1		967	1 252	75.9	251	2.2	753	5			1								11				
	C. C. C.	icm. J	COM-17	P75407	DISMICHT IS DISMOTTES	45110M510	16770M2791	IOWS(C	075M07791 D15M07752	P75M07	075M07787	10WSLO	DISMO7.	D75M07	DISMO7	D75M07755	015MIOLU2-15 075M07759		11-1 D75M19967 A	104510	1000510	LOWSIG	104510	10mstd	DISMO7	D75MO7	DTSM07	015407	D75M07	DISMO7	D75M07	D15MO7	015M01	075M07	LOW520	075M077	1204510	015407	DISMO7	736EIN570 2-11180M210	D75MILL	1774277	CIMSTO	075 MO7	DT5M07	015M01	DISMON	10M210	DT5MO7	LIOWSID	075M07756	D75MOT755	DTSMOT	ESLIOWSLO	LUNSLO	1111510	LUNISLO	015M07	UPAN'S	Dr.sam194				
		1	è ÷	17		51-215	9-219	2	015410412-0	1.213	• •			27	÷	D75MIO412-14	81-CP		D75MOBIL-1	61-610	-		7		1	8	-9-	=	77	1		94-		6	12.	14-	67.	- 14 - 14	\$2-610	2.11	1071-4	7-112	4-111	21-810	073-26	D75M09073-28	01 120	16-1104	66-610	56-610	11-110	96-610	01-220	D75M09072-11	1-720	11.0	ے ا	1-11-1	075AU9072 4	17.17				
1	33	cimero!			DIMS20	21-21901WSLD	D15MIO612-6	OIN270	DIASMO	D75MIO						DIMPRO	D15MID		75MC8	75M09	┽			\dagger	Η	+	\uparrow		1	Ħ	╋	┢╽			Ħ	1		+	25M09	DISMOB	P75M01	DWC/	D75MOB	20W 520	000510	DV5WO	DISMO	D75M05	D15M09	PPERSTO	POM210	LOWSID	DISAICO	POMPTO	D75M05	D75ALC	7/06.3145/0	075469	CHARGE CONTROL	D7-M13				
	MET CALL	85.81	14.9	8.8	0020MB	146200	IN ICOS	- 1 - 1	GOB3WI	IMING	INTATA OF	IMLINE	INIC	1 1 2 1		GW 6804	1.8.4		6062 W 5 D	2 1 10	2 4 12	41 74	1 4 2		2 4 21	6 A 2		44 7 2		2 M 32	2 W 33	2 W 36	2 W 5	8 3 2	2 1 2	5 M 40	44 M 2		2 W40	1 1	6043 W2	240		0 44	1 1 1	BOBAW 14	91 M	4 # 20	8	-	4440	1443	141	IWE	5422	12.1	125	12 11	ICLAN 20	15 4				
1			33	ŝ	3	503	3	8	20	9	3	00	608	509	609	88	4034 W		9	99	8		9		\$	\$	Ŏ	10	8	Š	0.0	8	000	0	ð	30		0	3	90	8		0	ò	8	Ś	88	8	\$	ŝ	89	ļ,	ğ	9000	3			3	133	3				
1	>																																																															

r ffr	-								•	-											-		In C	ينتز	ŧ			1 -	-		
																							E PHONES	CABLE.				<u>\</u>	13 H		
1		<u>.</u>															ł	i t													
																														CONTRACT NUMBER	
			·															There .													
						111				Ш	ŤΠ	ŦŦ	ПТ		Ш	T	TT				П			Т	Ш	TTI	Π				÷
	2 5-	-+	28 8																										_		
		┽╵┼╀╀╃																													
	ě ž																												╺┽┤┠ ╷╽┖		
	2	┽╍╺╂╋╂╂	0	1017 1013 11012	1024 1025 1025	021	10101		1610		100	0191		, ję	8001	2610	0100			101		A75M14085		4084					++	1	
			┿┿╋┿╄┤	75 A75MIIO17 75 A75MIIO1 75 A75MIIO1	79 A73MI024 196 A73MI025 196 A73MI025 196 A73MI022		ML VIEW	┼┼┼╸ ┼┼┿	TAL ATSMICHT		THE ATSMICIAL	1 1 .			100 A75MI006	1 A78MI0192	222 A75MID200		3/12	5044 A75MIIOIA 5063 A75MI4089		5084 A75MI 5084 A75MI 5085 A75MI		1 A76M14085						Magrams (Continued)	
			DITAD MACHI DIMO THA	DTANORDON DTAN ISON	9080-3 0734 (1219 9010-1 0734 (2016 9010-3 07340 (2016 9010-7 07340 (2016	1 01111 01111	A-1 DISMII		- * -	7 • •			₹₹₹				-10 075M1 972	- 	1 45 1 45 11446		+++	9 5 X		25 D?MI 2003							
	A TRANSA			D1944095	***	× 8	2 5																4 99	1 DISMO 3032						0119 TO 0	
	CINERAL TRAN		6155WI	- MO - MA			01 MIN	CICI W24	LICI WET	UM 1212	CILINTS.	6161 WA1	CICI W 44	6161 W 40	CI61 W 3	6101W5	2161 ¥ 1913 6161 ¥ 1913	6101 NO	10 1 10 10 10 10 10 10 10 10 10 10 10 10	CIGIANW	CICIANS	CILIAN CILIASH	6161A5W	CICINCA						nnect Dis	
	2													9				0					- 9	ŀ						Junoo-	
																														Cable Interco	
	Cantras																													Cable	
	VELUNE	:																												e 6-6	
	States and a state of the state		A75MI0100						CONICIAL CON	16101191	A79410160			A75WIOIBO	18101MSZV		NIO	A73MIDIEO	CONOINCT	A75MIIOI7 A75MIOI40		A75MIOIBO	TSMIOIT		1					Figure 6-6	
1 4 4 4	11-10-11-0 Michaela Michaela			ÊÉĒĒĒ	<u>e</u> tte				1134 1135	111				17136 1 7736 A 1735	V 01.104		1118 1114 1114		4517 AETT AETT AETTOWE	SMI 1296		A 0421124	SNI 12%	++							1
	6			ŢŢŢ₹Ţ	• 9 = 7	¢ = = # •		3. 	10 45 500 0410			R 2 -	;;;;¥				+ - + 	01-9200	124 124 124 124 124 124 124 124 124 124	9077-1 DI		10 2 100	011-3 Di		• • •						
		111 111 111		0142 WIT	61 42 W24	6145 W35 6142 W35 6142 W35 6145 W35 6145 W35 6145 W35 6145 W35 6145 W35 6145 W35 6145 W35 6145 W35 6145 W35 6145 W35 6145 W35 6145 W35 6145 W35 6145 W35 6145 W35 7000000000000000000000000000000000000	86 M 28 19	142W44	6142W41	1111		6144W18	11. 11. 10. 10. 10. 10. 10. 10.	6141W41 6144W41	bigum I		6146¥5 6146¥6 6146¥6	614 BW I	614 BW21 614 BW22 614 BW23	6148 M24	1 4 8 4 4 4 4 4 4 4 4 4 4 4 4 4 4 4 4 4	140410	140441 075								
						1-1-1-1				<u>: 1</u> 1	.17171		111	<u> </u>	111	11			<u>ו</u> ני				ĽĽ	11	111	نلل	11-	<u>· 1</u>	<u>1</u>		
				·		ه																									

		•		•									•					· .	<u> </u>		<u></u>	an ș																							_		•							E BMOMBG		INDEX							T	<u>^</u>	33						
				Þ																	-																																														TO LAURANCE NEWUNCH	EQUINENT, LCH	ALL ALL ALLA		
			• 1	11		۰T	1		ŗ		Т	1 1	Ţ	ŢŢ	-1	П		T	П	4	П	Т	T	П	Т	П		T	П	ग	ŤT	- T	T	T	П		Т	гт	т	П	Т	гт	T	E Q Q		П		71	T	П		П	 _	1	П	11	ΓT				гт	П		1	r r					12.20	
	ALC ACC				+						+		ľ		+		F	~		+		+	╞┼╸		╞		7					+	ŕ	<u>م</u>	ľ		╞	┝┥	- 2	- -			┼		╋				╺┥╕	8	2=		+							$\frac{1}{1}$			-+-				.	Ί			
	All COMMANDA				-																																																																		
	CONVECTOR																										-																																												
- 1	CONNECTOR O	i			 						 																																																												
	CALLE	A75M (0160											INCIMERA	9910			AT/WOIRO	ABMIOLOO			A794101BO	A19MIIDU7	101	310	200				H	0100	\mathbb{H}			000	101	88	LIOI	80									0190	101				₩	0180	A79MIJOI7															(Locial)	nanu	
CUNT	ALLENGLY RG-ALEN	1110m110									-		DISMINUT	DISMOTISI			D75M0 7751	B73MO 7796	1.1.1	Side of	D19MOT191	Contra 1 (16)	DIMINOS	52220WS20 11-1806	COLUMNER .					1011 M 110	58-1			1217 OM610	D75M11296	D15N07751	D75M11296	D71M07756	1.51	1796			1512	1317	4411	7754	D15411304	D79M11304	PTSMOT74	7762	7762	+5/1	1 9082-97 DIIMO 7754	515WILLOOF															Hace/	Caule Interconnect Diagrams (Continued)	
	A TO AGO AND A	- TROMON											DISMO BUT 41				at Wittomsta	PRM0305.1			Drawosocs-4	DTSMO PORO	+1 2000 1	11-1906	4	1 1	- E.		1 3	- 2 - 3		- 4	1 1		П	T	90-0906	1	П	T	П	T	Ħ	81.4-18	T	H	╈	Ħ	16-36-	H	56	1-26	9060-51	019 NO 5060 24																auna 181	
2			-										I M GHA	I A 2 71 7	SI M C SI O	HWGOIO	0 A 80 9	PIETA IWI	CICTMTIWI	14722 W	146759WI	A SA	AISBWE	5 A 9 1 1	S M S SIA	6-1 6 0 0	4148W 7	- 6 49 9 -	61494 10		LI MBOIS	CINDAIN	61484 20	6166W 22	614 0 W 23	LIABW 24	6168W 26	6168W27	ALAW BI	6148 W 32	616BW34	414BWBB	614 BV 37	4148W40	11 84 15	414 8A 23W	6100 A 2841	GIABA SIWI	0100 A72WE	W18 4 8714	61684844	8168 A 66W	6168 A 89W1	616 BAJOBUI															ont Die		
		-		+		Ţ	H		Π	Ţ			H		T		T]	T		F	-	ł		Ц	Ţ			ŀ	Η	T		ļ	Π	Ŧ	Η	4	Π		╞	1	-	H	T	H	1	Π	Ŧ	Π	Ţ	Π	Ŧ	Η	+	Η	Ţ	Π	T	Π	-		Ŧ	Η	-				200	CULLI	
																																																																					a Inton	TATIT A	
2				. I.																																													·																						
ł	Martine Contract	•		•	i 			!																																																													no 6_6	0-0 a.Instr	
ANK A		800	8			1	<u> </u>	-			+						-			┦		· ·	PMIDI91	89								$\frac{1}{1}$								-	APHIONO		0100	0100													İ		+-						AT MINING				ie Li	T 18u	
		Derrowing	- 15 - 1	121			7156			- 22	711	1111	7155	Ē			-									77.9.6	134	1517		84	1116	554		1511	7156	7755	7755	+	1 7756						7754	1153	1.1.1	Î.	154.4	441	*C11	1154	1155	111	+ 1512 +	181	+ + + + + + + + + + + + + + + + + + + +	1 755		- 1191		1754			÷.	1					
	1	silvine				?!? 		-		+	: ; ; T	¥ 4	2	-	: :	┝┼	+	╉╋	+		13-1-1	A BOLDOLOWICH	DTMOBILE 0 0	0.11-10 000-11	<u>1</u>		***	39-	9		4511 15.	6.	0.			77			19.		DYMONOR'SO D	IN ACTING	NIA 15 DI	15-7906			95.	96.	9 1 -	19-	35	;;	3	6	6.6	12	<i>n</i>			2.			ş ş		UNME SCAL-67 DISMC705						
	_		1	6142 W 19	6163 W IS	6162 M 1	6142 M 10	6142 W P2	63 M 1914	614.24 E	6162 W 26	6162 W27	6162 W 29	6-62 × 34	-6 A 3919	6169 W 37	0CA2419	6162W 44	6162W45	61 62 WAY	6162 W ⁴ 8	4. 4 7010		61634 9 61634 0						616 TV 20		616 TH 21	6149W24	616 BW29	6143 N 30	010343	GIUSW35	6169 ¥34	GILSWAI	B16 3 442	HWERIS	1 1 1 1 1 1	6164WS	GK4W 0	6164W12	DI64W13	6104W15	6164W17	61644 19		6164W41	61644 23	6164W24	6164M26	6164W21	6144W 29	6164W35	6164W32	6164W33	E164 W 32	6144W 34	210 JW41	6.16 4W41	· · · .	2 1 2 X 2 1 2						

			1.		
	•			• •	LE NUBCONE
		······································		****	
			<u> </u>		
Curl and read	Atsentoid Atsentoid Atsentoid Atsentoid		475410160 475410160 475410160 774410160 774410160 755410160 755410160 755410160 755410160 755410160 755410160 755410160 755410160		
				11111111111111111111111111111111111111	Diagrams (Continued)
		\$\$\$\$\$\$\$\$\$ \$ \$\$\$\$ \$ \$ \$ \$ \$ \$ \$ \$ \$ \$ \$	101 101 101 101 101 101 101 101	243 Presented al P	Diagrams
	4103/media 4103/media		14 6.004.040 0.004.040 0.004.040 0.004.040 0.004.040 0.004.040 0.004.040 0.004.040 0.0004.0 0.00		
	13544(0) 14544(0) 15544(0) 15444(0) 15444(0) 15444(0) 15444(0) 15444(0) 15444(0)	479 94(10) 4794(10) 4	1154410191		Figure 6-6
			Listenser Pressent (1970) Presse	I SALOVÍA	
Print Print		B.201m D75m00064-1 D B.201m2 D75m00064-1 D B.201m2 D75m00064-1 D B.201m2 D75m00064-1 D B.201m2 D75m00064-1 D B.201m2 D75m00064-1 D B.201m2 D75m001015-1 D B.201m2 D75m00115-1 D B.201m3 D75m00115-1 D B.201m3 D75m00115-1 D B.201m3 D75m00115-1 D B.201m3 D75m00115-1 D B.201m3 D75m00115-1 D B.201m3 D75m00115-1 D B.201m3 D75m00115-1 D B.201m3 D75m00115-1 D B.201m3 D75m00115-1 D <td< th=""><th>1202.000 1202.0000 1202.0000 1202.0000 1202.0000 1202.0000 1202.0000</th><th></th><th></th></td<>	1202.000 1202.0000 1202.0000 1202.0000 1202.0000 1202.0000 1202.0000		
		1:1:1:1:1:1:1:1:1:1:1:1:1:1:1:1:1:1:1:			
6-178					

)																																										a verse and ar a .		-									<u>.</u>													ENHAUENT L'TRA
	T	11		T	ŦŢ	—		Ŧ		- T	T		- T	Ŧ		-r	1		1 1		•	-r	- -	ŕт	-	. .	- -			ਸ	ŕт	- - -		•		_	* -		-				-	: ESHOV	AN APPOINT	••				•				_	-														and the second second second
AN 10	_	;						+	H	+	╞	H	+	╞	H	+			╞┤		H	*	ĥ		╞	H	+							ļ		\downarrow		╞	È	-	Ľ		+	₽	+			Ļ	╢	H							╞	\parallel	╁	4	*		ļĮ			IJ		1	Ί
																									ŀ																		.																		ŀ								1
4	No. 2			t		t		t	Ħ	Ť	t	H	Ť	t		╋	Ħ	t	Ħ	\dagger		╈	H	H	t	H				$\ $	H					+	Ħ	╏		\dagger		ľ	1.			\parallel	$\frac{1}{1}$	$\frac{1}{1}$					╎	+					+		1	<u>!</u> 		+					
	-	i		+		ŀ	\downarrow		H	+			╀		H	+	H	+		+		+							╀	H						╞		+	╢	╡			-		╀	╢	\downarrow					· +		+			+		╞		+		╟	╞					
Z																																																																				18	1
AUC	Τ	-+					T			T			T	ſ			Π	Ī		T	Π	T	Π					1			Ť			ŀ	·	1		t		t	ŀ	+	Ť		T		+			+				+		1	t	Ħ			+			\dagger	\parallel	╞╋			
	1	ż				į		<u> </u>	÷,											-		+	÷										<u>ار</u>									 			 -							 .		 .								H		+	H				
		0782				1 3	1								4	Ţ						LUG M SLLD	-196EIWSLD	0176-	011	16	ano ano	6		410	Ê E		6	110	210		61	6	DT-MOTTLe	DISMISSIO	10TIBL	6	0112	7410	T	H	0110	0115	0119	0115	01%	TLO 0	- 1	1210	WI LO	16110	SLL0	011	-1110	-110	ALL OWNER								
201		OULL N				*	+	-	ş	=	-	*	¢ i	<u>9</u>	? [17	22		¢.		12-06-06-0W	106116-5	110 HI-4000	09090	207				¥.	*		3	ŝ	ş	ŦŦ	\$ -		1-06060	08116-3	0816-16	14-04060	ŗ	9 :	**		₹ ;	*	<u>ج</u> ا	¥ 3	3	• •	3	1	\$	\$ \$	3	ş	; ?		2-05000		I	T					
10 mm			=		£	-		32			4										•	•	1 1	- 1	11	- 1	1 1			4 I	- F	55	62 C	Ē	2	1	Į			15.	┝┥	╡	H	$\left + \right $	╇	H		22		5						3 5						Η	╟	╀	$\left \right $				
		612240	1223	N2227	6223	62134	12229	6232	6222M	12129	6227W	4224	12230	6222	62221	61224	84M2229	12230	12210	12280	8+ ME 239	42229	W8229	GIT BUT	C12PV	622.81	11222	16229	W6229	622 M 20	6229W	6146119	1229	1223	1227	ME229	ITWE 229	6559 m	(223)	6224W	4229	6224V	C224W	6224	6224	624419	6234V	1.229	6234W23	A+224	62.44	6224V	6214289	WA230	MEET	C224W	V1223	10229	6224W42	C TAN		Ш		Ŀ					
2		1	r				138	ň			2];			ļ	Ţ	μ		39	ļ	F	Ţ	×	ľ		×	4	į	-	Π	-	Π	+	Π		Ţ	H	Ţ	ļ	Ţ	Π	Ţ	Π	H	Ţ	Ħ	T	ÍП	Π	Ţ	Ĥ	T			Π	I	1		Т :	Ţ		[]	•	Π	Π	T]		
PRINSIED BY		-					T				Π	•	Ī	Π	I	ſ	•	T				T				"		T			T		Ţ					T									Ī					t							+	-+				Ħ					
					_	- -	-		+	1			╞		+				ļļ	+		+		+		. 	$\left \right $		-		+		+									+	H		+				+							-	Ļ	-	•				- 	Ļ					
Constraints							:													ļ																																					1												
mular me						•				_																·															T					\prod		Π		Π						-	1				1	•	T	Γ	•				
	1						• •	• 	-	t		ł	1		1					╀		+		+	╟	╁		+	+		+		╀		+	$\left \right $		+				+		+					+							+	+	!	; 		1		t	1	، بر ز	+			
	1					•		1				\downarrow		Ц		İ.								ŀ				ļ		Π																										1	 		1										
Indianaly CARLY	: 					.	-INLUM	-+silo	6	36	MUL	1	MI5182-	01172-					DTSMOTOH- IL DTSMITT	· ·	- 46-211W	-ONE GOIN	MOTHN-	CTSMOS NO-						-6110	01169	0169.	01195	-1961	-			1	13941	14103-	-1161	10101	15392			15.392 -	1001	-580%	1004		15083		- 19085.					•	,				ł		; (
ier le	i					-+	AL- 22 D15	K-180		2 #	Ma 1 .000	+	DA. E. DF.	1	8.9	£			ELC I- SO	-	104-190	DTSHOPPOP IN DTSHOP SKO	10 . 11			107-100	DISMOSOBE ZNDTSHIBALE	111 P.11	1-1000	2-880	101 2		2-860		~ 4	27	4	22	22	ş	e l	e R	9	<u>,</u>	2		+				•		1910 B-10				! -	+	, 	Ī		+	+			+			
DESIGNATION ASSEMBLY							L'TE MUM	6203W3 01001-1	-	+	NowLa .	+	NOW-SLO	8			1		DISMON	1	COMPLO	Dismon	LOWIGL	DTSMO9960-5		2-10060-4LIG	COMPLE	TRACK.	8	8	01012		60	8	Ţ				Ţ		Ţ	Ţ			Г	08116-	П	Π	$\frac{1}{1}$										-	•			,			+			
8										Ł.	3	1	1	1	15	15	2		6216ARUI			1	11	1	11		11	1	1	11	1	11	1.	ы	1		L.L.	1.	1	10	_		Ы		1	1	l,n	i.	و ام		-		CZZIASWS E	1		1	11	1	ιÍ	Í.	1 i	11	1	11	11	1			

.

	9 	E TEMOROS	
			ELECTRONICI
Mark Mark 1 1 1 <		7184. 01800 7719. 01800 7719. 01800 7719. 1719.	Continued)
			nect Diagrams (Continued)
			terconnect
			6 Cable Intercon
	ATMINOT ATMINOT ATMINOT ATMINOT ATMINOT ATMINOT ATMINOT	очи сочи сочи сочи сочи сочи сочи сочи с	Figure 6-6
	1 1		
	LATANAS LAT	Control of the c	
6-180		· .	

ž

•			•																																				-																9	E TSMOP493	1. 201 1.02	CABLE	INDEX						-			TONECI II			ł
	e	3																																																										•								THE MONET			
2		• E	¢ ŗ		E.		5			2	5	ş	T	5	96				9	9	ş	\$ \$	\$	40	\$	4		9 9	ş	9	4	39	9	9	39	9	-	40	\$	40	\$ {	ę	8	2 4	П	8		8	4 4	Ş	\$	•	\$	•			T		Π	T	Γ		T	T		T	П				
							Ť						T																																																								l İ		
ADT TOWAD																																												Ţ		Ţ			Ţ					Ţ			Ī						F								
COMPETER																																												+			-																+					Ц		1	
	aryawiqueo		T	00101W54V	A75M11017		1	A75MI0204	- 1		1 1			A75MI1017	11		475MIDI80	11			11	BOOIINSLA	11	1			- 1	A75M11014	A75M11024	A 75M11026	475MII026				. A75M1026		- E -	56101H51Y	A75MIOI95	£ 6101W52 ¥	A76.410191	161015.517	415MI0181	A 75MIO192	475410192	A75MI0192	A75MI0192	ATSMIDI92	A75HI0191	Г			475MID10	274MU1037	П								+								
2011	151 015W01751	59210H520 3	10 075M07764	1 075401144	40 D75MH1296		11 015MISIO	5 D75M14529	57 D75MI4529	11 DISMINIST	1 DISMILES	614EIW510 4E	10 075441296	11 D75MI1296	35 DISMI196	14 DICLOBED	7 D75M07793		-7 D75409900	D DYSMIA103	11 075414103	2 D75MI4103	E0151W520 1.	15 075413942	17 D75M13942	2 2 2 5 MI 3 9 4 2		2 075411273	1 D75M07794	5 DISMOTTES	66110HS10	17 11 11 11 11 11 11 11 11 11 11 11 11 1	1 D75M11274		-14 DISMOTIPS	20 075407794	21 DISMUT	100114001	22 D75M13968	24 D75M13968	25 D75M13968	28 C75MI3167	29 C15M13967	-10 U75M15942	274EINS20 28	13 075M13942 -	12 075M13942	45 DTSM13942	47 D75M13967	211 SIWS20 61	50 D75MIS132		17 075MU298	19 075MISIB2	A VISMVILLE								+								
		1.5600 VA	-56060W510	-56060W5LD IM	D75M09092-40		D75M0902	D75MOBUT-5	D75MOBILT	VIA D75M09027-	MA D75M0901-	-26060H520 944	075.409092-1	2 075M07072	1 DISMOPOTA	17644084A.	4-74/2045/0		015409960-7		l i					11	- 1	1.240204120	E			╞	ŀ		ł		DTSMORTE - 2														Π	5-21190HSLO	11	D15409092-39	1		+						ł								
	6249W2	6269W4	6169M6	114411	6270WI		6771W16	6271438	6271W4	420/V/24	6271A10252	6271AI0252	42742W	6174A2W	64245478	10.25	4275W2		6276WI	AZBOWI	6180W1	6280W3	6200WS	4280M6	6280W1	6280H	į	12 10 10 10 10 10 10 10 10 10 10 10 10 10	6201W3	1 - 620IWS	62 JIWY	11074	1 Patient	6201M12	- 628IW14	5201W15	100miler	6201N21	6281W22	PEMIRE?	6281W25	6281W28	620IW19	6281W30	626/W32	628/W33	C201W35	62BIN36	- 628/W3)	6EM1829	6201W40	62 DIW41	141629	6291W24																	
191	33	2	2 2		8	8	2 5	36	=	, ,	2			96				-	=		8	2	£	¢ŗ	6	•	-		5	ŀ	-		3	ŝ		8		36	<u></u>	38		9	2	8	2	<u>۽</u>	20	2	2	•	=	8	2	2	8 9	\$	2	8	8	,	2	8									
																										, L																																													
A CONNETOR	-		+																																																												+								
CONVECTOR	•		1	÷																					· -				 			+																						-+							+-										
AL CABLE		091014547	•											╞	ŀ								A79MI0190	A75MI0200	ATEMIALED		•		A75MI0(00		A 75MIDIED	A75MIOH00	A 15MIIO17	A75MI0180	A 75MIOLOG	A 75MIDI 80	A 75MIOLBO	A 75MID180	A75MI0180	ATSMIIDI7	A75MIOIBO	A 75M11017	A 75MIDIGO												 		+	+		A75MI0180	A 79MILOUT	A75MILOIZ									
-	-1 (7/5403467 44 1/7/4/07743	42 0 7 7 40 7 7 4	95410H510 49-	-50 D75M07753	52 075407755	15110-510 13.	14 D75M07754	-14 015401755	57 DTSM07754	-50 D75M07754	1540W510 09	15110W510 19-	41 075M07751	-64 D75M07756	ES/104510 59-	-++ D19M01755	-67 075#07755	69 D75M07754	-70 D75M01753	-71 D15H07754	72 D75M07754	-74 DISMO7754	10 D75413967	D15M08117-89 075M07792	1 076407366	15110H210 2.	ESCLOWELD C.	-4 D75M07755	6 D754107764		ESLLOWSLO II-	10 D75M07753	16 DTSMI1303	PI 075M07751	92 D75M07751	13 075A107751	13 D75M07751	15 015401751	16 D75M07751	-27 D75M11296	15220W520 3.	-28 D75M11296	-4 D75M01756	154404540 11	151104510 11	18 D15M07151	12 075407756	35110W510 2-	-9 D75M07756	12/10/200 1-	15110M210 21-	16 D15M07751	ESTTOMETO PL	10 D75M07764	-76 D15M07154	11 DISHO1154	-78 D75M07762	-80 D75M07754	B1 275407754	-82 075M07754	27 UTSMINO	POCHINSTO 1	T								
MON ASSEMBLY	Concernance of	604-0445	4	+	+	$\left \right $	+	H		+												5 075M09094 -74 D	075M06117-9	-LINOWSLO	1	1			1		1 1	- 1	D75M07012-	075MOBIL7-	075M00117-	- LIIBOWSLO	£1908117-	1 075M0817	1 D75MOBILT-	1 D75M09092	1012m0101	· 075409092	7 D75M09093	6 075MC4093	0 075400117-17 0	D15MOBIL	64040451 D					7 075409093-16	1 D75M09094-				+	+	112	1W1 075M09094	WI 075407092	NI D75M07809-2									
ADLINING		1.5	104 M	111	144MI	1140924	2443	61 64 W1	5164 N2	249424	246.96	244M7	114 M 2	2 6.4 W 3	EW492	264145	26447	264W4	264W4	264 M4	26414	26414	14592	626543	1000	547774	145 W4	621615	246W7		26BW3	626.0W4	246746	148M2	268 44	MARK	248W17	2 LOWI	248W2	208W23	248W2	16 824	26842	268W2	2 LUW3	IEMO92	268W3	2.48.13	268W3	248W3	24843	248W3	-248W4	626644	6268472W2	2 uBAR4	P26HAHS	268484	88V. 11	24 8 4 F 2	CLARCE STOR	PIPAIE -									

1 1 <th></th> <th></th> <th></th> <th></th> <th></th> <th></th> <th></th> <th></th> <th></th> <th></th> <th>į</th> <th></th> <th>i</th> <th></th> <th>-</th>											į		i		-
		-		-				1.10.W.L.0		.			Π		-
							61041	HOLAN L	11	10000		╫			
				+					075W7915						
							6304		ifer sto						
							\$204ME	╢	9544M374						
		ŧ۴.	┥┥ ┥┥ ┥┥				6 304W14		P24M240						
		1					6104WE		ISTONETO						
		CO. PINSCO PC	1				6304M68		16414540						
		THECHNELD IC.	•				16 M 0 0 0	┝┼	075M7794			╫			
	Image: state in the state						6104019	$\left \right $	534LWSLD						
		S + 2 LINE CO 12	A 75MIOIS2			+	6104400		075M7753						
		.	A75M 4008				5104 MAR		075W7754			┼			
		1.1							12/1 11/1					-	
			1				6106W2	MONOWE20	075M7756						-
		¥I¥	A74 MILDAA				6201 N 4		075M7755						
	Rest Res Rest Rest	14	A75 MIQCB			ŦŦ	6306WS	╇	075M7755	-					
	Image: status Image: status<			·			6306 W 7.	SPSIMONONS	D75M7764						
	Rest Res Rest Res Rest Rest <						CMBOST	1- 19080WSU							
Result Result Result Result Result Result Result 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 <td< td=""><td></td><td></td><td></td><td></td><td></td><td></td><td>6 10 BW 4</td><td>1-10000W540</td><td>11</td><td>75MI0180</td><td></td><td></td><td></td><td></td><td></td></td<>							6 10 BW 4	1-10000W540	11	75MI0180					
Status Status	Result Result	- 1	11			F	6208W5	PERIOR IN CONCLUS		75MIID17					
Term Term						34	610BWIG	E BIIEDHER							
Result Result <td>Figure 6-1 Figure 6-1 Figure 6-1 Figure 6-1 Figure 6-1 Figure 6-1 Figure 6-1 Figure 6-1 Figure 6-1 Figure 6-1 Figure 6-1 Figure 6-1 Figure 6-1 Figure 6-1 Figure 6-1 Figure 6-1 Figure 6-1 Figure 6-1 Figure 6-1 Figure 6</td> <td></td> <td>11</td> <td></td> <td></td> <td></td> <td>6206W24</td> <td>1-100001510</td> <td>TM7761</td> <td></td> <td></td> <td></td> <td></td> <td></td> <td></td>	Figure 6-1 Figure 6-1 Figure 6-1 Figure 6-1 Figure 6-1 Figure 6-1 Figure 6-1 Figure 6-1 Figure 6-1 Figure 6-1 Figure 6-1 Figure 6-1 Figure 6-1 Figure 6-1 Figure 6-1 Figure 6-1 Figure 6-1 Figure 6-1 Figure 6-1 Figure 6		11				6206W24	1-100001510	TM7761						
	James James <td< td=""><td>1075M7751</td><td></td><td></td><td></td><td></td><td>930AVES</td><td>- 18060MS4</td><td>2547M25</td><td>OF IOIMEL</td><td></td><td>╉╉</td><td></td><td></td><td></td></td<>	1075M7751					930AVES	- 18060MS4	2547M25	OF IOIMEL		╉╉			
Result Result <td>Barrier Barrier td>3 D75W7754</td> <td>┿╺╋ ┥ ┥ ┥</td> <td></td> <td></td> <td></td> <td>A BOBWAT</td> <td>- 46050W540</td> <td>7871756</td> <td>2 DINING</td> <td></td> <td></td> <td></td> <td></td> <td></td>	Barrier Barrier	3 D75W7754	┿╺╋ ┥ ┥ ┥				A BOBWAT	- 46050W540	7871756	2 DINING					
Image: second	Jacobie Jacobie	1544MS40 5-					6 20 B WEA	175000017	7547751					Π	
Rights Rights	Figure 1 Figure 1 <td< td=""><td>+624M540 9-</td><td></td><td></td><td></td><td></td><td>6 SEAWID</td><td>I- BIIBONGL</td><td>1617427</td><td></td><td></td><td></td><td></td><td>1</td><td></td></td<>	+624M540 9-					6 SEAWID	I- BIIBONGL	1617427					1	
Reserved Reserved <td< td=""><td>Plane Plane Plane Plane Plane Plane Plane Plane Plane Plane Plane <td< td=""><td>1 075M7754</td><td></td><td></td><td></td><td></td><td>6398W3E</td><td>18080W84</td><td>9522WSLC</td><td></td><td></td><td></td><td></td><td>T</td><td></td></td<></td></td<>	Plane Plane Plane Plane Plane Plane Plane Plane Plane Plane Plane <td< td=""><td>1 075M7754</td><td></td><td></td><td></td><td></td><td>6398W3E</td><td>18080W84</td><td>9522WSLC</td><td></td><td></td><td></td><td></td><td>T</td><td></td></td<>	1 075M7754					6398W3E	18080W84	9522WSLC					T	
Figure 6-6 Cable Interconcert Data	Figures Figures	9 D75M775					TCN BOL 3	7 7 -						Π	
Times Times <th< td=""><td>Figure 6-6 Contract <</td><td>IG LUSWLIN</td><td></td><td></td><td></td><td></td><td>\$108415</td><td>6</td><td></td><td></td><td></td><td></td><td></td><td>Ŧ</td><td>÷</td></th<>	Figure 6-6 Contract <	IG LUSWLIN					\$108415	6						Ŧ	÷
Figure Figure	Figure 1 Figure 1 <td< td=""><td>1017M20</td><td></td><td></td><td></td><td></td><td>- 630BW34</td><td></td><td>9627.M27</td><td></td><td></td><td></td><td></td><td></td><td></td></td<>	1017M20					- 630BW34		9627.M27						
Image: state Image: state <th< td=""><td>Figure 6.6 Contract from the figure from the fig</td><td>15 075M7751</td><td></td><td></td><td></td><td></td><td>1 100 M 10</td><td></td><td>- Internet</td><td></td><td>+</td><td></td><td></td><td></td><td></td></th<>	Figure 6.6 Contract from the figure from the fig	15 075M7751					1 100 M 10		- Internet		+				
Tight Tight <th< td=""><td>Image: state of the state</td><td>AATTMCTU PI</td><td></td><td></td><td></td><td>F-</td><td>e total</td><td>075M09097 -10</td><td></td><td></td><td></td><td></td><td></td><td>11</td><td>Ŧ</td></th<>	Image: state of the state	AATTMCTU PI				F -	e total	075M09097 -10						11	Ŧ
Reserve in the state in th	Name Name	4917MCT0 41-					6308W49	D75MQ411	1512W31						
Reference Reference	Harten Harten	1844 WS20 (1					630BW42		75w776			+		T	
Figure 6-6 Cable Interconnect Diagram Continue Con	Reserve Reserve						6 308A78W2	3	7547754				+		
Image: state in the state i	Figure 6-6 Control <td>8</td> <td></td> <td></td> <td></td> <td>Ţ</td> <td>6 306A87W2</td> <td></td> <td></td> <td></td> <td></td> <td></td> <td>Π</td> <td>11</td> <td></td>	8				Ţ	6 306A87W2						Π	11	
Here Here	Figure 6-6 Cable Interconnection Contract Annual Contranua Contract Annual Contr	10					G JOSA PEWE	75M0909.40	Γ	SMIDIRO				<u> </u>	
Image: State in the state i	Here Here	10					1MLBINGOES	1 46060WGL	Π	SMILDIZ			2	<u> </u>	
Image: State of the state o	Image: state in the state	24					2100M 2	PENDANA IS	T						•
Nature Nature <td>Martin Martin Martin Martin Martin Martin State State Martin Martin Martin Martin Martin State State Martin Martin Martin Martin Martin State State Martin Martin Martin Martin Martin State Martin Martin Martin Martin Martin Martin State Martin Martin Martin Martin Martin Martin State Martin Martin Martin Martin Martin Martin State Martin Martin Martin Martin Martin Martin State Martin Martin Martin Martin Martin Martin State Martin Martin Martin Martin Martin Martin State Martin Martin Martin Martin Martin Martin State Martin Martin Martin Martin Martin Martin State Martin Martin Martin Martin Martin Martin State Martin Martin<td></td><td></td><td></td><td></td><td></td><td>6109W 3</td><td>075M09099</td><td>T</td><td>- Animi we</td><td></td><td></td><td></td><td>T</td><td>- •</td></td>	Martin Martin Martin Martin Martin Martin State State Martin Martin Martin Martin Martin State State Martin Martin Martin Martin Martin State State Martin Martin Martin Martin Martin State Martin Martin Martin Martin Martin Martin State Martin Martin Martin Martin Martin Martin State Martin Martin Martin Martin Martin Martin State Martin Martin Martin Martin Martin Martin State Martin Martin Martin Martin Martin Martin State Martin Martin Martin Martin Martin Martin State Martin Martin Martin Martin Martin Martin State Martin Martin Martin Martin Martin Martin State Martin Martin Martin Martin Martin Martin State Martin Martin <td></td> <td></td> <td></td> <td></td> <td></td> <td>6109W 3</td> <td>075M09099</td> <td>T</td> <td>- Animi we</td> <td></td> <td></td> <td></td> <td>T</td> <td>- •</td>						6109W 3	075M09099	T	- Animi we				T	- •
Name Name	Res R	2	A74410160				CHEOCS		75M7764 ~				<u>v</u>	Π	
Restart Control <t< td=""><td>Result Result</td><td></td><td></td><td></td><td>-</td><td></td><td>630945</td><td>+</td><td>75M7754</td><td></td><td></td><td></td><td>Ī</td><td>П</td><td></td></t<>	Result Result				-		630945	+	75M7754				Ī	П	
Name Name Name Name Name 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 <td>Telescol Contrast <th< td=""><td>11</td><td>A75MIOISI</td><td>Ĵ</td><td></td><td></td><td>4309A22MN</td><td>DT5 MONONS</td><td>Т</td><td>SMIDIBO</td><td></td><td></td><td>•</td><td>Т</td><td></td></th<></td>	Telescol Contrast <th< td=""><td>11</td><td>A75MIOISI</td><td>Ĵ</td><td></td><td></td><td>4309A22MN</td><td>DT5 MONONS</td><td>Т</td><td>SMIDIBO</td><td></td><td></td><td>•</td><td>Т</td><td></td></th<>	11	A75MIOISI	Ĵ			4309A22MN	DT5 MONONS	Т	SMIDIBO			•	Т	
Name Name Name Name 1000001 100000 100000 100000 100000 1000001 100000 100000 100000 100000 1000001 100000 100000 100000 100000 1000001 100000 100000 100000 100000 1000001 100000 100000 100000 100000 1000001 100000 100000 100000 100000 1000001 100000 100000 100000 100000 1000001 100000 100000 100000 100000 1000001 100000 100000 100000 100000 1000001 100000 100000 100000 100000 1000001 1000000 1000000 1000000 100000 10000000 1000000 1000000 1000000 1000000 10000000 1000000 1000000 1000000 1000000 10000000 1000000 1000000 1000000 1000000 10000000 1000000 1000000 1000000 1000000 10000000 1000000 1000000 1000000 1000000 10000000 1000000 1000000	Reserve to the state of th	η.	479MI0191						11					1	
Home Home Home Home 1010000000000000000000000000000000000	Image: Product in the image of the imag	17	0 PIOING LA		+		AIAOOTICS	D75M09094		210/144			Ŧ		<u> </u>
Figure 6-6 Cable Interconnect District interconnect	Image: state of the state	171						-		NO LINC		╀	•	T	
Figure 6-6 Cable Interconnect Diagrams Continue Contin Continue Cont	Figure 6-6 Cable Intercont Entrant Ent	24 075 M7755					11	1021- 0466 ONSLO				╞		1	
Result Result Result Result Result 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1	Riversion Autori Insento Autori Insento Autori Insento Autori Insento 1 <t< td=""><td>1075M7756</td><td></td><td></td><td></td><td></td><td></td><td>015M07793 6D</td><td></td><td>3MI0160</td><td></td><td></td><td>4</td><td>1</td><td>-</td></t<>	1075M7756						015M07793 6D		3MI0160			4	1	-
Figure 6-6 Cable Interconnect Diagram Continue Continte Continte Con	Right Right <th< td=""><td>2075M2735</td><td></td><td></td><td></td><td></td><td>T</td><td>1</td><td>OAPSAMP</td><td></td><td></td><td></td><td></td><td></td><td></td></th<>	2075M2735					T	1	OAPSAMP						
Figure 6-6 Cable Interconnect Distants Continue Contin Continue Continue <td>Figure 6-6 Cable Interconnect Disprise Continue<</td> <td>9524W520 04-</td> <td></td> <td></td> <td></td> <td></td> <td></td> <td></td> <td>A8227.007</td> <td></td> <td></td> <td></td> <td></td> <td>1</td> <td>•</td>	Figure 6-6 Cable Interconnect Disprise Continue<	9524W520 04-							A8227.007					1	•
Printing Printing Figure 6-6 Cable Interconnect Diagrams (Continued)	Figure 6-6 Cable Interconnect Dispute filter Continued Continued Figure 6-6 Cable Interconnect Disgrams Continued Continued	Ĩ					6120WI	9- 00160W540						Ţ	
Figure 6-6 Cable Intercomect Diagrams (Continued)	Right Right Right Right Figure 6-6 Cable Interconnect Diagrams (Continued) Right Right	1	APSMIOIBO				200213								FELSTINE.
Figure 6-6 Cable Interconect Displaying Displa	Figure 6-6 Cable Interconnect Diagrams (Continued)	1					112044	+							SPONGA
Figure 6-6 Cable Interconnect Diagrams (Continued)	Figure 6-6 Cable Interconnect Diagrams (Continued) Continued) Continued) Continued) Continued) Figure 6-6 Cable Interconnect Diagrams (Continued) Continued) Continued) Continued) Continued)	1					CN0859		+			╀			
Figure 6-6 Cable Intercomect Diagrams (Continued) Continued) Continued) Figure 6-6 Cable Intercomect Diagrams (Continued) Continued) Continued)	Figure 6-6 Cable Interconnect Diagrams (Continued)	-					e / E O We					$\left \right $		T	CABLE.
Figure 6-6 Cable Interconnect Diagrams Continued	Figure 6-6 Cable Interconnect Diagrams (Continued) Figure 6-6 Cable Interconnect Diagrams (Continued)	:		+ + + + + + + + + + + + + + + + + + + +			ZMORC 9	_							
Figure 6-6 Cable Interconnect Diagrams (Continued)	Figure 6-6 Cable Interconnect Diagrams (Continued)						BMDXc 9	ZI- DOIROWEZD	Ī						
Figure 6-6 Cable Interconnect Diagrams (Continued) Continued)	Figure 6-6 Cable Interconnect Diagrams (Continued)					 						$\left \right $	╉	Ŧ	
Figure 6-6 Cable Interconnect Diagrams (Continued) Continued) Continued	Figure 6-6 Cable Interconnect Diagram Continued Figure 6-6 Cable Interconnect Diagram Continued				+									1	
Figure 6-6 Cable Interconnect Diagrams (Continued)	Figure 6-6 Cable Interconnect Diagrams (Continued)						1174677	THEM AND IN THE A	- 1						•
Figure 6-6 Cable Interconnect Diagrams (Continued)	Figure 6-6 Cable Interconnect Diagrams (Continued)						6722W3	075M00119 3 01	11	MI0200				- T	
Figure 6-6 Cable Interconnect Diagrams (Continued)	Figure 6-6 Cable Interconnect Diagrams (Continued)					1	0172213	075MDBIOA 47 D		MIGLEO				1	
Figure 6-6 Cable Interconnect Diagrams (Continued)	Figure 6-6 Cable Interconnect Diagrams (Continued)					T	5122Wit							ĒŢ	
Figure 6-6 Cable Interconnect Diagrams (Continued) Entropy (Continued)	Figure 6-6 Cable Interconnect Diagrams (Continued)	!	•				1178717							-	
Figure 6-6 Cable Interconnect Diagrams (Continued)	Figure 6-6 Cable Interconnect Diagrams (Continued)						C S S WILL D	1015 R01507154	5M07751 775	4 DIAD				TT	0.
Figure 6-6 Cable Interconnect Diagrams (Continued)	Figure 6-6 Cable Interconnect Diagrams (Continued)														FI.
Figure 6-6 Cable Interconnect Diagrams (Continued)	Figure 6-6 Cable Interconnect Diagrams (Continued)											jji	, I		
Cable Interconnect Diagrams (Continued)	Cable Interconnect Diagrams (Continued)		Ē				i			1				1	
			Ingl'4	-			nect Dia	rams	(Contin	nued)					
			•					5							
												2			
						Figure 6-6	Relation Control 1 1	Relation Control 1 1	File File File File File File File File File File File File File File File File File File File File File File File File File File File File File File File File File File	Relation Comparison State State State </th <th>Figure 6 Cabine 1 Figure 6 Control of the figure 6 Figure 6 Cabine 1 Figure 6 Cabine 1 Figure 6 Cabine 1 Figure 6 Cabine 1 Figure 6 Cabine 1 Figure 6 Cabine 1</th> <th>Ref Ref Ref Ref Ref Ref Ref</th> <th>Ref Ref Ref Ref Ref Ref Ref</th> <th>Risting International Stating International Internation</th> <th>Signal Final Final Final Signa</th>	Figure 6 Cabine 1 Figure 6 Control of the figure 6 Figure 6 Cabine 1 Figure 6 Cabine 1 Figure 6 Cabine 1 Figure 6 Cabine 1 Figure 6 Cabine 1 Figure 6 Cabine 1	Ref Ref Ref Ref Ref Ref Ref	Ref Ref Ref Ref Ref Ref Ref	Risting International Stating International Internation	Signal Final Final Final Signa

Ż

-			•																																																								SONOICE -	112 12 12 12 12	CABLE.									L	·					AAS I TO A
		Ď																																													•	THE REAL PROPERTY I																										E SOALE INTERCON	THE EQUINENT, LL	
	T		П	L	Ţ		E I		Ţ	П	T	T	Π	Ţ	Π	Т	ł	Π		1.	T	П	Π	T	T	T	T	Π	П	Π	7	T	Π	Π	T			1	T	T		T	Ť	Π	Т	1			Т	Т	Т	Т	T	Π	Т	T	П	Т	Ţ	Г	Π	Т	T	П	Т	L	Г		Π	Ţ	Т	П	ור			į
The second second				\$	*							•		7						T																																												27			23	5 7	Ż	2				8		
DWACTOR																																																																												
COMPET TOP																																									2 872																																L 	Ш	L	J
POU CABLE			A- A75M/0(80		29- A75MIDIBO	Т	Π	A75MIOI40		•	-	51- A75M 10160	-	SB- A78 MIDIAD									3	ģ		+		5-		- 15		-		-	CLICINGLE - TUN	8				DELOTWIKY -L	_					+								•					-	4. A7SMIDIBO		04- A73MI0191		0£-	•	•		02-	i							
SCAPLY AG-AND-		14 2N 20	IT DIVING - CELT	-Serionero o- rectionero	- CELICATION - IL CLIMACITO			ALLOWING 1-2551 WSLO	•			AM2-1 DTSMOTTH-		-164.00 1 6- 1	1 -10 1 011	LOWILD 11-2654	-INLUMBLO STOLD		2	\$		Н	12-10-0	╉		T	Т	DICA-I TTAMOTISI-	Н	ADAGTO -ADTSMOTTEL	2				· Ŧ					I DIMOTINI		STIDHEID COKON		ę	\$		65-		56-				-	410 14-		9	Ŧ		ŶĬ	DISMORDIO 45015401154		015M05010 - 14 015M15805-		-100414001-12Kg				4	-1664. 1 1-	ALIWALD S-LENG						
PERSONA CONTRACTOR		(110 56		M410 34 242 4	ONCIO INVERT			MSLO INOT	Г	T	Т	Г	Π	ESELVIE DIDE	П	1	LADAUI DIANC	Н	Н	+	╀	Н	Н	╉	PHONEN A	CAPAGE I	Increase in the second	HOMITINI DISIN		ž	-	C ODAG Z	1	6105W6	1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1	6405A2M1	INEVEDA	174404		Ē		MALLO INIGINI			6406ABBINI	LADLASSAW	LAOLAN SLW	6406A534WI	CADLASSAL	6406A45121	C 40LATSIWI	4406 ATS2 WI	IN ISON AGEN	INTERVIOR	INT TOOL	CADA LOW!	CADELANI	CAOS LOW	CADELEW!	MALENI DISM		MALO INTON	L		CAISWE	1		1	6413W13							
		!	1		П	1	Π	Ţ	Γ	H	T	L	Π	Ţ	Π		4			Ţ	-	Π	\$	ļ	ļ	Ţ	Ţ	L	Π	Ĥ		Ţ		Π	*	F	Á		Ţ	Ţ		1	Į	H	H	1	Γ	П		Ţ	Ţ	Í	F	Π	1	ļ				Ţ	Π	Ţ	T	Ц		Ţ	Ţ	Ţ		Ţ	ļ	μ				
× T		•																																			,																																							
e conneron		: ; ;	!								-	P HEROLENA-15																																																																
CONFC TOR				-								4-WANDA SH																																																+++																
LEMONT AND AND A								_					-Sett		-1811	*	-161-				174- A75MD40		19101W544 -17541	7162 - 475MIDE00	TIGI- APSMIOIOC			+	1			*	794- A76MIOIBO		DTM SO 401.	1345- A75MI0196																						-	11753- A75MIQ100		0175	-962		1		DIDIMONO	TOTAL ATAMINAD			-+++		01754	1754- AZ2MIQ40			
ASSEMBLY BO-A		4410	a = 1		11			ţ	1				1		0	D -	-16119 - 1919-	9	0 N.	22	TOMPTON AND TOMON		-198 CIMELO 2-61180M	0815-4 0	0002-36 0			R 5	0	4	72	-*1.6	HONOT - MUTHU		SWELD 1-0000MLD	1 5-00140																							MALA - IDEN		0 5-10/60			9	9	9-10-60	1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1		1 1 1	5	+ **		BHRID TO FINISH			
Province AS	1	2 Min	2~10	11/12	6323 W24	22W25	12M22	22 M 20	100	1 I M 2	2412		21.55		. 00	19 14	142	1	14P	57	100		110	E My	44.80				100	444	42.46	6324W47	4W49 074	-	LICAI DI	1 CWS	eve	S GAIWZO	SCAINCS	CALCE!	2 GAIVB2	CAN N3	CAV'S	CAIW34	TEWA	ALVOR	AIV40	AIWAI	-AIWAR	CPWIA	AW44	SAW 45	CAIWAG	6326AIW48	CAIWA9	L NIW DO	C AZUZ															63264 35	Ę			

					ÿ																										-													1									E TOMORAGE	11 21 10	CANLE	LINUCA						T	٥	•	CONCOL			
		-	Ŧ			Ŧ		Ŧ		- -	.		.	+			-	T		Ŧ	.	T	Ţ	T	Ŧ		Ŧ	· T	•		ग	ŧ,	Ŧ	ľ I	Ŧ		T				1		1		AN ATTACAL MIN I		ΓŤ	11		ŤŦ		- T -	T					1-1	-1-	ŦŦ			T-T	- T -	FEEDWARD NEDWOOD			
111	Ŀ	1	+	2	Ť	#		+	μ	╡		2	-	1	1	25	**	12			12		μ	4	μ		\parallel		×	╞╢	+	╞┥	1		*	25			╇┥	+	╢		╢	+	1	Fi H	┢┼	╢		⋕	$\downarrow \downarrow$	╪	╟	╡	╟	-5					╢	-	\prod			1		lli
		\dagger	ł	Η	┥	ł		+	H	+		H				┥	╞			+		+		+			Ħ		Ħ		ł		╉		┥	H			Η	Ŧ		Ť	Ħ		₩	+	╟					╋	╞┼		╞┼				+		╂┤		╫					
т		$\left \right $		Ц	+	ļ	μ	╀	Ц	$\left \right $			-		H					+		\downarrow		+				4			+		+	H	+		μ			+	$\left \right $		μ						+	┼┤	┼╿	+		\parallel				H		μ		+	╟	+		ļ	H	
	1																																																																ľ	88	1	Ţ
1	Numbra	ANUMAL AND			+	1				+		A19MMO19	+	ļ	Ħ		+			$\frac{1}{1}$		101		+		ATSWI DLOD	10100	0100	4075				t		+			ATMIPAG							CRIMINER	10160	0910	900	610	010	0910	0100		0100	0.00	200			+			\uparrow					•	(pon
		Т		Ц	+				H	$\frac{1}{1}$	Π		╉				╈			+	┼┥	+		+	H			1	1	П	+	╞┼	╈	╞┤	+	H		-		+					Ħ	Т	1 1		1							IMELY	-		+				$\prod_{i=1}^{n}$					oomoot Namama (Continued)
	110	+			+		4004	D11100				DISKINSIO	╉	+		+	+	Ħ		+	╞┥			+			DTSKI 19270		39.6	┝╉	+	╞┼		╞┤	+			STITEMENT OF	19 <u>1</u>		111				12110M5/0	075M112 09	DISMO7751	DISMOTO	D75MILE 19	15170M277	DT3MOT731	DTSWIELDS	DAWI TZUS	CITINS/G	15/CONSED	DISMOTING												
The state of the s	1490		12:35			1	ų	00.1100		T		133-010601		2	ş	Ż	ų.	č	4		Į	2	ş	ę ș	111-	111-01040	ALCODED	Ę	LL-	92	9	F		ā		NA I		5-HORON	-		÷ No	9010-01			DISMOSCIO-N.	31-02060 M		1. 1. 1. 1. 1. 1. 1. 1. 1. 1. 1. 1. 1. 1	N .	N.	- 10	<u>.</u>			1				Ť			Ť						
	10					+			H	+		4510 11	+				+		9	+		+		=		Had on	11 D19M								+++**			10							15. 101.1	PHO 2	Π				2121	2				LOSOMSIC BIN	+		+		H	+	μ					202
			3	3			INANA			6449Wi		11659	11657	11653	THE P	11110	11652	11653	11059		6591		1053	571653		1169	65422		1659	F		í.	65961	1362.2	16385	A SWART		ALL DOOD	66035	605	1033	1027	1003	1000	1405W 2	13023	1000	- COL	1097	101	149039	1000	19091	1000	6604	6606N10												
			1		1				5		F	Π	+		÷	1	 	Π	2		Ħ	7			Π	2	Π	4	1	Π		ļ	-	Π		2			11	T		5	H	Ţ	H	2	Π	Π	I.	2	2		-		2	+4/62			: .	-	2	T						
		7 	:		Ī	ſ		T					Ì				T					T	Π				Π						T	Π			ſ	I		T	T	T				T						T	Ħ		I					I	Π	T						wothin Taton
	-	İ		1	1			╀		+	H		+				+	H		+		+	$\ $	+	H	H	$\ $		\parallel	Ц	+		+	H	+			μ			\prod		Η			╉				$\prod_{i=1}^{i}$	+	+	┞	$\left \right $	H				+	H	╢			+			1	
		•	•		ļ	:							Ī							T		T		T		I					T		ļ		T															Π			Π	Π	T	T												The second of G
	-			1	1	╞		∔-	H		8				9		-				8		9			╟╋	H	4	$\left \right $		+		╀		╀			╎╎	$\left \right $	H	+	H	Η			+	╟			╢	$\frac{1}{1}$	+	╟	╢					+				╢					
					-	Ļ			AT3MIO					7.42.0	APSN10	ð	DAIG IMMA		ATNI O	*	80	VINUS	ABMIO	ATMU DIZ							\downarrow			Π																		 					; ; ;	• •	!						İ			•
	14.604			-	Ş i	In 1991		17411644	3MILLER	NI 1240	111101	121	1. 1.1	JMO LLJI	1611046	MIRM	Suits?		MILEN	SMI LEO	0423W3 1 7012 4 0740713	ALL THAT	121704					MILEN	DISMILL DO		14/12 00	DISUTIES	DTSM112 CA		DISMILEM	ICOL IN			1945	1004	M14004	1001	1 1			1000	MI 4004	M14904		INCLO	DI3MILEON	MI12.08	DISULUTED.		07544112.66	DISMILLA	DISNI LEAD		INVIO	141 141	I.L.M	;						
				Ţ		a series			10 21-21.04	10 40 40 40	10 51	-	1 - 2 - 1	10 0-310	10 1-3101	14 D1	CATTAL 10100001		101.1.101	0001100	DIE 4 DT	10 5-110	012 4 015	CHINELO DI SIGNATO DI SI CONTRA CONTR				DISKO BIST & DISKI			D75MO 9117-11 075M11266	10 11-1516			111-14 D1	ISSUNTE SI-TER DARTO			01		210 12-1610	10 10 10	1	1	12	\$ 9	10 11.11	PTIME DIST. 14 DISMISSO	164										D75MO 913617 D7	10 11.10	-176-20 DT	-	L i	+-				
	-		~				- г		OW A	+			1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1	DWEIN -	DISMO		DIAMO		DING			DIBMO	D19MO	P19MO				T	01-12100121-10	11	-	11-1516 DMLD	VI-LUISONNIC	_	D79408137-14	DIMU	Ţ				D75MON	Pare 10			92					DISMO	C-BEIPONETO	-	Ti	Π	DTSMC 9130-12	DTSMO9116-13	D75MO913A-M	1	01540	CINNO	01210	, 			•			
	1							A 2 1 5 4	ALC W	1000	SUCUS	AIGWG	641647		1 M 1 1 1	AWILL			146344	12 3 W.	12343	12 3WG	1424WI	5 m 62 4		C SCWI		24145	6432W2		111W2	TASAVE	11111		6436WZ	1.01.01	122 M	6437WA	2222	Inite	71.5		37914		CA16W3	436K6	43044	543543	10044	64 10 MS	641 W.C	SUSPAS	1.44.44.1		6444W2	643WZ	LABLUZ		C447W2	4114	11111							

の時代を

			•																																																	E73MOR93	1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1	INDEX							L	0		
145																																						-																										TARE NERONACI
	1	гт	T		T1		Ť	TT	T	Т	T1		7	T T	Т		-	ń		1	П	1			Π		•	ГТ	T	7			Т	-	ΓŤ	T	-	11	Ŧ	гт	T					İT	Ŧ	ГT	T	гт	11	T	17		T	T	- T -	П	T		T		TI	
JAN IS UNIVERSITY					44		*				34							-3					۹ ۲			*				-\$							1								8			4	*					33			11							
DANK TOR																																																																
COMMETTER .													-											Ţ					F										- -																									
Sau Curr			9-1											5												, ,		<u></u>																																				
A RE-AUG			45 DTSMI5495-8	66EINS20 57				-26 075415495	27		-10			-14 DISHIE495	5 DISH1399	 		1			14 D75M3992	35 07541543		20	40-1			44 DISMIS495	411-10 5	1		02	21	23 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1													+		-															
A TRANSFE	 [D75M14067-45				╇						+		+																			D75M1404534																														
PENGWATT	6422M	662249	6612WIO	6622WIG	6422M10	61ME299	6622W20	6622W23	66422M23	3CM1 C77	4612W26	6622W37	66221W28	6632830	6623W31	6632W38	FENTERS	9EM2299	16W1639	6412M3	6622W40	6622W4	6622 W4	6422W4	9442239	6432W47	6622W4	6423W-	5622W	0612 W. J	6622W5	6623W 56	6622W 50	6622W [21		6613177	66134/9	1214249	66231 24	66131/26	66231623	EA16299	6623 130 6623 132	((1 (7))	167 179 -	6623 N 30	6623 M39	+ 6623 W41	6623W42	662 J 1/44	662 W47	667: W48	15W 534	662 1W5	467 3W54	662 1W36	15M6299							
757							-2								2	=						-			ļ	3	F						Ī		ł			ļ		E	1		$\frac{1}{1}$		-		2		Ī	Π		$\left \right $		$\frac{1}{1}$	ļ	5			Ţ			Π	\prod	
		!										•																																																				
COMPETER	_								T						T																					Ī					Ī																			-				
DWAR TOA				1																																																												-
T CALL																																																						İ				4						
м,	- CARELAN-						1 07544 1993		246EIWSLO 1						266614540	D75413992				076412003		26961M270				249EIM270	D164412402								2 075M12992	D75411992			5 5 5	16461W540 L	1 D75M13992				7 D15M13992		D75M09801-1 D75M13992							+		246EIW510	26481454	1	+		744519510			
N ASSEMBLY				 •		; • 	01-14-10		- 16190M210	- - -	-7 			+	1-96260W510	1-16180W510				010100000	1 12/2/02/1	075M0779-1		•		D75M09798-7	1-00104140-1	f		+		Η	╀		E1- 641404310	D76M06794 - 1			5. 5.	-225M09794-	D75M09600-						075M09801-		•••		\dagger		t	77		D75M09801 -15							 	
NOLIMIES	1.1.1.4		Serjawiz Lecijawiz	Birl Juri	661 1W 21	6613N33	6613W41		6614W4	+114 MG	6014W6	6614WII	6LIAWIZ	6614W35	6614 W42	6615 W4	545179	6415WG	SEMS179	6615W41					6414W35			6417W5	DLI TAK	PEI 7W7	PLUT I	11 1 127	6617W20	6 CT 1 40	6617m42	441BWA	54197	1149179	6618W35	6418W42	2119112	6119N13	6019W19	6419W21	6019 M41		6610W4	64.70WG	6630WH	6+20W14	6620W27	6620W28	66.20W33	16 400 19	46 20W41	16 10W42	66.11W4	6. 21 W 6	4621W11	6621W41	+621W42			

_

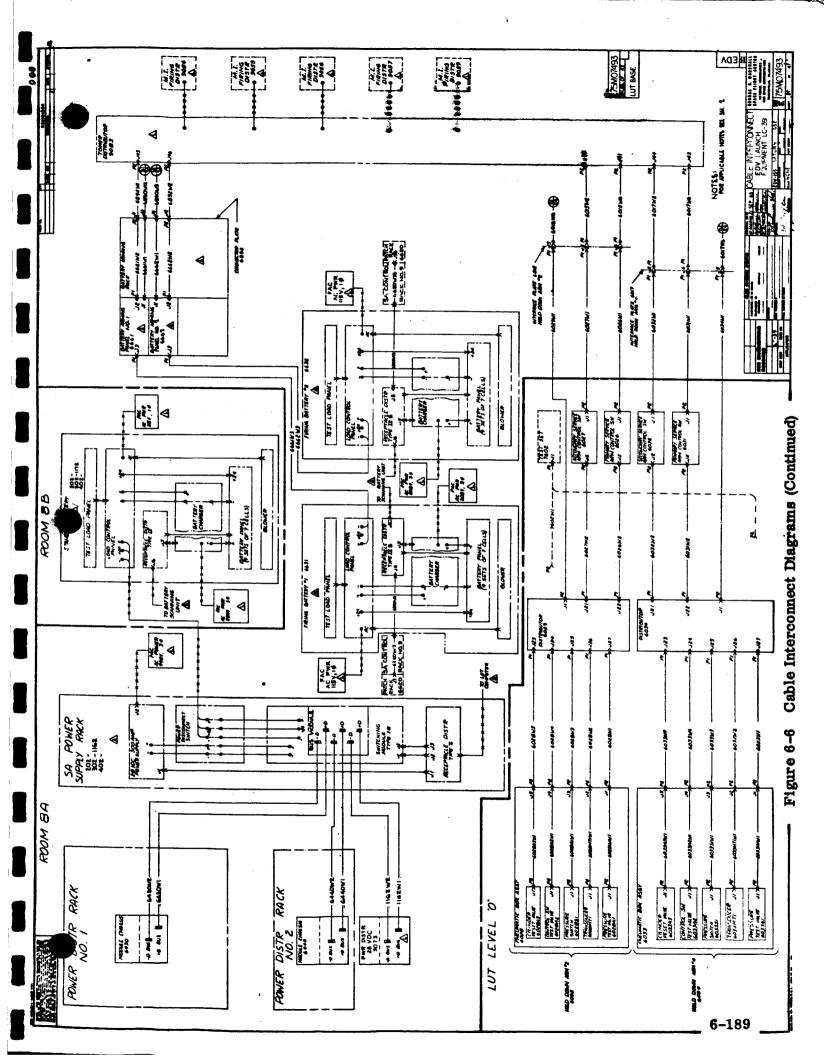
New level New l																	· · ·			A NUMBER OF TAXABLE PARTY.	
				(NOW	CAN'S	CARE	1		Z	1000	1	Nr.]	- INNO	CANE.			2				
													1- LOWING								
				at the t	B. SALALAN J.							100 100 100 100 100 100 100 100 100 100									
Partial international inter										i	r	12.41000									
Province Provi								<u> </u>					X								
Printing Prin Prin Printin			Musical barrer Music	125			1					+	╀	.†.					28		
Partial intermentation P		Pictor of the pictor Pictor of the pictor Pictor of the pictor Pictor of the pictor Pictor of the pictor Pictor of the pictor Pictor of the pictor Pictor of the pictor Pictor of the pictor Pictor of the pictor Pictor of the pictor Pictor of the pictor Pictor of the pictor Pictor of the pictor Pictor of the pictor Pictor of the pictor Pictor of the pictor Pictor of the pictor Pictor Pictor of the pictor		tellue	Dimonin.						П	K SIVK	$\left \right $						П		
				EVANT.	OTHORNE B	D'SMINDI-					Т			- 					Ŧ		1
											П	SEWI4	╟								
Terrer Terrer											Т	101 M 10	$\frac{1}{1}$	•					261		**
Tight bill Tight bill <td></td> <td></td> <td></td> <th>6440MI</th> <td>N OLUCOWILD</td> <td>-CHLOWID</td> <td></td> <td></td> <td></td> <td></td> <td></td> <td>IFWIC 5</td> <td>5.</td> <td>•</td> <td></td> <td></td> <td></td> <td></td> <td>36</td> <td></td> <td>-</td>				6440MI	N OLUCOWILD	-CHLOWID						IFWIC 5	5.	•					36		-
Total of the structure in				LC40WE	21-04-050	1991					-	A SINAL		-TAALINAL					26		
				51013	0.000	01159-	ī					П	DI-18680WSL								
Image: second				14044		.					T	Т			V/Dm/wtoo		Ť				
Image: second				6640M6							Π	П									
Image: second				6 COWT			101010	+			T			-ICCLINE	178 MID 200				21		
				664049	18.4	1995	SMIDIME				Γ	1	0 9-90 10-00 L	-1866.W4					21		<u></u>
Plane Plane Plane Plane Plane Plane Plane Plane Plane Plane Plane <td< td=""><td></td><td></td><td></td><th>6140410</th><td>*</td><td>5</td><td>34(019)</td><td></td><td></td><td></td><td></td><td></td><td></td><td></td><td></td><td></td><td></td><td></td><td></td><td></td><td><u>.</u></td></td<>				6140410	*	5	34(019)														<u>.</u>
Image: interment intermen	Figure 6. Contrast Cont			6640WH			5MI0192			Ŧ	<u>_</u>	Manal Manal	146-01060WBL	- HARRING							
Hart Arrest freeded Hart Arrest Arrest freeded Hart Arrest f				6640W13			SMIOL 91		┟		Γ	14599	6	-04211					4		<u>.</u>
Image: state in the state			Plane Plane Plane Plane Plane Plane Plane Plane Plane Plane Plane <td< td=""><th>LC AOWIA</th><td></td><td></td><td>1 BIOINS</td><td></td><td></td><td></td><td></td><td>+ 48 W 4</td><td>K-</td><td>-9821</td><td></td><td></td><td></td><td></td><td>44/</td><td></td><td>.</td></td<>	LC AOWIA			1 BIOINS					+ 48 W 4	K -	-9821					44/		.
The strength of the strength of		Bits Bits		66 40W18			1610IW S				Ĩ	2w2444	-112	11280					/46		
	Rate of a line Image of a line I		Bits Bits	Inc Could			SMIOI92				T		X :	-PGEL					Ţ		H.J
The state of the state of	Plane de la plane Pl			1040M25				+			Ţ	1000 M2		11280-					Ţ		<u></u>
	Figure 6. Contrast <t< td=""><td></td><td></td><th>100 100 100 100 100 100 100 100 100 100</th><td>Ţ</td><td></td><td></td><td></td><td></td><td></td><td></td><td>CARANT C</td><td></td><td>111-0-</td><td></td><td></td><td>Ì</td><td></td><td><u></u></td><td></td><td><u></u></td></t<>			100 100 100 100 100 100 100 100 100 100	Ţ							CARANT C		111-0-			Ì		<u></u>		<u></u>
	Image: 1 Image: 1 <td< td=""><td></td><td></td><th>640W28</th><td></td><td>1394</td><td>75M0192</td><td></td><td></td><td></td><td></td><td>1145335</td><td>3</td><td>-9665</td><td></td><td></td><td></td><td></td><td></td><td></td><td>.2.</td></td<>			640W28		1394	75M0192					1145335	3	-9665							.2.
Image: state in the state				62/104.99	L	A -41961	SNICIO3					1-	Q -	11280-							
	Rute: Rute: <th< td=""><td></td><td></td><th>0E-MOP-9-9</th><td></td><td>1.946</td><td>75MIOI93</td><td></td><td>·</td><td>•</td><td></td><td>INS 43</td><td>617</td><td>11260-</td><td></td><td></td><td></td><td></td><td></td><td></td><td></td></th<>			0E-MOP-9-9		1.946	75MIOI93		·	•		INS 43	617	11260-							
	Image: Sector of the sector			4 CHOM SI		10940- 4	75MI0193			2		14 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1	1. 1						6		
	Mathematical Mathematical <th< td=""><td></td><td></td><th>2.0000</th><td></td><td></td><td></td><td></td><td></td><td></td><td></td><td></td><td>DING THE OWNER</td><td>- ACCERT</td><td></td><td></td><td>t</td><td></td><td>Ŧ</td><td></td><td></td></th<>			2.0000									DING THE OWNER	- ACCERT			t		Ŧ		
	Mathematical interpretation in the interpretatint in the interpretation in the interpretation in the			Le toway		-	I A JOI WC				-Ŀ		COLORADO STATE				Ť		Ţ		6
Fight	Pietro II Pietro II Pietro II Pietro II Pietro II Pietro II Pietro II Pietro II Pietro IIIIIIIIIIIIIIIIIIIIIIIIIIIIIIIIIIII	Reference Reference		10 MON 01		- 1 M					Ţ			-01210			Ì				
Figure 1 1<	Telephone Image: second se	Image: image:		60404103			-				Ē	CCCW EY	14				Ì		27		
The stand structure The stand structure<	Start Start <th< td=""><td>Image: second</td><td></td><th>6640M104</th><td>2</td><td></td><td></td><td></td><td></td><td></td><td></td><td>1 1 1 1 1</td><td>2</td><td></td><td></td><td></td><td></td><td></td><td>16/16</td><td></td><td></td></th<>	Image: second		6640M104	2							1 1 1 1 1	2						16/16		
Picture Picture	Figure 6. Contract of the second	Filter Filter		101100	9 4 -	•	i					· + ~ ~ .	4 11-	•					46		
The start of the start of	Image: 1 Image: 1 <th< td=""><td>Print Hall Print Hall<td></td><th>44 AQUIDE</th><td>-W-</td><td>-</td><td></td><td></td><td></td><td></td><td></td><td></td><td>Ť</td><td>11280-</td><td></td><td></td><td></td><td></td><td></td><td></td><td>•</td></td></th<>	Print Hall Print Hall <td></td> <th>44 AQUIDE</th> <td>-W-</td> <td>-</td> <td></td> <td></td> <td></td> <td></td> <td></td> <td></td> <td>Ť</td> <td>11280-</td> <td></td> <td></td> <td></td> <td></td> <td></td> <td></td> <td>•</td>		44 AQUIDE	-W-	-							Ť	11280-							•
Transmission Transmission <th< td=""><td></td><td></td><td></td><th>FOIL WORLDT.</th><td>IN:</td><td></td><td></td><td></td><td></td><td></td><td></td><td></td><td>2</td><td>- 4</td><td></td><td></td><td></td><td></td><td></td><td></td><td></td></th<>				FOIL WORLDT.	IN:								2	- 4							
The form The form <th< td=""><td>Tight of the stress of the</td><td>The state of the state of</td><td></td><th>00M0099</th><td>1 1 35</td><td></td><td></td><td></td><td></td><td></td><td>Ţ</td><td>POCEN -</td><td></td><td></td><td></td><td>- 1</td><td></td><td></td><td></td><td></td><td></td></th<>	Tight of the stress of the	The state of the state of		00M0099	1 1 35						Ţ	POCEN -				- 1					
Flore 6-0 Control	Protocol Protocol <th< td=""><td>Protection Protection<td></td><th></th><td>A DISCOURSE</td><td>-291 1044</td><td></td><td></td><td></td><td>•</td><td>Ţ</td><td></td><td></td><td></td><td>-</td><td></td><td></td><td></td><td></td><td></td><td></td></td></th<>	Protection Protection <td></td> <th></th> <td>A DISCOURSE</td> <td>-291 1044</td> <td></td> <td></td> <td></td> <td>•</td> <td>Ţ</td> <td></td> <td></td> <td></td> <td>-</td> <td></td> <td></td> <td></td> <td></td> <td></td> <td></td>			A DISCOURSE	-291 1044				•	Ţ				-						
Bits if Constraint	Bottom Control Contro Control Control	Thrust Thrust	Home Home	641W1	DISMOSOTO-SM	HOTTS9-	12 MIOIBI				10	. C. C. 10	9 . .						26		
Tight of the second	Team Team	Stress Attribute	Result Result	6641W2	14-01060	-65110	18IOIWS					(CON I	-114						26/46		
Tight of the stand Tight of	Figure 6- Contrained <td>The state The state</td> <td>Provinci <th< td=""><th>66412</th><td>8-14160</td><td>-16661</td><td></td><td></td><td></td><td></td><td>2</td><td>6666 13</td><td>070</td><td>•</td><td></td><td></td><td></td><td></td><td>27</td><td></td><td></td></th<></td>	The state The state	Provinci Provinci <th< td=""><th>66412</th><td>8-14160</td><td>-16661</td><td></td><td></td><td></td><td></td><td>2</td><td>6666 13</td><td>070</td><td>•</td><td></td><td></td><td></td><td></td><td>27</td><td></td><td></td></th<>	66412	8-14160	-16661					2	6666 13	070	•					27		
Right of the strength private interval of the strength prive strength private interval of the strengt	Tight of the strength of the strengt of the strength of the strength of the str	The state The state	Res R	CCAIN B	EN-01060	5	SMIDI91					646×14							27		
Tight of the strength of the strengt of the strength of the strength of the str	Tight of the strength of the st	Right of the strength of the st		SCAWA	<u></u>	-]							-	26	:	
The state is a strain between the strai	Tight best best best best best best best bes	Night Streich Night Streich<	Image: state in the state	6441M2	2						Ţ			1180-					12		
House House <th< td=""><td>Tight 6 Contract of the state of the</td><td>Image: Second second</td><td>Bit B</td><th>664 N 6</th><td>5</td><td>41.6</td><td>Intoiws</td><td>ł</td><td></td><td></td><td>1</td><td></td><td></td><td></td><td></td><td></td><td></td><td></td><td></td><td></td><td>-</td></th<>	Tight 6 Contract of the state of the	Image: Second second	Bit B	664 N 6	5	41.6	Intoiws	ł			1										-
Figure 6-6 Cable Intercent Diagram Contrant Contra	Tight bill Tight bill <td>Image: state of the state</td> <td>Image: state of the state</td> <th>541W7</th> <td></td> <td>-24681 +</td> <td>SMICIPE</td> <td></td> <td></td> <td></td> <td><u> </u></td> <td>D 02 7 99 9</td> <td>A MONOLOGINA</td> <td></td> <td>TZENINEZ</td> <td></td> <td>ļ</td> <td></td> <td></td> <td></td> <td></td>	Image: state of the state	Image: state of the state	541W7		-24681 +	SMICIPE				<u> </u>	D 02 7 99 9	A MONOLOGINA		TZENINEZ		ļ				
Figure 6-6 Continued Cont	Tight Tight <th< td=""><td>Image: Second second</td><td>Bits Bits</td><th>6491M</th><td></td><td>11222</td><td>2610145</td><td></td><td></td><td>$\frac{1}{1}$</td><td>ŗ Ţ</td><td></td><td></td><td>-</td><td>TANK OF A</td><td></td><td></td><td></td><td></td><td></td><td></td></th<>	Image: Second second	Bits Bits	6491M		11222	2610145			$\frac{1}{1}$	ŗ Ţ			-	TANK OF A						
Filter Filter Filter Filter Filter Filter Filter 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 <td< td=""><td>Figure 6 Inter 7</td><td>Image: District of the state of the sta</td><td>Tight of the strength of the st</td><th>C M M S</th><td></td><td>-2002</td><td>Selons.</td><td></td><td></td><td></td><td><u>я</u></td><td>-</td><td></td><td>+</td><td>16101 #422</td><td>ŀ</td><td></td><td></td><td>27</td><td></td><td>-</td></td<>	Figure 6 Inter 7	Image: District of the state of the sta	Tight of the strength of the st	C M M S		-2002	Selons.				<u>я</u>	-		+	16101 #422	ŀ			27		-
Figure 6-6 Cable Interconnection Continue	Figure 6-6 Continue Conten Conten Conten Con	Figure 6-6 Contract interaction Contract interacion Contraction Contract interactio	Figure 6-6 Contract interaction Contract interacion Contraction Contract interactio	GEALWID			1010WE				 _	+		T					T		
Image: Non- Image: Non-	Figure 6-6 Collegiant <td>Figure 6-6 Cable Interconnect Diagram Construction<td>Figure 6-6 Contrained<th>6641WII</th><td></td><td>2</td><td>1610/WS</td><td></td><td></td><td></td><td></td><td></td><td>A RE- DENOVAL</td><td>T</td><td>3200182</td><td></td><td></td><td></td><td>T</td><td></td><td></td></td></td>	Figure 6-6 Cable Interconnect Diagram Construction <td>Figure 6-6 Contrained<th>6641WII</th><td></td><td>2</td><td>1610/WS</td><td></td><td></td><td></td><td></td><td></td><td>A RE- DENOVAL</td><td>T</td><td>3200182</td><td></td><td></td><td></td><td>T</td><td></td><td></td></td>	Figure 6-6 Contrained <th>6641WII</th> <td></td> <td>2</td> <td>1610/WS</td> <td></td> <td></td> <td></td> <td></td> <td></td> <td>A RE- DENOVAL</td> <td>T</td> <td>3200182</td> <td></td> <td></td> <td></td> <td>T</td> <td></td> <td></td>	6641WII		2	1610/WS						A RE- DENOVAL	T	3200182				T		
Tight of the stress of the	Image: state in the state	Image: Non- Image: Non-		6641MIZ			Smith P									Ť			:		
Figure 6-6 Cable Interconnect Diagrams (Continued)	Right State <th< td=""><td>The state State</td><td>Figure 6-6 Continue <</td><th>EINIPAP</th><td></td><td></td><td>16ioms</td><td></td><td></td><td></td><td>Ţ</td><td></td><td>a - anticu</td><td>THOMAS -</td><td></td><td></td><td></td><td></td><td>THE ANALYMENT AN</td><td></td><td></td></th<>	The state State	Figure 6-6 Continue <	EINIPAP			16ioms				Ţ		a - anticu	THOMAS -					THE ANALYMENT AN		
Figure 8-6 Continue Continue Continue Continue Continue Figure 8-6 Continue Continue Continue Continue Continue	Image: state of the state o	Tight Tight <th< td=""><td>Tight of the state of the</td><th>6-41 MIA</th><td></td><td>01150</td><td></td><td></td><td></td><td></td><td></td><td>100</td><td></td><td></td><td></td><td></td><td></td><td></td><td>-</td><td></td><td></td></th<>	Tight of the state of the	6-41 MIA		01150						100							-		
Figure 6-6 Continue <	Image: state	Image: constraint of the state of the st	Image: state in the state				EMIDIAL									t			Ţ	-	
Hart Hart Print P	Harding Harding Harding Harding Harding Introfteet Freedom Introfteet Freedom Introfteet Freedom Introfteet Freedom Introfteet Freedom Introfteet Freedom Introfteet Freedom Introfteet Freedom Introfteet Freedom Introfteet Freedom Introfteet Freedom Introfteet Freedom Introfteet Freedom Introfteet Freedom Introfteet Freedom Introfteet Freedom Introfteet Freedom Introfteet Freedom Introfteet Freedom Introfteet Freedom Introfteet Freedom Introfteet Freedom Introfteet Freedom Introfteet Freedom Introfteet Freedom Introfteet Freedom Introfteet Freedom Introfteet Freedom Introfteet Freedom Introfteet Freedom Introfteet Freedom Introfteet Freedom Introfteet Freedom Introfteet Freedom Introfteet Freedom Introfteet Freedom Introfteet Freedom Introfteet Freedom Introfteet Freedom Introfteet Freedom Introfteet Freedom Introfteet Freedom Introfteet Freedom Introfteet Freedom Introfteet Freedom Introfteet Freedom Introfteet Freedom Introfteet Freedom Introfteet Freedom Introfteet Freedom Introfteet Freedom	Thrust in the second	Figure 6-6 Cable Interconnect Display Continue Display <thdisplay< th=""> Display <thd< td=""><th>66 1W 24</th><td></td><td></td><td>I I I I I I I I I I I I I I I I I I I</td><td></td><td></td><td></td><td></td><td></td><td></td><td> </td><td>t</td><td></td><td></td><td></td><td>Ţ</td><td></td><td></td></thd<></thdisplay<>	66 1W 24			I I I I I I I I I I I I I I I I I I I							 	t				Ţ		
Figure 6-6 Cable Interconnect Diagram Contrant Con	Printerio in Priorite: Childrene in Priorite: Childrene in Priorite: Childrene in Priorite: Priorite: 1 1 1 1 Priorite: 1 1 1 1 Priorite: 1 1 1 1 Priorite: 1 1 1 1 Priorite: 1 1 1 1 Priorite: 1 1 1 1 Priorite: 1 1 1 1 Priorite: 1 1 1 1 Priorite: 1 1 1 1 Priorite: 1 1 1 1 Priorite: 1 1 1 1 Priorite: 1 1 1 1 Priorite: 1 1 1 1 Priorite: 1 1 1 1 Priorite: 1 1 1 1 Priorite: 1 1 1 1 Priorite: 1 1 1 1 Priorite: 1 1 1 1 Priorite: 1 1 1 1 Priorite	Printerior in fractions Printerior in fractions Printerior in fractions Interface is fractions Interface is fractions Interface is fractions Interface is fractions Interface is fractions Interface is fractions Interface is fractions Interface is fractions Interface is fractions Interface is fractions Interface is fractions Interface is fractions Interface is fractions Interface is fractions Interface is fractions Interface is fractions Interface is fractions Interface is fractions Interface is fractions Interface is fractions Interface is fractions Interface is fractions Interface is fractions Interface is fractions Interface is fractions Interface is fractions Interface is fractions Interface is fractions Interface is fractions Interface is fractions Interface is fractions Interface is fractions Interface is fractions Interface is fractions Interface is fractions Interface is fractions Interface is fractions Interface is fractions Interface is fractions Interface is fractions Interface is fractions Interface is fractions Interface is fractions Interface is fractions Interface is fractions Interface is fractions Interface is fractions	Priority in the function Priority in the function Priority in the function Priority in the functin Priority in the function <th>66414 102</th> <td></td> <td>1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1</td> <td>5MI0 200</td> <td></td> <td></td> <td></td> <td>T</td> <td></td> <td></td> <td></td> <td></td> <td></td> <td>Ì</td> <td></td> <td>T</td> <td></td> <td></td>	66414 102		1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1	5MI0 200				T						Ì		T		
Figure 6-6 Continues Present<	Protocol Protocol <th< td=""><td>Rise Rise</td><td>Mindless Frankline Prime</td><th>6641WIOB</th><td></td><td>PHMOTTR2- A</td><td>9MI0 200</td><td></td><td></td><td></td><td></td><td>TURNIT</td><td></td><td></td><td></td><td></td><td></td><td></td><td>Ţ</td><td></td><td></td></th<>	Rise Rise	Mindless Frankline Prime	6641WIOB		PHMOTTR2- A	9MI0 200					TURNIT							Ţ		
Triputores Triputores <td>Figure 6-6 Contractors Contr</td> <td>Privation Privation Privation</td> <td>Ministrict Ministrin Ministrict Ministrict<th></th><td></td><td></td><td></td><td></td><td></td><td></td><td></td><td>8119QL</td><td>4-</td><td></td><td></td><td></td><td></td><td></td><td>ŀ</td><td></td><td></td></td>	Figure 6-6 Contractors Contr	Privation Privation Privation	Ministrict Ministrin Ministrict Ministrict <th></th> <td></td> <td></td> <td></td> <td></td> <td></td> <td></td> <td></td> <td>8119QL</td> <td>4-</td> <td></td> <td></td> <td></td> <td></td> <td></td> <td>ŀ</td> <td></td> <td></td>									8119QL	4-						ŀ		
Privation Privation	Privation Priva	Priversion of the private strategy of the private stra	Private Privat Private Private	5642W3		-ILLEIMEL			-	-	-	100/A	ž	•							
Tigure 6-6 Cable Interconnect Diagrams (Continued) Continued (Continued) Continued (Continued) Continued (Continued) Tigure 6-6 Cable Interconnect Diagrams (Continued) Continued) Continued (Continued) Continued (Continued) Continued (Continued)	Figure 6-6 Cable Interconnect Diagrams Continued	Figure 6-6 Cable Interconnect Diagrams (Continued)	Figure 6-6 Cable Interconnect Diagram Continued Figure 6-6 Cable Interconnect Diagram Continued			+			+	Ī	Ţ	100/100	-			-					
Figure 6-6 Cable Interconnect Diagrams (Continued)	Privatora i Distribution Privatora Distribution 1 1 1 1	Figure 6-6 Cable Interconnect Diagrams (Continued)	Figure 6-6 Cable Interconnect Diagram Continued Continued Continued 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1	24643		-1226 144617					Ţ	TOB/ N				-					
Figure 6-6 Cable Interconnect Diagrams (Continued) Figure 6-6 Cable Interconnect Diagrams (Continued)	Figure 6-6 Cable Interconnect Diagrams Continued Continued Continued Figure 6-6 Cable Interconnect Diagrams Continued Continued Continued	Figure 6-6 Cable Interconnect Diagrams (Continued) Figure 6-6 Cable Interconnect Diagrams (Continued)	F F	SMOK9	013MI0606-5	-2866IWSLC	-				<u> </u>	0 411/00		SHORE A							,
Figure 6-6 Cable Interconnect Diagrams (Continued)	Rest view Access	Figure 6-6 Cable Interconnect Diagrams (Continued)	Figure 6-6 Cable Interconnect Diagrams (Continued)	GUDOWS	ڊ	••															
Figure 6-6 Cable Interconnect Diagrams (Continued)	Figure 6-6 Cable Interconnect Diagrams (Continued)	Figure 6-6 Cable Interconnect Diagrams (Continued)	Figure 6-6 Cable Interconnect Diagram Continued Figure 6-6 Cable Interconnect Diagrams Continued	5 WOLD		•					4	DO NAME OF		1 - ANOMAI					2		
Figure 6-6 Cable Interconnect Dagrams (Continued)	Figure 6-6 Cable Interconnect Diagrams (Continued)	Figure 6-6 Cable Interconnect Diagrams (Continued)	Figure 6-6 Cable Interconnect Diagrams (Continued)	51×0499	-	+						1021 JML	_						Ţ		
Figure 6-6 Cable Interconnect Diagrams (Continued)	Figure 6-6 Cable Interconnect Diagrams (Continued)	Figure 6-6 Cable Interconnect Diagrams (Continued)	Figure 6-6 Cable Interconnect Diagrams (Continued)			+				ł				.†.							
Figure 6-6 Cable Interconnect Diagrams (Continued)	Figure 6-6 Cable Interconnect Diagrams (Continued)	Figure 6-6 Continued Figure 6-8 Cable Interconnect Diagrams (Continued)	Figure 6-6 Cable Interconnect Diagrams (Continued)			÷						INC. COL							I	-	
Figure 6-6 Cable Interconnect Diagrams (Continued)	Image: state of the state o	Figure 6-6 Cable Interconnect Diagrams (Continued)	Figure 6-6 Cable Interconnect Diagrams (Continued) Figure 6-6 Cable Interconnect Diagrams (Continued)	12/10/33	_	-	1	. 1				INDUCOL	a	•							-
Primaria Primaria Primaria Primaria Primaria Primaria Primaria Primaria Primaria Primaria Primaria Primaria Primaria Primaria Primaria Primaria Primaria Primaria Primaria Primaria Primaria Primaria Primaria Primaria Primaria Primaria Primaria Primaria Primaria Primaria Primaria Primaria Primaria Primaria Primaria Primaria Primaria Primaria Primaria Primaria Primaria Primaria Primaria Primaria Primaria Primaria Primaria Primaria Primaria Primaria Primaria Primaria Primaria Primaria Primaria Primaria Primaria Primaria Primaria Primaria Primaria Primaria Primaria Primaria Primaria Primaria Primaria Primaria Primaria Primaria Primaria Primaria Primaria Primaria Primaria Primaria Primaria Primaria Primaria Primari Primaria Primaria P	Private 1 1 1 Privati 1 1 1 Priv	Printer 10 Printer 10	Primate i 1 1 1 Primate i 1 <td< td=""><th>SEMOS75</th><td></td><td>•</td><td></td><td>ł</td><td>•</td><td></td><td>J</td><td>INLIGO</td><td></td><td></td><td></td><td></td><td></td><td></td><td></td><td></td><td>•</td></td<>	SEMOS75		•		ł	•		J	INLIGO									•
Figure 6-6 Cable Interconnect Diagrams (Continued)	Figure 6-6 Cable Interconnect Diagrams (Continued)	Private Private Private Private Private Private	Figure 6-6 Cable Interconnect Diagrams (Continued)	66200 37	i	+		ļ		+	9 T	1201		THOMAS							
Private de la Cable Interconnect Diagrams (Continued)	Opwinke Image: Construct in the second s	Private a low with the second	Tigure 6-6 Cable Interconnect Diagrams (Continued)			-				ł	T	ł									-
Figure 6-6 Cable Interconnect Diagrams (Continued)	Figure 6-6 Cable Interconnect Diagrams (Continued)	Figure 6-6 Cable Interconnect Diagrams (Continued)	Figure 6-6 Cable Interconnect Diagrams (Continued)	T LINNAL	01- 40401MGL0		-		-	-	Ţ			-					Ţ		-
Figure 6-6 Cable Interconnect Diagrams (Continued)	Privation 11	Figure 6-6 Cable Interconnect Diagrams (Continued)	Figure 6-6 Cable Interconnect Diagrams (Continued)									Η									
Figure 6-6 Cable Interconnect Diagrams (Continued)	Figure 6-6 Cable Interconnect Diagrams (Continued)	Figure 6-6 Cable Interconnect Diagrams (Continued)	Figure 6-6 Cable Interconnect Diagrams (Continued)					ł			+		+					Ť	Ī		
Figure 6-6 Cable Interconnect Diagrams (Continued)	Figure 6-6 Cable Interconnect Diagrams (Continued)	Figure 6-6 Cable Interconnect Diagrams (Continued)	Figure 6-6 Cable Interconnect Diagrams (Continued)		DTEMICIOT 12	TEAM APPEN				~	Ļ	+-									Į
Figure 6-6 Cable Interconnect Diagrams (Continued)	Figure 6-6 Cable Interconnect Diagrams (Continued)	Figure 6-6 Cable Interconnect Diagrams (Continued)	Figure 6-6 Cable Interconnect Diagrams (Continued)			Н															1
Figure 6-6 Cable Interconnect Diagrams (Continued)	Figure 6-6 Cable Interconnect Diagrams (Continued)	Figure 6-6 Cable Interconnect Diagrams (Continued)	Figure 6-6 Cable Interconnect Diagrams (Continued)	141541	<u>.</u>					~											•
Cable Interconnect Diagrams (Continued)	Cable Interconnect Diagrams (Continued)	Cable Interconnect Diagrams (Continued)	Cable Interconnect Diagrams (Continued)	21215-2	Q-1090	THEFT]					-1[Ţ		E
Cable Interconnect Diagrams (Continued)	Cable Interconnect Diagrams (Continued)	Cable Interconnect Diagrams (Continued)	Cable Interconnect Diagrams (Continued)															ļ	THE PART OF THE PART	NIERONECI	
Cable Interconnect Diagrams (Continued)	Cable Interconnect Diagrams (Continued)	Cable Interconnect Diagrams (Continued)	Cable Interconnect Diagrams (Continued)																		
Cable Interconnect Diagrams (Continued)	Cable Interconnect Diagrams (Continued)	Cable Interconnect Diagrams (Continued)	Cable Interconnect Diagrams (Continued)																		}
								re R-R		a Inter	oonne.		oromo		Thursd						
																Ĺ			11 Z 4 - P		
														1	5						I

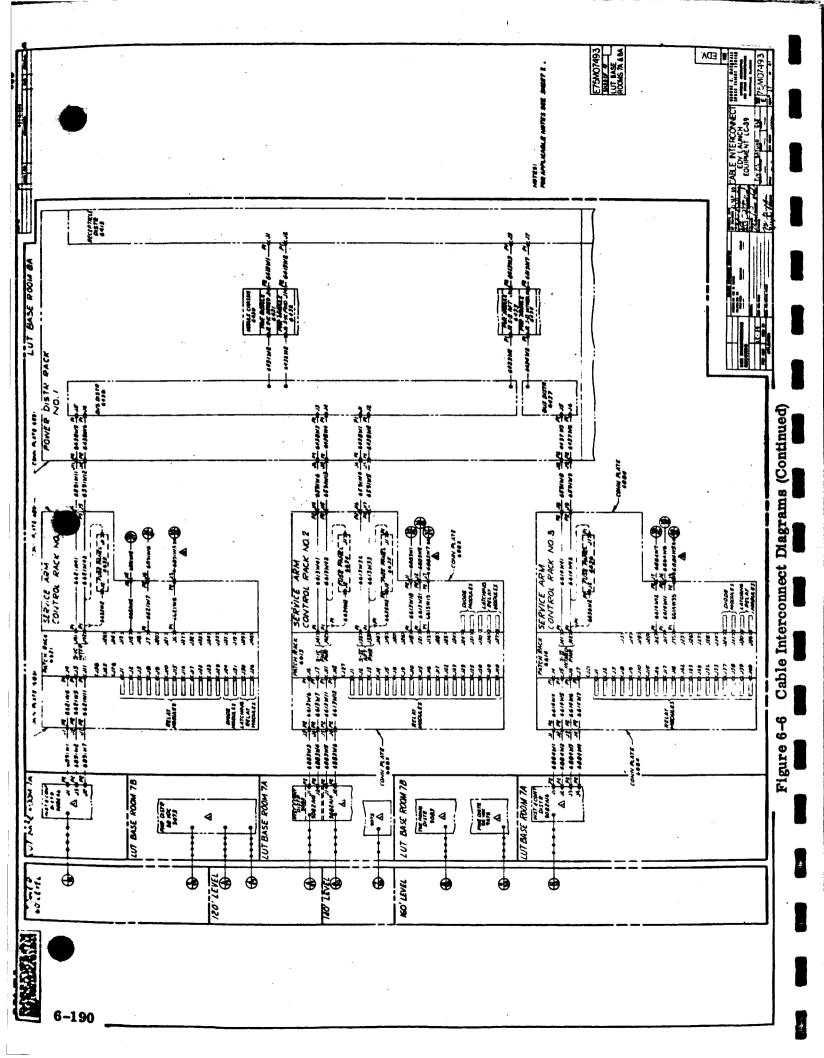
				•										- 7.4		2-4																	1		AOTES : Ang Arriversita Antres and an ar a							E CHOINE		INDEX							<u>-</u>	A MARA CARE INTERCONCOL	EQUIPMENT, LC39		
	N A										Π	Π			Π			Π	Π		Τ	Π	Π	TT	Π		Π	T	Π	Π	Ť	Π	Π	Π	20			Π	Π	Π	Π	Π	Π	Π	Π			Ţ	Ш	Π	Π				
	PLANNING BY SH															21																																					 		
	Annual ma																																																						
	Configuration of the				-																																															Ш		ł	
	- BULK	onlowly	9910	2 2 0 0		0180	A11MI DIBO	ATSMICIA	A75MICI91	ATMION!	ATCHIOLED	11	ATM 9101	11	ATMIOLOS	A75MI OBOI	AISMICEON																			A75MID191		A1610191	ANMIDEDI	NIMI DEGI	A73410201						, ,							rconnect Diagrams (Continued)	
	V Ne CANG	a 	TT	111			11 1 111	estr owerd i	E DIANO 775	1 D75M14947	A PREMA PERI				DIMUMAAS DIMOTIS	7 DISMO 7714		DIMONT		+	+	4 8663	H	$^{++}$		10 Du45		7 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1	4754 F7		200	0,0		2968 42	19 I I I I I I I I I I I I I I I I I I I	38 1 7759	X	Strought to	5 DISMOTTIA	DIBMOT714	MITOMETO II													ns (Coi	
	A CASH	No.			1	2M4	NELO ZI-QONOINELO IMI	Prswi 1604	I D'IMI JACA-	TACHUR DIMINOL DIMINI		1 1		1 1			- DIMONGIA	A DINOULU		41		+	Ħ							63							41 1 34 142 1 35	-12 BPINDBA71-	11		11-401604640													Diagrar	
			7200AIL	10001	72004	7200A2W5	7200AWI	7202WI	W(022				1001	A ON	ENION:	TEDZWI	TEORY	VYOYL	Thomas a	14037		TIGIAIVA	INVRONT		THE PARTY	NO10	TAWAGON .	VIACONT.	THANKAR	MON	THOMAN	TGONAMIA	ACONT	HYCOX	CONVION.	014440	THOMANAL	TOWNY	TOWN	165AV	2495%													nnect]	
	din a mar	2 :	1	12	×	77	77	*	**		7	37.44	***		27,46	- FR	72	17	27.22		2	77 77 77	1741	*		23.46	1		11.4	27.46	*	7	772	**	24	5 7 7	*	a a	22	27		31	**				74		+					Interco	
	Configure Press	L																																																				Cable Inter	
	ALTER CON																											-						-		-																			
	ANLE ON	╞	A19MIIBIT		·							1101			4015	101					5		100	ATEMI4OTS		1/04	100			101				110	101		4015	87.04		4014	1076		1	Ħ		0100	4014							Figure 6-6	
	a cliffer A	19451	MEA BURNER	, ,	1110	DISMSONS	D79M50454	1580	DTIMSO411		1800	TRUE ATTINIES			ILEO ATENIAOTE	1200 ATSM14075				Π	TON MUTA		300 AIMI405	INSLV OVZINKQ		DISMIRED DESIMATE	STOPHINETA DESI	•		1100 A114 A014	П	Ħ	П		TOP ATSWHOTS			STOPHNELA AGT	- 93	1640 A15M14075		Π	1 1001	11		PH ATSHIDLOD								1	
		Pireminio-E bisu	10101111111111111111111111111111111111	MPT 1 0015 IM PT		M810 0.001 INSI	MELO 02-00121MELO	M10 5 00 8 M	DTSNIFIOO-25 DTSN	1.1.0.1	DTSMIRIOG-28 D79MI 1266	OF STIMELO IN - ALCONSIC		ş.	COSTINITE SALES	405010-187 D15 M1 12 B0	01)- 1	2	10-1	WEID SEL-DIDLO	DISMONDO-01 DISMONDIA	561-9	COCO-IN DIMIIAN	INSUG LEI-OLOGOWSLO		NS10 008-01060	ADDOTO-BOL DITAN LEDO	ter ter		AN WORLD AND AND AND AND AND AND AND AND AND AN	Cano be men		, , ,	01010-101 0144	111-11	1 -10-	MSL0 (17-01060)	Bearinged HE- DIDE ented	1	01340010-214 D19401 094	I INITIAL OF OTOPO	84	161-1	The second		braudio Diski Hara	00 211W 510 482-0106 04 510								
	MER CARE		1-150604610 MASSIMIN	Ma INFITA					6717WE 0751	T	6718WE DTSI				LOONE DIS			П	14499		LEASWE DISA			11	11	CARLES DISK	11		6887N4 6887N4	1 1		6005W2	11	tenewe DTH	6009W1 075	648944 648944	1510 2 m 6194		CAPOW6 CAPOWF	6690W6 5140643	-		689146	< i.	66224E	_	60-51.45 DISM								
A CONTRACT											<u>.</u>		<u> </u>	ن م ت	<u>i i</u>			<u>.</u>					د ا	<u> </u>		#	I		<u></u>	<u></u>	<u>.</u>	<u> </u>	الب ال	<u> </u>		ني د	. لا	.	ε. <u>ι</u> .	• - • • •	▲↓	е 4	- b - b - b	- 4L		ل ــــ					• • '				

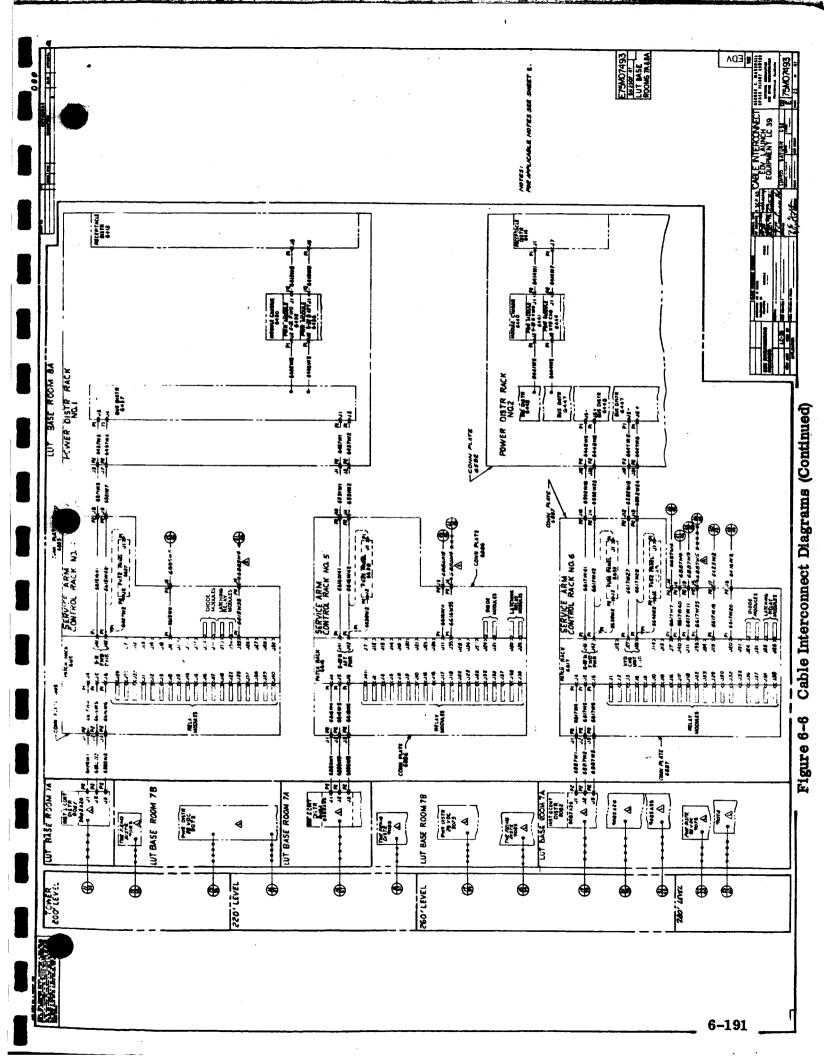
E.

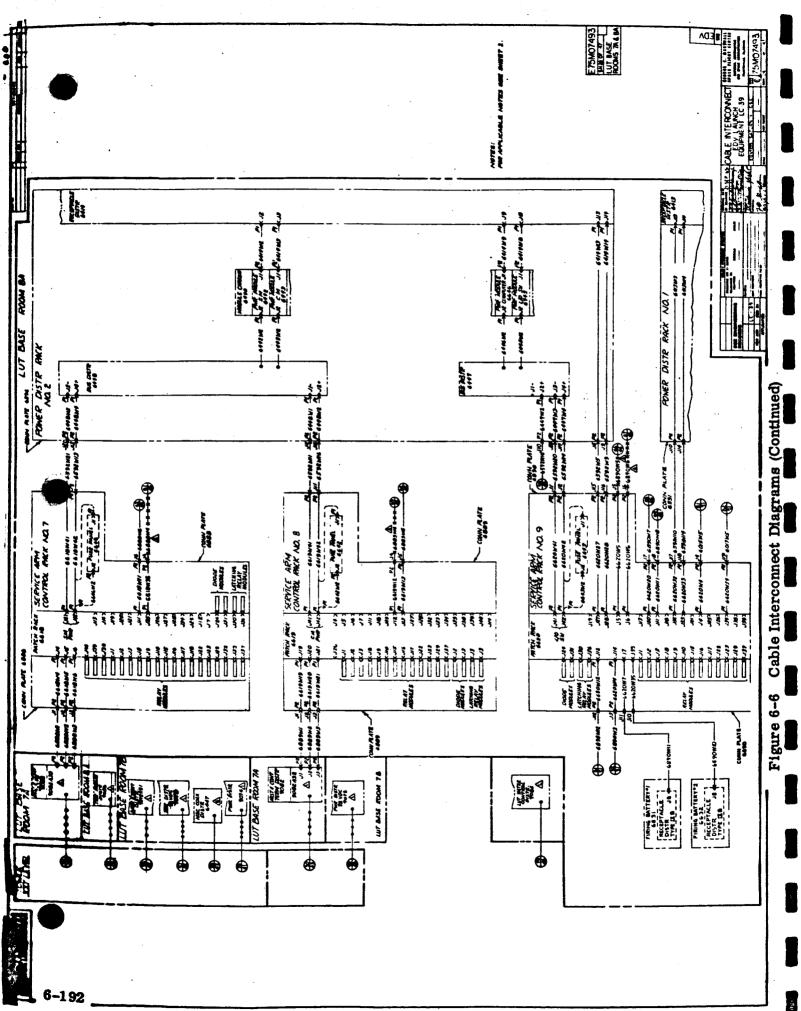
6-188

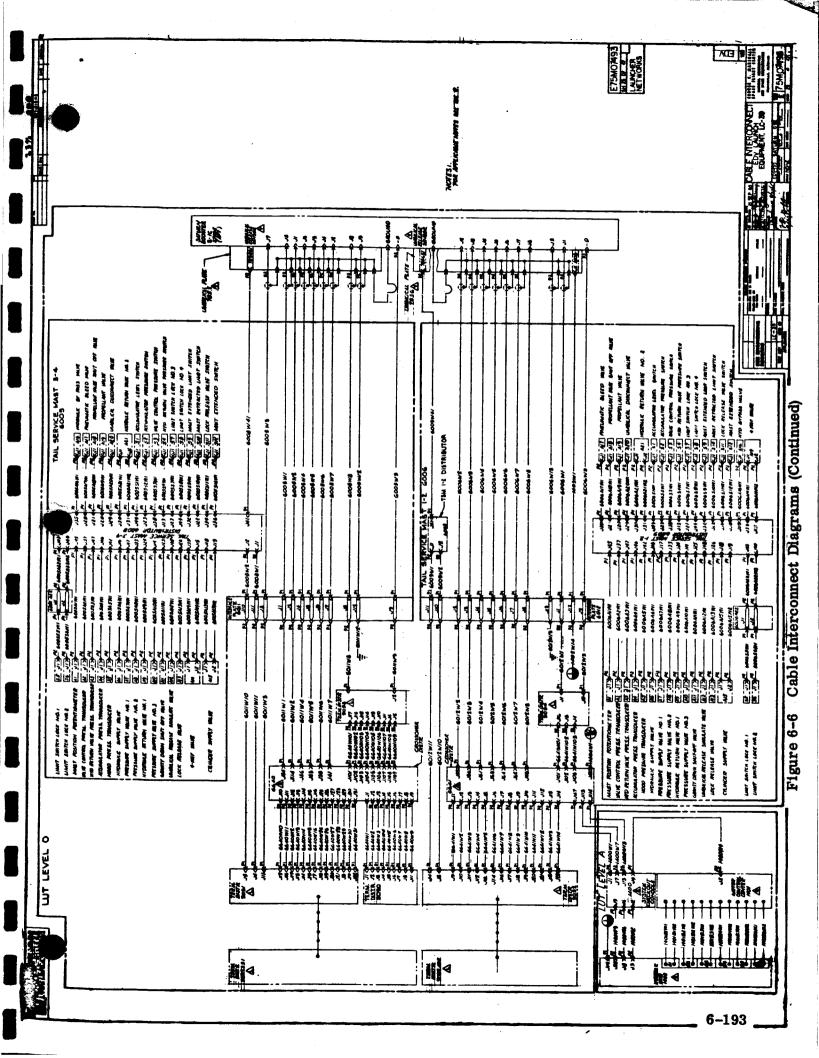
Ę

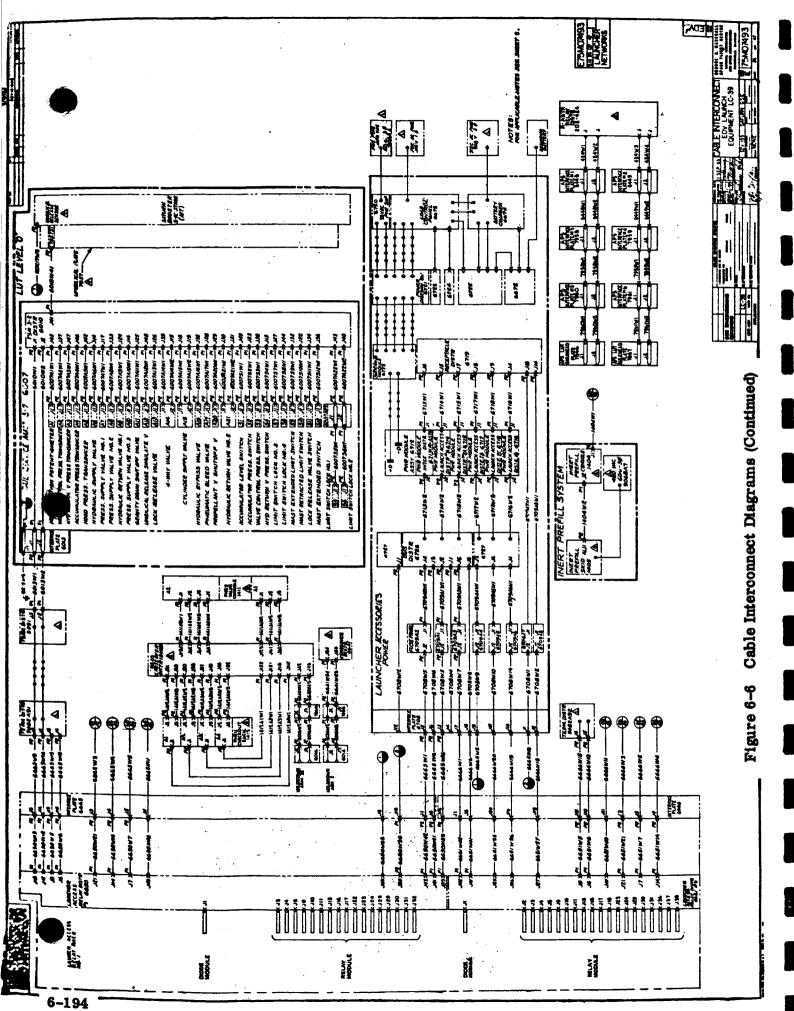


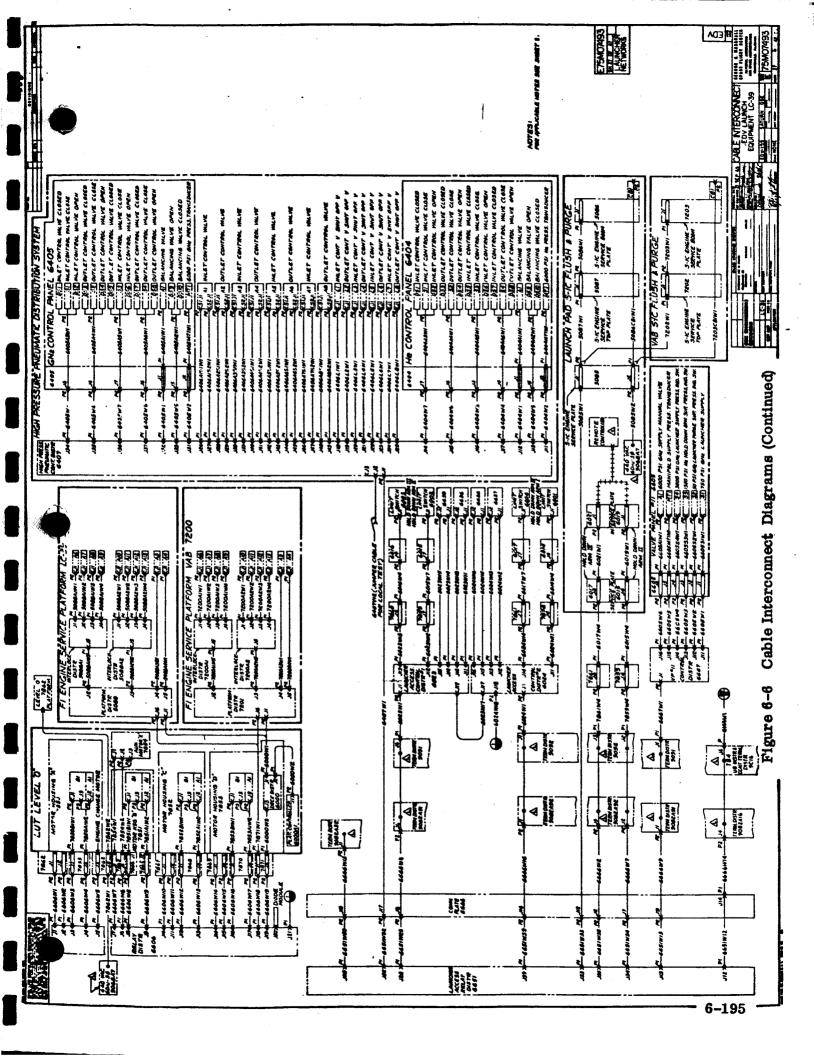


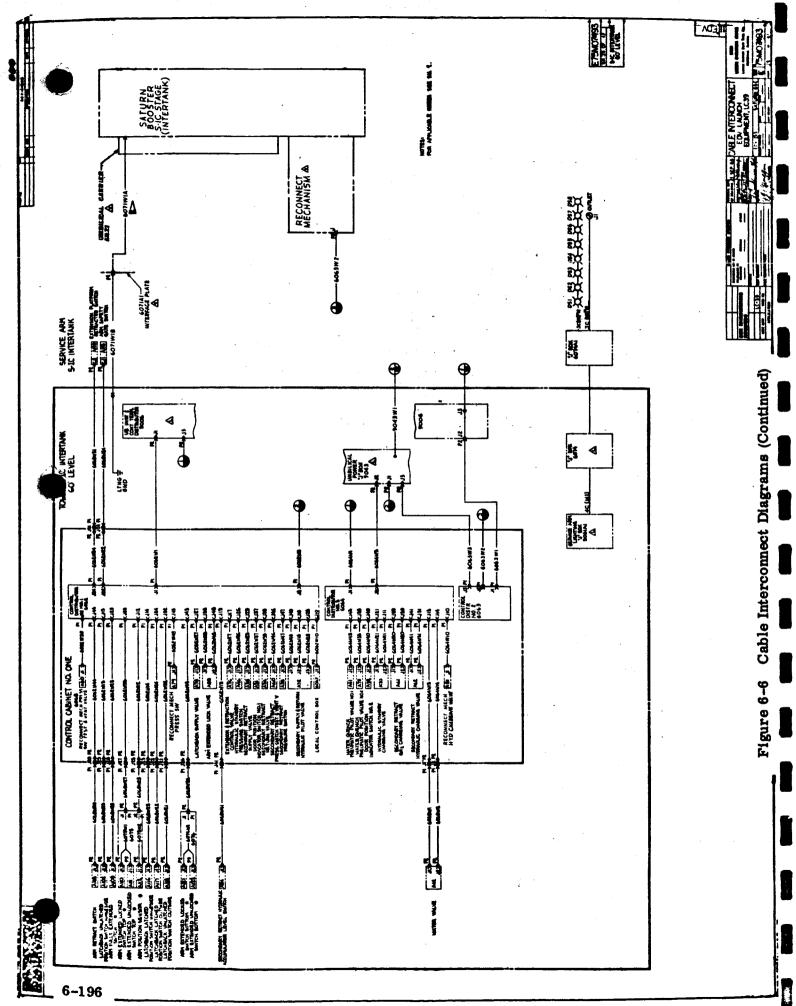


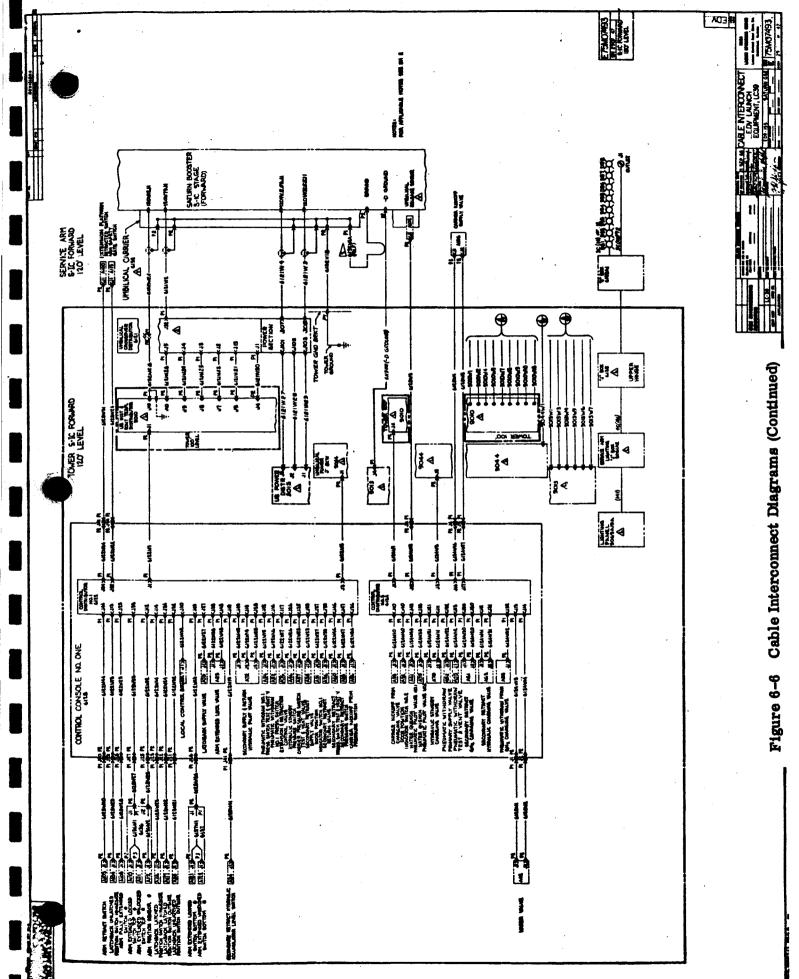


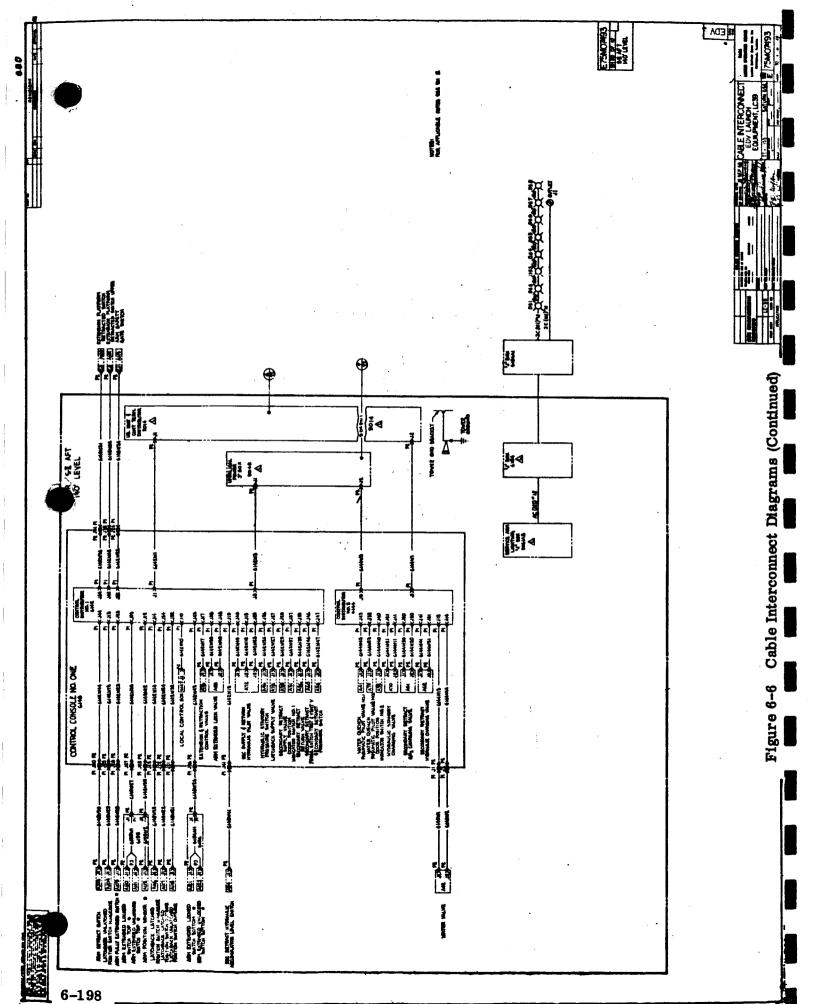


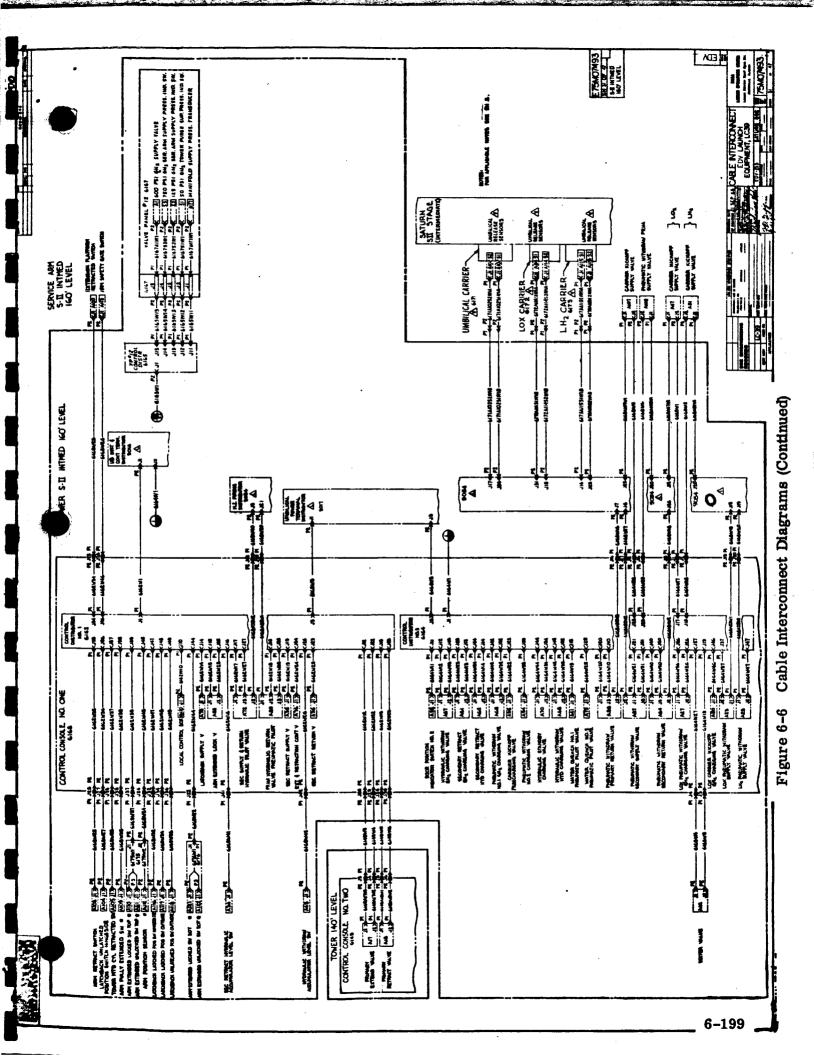


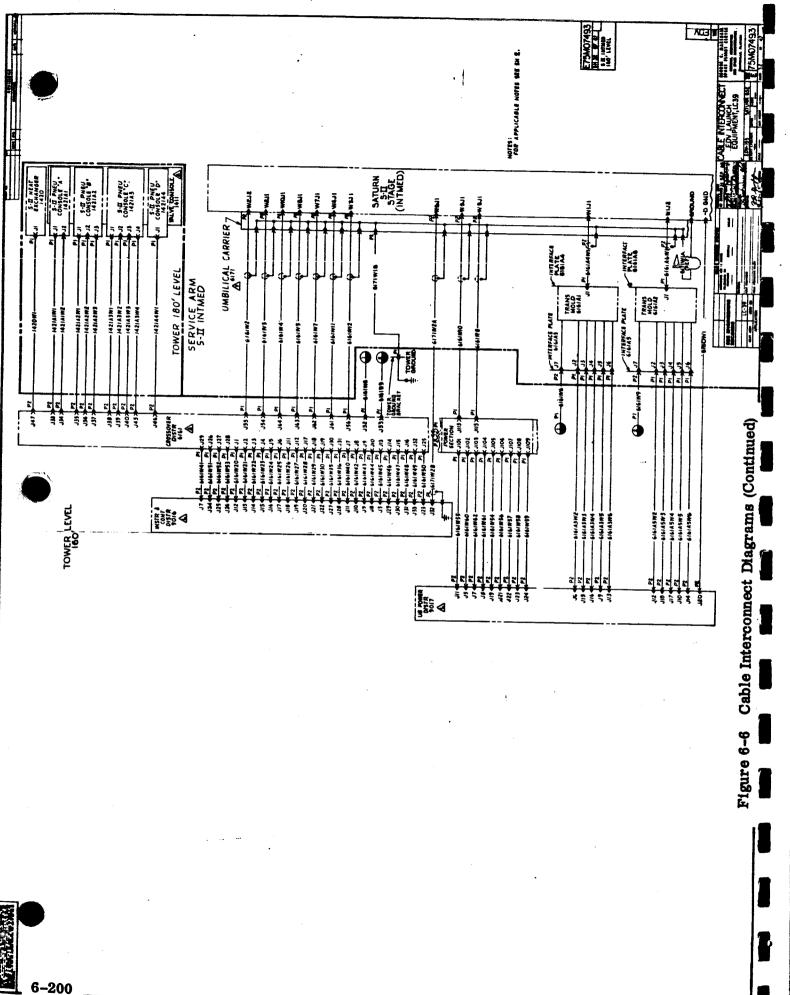


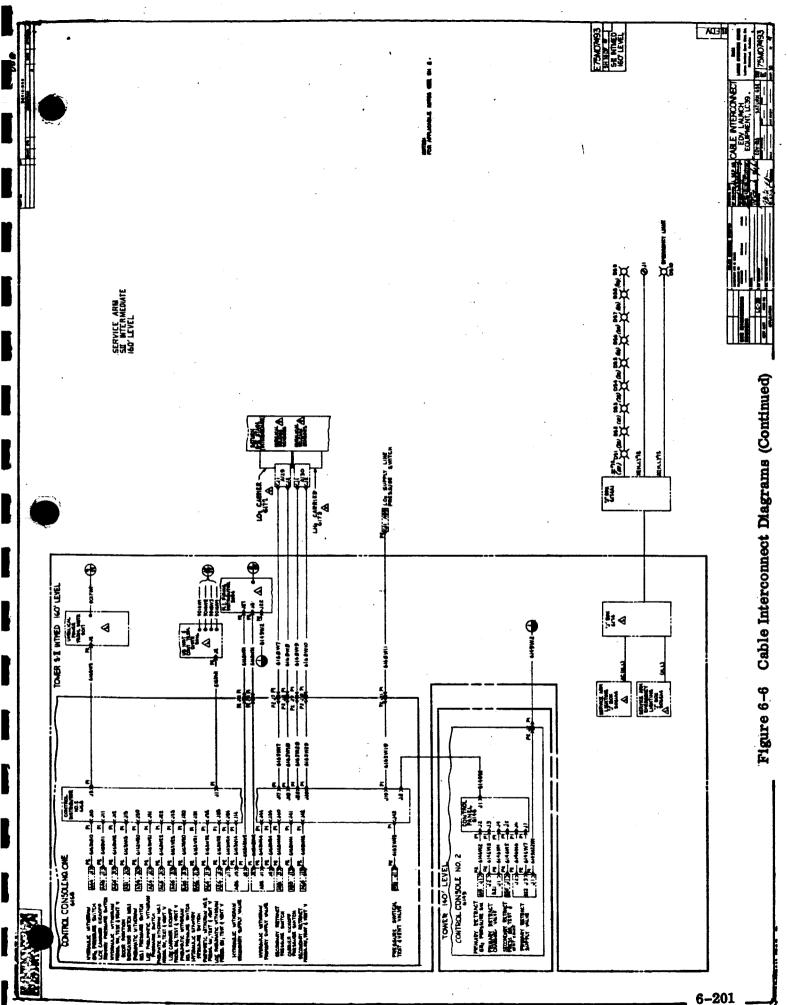


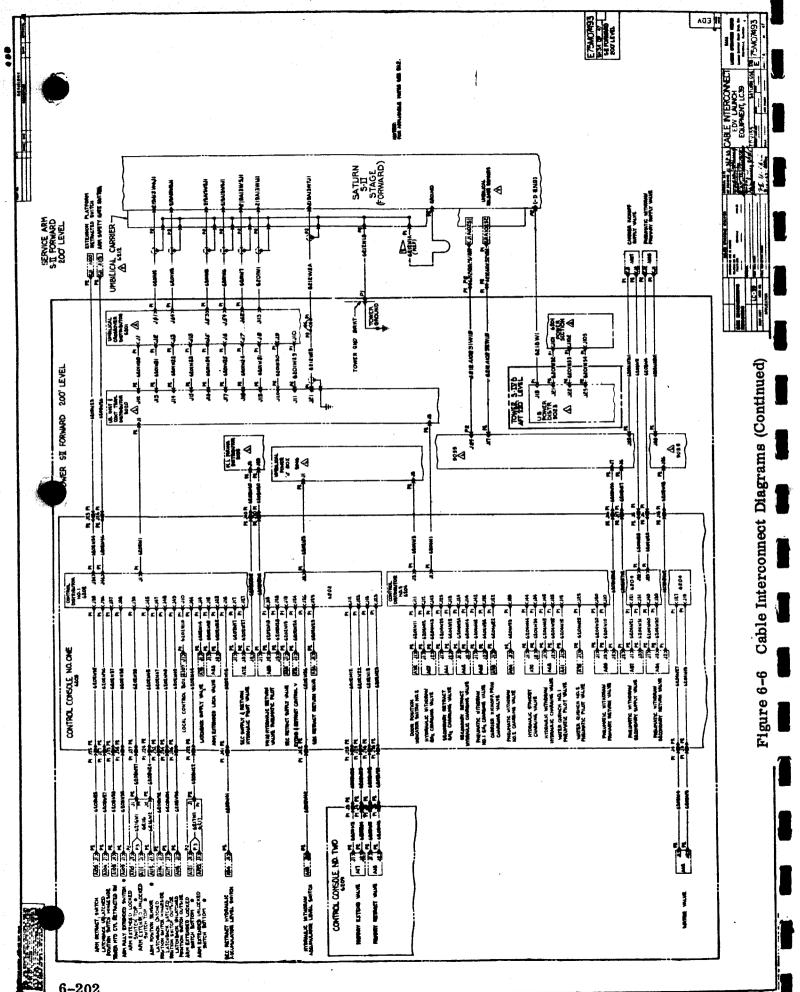


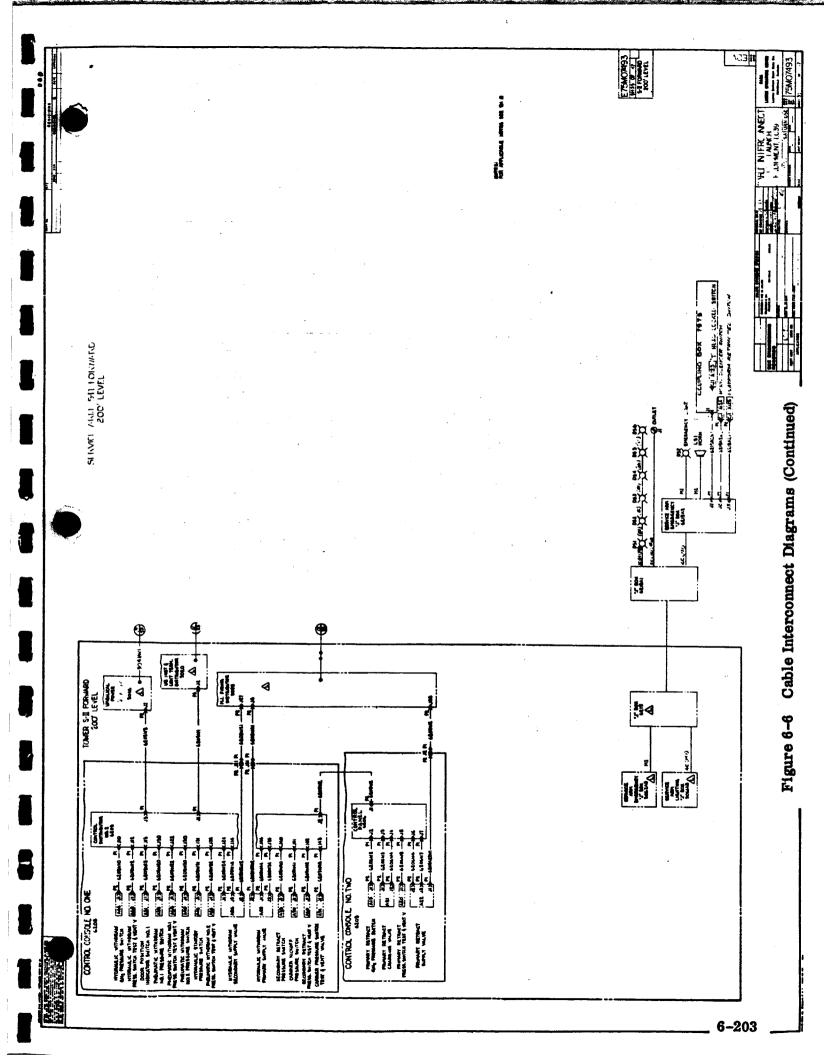


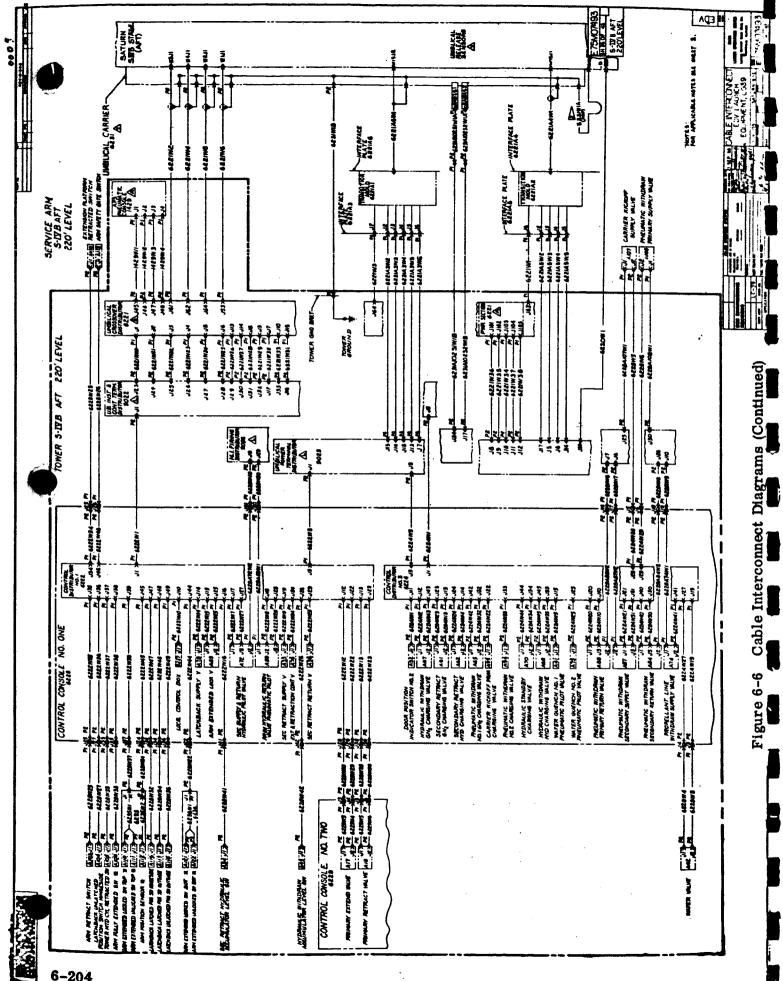




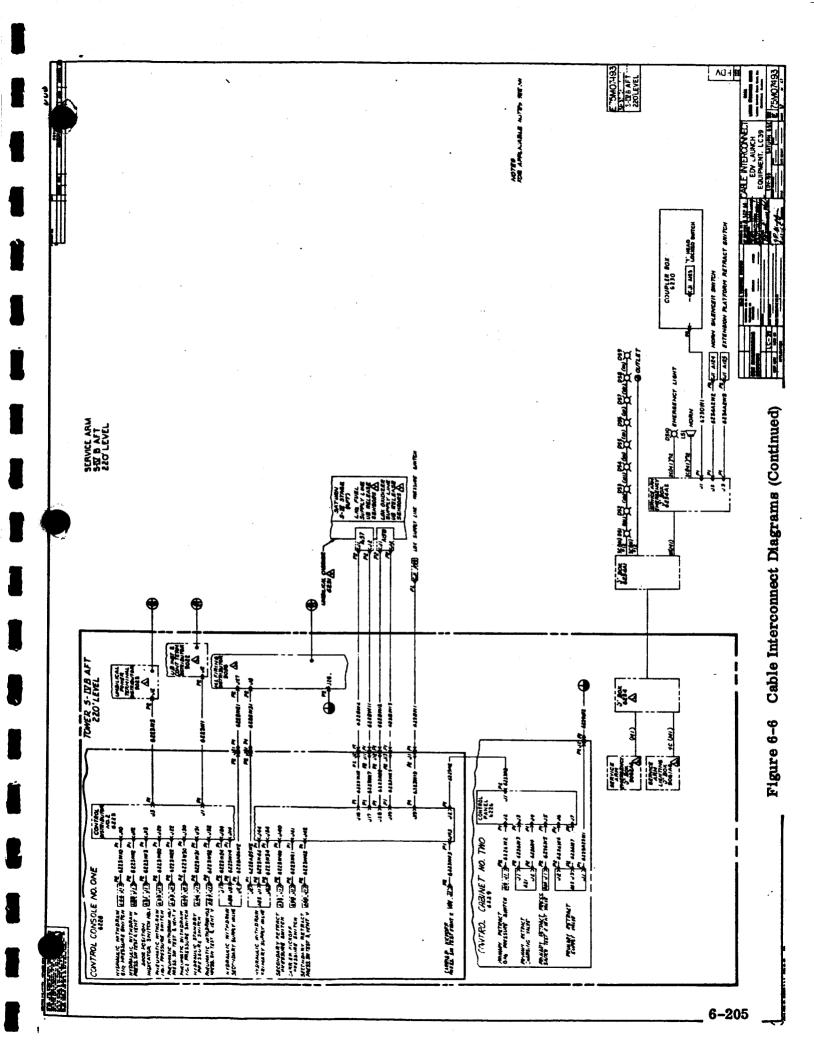


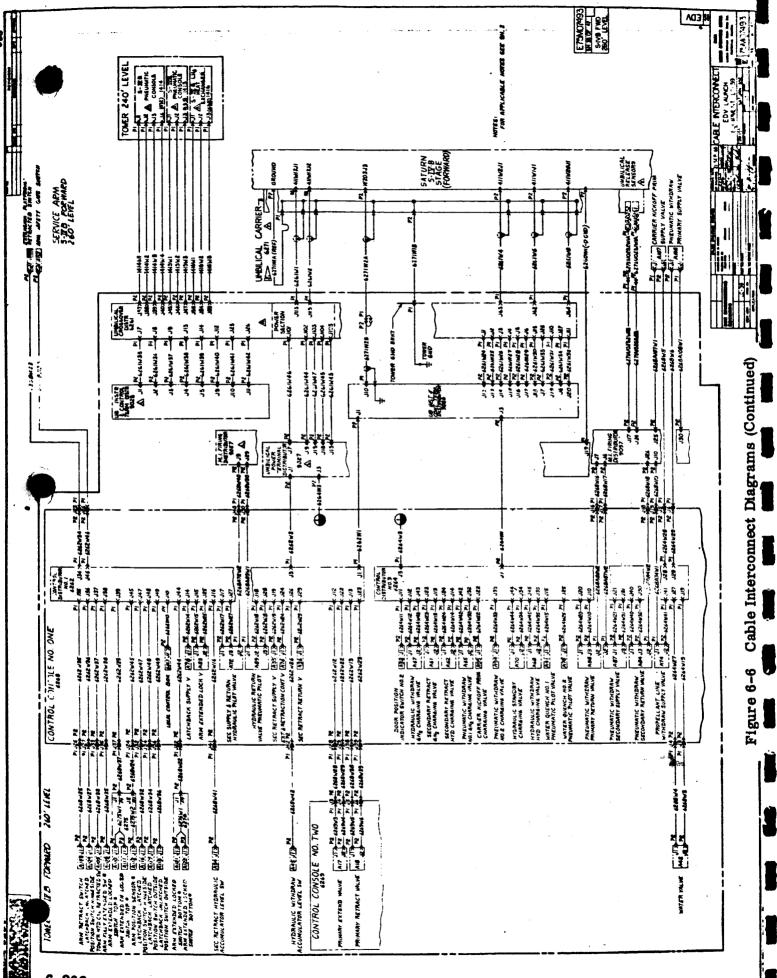


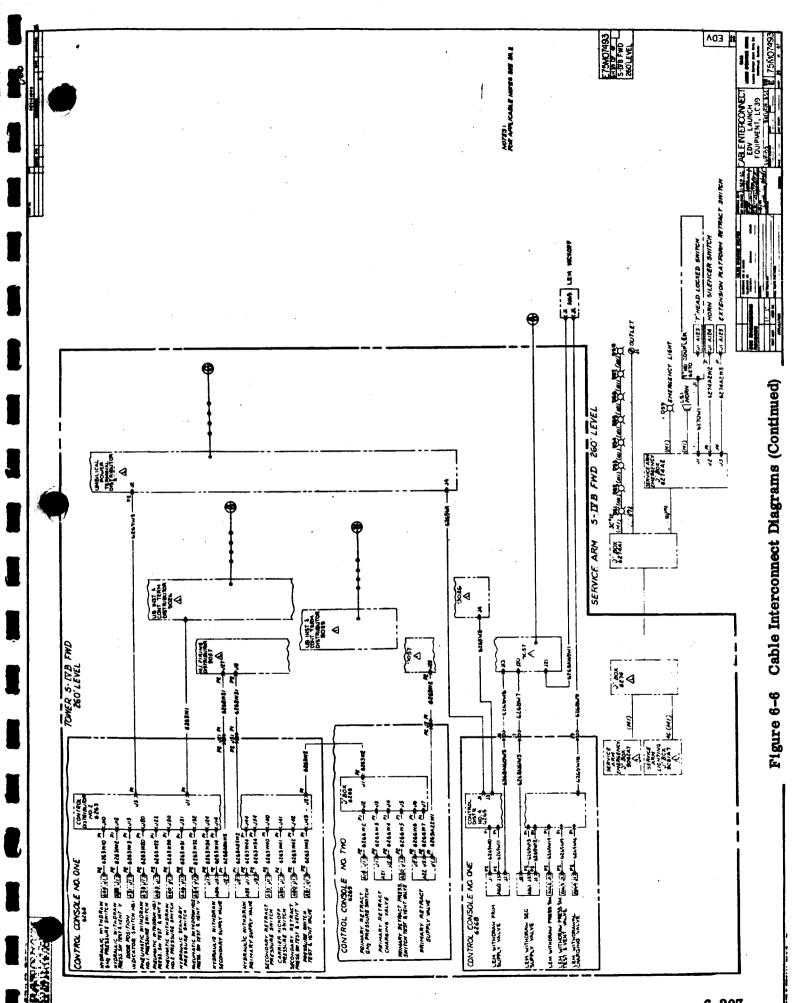


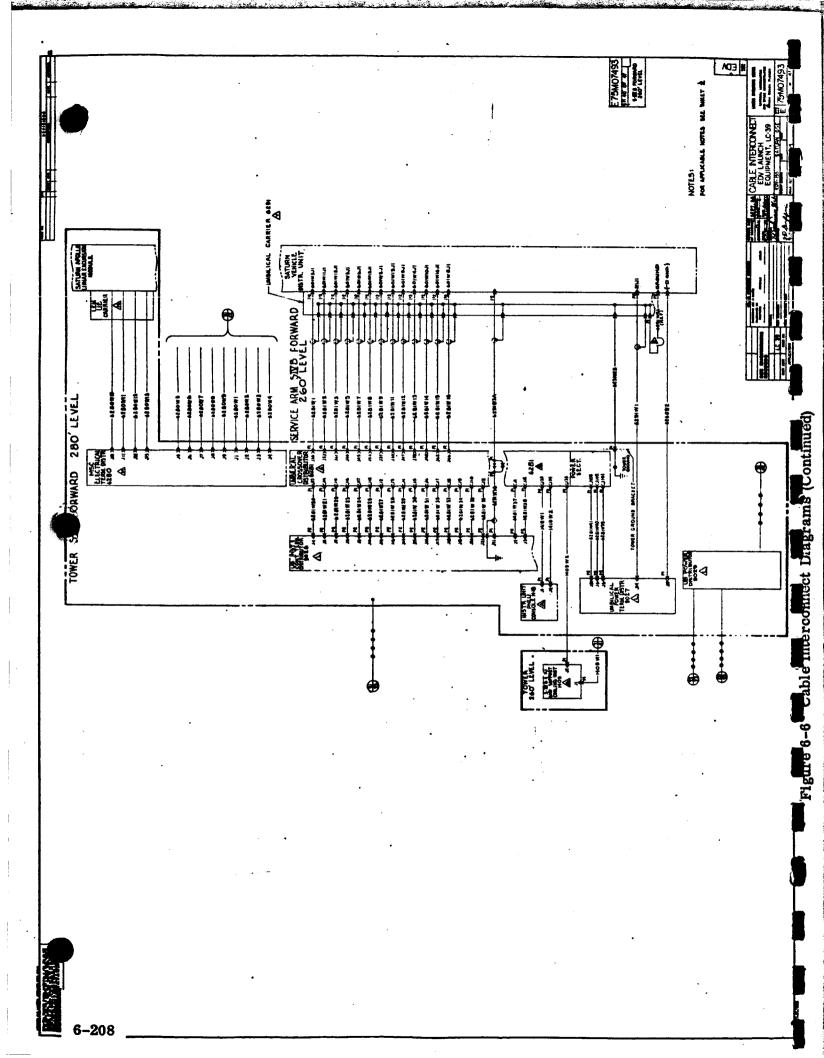


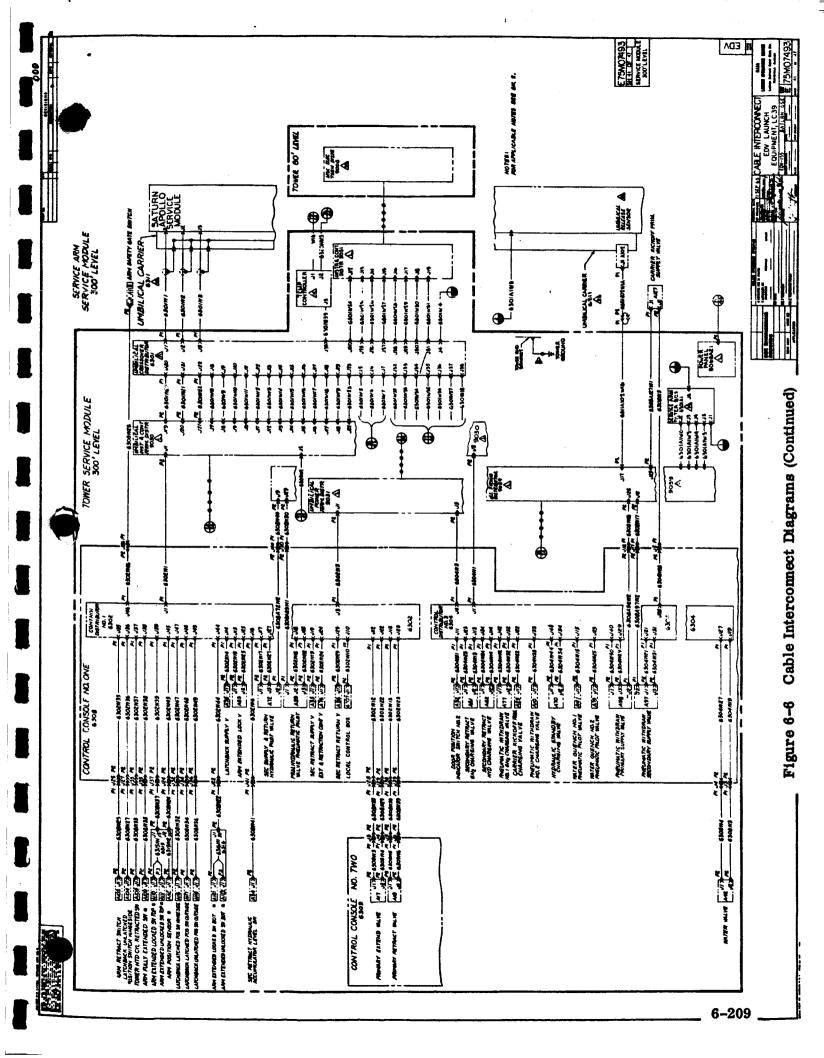
6

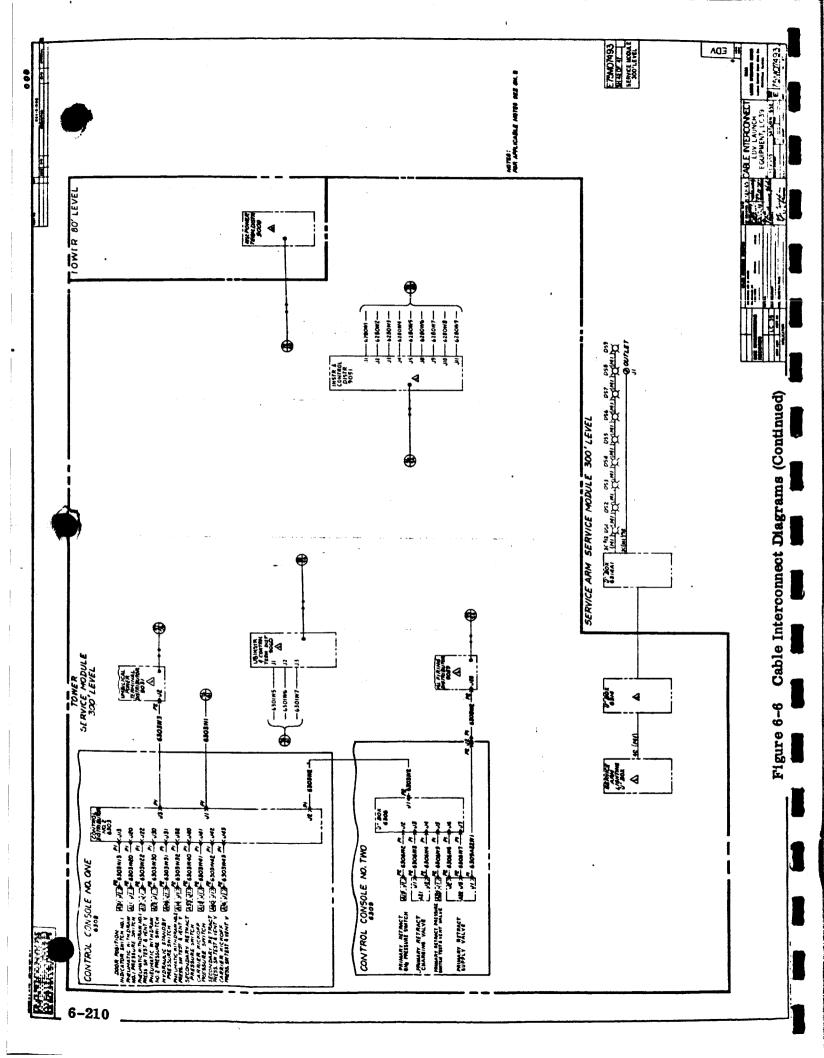


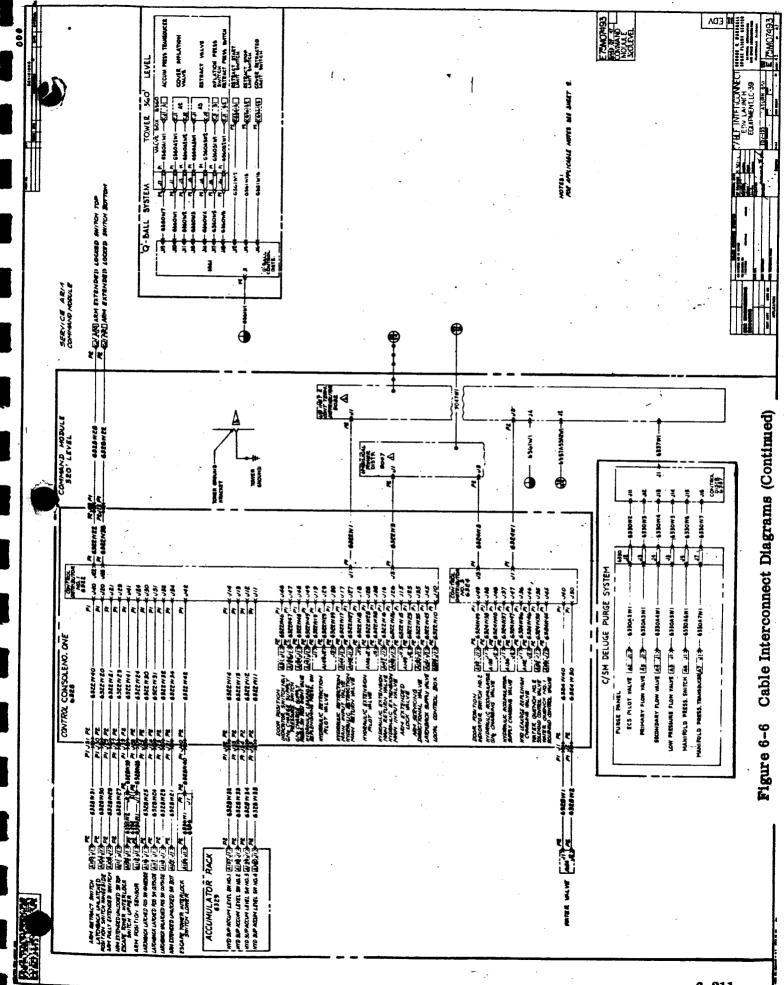


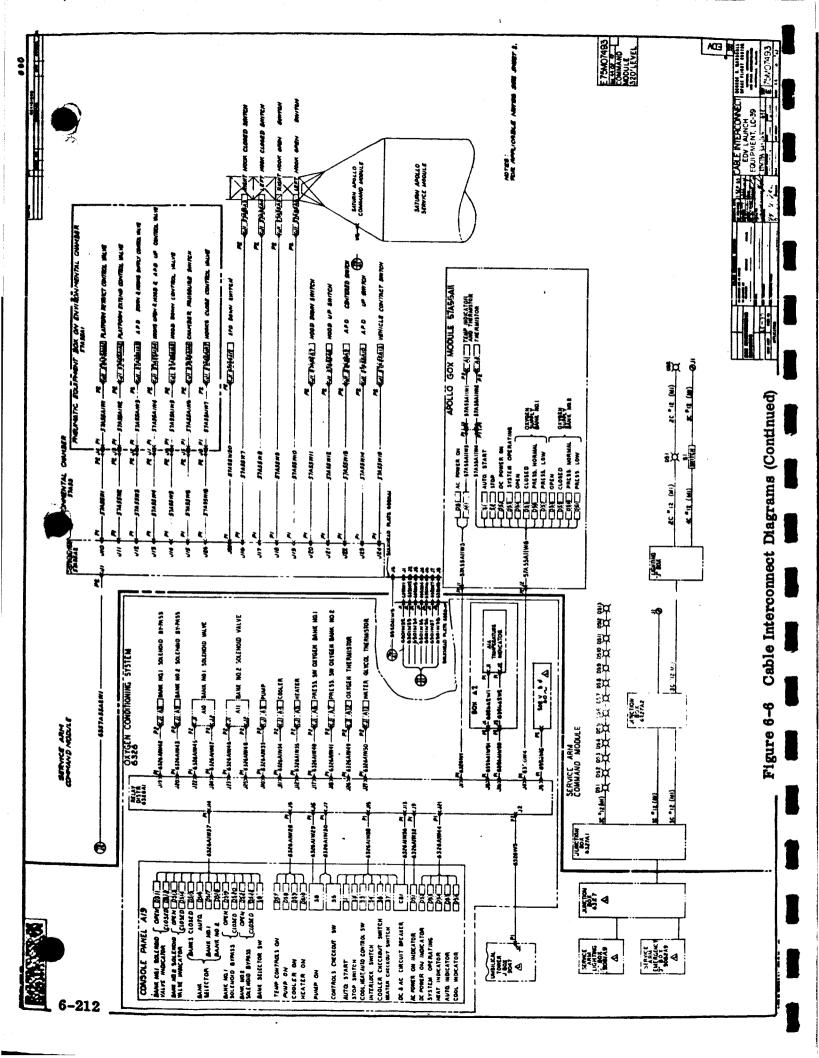


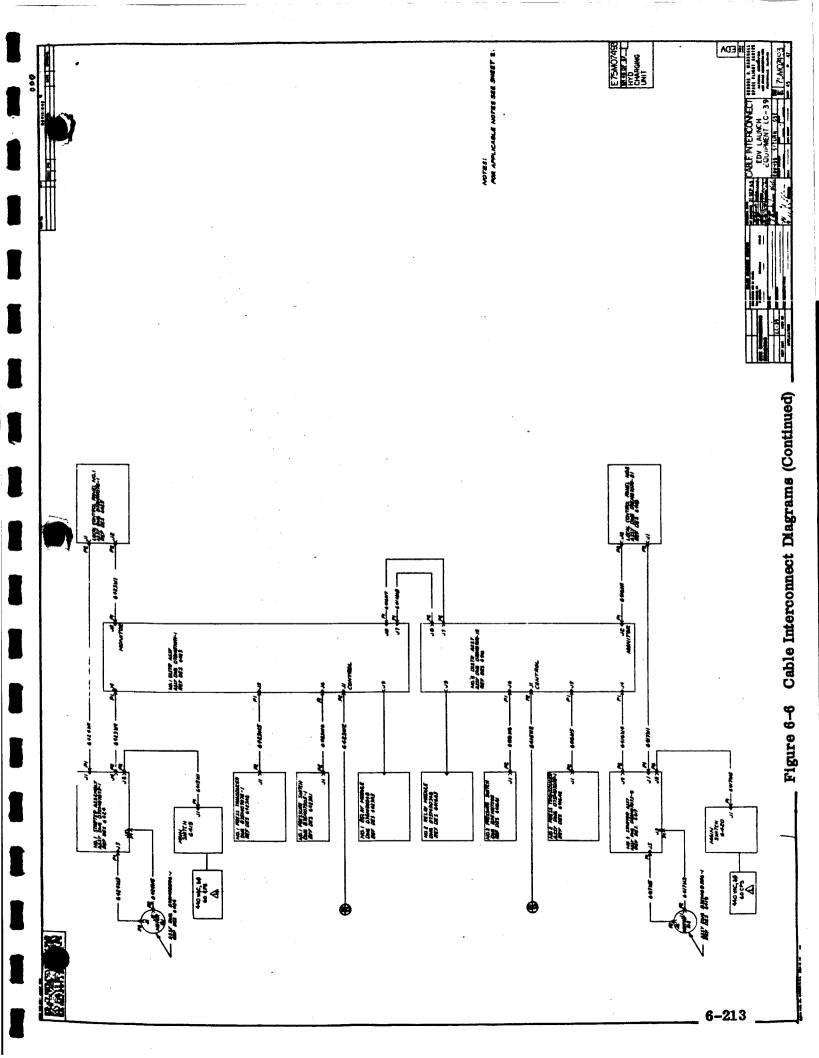












LUT Cable Interconnect

Information

to be

Supplied

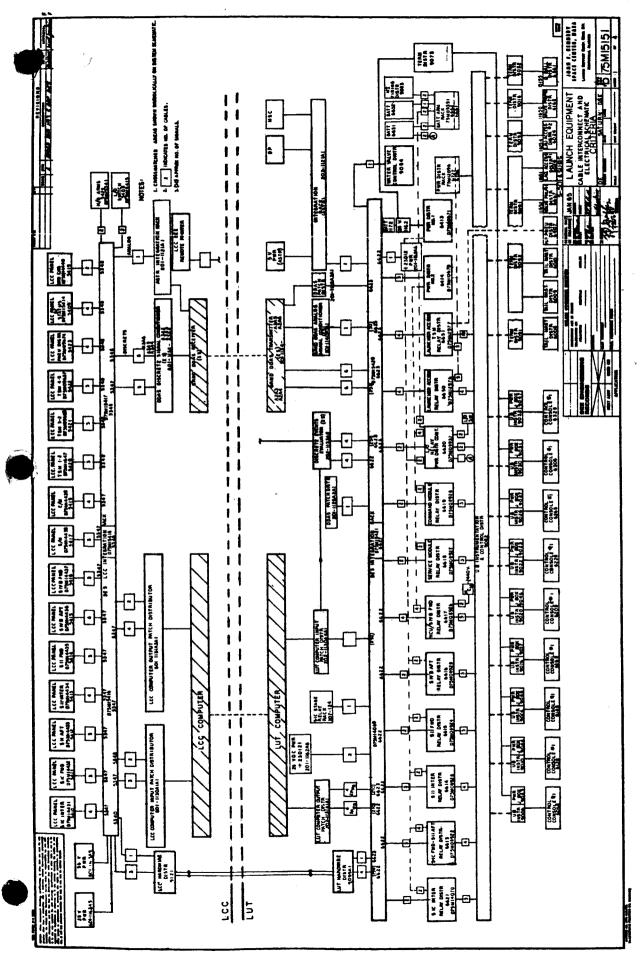
LCC Cable Interconnect

Î

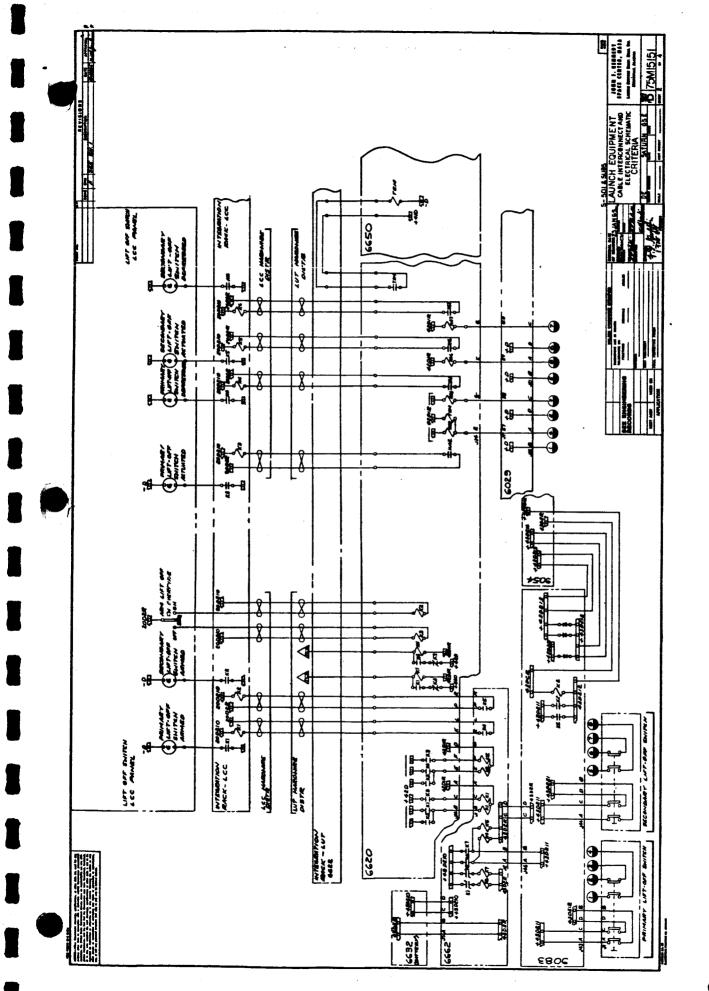
Information

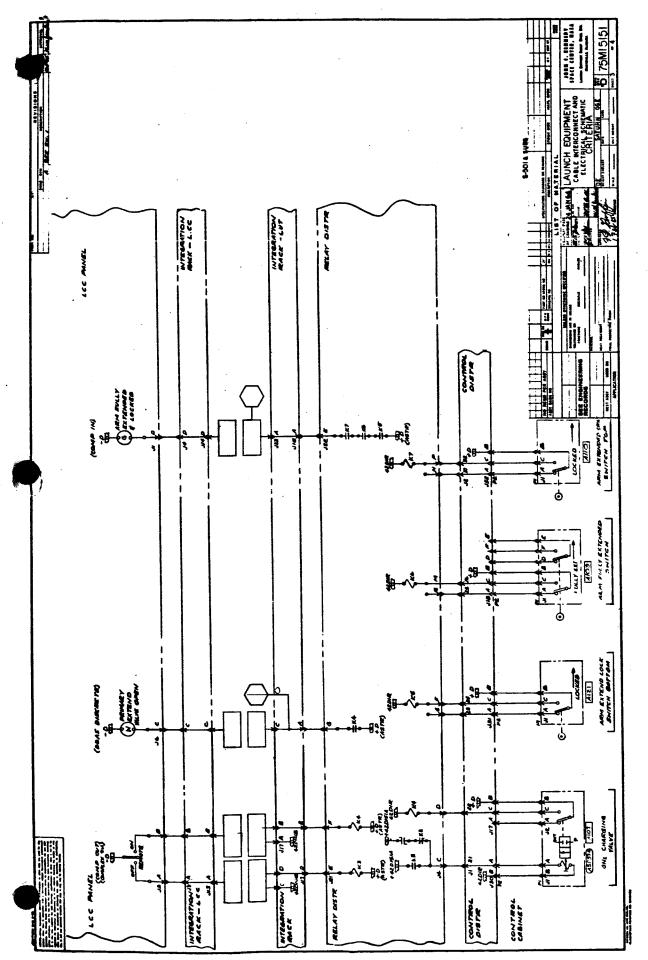
to be

Supplied

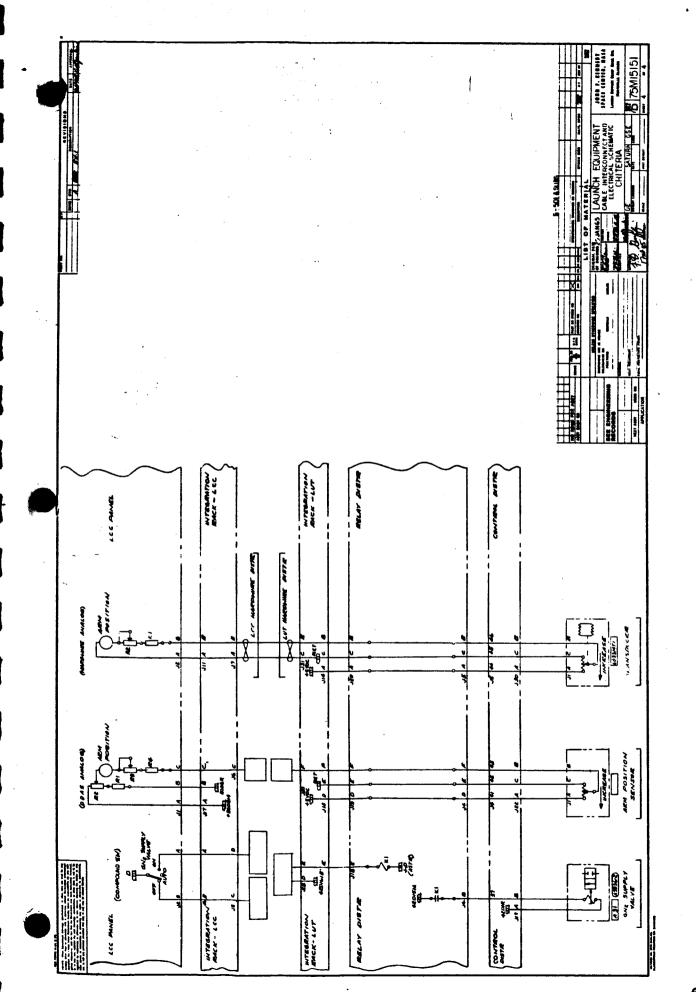


ł





1.1



5.

REFERENCE DESIGNATION NUMBERS VS. ASSY DRAWING NUMBERS.

The ectrical reference designation numbers in Table 6-4 are assigned to ESE assemblies or to assemblies carrying ESE cable. The drawing given for each assembly should call out other drawings showing any assembly or installation information required.

Reference Designation		Assembly Drawing or Installation
Number	Assembly	Drawing
5005	9 10 Endre Semicing Diste	J75M09312
5085 5086	S-IC Engine Servicing Plate	J75M09312-3A
5086 5087	S-IC Engine Servicing Boom Plate S-IC Engine Servicing Top Plate	J75M09312-3A J75M09312-1A
	C C C	\$E75M51238
5088 5080	S-IC Engine Servicing Platform (Pad)	
5089	F-1 Engine Skirt Instl. Distr.	E75M13209
5347	LCC Integration Rack Assy #1	D75M15416
5348	LCC Integration Rack Assy #2	D75M15417
349	ICC Cable Assy Installation	D75M15419
6000	Deals Distributor E 1 Engine	
0000	Deck Distributor, F-1 Engine	E75M13211
6000A1	Servicing Platform	E75M13211
OUUAI	Platform Controller, F-1 Engine	
6005	Servicing Platform	D75M13570
	Tail Service Mast 3-4	*J75M12019
6006 6007	Tail Service Mast 1-2	*J75M12023
6007	Tail Service Mast 3-2	*J75M12027
6008	Control Distributor, TSM 3-4	F75M12017
6009	Control Distributor, TSM 1-2	F75M12025
6010	Control Distributor, TSM 3-2	F75M12029
6011	Interface Plate, TSM 3-4	J75M11882
6012	Interface Plate, TSM 1-2	J75M11902
6013	Interface Plate, TSM 3-2	J75M11932
6014	Interface Plate, Holddown Arm #1	J75M05972-677
6015	Interface Plate, Holddown Arm #2	J75M05972-681
6016	Interface Plate, Holddown Arm #3	J75M05972-678
6017	Interface Plate, Holddown Arm #4	J75M05972-682
6019	Service Plate, Holddown Arm #2	E75M06443
6021	Service Plate, Holddown Arm #4	E75M06443
<u>6023</u>	Launcher Accessories Control	
	Distributor #1	F75M14254
	* Electrical Installation Drawing	

Table 6-4 REFERENCE DESIGNATION NUMBERS VERSUS ASSY DRAWING NUMBERS

	Reference Designation Number	Assembly	Assembly Drawing or Installation Drawing
	6024	Launcher Accessories Control Distributor #2	F75M11651
	6026	Primary Liftoff Switch, Holddown Arm #2	J75M12231
	6027 ·	Secondary Liftoff Switch, Holddown Arm #2	J75M12231
	6028	Liftoff Switch Pneumatic Assy, Holddown Arm #2	*J75M11078
	6029	Liftoff Switch Distr. Assy, Holddown Arm #2	D75M11083
	6031	Primary Liftoff Switch, Holddown Arm #4	J75M12231
	6032	Secondary Liftoff Switch, Holddown Arm #4	J75M12231
	6033	Liftoff Switch Pneumatic Assy, Holddown Arm #4	*J75M11078
	6034	Liftoff Switch Distributor Assy, Holddown Arm #4	D75M11084
	6062	Control Distr. #1	Assy.
		,	J75M10030
			Wiring
			A75M10558
	6063	Control Distr. #2	Assy.
			J75M10030
			Wiring
			A75M10557
	6064	Control Distr. #3	Assy.
			J75M10030-3
			Wiring
			A75M10561
	6067	Local Control Unit, S-IC Int.	Assy.
			J75M10898
			Instl.
	n na∰rak s		J75M07578-1
	6068	Control Console No. 1, S-IC Intertank	*J75M08751
	6070	Reconnect Mech. S-IC Intertank	J75M08754
1	6071	Umb. Carrier, S-IC Intertank	Cable Assy.
			D75M09071
	6072	LOX Carrier, S-IC Intertank	Cable Assy.
			D75M09071
	6073	LH ₂ Carrier, S-IC Intertank	Cable Assy.
		4	D75M09071
	6074	Service Arm Power "J" Box	Cable Assy.
			D75M09071
		*Electrical Installation Drawing	l

Table 6-4 REFERENCE DESIGNATION NUMBERS VERSUS ASSY DRAWING NUMBERS (Continued)

1	7

(1
Reference Designation Number	Assembly	Assembly Drawing or Installation Drawing
6075	Hinge Upper, S-IC Int.	J75M08260-1
6076	Hinge Lower, S-IC Int.	J75M08261-1
6122	Control Distr. #1	Assy.
		J75M10030-5
		Wiring
		A75M10559
6124	Control Distr. #3	Assy.
		J75M10030-7
		Wiring
		A75M10562
6127	Local Control Unit, S-IC FWD	Assy.
		J75M10898
		*J75M07578-3
6128	Control Console No. 1	*J75M08746
6132	Umb. Carrier, S-IC Fwd	Cable Assy.
		D75M09074
6133	LOX Carrier, S-IC Fwd	Cable Assy.
		D75M09074
6134	LH ₂ Carrier, S-IC Fwd	Cable Assy.
		D75M09074
6135	Service Arm, Power "J" Box	Cable Assy.
		D75M09074
6136	Hinge Upper, S-IC Fwd	J75M08260-1
6137	Hinge Lower, S-IC Fwd	J75M08261-1
6142	Control Distr. #1	Assy.
		J75M10030-9
		Wiring
6144		A75M10560
6144	Control Distr. #3	Assy.
		J75M10030-11
		Wiring
6146		A75M10563
01.40	Console No. 2 "J" Box	Assy
J		J75M07975
		Harness
	* Flootwight Installation Drawing	D75M11336
	* Electrical Installation Drawing	

Table 6-4 REFERENCE DESIGNATION NUMBERS VERSUS ASSY DRAWING NUMBERS (Continued)

 	Reference Designation Number	Assembly	Assembly Drawing or Installation Drawing
	C1 47	Local Control Unit, S-II Aft	`Assy.
	6147	Local Control Unit, 5-11 Alt	J75M10898 *J75M07578-5
	6148	Control Console No. 1	*J75M08741
	6149	Control Console No. 2	*J75M07976
-	6151	Umb. Carrier, S-II Aft	Cable Assy.
_	0101	Unit. Carrier, 5 m mit	D75M09077
	6152	LOX Carrier, S-II Aft	Cable Assy.
-	0102		D75M09077
	6153	LH ₂ Carrier, S-II Aft	Cable Assy.
	0100	$\frac{1}{2}$	D75M09077
-	6154	Service Arm Power "J" Box	Cable Assy.
	0101		D75M09077
	6155	Hinge Upper, S-II Aft	J75M08260-1
	6156	Hinge Lower, S-II Aft	J75M08261-1
	6162	Control Distr. #1	Assy.
	0101		J75M10030-13
-			Wiring
			A75M10546
	6163	Control Distr. #2	Assy.
_	0100		J75M10030-15
		· • •	Wiring
			A75M10550
_	6164	Control Distr. #3	Assy.
	0202		J75M10030-17
			Wiring
			A75M10554
	6165	Valve Panel No. 12 Control Distr.	Assy.
		· · · · · · · · · · · · · · · · · · ·	F75M11650
			Wiring
			A75M11644
	6167	Valve Panel #12	J75M05958
_	6168	Control Console #1	J75M08735
	6170	Local Control Unit, S-II Int.	Assy.
•			J75M10898
			*J75M07578-7
		* Electrical Installation Drawing	
- 1			

Table 6-4 REFERENCE DESIGNATION NUMBERS VERSUS ASSY DRAWING NUMBERS (Continued)

Ĺ

Ē



Reference Designation Number	Assembly	Assembly Drawing or Installation Drawing
6171	Umb. Carrier, S-II Int.	Cable Assy.
		D75M09080
6172	LOX Carrier, S-II Int.	Cable Assy.
		D75M09080
6173	LH, Carrier, S-II Int.	Cable Assy.
	2	D75M09080
6174	Service Arm Power "J" Box	Cable Assy
		D75M09080
6175	Hinge Upper, S-II Int.	J75M08260-1
6176	Hinge Lower, S-II Int.	J75M08261-1
6202	Control Distr. #1	Assy.
		J75M10030-19
		Wiring
		A75M10547
6203	Control Distr. #2	Assy.
		J75M10030-21
		Wiring
		A75M10551
6204	Control Distr. #3	Assy.
		J75M10030-23
		Wiring
		A75M10555
6206	Console No. 2 "J" Box	Assy.
		J75M07975
		Harness
		D75M11366
6207	Local Control Unit, S-II Fwd	Assy.
		J75M10898
		*J75M07578-9
6208	Control Console #1	*J75M08728
6209	Control Console #2	*J75M07977
6212	Umb. Carrier, S-II Fwd	Cable Assy.
0010		D75M09084
6213 .	LOX Carrier, S-II Fwd	Cable Assy.
	I He Commission & H David	D75M09084
6214	LH ₂ Carrier, S-II Fwd	Cable Assy.
	* Electrical Installation Drawing	D75M09084

Table 6-4 REFERENCE DESIGNATION NUMBERS VERSUS ASSY DRAWING NUMBERS (Continued)

		· · · · · · · · · · · · · · · · · · ·
Deference		Assembly Drawing or
Reference		
Designation	·	Installation
Number	Assembly	Drawing
6215	Service Arm Power "J" Box	Cable Assy.
	· ·	D75M09084
6216	Hinge Upper, S-II Fwd	J75M08260-1
6217	Hinge Lower, S-II Fwd	J75M08261-1
6222	Control Distr. #1	Assy.
		J75M10030-25
		Wiring
		A75M09346
6223	Control Distr. #2	Assy.
		J75M10030-27
		Wiring
		A75M09347
6224	Control Distr. #3	Assy.
		J75M10030-29
		Wiring
		A75M09348
6226	Console #2 "J" Box	Assy
		J75M07975
		Harness
		D75M11366
6227	Local Control Unit, S-IVB Aft	Assy.
		J75M10898
		*J75M07578-11
6228	Control Console #1	*J75M08726
6229	Control Console #1	J75M07978
6231	Umb. Carrier, S-IVB Aft	Cable Assy.
		D75M09088
6232	LOX Carrier, S-IVB Aft	Cable Assy.
		D75M09088
6233	LH, Carrier, S-IVB Aft	Cable Assy.
	2	D75M09088
6234	Service Arm Power "J" Box	Cable Assy.
		D75M09088
6235	Hinge Upper, S-IVB Aft	J75M08260-1
6236	Hinge Lower, S-IVB Aft	J75M08261-1
	* Electrical Installation Drawing	

Table 6-4 REFERENCE DESIGNATION NUMBERS VERSUS ASSY DRAWING NUMBERS (Continued)

Reference Designation Number	Assembly	Assembly Drawing or Installation Drawing
6262	Control Distr. #1	Assy. J75M10030-31
		Wiring
		A75M10548
6263	Control Distr. #2	Assy.
		J75M10030-33
		Wiring A75M10552
6264	Control Distr. #3	A75M10552 Assy.
0201		J75M10030-35
		Wiring
		A75M10556
266	Console #2 "J" Box	Assy.
		J75M07975
		Harness
		D75M11366
6267	Local Control Unit, S-IVB Fwd	Assy.
		J75M10898
60 00		*J75M07578-13
6268 6269	Control Console #1	*J75M08721
6269 6271	Control Console #2	*J75M07979
0411	Umb. Carrier, S-IVB Fwd	Cable Assy.
6272	LOX Carrier, S-IVB Fwd	D75M09092 Cable Assy.
0412	LOA Callier, S-IVB F wu	D75M09092
6273	LH ₂ Carrier, S-IVB Fwd	Cable Assy.
		D75M09092
6274	Service Arm Power "J" Box	Cable Assy.
		D75M09092
6275	Hinge Upper, S/M	J75M08260-1
6276	Hinge Lower, S/M	J75M08261-1
6291	Umb. Plate, Inst. Unit S-IVB Fwd	Cable Assy.
		D75M09092
	* Electrical Installation Drawing	

1

Table 6-4 REFERENCE DESIGNATION NUMBERS VERSUS ASSY DRAWING NUMBERS (Continued)

į

6-226

,

······	DRAWING NUMBERS (Continued)	
Reference Designation Number	Assembly	Assembly Drawing or Installation Drawing
6302	Control Distr. #1	Assy. J75M10030-37 Wiring
6303	Control Distr. #2	A75M10549 Assy. J75M10030-39 Wiring
6304	Control Distr. #3	A75M10553 Assy. J75M10030-41 Wiring A75M10557
6306	Console #2 "J" Box	
6307	Local Control Unit, S/M	*J75M07975
0301	Local Control Unit, S/M	Assy.
		J75M10898
6308	Control Console #1	*J75M07578-15
6309	Control Console #2	*J75M09055
6311		*J75M09052
0311	Umb. Carrier, S/M	Cable Assy.
6312		D75M09096
0312	LOX Carrier, S/M	Cable Assy.
6313	TT Comin C/M	D75M09096
0313	LH ₂ Carrier, S/M	Cable Assy.
6314	Service Arm Power ''J'' Box	D75M09096
0014	Service Arm Power "J" Box	Cable Assy.
6315	Hinge Upper, C/M	D75M09096
6316	Hinge Lower, C/M	J75M12139-1 J75M12139-3
6322	Control Distr. #1	
		Assy. J75M10030-49
		Wiring A75M12674
6324	Control Distr. #3	Arsmizor4 Assy.
5021		ASSY. J75M10030-51
		Wiring
		A75M12675
	* Electrical Installation Drawing	A101112010

Table 6-4 REFERENCE DESIGNATION NUMBERS VERSUS ASSY DRAWING NUMBERS (Continued)

ſ

Reference Designation Number	Assembly	Assembly Drawing or Installation Drawing
6325	Local Cont. Unit, C/M	Assy.
		J75M10898
		*J75M07578-17
6326	Oxygen Conditioning Console	J75M11608
6327	Service Arm Lighting "J" Box	Cable Assy.
		D75M09100
6328	Control Console #1	*J75M08722
6329	Accumulator Bottles	J75M05974
6357	Environmental Chamber	*J75M07104
6360	Q-Ball Control Panel	D75M06150
6361	Q-Ball Cover Control Distr.	F75M13155
6401	Flush & Purge Boom	delete
6402	Flush & Purge Distr.	delete
6404	He Distribution Panel	J75M07022
6405	GN ₂ Distribution Panel	J75M07021
6406	High Press Pneu. Distr. System	delete
6407	High Pressure Pneumatics Control Distr.	J75M11649
6413	Power Distr. S/A	J75M10671
6414	Power Distr. S/A	J75M10670
6415	Safety Sw.	D75M12973
6416	Distr. Assy Hyd Unit #2	D75M07071
6417	Starter Assy Hyd Unit #2	D75M07073
6418	Control Pnl Hyd Unit #2	D75M07070
6419	Motor Assy Hyd Unit #2	B75M08506
6420	Safety Sw. Hyd Unit #2	D75M12973
6421 6422	Pwr Supply Launcher Acc.	delete
6422 6422	Cable Installation Rack	J75M12096
6423 6424	Distr. Assy Hyd Unit #1	D75M07071
6424 6425	Starter Assy Hyd Unit #1	D75M07073
· 6425	Cont. Pnl. Assy Hyd Unit #1	D75M07070
· 6426 6427	Motor Assy Hyd Unit #1 Fuse Panel	B75M08506
6428	ruse Panel Fuse Panel	delete
6429		delete
	Interface Plate, Launcher Pwr Distr	delete
7	*Electrical Installation Drawing	

Table 6-4 REFERENCE DESIGNATION NUMBERS VERSUS ASSY DRAWING NUMBERS (Continued)

Reference Designation Number	Assembly	Assembly Drawing or Installation Drawing
6430	Pwr Module, Launcher Pwr Distr. 2	delete
6431	Pwr Module Launcher Pwr Distr. 1	delete
6432	Pwr Module, Spare	delete
6433	Power Transfer Module Chassis	delete
6434	Pwr Transfer Unit Batt. Rack	delete
6604	Deluge Purge Panel	delete
6605	Valve Panel #11	J75M07433
6606	Engine Service Platform Relay Distr.	E75M13471
6611	Launch Aux. Relay Patch Distr.	
6612	Flush & Purge Power Supply	
6613	Service Arm Control Rack #2	D75M08922
6614	Service Arm Control Rack #3	D75M08923
6615	Service Arm Control Rack #4	D75M08924
6616	Service Arm Control Rack #5	D75M08925
6617	Service Arm Control Rack #6	D75M08926
6618	Service Arm Control Rack #7	D75M08927
6619	Service Arm Control Rack #9	D75M08928
6620	Service Arm Control Rack #8	D75M08931
6621	Relay Distr. (Spare)	F75M04681
6622	LUT Integration Rack #1	D75M14069
6623	LUT Integration Rack #2	D75M15420
6629	Meter Panel, Integration Rack	J75M10165
6636	Purge Valve #1	B75M14551-1
6637	Purge Valve #2	B75M14551-1
6638	Purge Valve #3	B75M14551-1
6639	Purge Valve #4	B75M14551-1
6640	Crossover Distr. S-IC, Room 13-A	delete
6641	Crossover Distr. S-IC, Room 5-A	delete
6650	Launcher Access. Relay Rack #1	J75M12976
6651	Launcher Access. Relay Rack #2	J75M12977
6652	Hydraulic Charging Unit Electric Distr. #2	D75M07071
6653	Hydraulic Charging Unit Starter Panel #2	D75M07073
6654	Hydraulic Charging Unit Control Panel #2	D75M07070
6655	Hydraulic Charging Unit Pump Motor #2	Cable Assy.
0000	nyuraune Charging one rump motor "A	D75M07072
	* Electrical Installation Drawing	

Table 6-4 REFERENCE DESIGNATION NUMBERS VERSUS ASSY DRAWING NUMBERS (Continued)

	DRAWING NUMBERS (Continued)	
References Designed on Number	1. sembly	Assembly Drawing or Installation Drawing
665 (*	Connector Plate, Rela, Rack	Assy
66 57	Connector Plate, Relay Rack	J75M09711 Assy.
6658	Connector Plate, Relay Rack	J75M09711 Assy.
6 655	Connector Plate, Relay Rack	J75M09711 Assy.
46 04	Connector Plate, Relay Back	J75M09711 Assy.
8601	Connector Plate, Relay Rack	J75M09711
	Connector Plate, Relay Rack	Assy. J75M09711
Sterring States	T. S. Mast Eelay Distr. Int. Plate	Assy. J75M09711
		Assy. J75M09711
	Launcher Access Distr. Int. Plate	Assy. J75M09711
665.7 0736.	Valve Panel #11 Control Distr. Interface Plate, Launcher Power Rack	F75M11652 Assy.
610	Fuse Panel Assy	J75M09711-1 *J75M12096 Assy. J75M10163-1
671.2	Power Module Frame Assy	*J75M12096 D75M12301
.69.6	Power Module R/ASTP Integration	Assy.
0724	Power Module, Launcher Accessories Distr. #1	*J75M12096 Assy.
Star Sec.	Power Module, Launcher Accessories Distr. #2	*J75M12096 Assy.
.671 <i>6</i> 9	Power Module Spare	*J75M12096 Assy. *J75M12096
	* Electrical Installation Drawing	

Table 6-4 REFERENCI: DESIGNATION NUMBERS VERSUS ASSY DRAWING NUMBERS (Continued)

7 6-236

Reference Designation Number	Assembly	Assembly Drawing or Installation Drawing
6717	Power Module GETS #1	Assy.
		*J75M12096
6718	Power Module GETS #2	Assy.
		*J75M12096
6726	Bus Distributor	*J75M12096
6727	Bus Distributor	*J75M12096
7200	F1 Engine Change Platform (VAB)	*J75M51342
7201	F1 ESP Platform Distr.	E75M13209
7202	S-IC Engine Service Top Plate (VAB)	J75M09312-5A
7203	S-IC Engine Service Boom Plate (VAB)	J75M09312-3A
7204	Distribution Box, S-II Engine Servicing Platform	B75M14680
7205	"J" Box Winch #1	B75M14683
7206	"J" Box Winch #2	B75M14683
7207	"J" Box Winch #3	B75M14683
7208	"J" Box Winch #4	B75M14683
7601	Tail Mast Test Set	E75M07658
7602	Launcher Accessories Test Set No. 2	J75M11087
7603	Portable Arm Control Console	J75M07542
7604	Portable Control Unit Distr.	delete
7605	Portable Control Unit Power Supply	401010
7606	Tower Test Set	E75M14137
7607	Portable Power Supply (Pneumatic)	
7650	Launcher Acc. Test Set No. 1	E75M14009
7656	GETS	E75M13286
7657	GETS Simulator	E75M13286
7658	Launcher Test Set	E101110200
7850	Platform Motor & Brake Assy #1	*J75M51413
7851	Platform Motor & Brake Assy #2	*J75M51413
7852	Platform Motor & Brake Assy #3	*J75M51413
7853	Platform Motor & Brake Assy #4	*J75M51413
7854	Platform Motor & Brake Assy #5	*J75M13064
7855	Umbilical Plate, TSM 3-4	J65B80001
7856	Umbilical Plate, TSM 1-2	J65B80002
7857	Umbilical Plate, TSM 3-2	J65B80003
	U	00000000
	* Electrical Installation Drawing	

Table 6-4 REFERENCE DESIGNATION NUMBERS VERSUS ASSY DRAWING NUMBERS (Continued)



DRAWING NUMBERS (Continued)		
Reference Designation Number	Assembly	Assembly Drawing or Installation Drawing
7858	Haunch Plate, Holddown Arm #1	J75M05972
7859	Haunch Plate, Holddown Arm #2	J75M05972
7860	Haunch Plate, Holddown Arm #3	J75M05972
7861	Haunch Plate, Holddown Arm #4	J75M05972
7862	LUT Level Interface Plate Aux Motor "A"	*J75M51413
7863	LUT Level Interface Plate Motor Brake "A"	*J75M51413
7865	LUT Level Interface Plate Aux Motor "B"	*J75M51413
7866	LUT Level Interface Plate Motor & Brake "B"	*J75M51413
7867	LUT Level Interface Plate Aux Motor "C"	*J75M51413
7868	LUT Level Interface Plate Motor & Brake "C"	*J75M51413
7869	LUT Level Interface Plate Aux Motor "D"	*J75M51413
7870	LUT Level Interface Plate Motor & Brake "D"	*J75M51413
2872	LUT Level "0" Platform Elect. System	*J75M51639
950	Distributor Box #1, Platform Transporter Control	
	North	B75M14679-1
7951	Distributor Box #2, Platform Transporter Control	
	North	B75M14679-2
7952	Portable Controller, Platform Transporter Control	B75M14682
7953	Distributor Box #1, Platform Transporter Control	
	South	B75M14679-3
7954	Distributor Box #2, Platform Transporter Control	
	South -	B75M14679-4
7955	Portable Controller, Platform Transporter Control	
	South	B75M14682
7956	Motor/Winch Bulkhead, Platform Transporter	
	Control #1	J75M14891
7957	Motor/Winch Bulkhead, Platform Transporter	
	Control #2	J75M14891
9001	LUT - Pad Interface, Power	F75M12270
9002	LUT - Crawler Interface, PWR	F75M05121
9006	UB. Instr. & Control Term Distr	F75M05121
9008 9000	UB. Instr & Control Term Distr-80' (MSC)	F75M12270
9009 9010	UB. Power Term Distr-80' (MSC)	F75M12270
	UB. Instr. & Control Term Distr-100'	F75M05121
	*Electrical Installation Drawing	

Table 6-4 REFERENCE DESIGNATION NUMBERS VERSUS ASSY DRAWING NUMBERS (Continued)

Reference Designation Number	Assembly	Assembly Drawing or Installation Drawing
9013	UB. Power Term Distr-120' (S-IC FWD)	F75M05121
9014	UB. Instr & Control Term Distr - 140'	F75M05121
9016	UB. Instr & Control Term Distr - 160'	F75M05121
9017	UB. Power Term Distr - 160'	F75M05121
9020	UB. Instr & Control Term Distr-200'	F75M05121
9022	UB. Instr & Control Term Distr-220'	F75M05121
9023	UB. Power Term Distr-220'	F75M05121
9025	UB. Instr & Control Term Distr-260'	F75M05121
9026	UB. Instr & Control Term Distr-260'	F75M05121
9027	UB. Power Term Distr-260'	F75M05121
9028	UB. Instr & Control Term Distr-280'	F75M05121
9029	UB. Power Term Distr-280'	F75M05121
9030	UB. Instr & Control Term Distr-300'	F75M05121
9031	UB. Power Term Distr-300'	F75M05121
9032	UB. Instr & Control Term Distr-320'	F75M05121
9033	Prop. Loading A.C. PWR "J" Box-30'	F75M05121
9034	Prop. Loading A.C. PWR "J" Box-120'	F75M05121
9035	Prop. Loading A.C. PWR "J" Box-200'	F75M05121
9038	Crane	F75M05120
9042	UB. Power "J" Box 30' Tower Level	F75M05121
9043	UB. Power "J" Box 60' Tower Level	F75M05121
9044	UB. Power "J" Box 120' Tower Level	F75M05121
9045	UB. Power "J" Box 140' Tower Level	F75M05121
9046	UB. Power "J" Box 200' Tower Level	F75M05121
9047	UB. Power "J" Box 320' Tower Level	F75M05121
9048	OTV Cables, Phase III	F75M12270
9049	Measuring Acquisition Term Distrs.	F75M05121
9050	Measuring Coax Term. Distrs.	F75M05121
9051	UB. Instr & Control Distr-300' (MSC)	F75M12270
9052	Service Arm MI Term Distr-120' Level	F75M05121
9054	Service Arm MI Term Distr-160' Level	F75M05121
9055	Service Arm MI Term Distr-200' Level	F75M05121
9056	Service Arm MI Term Distr-220' Level	F75M05121
9057	Service Arm MI Term Distr-260' Level	F75M05121
	* Electrical Installation Drawing	

Table 6-4 REFERENCE DESIGNATION NUMBERS VERSUS ASSY DRAWING NUMBERS (Continued)

.

DRAWING NUMBERS (Continued)		
Reference Designation Number	Assembly	Assembly Drawing or Installation Drawing
9059 9060 9061 9062 9063 9064 9065 9068 9069 9070 9072 9073 9072 9073 9076 9077 9078 9076 9077 9078 9077 9078 9081 9082 9083 9084 9085 9084 9085 9086 9087 9088 9088 9088 9088 9089 9090 9091 9092	Service Arm MI Term Distr-300' Level Coax Distr - MSC. 300' Level Service Arm Lighting Junction Boxes Service Arm Emergency PWR. "J" Boxes "J" Box, S-IC FWD Foamflex (120') "J" Box, S-IC FWD Foamflex (200') "J" Box, S-IVB FWD Foamflex (260') Power Panel ("A" Suffixes) Lighting Panel ("A" Suffixes) Term. Rm. Instr & Control Distr. Rm. 1-A Instr & Control Distr. Rm. 7-B Power Term Distr. MSC PWR Distr Rm. 1-A Anem & Film Camera Distr 9-B Coax Cable, Meas, 10B Rack 42 Pair Cables, Meas, 10B Rack 43 S-IC Eng. Ign. Term. Distr, Rm. 7-B Rm. 8-A Instr & Control Distr. Firing Distributor ML Water Control Cabinet Prop Loading AC PWR "J" Box, Rm. 7-A Inst Gnd Plates, MSC, Rm. 1-A Computer Interface, Rm. 7-A No. 1 Tail Service Mast Instr & Control No. 1 Tail Service Mast Instr & Control No. 2 Tail Service Mast Instr & Control Term Distr Rm. 5-A S-IC AFT PWR, Distr No. 1 Rm. 13-A S-IC AFT PWR, Distr No. 2 Rm. 5-A Prop Loading AC PWR "J" Box, Rm. 4-A PTCR-LUT Instr & Control Interface * Electrical Installation Drawing	F75M05121 F75M12270 F75M12270 F75M12270 F75M12270 F75M05121 F75M05121 F75M05121 F75M05121 F75M12270 F75M12270 F75M12270 F75M05121 F75M05121 F75M05121 F75M05121
	* Electrical Installation Drawing	

Table 6-4 REFERENCE DESIGNATION NUMBERS VERSUS ASSY DRAWING NUMBERS (Continued)

			•		
5000-5049	ENVIRONMENTAL CONTROL SYSTEM		6380-6399	3 80'	
5050	LH ₂ STORAGE AREA		6360-6379	2 60'	
5051	LOX STORAGE AREA		6340-0359	3 40'	
5052	RP-1 STORAGE AREA		6320-6320	201	
5053-5054	(FUTURE)		6300-6319	[⊗] (101	
5055-5079	PNEUMATICS SYSTEM (CCF)		6280-6299	≥901	
5080-5099	PAD & PERIMETER AREA	,	6260-6279	2601	
5100-5299	PTCR (1st & 2nd FLOOR)		6240-6259	240'	
5300-5999	LCC (FIRING ROOMS 1 THRU 4)		6220-6239	220'	
			6200-6219	200'	
6900-6999	(FUTURE)		6180-6199	° 801	
7000-7199	ARMING TOWER		<u>9719</u>	1 50'	
7200 7499	VAB		· · · · · · · · · · · · · · · · · · ·	40'	
7500-7549	MOBILE LAUNCHER ERECT. & REFURE	3.		_20'	
7550-7599	CRAWLER - TRANSPORTER		6100-6119	100f	
7600-7799	PORTABLE TEST EQUIPMENT		6080-9099	:: 0 1	
7800-7814	PNEUMATICS STORAGE BATTERY (VAE	3)	6069-6079	30'	
7815-78 29	PNEUMATICS STORAGE BATTERY (PAI))	6020-6059	0'	
7830-89 99	(FUTURE)		6000-6029	0'	
		LEVI	EL " 6600-68.		,
9000-9099	MOBILE LAUNCHER TERM. DISTR.	LEVI	EL"E" 6400-655		
9100-9299	LCC TERMINAL DISTRIBUTORS				
9300-9499	VAB TERMINAL DISTRIBUTORS				
9500-9699	PTCR TERMINAL DISTRIBUTORS				
9700-9799	ARMING TOWER TERM. DISTR.				
9800-9899	OTHER LAUNCH AREA TERM. DISTR.				
9900-9999	(FUTURE)				

Figure 6-7 Reference Designation Assignment Areas, LC-39

ند. 120

APPENDIX A

GLOSSARY

C/M	Command Module, Apollo spacecraft
DDAS	Digital Data Acquisition System
ESE	Electrical Support Equipment (ESE) which is responsibility of Launch Equipment Branch
DLTR	Digital Link Transmission Repeater
Firing Batteries	See Service Arms Firing Circuits, par. 3-25
GETS	Ground Equipment Test Set
GN ₂	Gaseous Nitrogen
GOX	Gaseous Oxygen
GSE	Ground Support Equipment
Не	Helium
H.P. Gas	High Pressure Gas
Inflight Arms	Service Arms which retract during launch
. •	Service Arms which retract during launch Kennedy Space Center
Arms	-
Arms KSC	Kennedy Space Center
Arms KSC LCC	Kennedy Space Center Launch Control Center
Arms KSC LCC LC-39	Kennedy Space Center Launch Control Center Launch Complex 39
Arms KSC LCC LC-39 LEM	Kennedy Space Center Launch Control Center Launch Complex 39 Lunar Excursion Module
Arms KSC LCC LC-39 LEM LES	Kennedy Space Center Launch Control Center Launch Complex 39 Lunar Excursion Module Launch Escape System
Arms KSC LCC LC-39 LEM LES LH ₂	Kennedy Space Center Launch Control Center Launch Complex 39 Lunar Excursion Module Launch Escape System Liquid Hydrogen
Arms KSC LCC LC-39 LEM LES LH ₂ LOX	Kennedy Space Center Launch Control Center Launch Complex 39 Lunar Excursion Module Launch Escape System Liquid Hydrogen Liquid Oxygen
Arms KSC LCC LC-39 LEM LES LH ₂ LOX MI Distributor	Kennedy Space Center Launch Control Center Launch Complex 39 Lunar Excursion Module Launch Escape System Liquid Hydrogen Liquid Oxygen Terminating Mineral Insulated Cable

GLOSSARY (Continued)

Preflight Arms	Service Arms which retract prior to launch
Q-Ball	A device on the vehicle nose for measuring temperature or aerodynamic pressure
R. F. I.	Radio Frequency Interference
SCR	Silicon Control Rectifier
S-IC	Saturn V first stage
S-II	Saturn V second stage
S-IVB	Saturn V third stage
SM	Service Module, Apollo spacecraft
TCS	Terminal Countdown Sequencer
TSM	Tail Service Mast
UB	Umbilical
VAB	Vehicle Assembly Building

i t

A-2